

**AIRWORTHINESS DIRECTIVES & SERVICE BULLETIN - STATUS REPORT**

Rev. 1  
Revisado: 25-Ene-19

<b>A/C REGISTER</b>		<b>REPORT DATE</b>	2-sep-21	<b>PAGE N.</b>	1 de 73
<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>APU</b>	<b>CSO</b>
					N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
68-25-02 SB 24-1009 R1	10-dic-68	<b>INTERPHONE CIRCUIT CARD.</b> N/A por efectividad del SB 24-1009 R1, este solo aplica a los series 100 y 200								-	-	-	-	-	-	-	-	-	-	-	-
69-17-01 SB 49A1760	Upon Receipt	<b>APU IMPELLER BLADES FAILURE.</b> N/A por P/N del APU según SB 49A1760 (Julio 01, 1969).								-	-	-	-	-	-	-	-	-	-	-	-
69-20-06 SB 24-1011 R1	30-oct-69	<b>PREVENT OVERLOAD OF CIRCUITS USING P/N BAC-C18L CB'S.</b> N/A por efectividad del SB 24-1011 R1 (mayo 21- 1969), este solo aplica a las series 100 y 200.								-	-	-	-	-	-	-	-	-	-	-	-
70-04-03 SB 36-1003 R2	9-jul-70	<b>APU BLEED AIR DUCT/APU FIRE EXTINGUISHER BOTTLE.</b> N/A por efectividad del SB 36-1003 R2 (JUNIO 12- 1970) y 26-1011 (Junio 9, 1970) estos solo aplican a las series 100 y 200.								-	-	-	-	-	-	-	-	-	-	-	-
70-06-03 SB 27-1031 R4	13-mar-70	<b>TAKE OFF WARNING SWITCH ACTUATING CAM REPLACEMENT AND RIGGING.</b> N/A por efectividad del SB 27-1031 R4 (Agosto 4- 1970), este solo aplica a las series 100 y 200.								-	-	-	-	-	-	-	-	-	-	-	-
70-09-01 SB 24-1010 R1	23-abr-70	<b>OPERATION OF BATTERY SWITCH.</b> N/A por efectividad del SB 24-1010 R1 (Abril 24- 1970), este solo aplica a las series 100 y 200.								-	-	-	-	-	-	-	-	-	-	-	-
70-18-06 SB 52-1034 R1	10-sep-70	<b>EMERGENCY HATCH HANDLE.</b> N/A por efectividad del SB 52-1034 R1 (Septiembre 24- 1970), este solo aplica a las series 100 y 200.								-	-	-	-	-	-	-	-	-	-	-	-
73-09-04 SB 52-1044 R1	Upon Receipt	<b>ENTRY DOOR HINGE.</b> N/A por efectividad del SB 52-1044 R1 (Junio 15- 1973), este solo aplica a las series 100 y 200.								-	-	-	-	-	-	-	-	-	-	-	-
74-08-09 R3	28-mar-12	<b>LAVATORIES WASTE RECEPTACLES.</b> Placard & Ahstray Installation and announcement Procedures. Ultimo cumplimiento mediante documento WO 0500 de GCA.											39896,2	57421	4-may-21	40896,20	-	-	656,12	-	-
74-09-05 SB 25-1101 R1	Upon Receipt	<b>EMERGENCY OVERWING EXITS.</b> N/A por efectividad del SB 25-1101 R1 (Mayo 22- 1974), este solo aplica a las series 200.								-	-	-	-	-	-	-	-	-	-	-	-
74-20-02 SB 34-1031 R2	Upon Receipt	<b>PITOT STATIC TUBING.</b> N/A por efectividad del SB 34-1031 R2 (Febrero 14- 1975), este solo aplica a las series 100 y 200.								-	-	-	-	-	-	-	-	-	-	-	-
74-21-03 SB 25-1096 R3	10-nov-75	<b>LAVATORY WASTE CONTAINERS.</b> N/A por efectividad del SB 25-1096 R3 (Octubre 31- 1975) y 25-1108 R3 (Octubre 31- 1975), estos solo aplican a las series 100 y 200.								-	-	-	-	-	-	-	-	-	-	-	-

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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>APU</b>	<b>CSO</b>
					N/A

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75-04-08 SB 29-1004 R1	10-mar-75	<b>B' HYDRAULIC SYSTEM ELECTRICAL WIRING.</b> N/A por efectividad del SB 29-1004 R1 (Octubre 11- 1968), este solo aplica a las series 100 y 200.								-	-	-	-	-	-	-	-	-	-	-	-
75-05-01 SB 27-1073 R1	7-abr-75	<b>CONTROL CABLE PULLEYS.</b> N/A por V/N y MSN en efectividad (párrafo I.A.1.) del SB 27-1073 R1 (MAR 4, 1975), solo aplica a aeronaves que tienen instaladas poleas marca Arvan Inc./Tansey con P/N MS 20220-3/4; SMS 20220-3/-4 (Solamente Color negro) (BAC P30F8 o BAC P30F9).								-	-	-	-	-	-	-	-	-	-	-	-
75-05-09 SB 28-1022	24-mar-75	<b>ENGINE FUEL SHUTOFF AND CROSSFEED VALVE WIRE BUNDLES.</b> N/A por efectividad del SB 28-1022 (Diciembre 20- 1974), este solo aplica a las series 100 y 200.								-	-	-	-	-	-	-	-	-	-	-	-
75-20-02 SB 27-1041 R4 SB 27-1076 R1	9-feb-76	<b>TRAILING EDGE FLAPS RETRACTION AND LEADING EDGE FLAP EXTENSION.</b> N/A por Line Number y por efectividad del SB 27-1041 R4 (Abril 4- 1972) y SB 27-1076 R1 (Agosto 29- 1975), estos solo aplican a las series 100 y 200.								-	-	-	-	-	-	-	-	-	-	-	-
75-25-02 SB 52-1004 R2 SB 52-1034 R1 SB 52-1054 R1	5-ene-76	<b>ESCAPE HATCH.</b> N/A por S/N y por efectividad del SB 52-1004 R2 (Noviembre 14, 1969), SB 52-1034 R1 (Septiembre 24 1970) y V/N en el SB 52-1054 R1 (NOV 8, 1990).								-	-	-	-	-	-	-	-	-	-	-	-
76-01-03 SB 53-1044	23-ene-76	<b>PASSENGER SEATS RESTRAINED.</b> N/A por efectividad del SB 53-1044 (AGO 29, 1975), este solo aplica a las series 100 y 200.								-	-	-	-	-	-	-	-	-	-	-	-
76-26-02 SB 52-1065 R6	2-mar-77	<b>CARGO DOORS.</b> N/A por V/N en efectividad del SB 52-1065 R6 (AGO 29, 1996).								-	-	-	-	-	-	-	-	-	-	-	-
78-13-07 SB 57-1082 R9 SB 57-1084 R6	12-jul-78	<b>TRAILING EDGE FLAP.</b> N/A por V/N en efectividad del SB 57-1082 R9 (SEP 11, 2003) y SB 57-1084 R6 (MAR 28, 1991)								-	-	-	-	-	-	-	-	-	-	-	-
79-07-03 SB 31-1026 R4	4-may-79	<b>THRUST LEVER OPERATION.</b> N/A por efectividad del SB 31-1026 R4 (ENE 29, 1987).								-	-	-	-	-	-	-	-	-	-	-	-
79-23-02 SB 35-A1014	15-nov-79	<b>CARRY-ALL' INTERIOR.</b> N/A por efectividad del SB 35-A1014 (JUL 6, 1979).								-	-	-	-	-	-	-	-	-	-	-	-
80-02-02 R2 SB 28-1032 R2	20-nov-80	<b>AUXILIARY BODY FUEL TANK INSTALLATION.</b> N/A por V/N en efectividad del SB 28-1032 R2 (MAY 4, 1995).								-	-	-	-	-	-	-	-	-	-	-	-
80-07-02	21-abr-80	<b>FLIGHT CONTROL SYSTEMS.</b> Manual input hardover test. N/A por P/N de componentes hidraulicos, según estan listados en el parrafo (b) de la AD.								-	-	-	-	-	-	-	-	-	-	-	-

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<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>	<b>CSO</b>
BOEING		40240,1	57663	N/A	N/A

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80-22-12 R2 SB 31-1038 R4	21-jun-83	<b>LEADING EDGE DEVICES.</b> N/A por V/N en efectividad del SB 31-1038 R4 (MAR 16, 1984).								-	-	-	-	-	-	-	-	-	-	-	-
80-26-51 SB 27-A1109 R1	13-dic-80	<b>CONTROL SYSTEM JAMMING.</b> N/A por Line Number y por efectividad del SB 27-A1109 R1 (DEC 11, 1980), este solo aplica para series 100 y 200.								-	-	-	-	-	-	-	-	-	-	-	-
81-20-04 R1 SB 21-1064	25-feb-85	<b>HAMILTON STANDARD PRESSURIZATION CONTROLLERS.</b> N/A por efectividad del SB 21-1064 (AGO 28, 1981), este solo aplica a las series 100 y 200. N/A por P/N del controlador de presión de cabina según IPC.								-	-	-	-	-	-	-	-	-	-	-	-
82-01-09 SB 53-A1042 R9	3-feb-86	<b>LOWER BODY SKINS CORROSION.</b> N/A por L/N en la efectividad del SB 53-A1042 R9 (JUL 25, 1991).								-	-	-	-	-	-	-	-	-	-	-	-
82-18-10	6-sep-85	<b>AIRCRAFT SEATS. TRACK LOCK FITTINGS REPLACEMENT.</b> N/A por S/N de las sillas instaladas de acuerdo a aplicabilidad de la AD, según documento "LOPA SEATS".								-	-	-	-	-	-	-	-	-	-	-	-
84-10-04 SB 24-1038 R1	12-oct-85	<b>APU FEEDER CABLE.</b> N/A por V/N en efectividad del SB 24-1038 R1 (JUN 29, 1984).								-	-	-	-	-	-	-	-	-	-	-	-
84-23-05 SB 55-A1029 R4	25-nov-85	<b>HORIZONTAL STABILIZER.</b> N/A por V/N en efectividad del SB 55-A1029 R4 (OCT 26, 1989).								-	-	-	-	-	-	-	-	-	-	-	-
85-01-06 SB 53-A1064 R6	30-jun-86	<b>FORWARD AIRSTAIR ADJACENT FRAMES INSP.</b> N/A por V/N en efectividad del SB 53-A1064 R6 (JUN 29, 1989).								-	-	-	-	-	-	-	-	-	-	-	-
85-01-07 SB 57-1087 R9	25-feb-85	<b>BBL 70.85 RIB UPPER CHORD INSPECTION.</b> N/A por V/N en efectividad del SB 57-1087 R9 (JUL 6, 1989).								-	-	-	-	-	-	-	-	-	-	-	-
85-03-06 R1 SB 53-1031 R4	3-feb-86	<b>UPPER DRAG ANGLES INSP.</b> N/A por V/N en efectividad del SB 53-1031 R4 (JUN 15, 1989).								-	-	-	-	-	-	-	-	-	-	-	-
85-16-05 SB 25-1171	6-sep-85	<b>FIRE PREVENTION - LAVATORIES.</b> N/A por L/N en efectividad del SB 25-1171 (AGO 16, 1984).								-	-	-	-	-	-	-	-	-	-	-	-
85-19-01 SB 27-A1126 R1	12-oct-85	<b>ROSEMOUNT AOA SENSORS.</b> N/A modelo de sensor AOA no instalado. N/A por V/N en efectividad del SB 27-A1126 R1 (OCT 25, 1985).								-	-	-	-	-	-	-	-	-	-	-	-
85-22-02 SB 57-1137 R6	25-nov-85	<b>BBL RIB UPPER CHORD INSP.</b> N/A por V/N en efectividad del SB 57-1137 R6 (JUN 14, 1990).								-	-	-	-	-	-	-	-	-	-	-	-
86-12-04 SB 55-1034 R2	30-jun-86	<b>HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD.</b> N/A por V/N en efectividad del SB 55-1034 R2 (MAY 10, 1990).								-	-	-	-	-	-	-	-	-	-	-	-
86-12-05 SB 55-A1033 R2	30-jun-86	<b>HORIZONTAL STABILIZER ATTACH LUG.</b> N/A por V/N en efectividad del SB 55-A1033 R2 (AGO 7, 2015).								-	-	-	-	-	-	-	-	-	-	-	-



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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>APU</b>	<b>CSO</b>
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89-04-03 SB 52-A1105 R2	10-mar-89	<b>COCKPIT DOOR BLOWOUT PANELS.</b> N/A por L/N en efectividad de la AD (aplica de la 1 a la 1587). N/A por V/N en efectividad del SB 52-A1105 R2 (JUL 13, 1989) (no hay L/N iniciados en PV).						-	-	-	-	-	-	-	-	-	-	-	-
89-07-13	24-ago-89	<b>GENERATOR CONTROL UNIT FILTER.</b> N/A GCU P/N 915F212- 5 y 948F458-1 no instalados.						-	-	-	-	-	-	-	-	-	-	-	-
89-08-12 R1 SB 26-1063	4-dic-89	<b>FIRE/OVERHEAT DETECTION MODULE.</b> N/A por V/N en efectividad del SB 26-1063 (MAY 18,1989)						-	-	-	-	-	-	-	-	-	-	-	-
89-09-03 S/B 53-A1039	16-feb-90	<b>LAP JOINTS.</b> N/A por L/N en efectividad de la AD y el S/B 53-A1039 (solo L/N del 1 al 519)						-	-	-	-	-	-	-	-	-	-	-	-
89-11-03	11-dic-89	<b>LAVATORY SERVICE SYSTEM.</b> supercedida por la AD 2004-06-18						-	-	-	-	-	-	-	-	-	-	-	-
89-11-06 R1 SB 53-A1027 R7	21-feb-90	<b>FUSELAGE SKIN INSPECTION.</b> N/A por V/N en efectividad del SB 53-A1027 R7 (DIC 14, 1989) (no hay V/N iniciados en PV).						-	-	-	-	-	-	-	-	-	-	-	-
89-12-02 SB 25-1231 R1	29-jun-89	<b>LAVATORY.</b> N/A por V/N en efectividad del SB 25-1231 R1 (ENE 12, 1989) (no hay V/N iniciados en PV)						-	-	-	-	-	-	-	-	-	-	-	-
89-14-11	7-ago-89	<b>SEAT PAN ROLLER.</b> N/A por P/N 90835 no instalado.						-	-	-	-	-	-	-	-	-	-	-	-
89-15-08	24-ago-89	<b>OBSERVERS SEAT.</b> N/A por L/N en efectividad de la AD y del SL-25-41-B (OCT 31, 1989) (solo L/N del 1 al 1581)						-	-	-	-	-	-	-	-	-	-	-	-
89-23-10	4-dic-89	<b>ENGINE ICE INGESTION.</b> Cumplida mediante actualizacion del AOM respecto a la ultima revision de My Boeing Fleet.						-	-	-	-	-	-	-	-	-	-	-	-
89-23-52	16-feb-90	<b>ENTRY DOOR BOLTS.</b> N/A por L/N en efectividad de la AD. (solo L/N entre 1770 y 1789)						-	-	-	-	-	-	-	-	-	-	-	-
89-25-02 SB 77-A1026 R2 SB 77-A1025 R2	11-dic-89	<b>ENGINE IDLE CIRCUITRY.</b> N/A por V/N en efectividad de los SB 77-A1026 R2 (FEB 1, 1990) Y 77-A1025 R2 ( FEB 1, 1990) ( Solo aplica hasta los V/N PV020)						-	-	-	-	-	-	-	-	-	-	-	-
89-26-06 R1 SB 35-A1029 R3	17-jul-90	<b>OXYGEN GENERATORS.</b> N/A por V/N en efectividad de SB 35-A1029 R3 (JUN 29, 1989) (No aplica a V/N iniciados en PV)						-	-	-	-	-	-	-	-	-	-	-	-
90-03-18 SB 31-1033 R3	7-mar-90	<b>TAKEOFF CONFIGURATION WARNING SYSTEMS.</b> N/A por V/N en la efectividad del SB 31-1033 R3 (OCT 14, 1993) (No aplica a V/N que inician en PV)						-	-	-	-	-	-	-	-	-	-	-	-
90-06-02	17-april-90	<b>STRUCTURAL MODIFICATIONS.</b> N/A por efectividad de la AD para modelos 737-400						-	-	-	-	-	-	-	-	-	-	-	-





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<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
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91-07-14 SB 25-1266	29-abr-91	<b>MAGNETIC STAND BY COMPASS LINER REPLACEMENT.</b> N/A por V/N en efectividad del SB 25-1266 (JUL 26, 1990)									-	-	-	-	-	-	-	-	-	-	-	-
91-08-12 SB 53-1124 SB 53-1076 R2	17-may-91	<b>BONDED DOUBLERS INSPECTION.</b> N/A por L/N en efectividad de la AD. Solo aplica a los L/N entre 520 y 610.									-	-	-	-	-	-	-	-	-	-	-	-
91-09-05 SB 23-1059 R1	28-may-91	<b>COMMUNICATIONS - INTERPHONE - CREW INTERPHONE SYSTEM - CREW CALL HORN CIRCUIT BREAKER INSTALLATION.</b> N/A por L/N en efectividad de la AD (Solo aplican los L/N anteriores al 2000). N/A por V/N en efectividad del SB 23-1059 R1 (OCT 3, 1991).									-	-	-	-	-	-	-	-	-	-	-	-
91-09-12 SB 26-1067 R1	28-may-91	<b>FIRE PROTECTION - ENGINE FIRE EXTINGUISHING SYSTEM - WIRING AND PLUMBING CHANGES.</b> N/A por V/N en la efectividad del SB 26-1067 R1 (NOV 14, 1991). Solo Aplica a los V/N del PV001-PV034.									-	-	-	-	-	-	-	-	-	-	-	-
91-09-17 SB 57-A1202 R1	3-jun-91	<b>WINGS - OUTBOARD FLAP INBOARD TRACK SUPPORT FITTING ATTACH BOLT AND NUT INSPECTION/REPLACEMENT.</b> N/A por V/A en efectividad del SB 57-A1202 R1 (ABR 26, 1990). Solo aplican los V/N del PV001 al PV020.									-	-	-	-	-	-	-	-	-	-	-	-
91-16-08 SB 28-1088	6-sep-91	<b>AUXILIARY FUEL SYSTEM.</b> N/A por V/N en efectividad del SB 28-1088 (SEP 6, 1990). No aplica a V/N iniciados en PV.									-	-	-	-	-	-	-	-	-	-	-	-
91-20-04 SB 27-1161 R2	7-nov-91	<b>FLIGHT CONTROLS - HORIZONTAL STABILIZERS - HORIZONTAL STABILIZER TRIM CONTROL SYSTEM - AUXILIARY BRAKE REPLACEMENT.</b> N/A por V/N en efectividad del SB 27-1161 R2 (SEP 30, 1993). Solo aplican los V/N del PV001-PV041.									-	-	-	-	-	-	-	-	-	-	-	-
91-21-12	12-nov-91	<b>UNCOMMANDED FREQUENCY CHANGES OF THE HIGH FREQUENCY (HF) AND VERY HIGH FREQUENCY (VHF) RADIOS.</b> N/A por panel de control de radios con P/N 285U0037 no instalado según IPC 285A1-00-01.									-	-	-	-	-	-	-	-	-	-	-	-
91-21-13 SB 76-1023	29-nov-91	<b>ENGINE CONTROLS - ENGINE CONTROL SYSTEM - REPLACEMENT OF ENGINE THROTTLE CONTROL CABLES.</b> N/A por V/N en efectividad del SB 76-1023 (FEB 14, 1991). Solo aplica a los V/N del PV001-PV052.									-	-	-	-	-	-	-	-	-	-	-	-
91-25-06 SB 78-A1048	18-dic-91	<b>EXHAUST - THRUST REVERSER - AUXILIARY SLIDER TRACK LINER INSPECTION AND REPLACEMENT.</b> N/A por V/N en efectividad del SB 78-A1048 (AGO 04, 2005). Solo aplica a los V/N del PV001-PV018.									-	-	-	-	-	-	-	-	-	-	-	-





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<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>APU</b>	<b>CSO</b>
				N/A	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
92-16-15	1-sep-92	<b>OXYGEN MASK CONNECTORS.</b> N/A por P/N 289-57 de conector de mascarera de Oxigeno no instalado, según IPC CAP 35-12/22/31.								-	-	-	-	-	-	-	-	-	-	-	-
92-19-16 SB 25-1294 SB 25-1291 R1 SB 25-A1283 R2	20-nov-92	<b>OVERHEAD STOWAGE BINS - CHANGE OF FORWARD RIGHT FORTY-FIVE INCH OVERHEAD STOWAGE COMPARTMENT ASSEMBLY.</b> N/A por V/N en efectividad del SB 25-1294 (FEB 20, 1992), no aplican V/N iniciados en PV. N/A por L/N en efectividad de los SB 25-1291 R1 (NOV 24, 1992) y SB 25-A1283 R2 (SEP 19, 1991), solo aplican hasta el L/N 2182.								-	-	-	-	-	-	-	-	-	-	-	-
92-21-04 SB 52-1060	24-nov-92	<b>MAIN DECK CARGO DOOR - VISUAL AND WARNING INDICATION MODIFICATIONS.</b> N/A por L/N en efectividad del SB 52-1060 (JUN 11, 1976). Solo aplica para aeronaves de configuracion carguera y hasta el L/N 443.								-	-	-	-	-	-	-	-	-	-	-	-
92-21-05	24-nov-92	<b>AUXILIARY POWER UNITS - DAMAGE TO THE TITANIUM APU FIRE WALL DUE TO APU TIESHAFT SEPARATION.</b> N/A por P/N del APU en efectividad de la AD según "Spec. Sheet". La AD aplica para el P/N GTCP 36-280[B] y la aeronave cuenta con GTCP85-129H.								-	-	-	-	-	-	-	-	-	-	-	-
92-24-04 SB 78-A1055 R3	14-dic-92	<b>THRUST REVERSER CONTROL SYSTEM - MODIFICATION OF TEE AND RESTRICTOR FITTINGS ON THRUST REVERSER - CFM56 ENGINES.</b> N/A por L/N en efectividad del SB 78-A1055 R3 (ENE 13, 2000). Solo aplica a los L/N del 1001 al 2227 y a los V/N del PV001 al PV055.								-	-	-	-	-	-	-	-	-	-	-	-
92-25-04 SB 35-A1037 R1 SB 35-A1038	23-dic-92	<b>TEST STOP PLUNGERS.</b> N/A por V/N en efectividad del SB 35-A1037 R1 (ABR 8, 1993) y SB 35-A1038 (MAR 19, 1992). Solo aplica a los V/N del PV001 al PV055.								-	-	-	-	-	-	-	-	-	-	-	-
92-25-09 SB 53-1076 R4	12-ene-93	<b>CORROSION INSPECTION - FUSELAGE - CIRCUMFERENTIAL BUTT SPLICES AND BONDED DOUBLERS INSPECTION, REPAIR AND MODIFICATION.</b> N/A por L/N en efectividad del SB 53-1076 R4 (SEP 26, 1991). Solo aplica a los L/N del 001 al 519.								-	-	-	-	-	-	-	-	-	-	-	-
93-01-27 SB 27-A1164 R2	9-mar-93	<b>FLIGHT CONTROLS - FLIGHT CONTROLS MULTIPLE USE SYSTEMS/UNITS - CONTROL CABLES GUARD REPLACEMENT.</b> N/A por L/N en efectividad del SB 27-A1164 R2 (AGO 4, 1994). Solo aplica hasta el L/N 1869								-	-	-	-	-	-	-	-	-	-	-	-





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<b>APPLICABILITY</b>		ENGINE LH		ENGINE RH	
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>CSO</b>	APU
				N/A	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE	INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O		FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS
95-15-06	23-ago-95	<b>ACTUATORS ON FUEL SYSTEM - IMPROPER FUNCTIONING.</b> N/A por P/N de fuel valve actuador.						-	-	-	-	-	-	-	-	-	-	-	-
95-22-01	4-dic-95	<b>FLUORESCENT LIGHTING SYSTEM - HIGH VOLTAGE ARCING.</b> N/A por P/Ns no instalados (de acuerdo a IPC FRONT MATTER ) en efectividad de la AD.						-	-	-	-	-	-	-	-	-	-	-	-
95-23-10	13-dic-95	<b>AUXILIARY POWER UNIT FIRE WARNING.</b> Cumplida mediante revision a la Pag 3, Seccion 2 (EMERGENCY PROCEDRES) del AFM.						-	-	-	-	-	-	-	-	-	-	-	-
95-26-15 R1	26-dic-96	<b>TRAFFIC ALERT AND COLLISION AVOIDANCE SYSTEMS - AUDIO OUTPUT OF THE CAS-81 TCAS FAILURE.</b> N/A por Marca y Modelo de TCAS en efectividad de la AD de acuerdo a "Spec Sheet" y "OCCM Component Status".						-	-	-	-	-	-	-	-	-	-	-	-
96-10-06 SB 52-A1124	24-may-96	<b>DOORS - FORWARD GALLEY SERVICE DOOR - LOWER GATE HINGE INSPECTION AND REPLACEMENT.</b> N/A por L/N y V/N en efectividad del SB 52-A1124 (ENE 11,1996), solo aplica a los L/N del 241 al 1727.						-	-	-	-	-	-	-	-	-	-	-	-
96-10-08	13-jun-96	<b>WIRE BUNDLE CLAMPS - LOWER 43 SECTION WIRE CHAFING.</b> N/A por L/N en efectividad del SL-24-106-C (AGO 20, 1997), solo aplican los L/N del 2039 al 2608, 2610 al 2654, 2656, 2659, 2662, 2663 y 2665.						-	-	-	-	-	-	-	-	-	-	-	-
96-18-04	4-oct-96	<b>GIRT RETAINING STRAPS - FAILURE OR INTERFERENCE OF OPENING OF THE FORWARD DOORS.</b> N/A por L/N en efectividad de la AD, solo aplican los L/N hasta el 2211.						-	-	-	-	-	-	-	-	-	-	-	-
96-25-17 SB 27-A1194 R2	3-feb-97	<b>TIE LINKS - HIGHER THAN NORMAL ELEVATOR CONTROL FORCES CAUSED BY DAMAGED TIE LINKS IN THE ELEVATOR CENTERING UNIT.</b> N/A por L/N en efectividad de la AD y el SB 27-A1194 R2 (JUN 19, 1997).						-	-	-	-	-	-	-	-	-	-	-	-
96-26-07		<b>REVISING THE FAA-APPROVED AIRPLANE FLIGHT MANUAL (AFM).</b> supercedida por la AD 2000-22-02 R1.						-	-	-	-	-	-	-	-	-	-	-	-
97-02-16 SB 27-A1199	4-mar-97	<b>PROXIMITY SWITCH - MODIFICATION OF THE SYSTEM THAT DETECTS A LOSS OF TENSION IN THE CABLE CONTROLLING THE FLAPS.</b> N/A por L/N en la efectividad de la AD, solo aplican los L/N del 1001 al 2765.						-	-	-	-	-	-	-	-	-	-	-	-









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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	
<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>CSO</b>	<b>CSO</b>
				N/A	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
98-25-06	-	DOORS - AFT CARGO - FRAME AND BEAM INSPECTION, REPAIR AND MODIFICATION. supercedida por AD 2012-12-12								-	-	-	-	-	-	-	-	-	-	-	-
99-01-16 SB 38-1043	12-feb-99	<b>WATER/WASTE - POTABLE WATER SYSTEM - PASSENGER WATER SYSTEM - INFLIGHT POTABLE WATER DRAINAGE CHANGE.</b>								-	-	-	-	-	-	-	-	-	-	-	-
		Placard Installation. Cumplida mediante EO 3811-01005	X		X					-	-	-	2401,53	3479	14-jun-98	-	-	-	-	-	-
		Lavatory Second Drain Valve Installation. Cumplida mediante EO 0025-01331.	X		X					-	-	-	8119,55	11886	14-may-01	-	-	-	-	-	-
99-03-04 SB 28-1131 SB 28-1157	9-mar-99	<b>FUEL SYSTEM WIRING.</b>								-	-	-	-	-	-	-	-	-	-	-	-
		Shielding and separation of the fuel system wiring. Cumplido mediante implementacion del STC ST00127BO (Boeing SB 737-28-1157), soportado en la EO 2841-01044.	X		X					-	-	-	11591,72	16916	15-feb-03	-	-	-	-	-	-
		Installation of flame arrestors and pressure relief valves. Cumplido mediante implementacion del SB 737-28-1131, soportado en la EO 2811-01017.	X		X					-	-	-	-	-	3-ago-00	-	-	-	-	-	-
99-04-10	18-mar-99	<b>PASSENGER CABIN FLUORESCENT LIGHT BALLASTS REPLACEMENT.</b> N/A por P/N no instalado, de acuerdo al IPC 33-50-50-01								-	-	-	-	-	-	-	-	-	-	-	-
99-05-12 SB 28-A1132 R1	-	CENTER FUEL TANK FLOAT SWITCH WIRING. supercedida por la AD 2004-15-04.								-	-	-	-	-	-	-	-	-	-	-	-
99-05-15	-	ELEVATOR TAB PUSH ROD ATTACHMENT NUTS INSPECTION AND REPLACEMENT. supercedida por la AD 2000-19-05.								-	-	-	-	-	-	-	-	-	-	-	-
99-08-16	-	AIRWORTHINESS LIMITATIONS SECTION OF ENGINE SHOP MANUAL REVISION. supercedida por la AD 2000-12-01.								-	-	-	-	-	-	-	-	-	-	-	-
99-08-23	-	REPETITIVE INSPECTIONS TO DETECT CRACKING IN THE WEB OF THE AFT PRESSURE BULKHEAD. supercedida por la AD 2012-18-13 R1.								-	-	-	-	-	-	-	-	-	-	-	-
99-10-12 SB 32-A1314 R1	27-may-99	<b>LANDING GEAR - MAIN GEAR AND DOORS - MAIN GEAR TRUNNION PIN, ACTUATOR BEAM ARM, AND ACTUATOR BOLT ASSEMBLIES REWORK.</b>								-	-	-	-	-	-	-	-	-	-	-	-
		Actuator Beam Arm LH y RH . Complied with E.W.O. 0174 by Cima and GCA WO 0518 / EW00196.		X	X					-	-	90	39954,5	57462	30-jul-21	-	-	28-oct-21	-	-	56



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<b>APPLICABILITY</b>		ENGINE LH		ENGINE RH	
<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>APU</b>	<b>CSO</b>
					N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
99-18-17	-	AIRPLANE BATTERY REPLACE - CONDITION TEST OF DIODE CR910 OF THE GCU. supercedida por la AD 99-24-08								-	-	-	-	-	-	-	-	-	-	-	-
99-19-27	20-oct-99	<b>TIE BOLTS OF MAIN LANDING GEAR (MLG) INSPECTION AND REPLACEMENT.</b> Cumplida mediante EO 3245-01015	X				X			-	-	-	3191,1	4669	18-nov-98	-	-	-	-	-	-
99-21-15	-	INSPECTION - FUEL BOOST PUMP WIRING. supercedida por la AD 2007-24-02								-	-	-	-	-	-	-	-	-	-	-	-
99-22-16 SB 52-1137 SB 52-1128	7-dic-99	<b>VISUAL INSPECTION OF THE UPPER DECOMPRESSION PANEL ON THE FLIGHT DECK DOOR, CORRECTIVE ACTION, IF NECESSARY.</b> Cumplida mediante EO 5251-01005	X				X			-	-	-	5010,9	7501	31-oct-99	-	-	-	-	-	-
99-23-20 SB 28-1115	27-dic-99	<b>FUEL - DISTRIBUTION - REPLACEMENT OF THE REFUEL VALUE ACTUATOR SUPPRESSION DIODES.</b> Cumplida mediante EO 2820-01004.	X				X			-	-	-	6432,4	9485	4-jul-00	-	-	-	-	-	-
99-23-22 R2	20-abr-00	<b>TRANSPONDERS MODE "C" TRANSPONDER SINGLE GILLHAM CODE ALTITUDE INPUT - REPETITIVE TEST.</b> N/A por P/N de ATC transponder en efectividad de la AD, de acuerdo a el listado "OCCM Component Status", la aeronave cuenta con Transponder modo "S".								-	-	-	-	-	-	-	-	-	-	-	-
99-24-06 SB 28-A1134 R1	3-ene-00	<b>FUEL TANK BOOST PUMPS - DISCREPANT BREATHER PLUGS.</b> N/A por V/N en efectividad del SB 28-A1134 R1 (Junio 10, 1999), no aplican V/N iniciados en PV. N/A debido a que la aeronave no cuenta con tanques auxiliares, según se indica en la AD.								-	-	-	-	-	-	-	-	-	-	-	-
99-24-08	align="center">16-sep-99	<b>ELECTRICALLY POWERED AIRPLANE SYSTEMS.</b> Cumplimiento inicial mediante EO 2400-01173.	X				X			-	-	-	3903,3	5714	26-mar-99	-	-	-	-	-	-
		<b>TEST CONDITION OF DIODE CR910 - GCU.</b> Ultimo cumplimiento mediante WO 0544 de GCA.						X	X	600	-	-	40044,2	57529	8-jun-21	40644,2	-	-	404,12	-	-
		<b>BATTERY REPLACE</b> Controlled in HT status								-	-	-	-	-	-	-	-	-	-	-	-
99-25-02 SB 27-A1213 R1	15-dic-99	<b>FLIGHT CONTROLS - FLIGHT SPOILER CONTROL SYSTEM - INSPECTION OF THE SPOILER MIXER FOR CORRECT INSTALLATION OF FASTENERS.</b> Cumplida mediante EO 2761-01008	X				X			-	-	-	5431,38	8035	4-ene-00	-	-	-	-	-	-





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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	
<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>CSO</b>	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING			
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS				
2001-08-24	10-may-01	<b>AIRPLANE FLIGHT MANUAL (AFM) - LIMITATIONS SECTION - FUEL PUMP OPERATION.</b> Cumplida mediante revisiones del AFM.					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	
2001-10-09	-	<b>INSPECT THE KC 225 AFCS TO DETERMINE THE COMPUTER MODIFICATIONS (MODS).</b> N/A por P/N de Automatic Flight Control System (AFCS) en efectividad de la AD, según IPC 22-11					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	
2001-10-14 SB 35-1076	8-jun-01	<b>OXYGEN - PASSENGER-PASSENGER OXYGEN SYSTEM- INSPECTION OF THE RELEASE PIN LOCATION IN THE OXYGEN GENERATOR FIRING PIN.</b> Cumplida con el cambio de los generadores de Oxigeno de acuerdo al AMM Task 35-22-11-202-083.					X	X														
2001-11-07 SB 29-A1078	16-jul-01	<b>HYDRAULIC POWER - DC MOTOR OPERATED SHUTOFF VALVE (SENSITIVE LOCATIONS) REPLACEMENT.</b> Cumplida accion terminante bajo EO 2915-01012.	X				X					9738,92	14264	15-mar-02	-	-	-	-	-	-	-	
2001-11-11 SB 25-1371 R3	16-jul-01	<b>EQUIPMENT/FURNISHINGS - PASSENGER COMPARTMENT - PASSENGER SEATS - AVIO INTERIORS SEAT TRACK FITTING INSPECTION.</b> N/A por V/N en efectividad del SB 25-1371 R3 (FEB 27, 2003)																				
2001-13-07 SB 28-1164	1-ago-01	<b>FUEL - ENGINE FUEL FEED SYSTEM.</b> Cumplida mediante EO 2822-01080	X				X						7108,73	10454	1-nov-00	-	-	-	-	-	-	
2001-14-06	-	<b>FUEL - ENGINE FUEL FEED SYSTEM - ENGINE FUEL SHUTOFF VALVE - WIRING CHANGES IN THE MLG MAIN WHEEL WELL.</b> supercedida por la AD 2005-10-11.																				
2001-15-01 SB 25-1405	28-ago-01	<b>EQUIPMENT/FURNISHINGS - EMERGENCY - DOOR-MOUNTED ESCAPE SYSTEM - LATCH ASSEMBLY MODIFICATION.</b> Cumplido mediante EO 2560-01044.	X				X						11245,82	16429	16-dic-02	-	-	-	-	-	-	
2001-20-10	20-nov-01	<b>MAIN WHEEL WELL JUNCTION BOX WIRE CHAFING.</b> Cumplida mediante EO 2400-01223	X				X						9738,22	14264	14-mar-02	-	-	-	-	-	-	
2001-22-13 SB 27-A1214 R1	6-dic-01	<b>FLIGHT CONTROLS - RUDDER AND TAB - RUDDER PEDAL PUSHROD FASTENER REPLACEMENT.</b> Cumplida mediante EO 2721-01042, soportada en la pagina 430944 del FLIGHT MAINTENANCE LOGBOOK de ASIANA AIRLINES.	X				X						6274,67	9263	5-jun-00	-	-	-	-	-	-	

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<b>APPLICABILITY</b>		ENGINE LH	ENGINE RH	APU	
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>CSO</b>	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE		STATE		INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS
2001-24-02	-	EQUIPMENT/FURNISHINGS - FLIGHT/PASSENGER COMPARTMENT - OBSERVER/ATTENDANT SEAT - CHANGE ATTACHMENT OF SHOULDER RESTRAINT SYSTEM. supercedida por la AD 2006-26-13					-	-	-	-	-	-	-	-	-	-	-	-
2002-01-01	-	STABILIZERS - ELEVATOR - TAB ASSEMBLY - INSPECTION OF ELEVATOR TAB HINGE AND ELEVATOR TAB CONTROL MECHANISM. supercedida por la AD 2006-12-23					-	-	-	-	-	-	-	-	-	-	-	-
2002-02-08 SB 57-A1260 R2	19-mar-02	<b>WINGS - MAIN LANDING GEAR SUPPORT BEAM RETAINING PIN - BOLT AND SELF-LOCKING NUT REPLACEMENT.</b> N/A por modelo de aeronave en efectividad de la AD y del SB 57-A1260 R2 (OCT 18, 2001), solo aplican las series 737-200/300/500.					-	-	-	-	-	-	-	-	-	-	-	-
2002-05-07	-	WINGS - MAIN LANDING GEAR SUPPORT BEAM RETAINING PIN - BOLT AND SELF-LOCKING NUT REPLACEMENT. supercedida por la AD 2013-09-02.					-	-	-	-	-	-	-	-	-	-	-	-
2002-07-08	17-may-02	FUSELAGE - SKIN - REPAIR METHODS FOR CRACKS IN LOWER SKINS AT LAP JOINTS AND WINDOW CORNERS. supercedida por las AD's 2013-18-058, 2013-09-01, 2015-21-06.					-	-	-	-	-	-	-	-	-	-	-	-
2002-07-10	17-may-02	FUSELAGE - SKIN - REPAIR METHODS FOR CRACKS IN LOWER SKINS AT LAP JOINTS AND WINDOW CORNERS. N/A por L/N en aplicabilidad de la AD, solo aplican los L/N del 292 al 2565.					-	-	-	-	-	-	-	-	-	-	-	-
2002-10-11 SB 53A-1075 R3	27-jun-02	CRACK & CORR' OF PRESS' BULKHEAD BS 1016 & FOLLOW. supercedida por la AD 2014-18-02.					-	-	-	-	-	-	-	-	-	-	-	-
2002-10-12 SB 27-1223 R0	27-jun-02	<b>FLIGHT CONTROLS - AILERON &amp; AILERON TRIM CONTROL SYSTEM, AILERON TAB CONTROL ROD BARREL INSPECTION OR REPLACEMENT.</b> Cumplida accion terminante mediante EO 2710-01004.	X			X	-	-	-	9738,92	14264	12-mar-02	-	-	-	-	-	-
2002-18-02 SB 27-A1238	10-oct-02	<b>FLIGHT CONTROLS - TRAILING EDGE FLAP AND HORIZONTAL STABILIZER TRIM SYSTEMS S245 FLAP LIMIT SWITCH INSPECTION AND R850 RELAY INSTALLATION.</b> Cumplida accion terminante mediante EO 2711-01025	X			X	-	-	-	10442,9	15285	28-jul-02	-	-	-	-	-	-











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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	
<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>APU</b>	<b>CSO</b>
				N/A	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
2004-23-07	20-dic-04	FUSELAGE - GALLEY NO. 2 UPPER ATTACHMENT SUPPORT - BODY STATIONS 344 TO 360 BETWEEN RIGHT STRINGERS 3 AND 7. N/A por capacidad y peso del Galley 2 en aplicabilidad de la AD.					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
2005-03-02 SB 21-1131 R2	9-mar-05	AIR CONDITIONING - DISTRIBUTION - PASSENGER CABIN CONDITIONED AIR DISTRIBUTION SYSTEM - OVERHEAD DUCT LANYARD INSTALLATION. N/A por V/N en efectividad del SB 21-1131 R2 (ABR 18, 2002), solo aplican los V/N PV 231 - PV237.					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
2005-04-01 SB 28-A1174 R1	23-mar-05	FUEL - ENGINE FUEL FEED SYSTEM, ADDITION OF BONDING BETWEEN ENGINE FUEL FEED LINE FITTING AND WING FRONT SPAR. Cumplida mediante EO 2822-01101.	X				X	-	-	-	16822,03	24474	2-sep-05	-	-	-	-	-	-	-	-
2005-05-17	18-abr-05	MODIFICATION OF THE SAFE SIDE HARNESS CONECTORS AT TANK PENETRATIONS						-	-	-	-	-	-	-	-	-	-	-	-	-	-
		Tanque Central - Cumplida mediante MEO SB 737-300766-28-2.000.06 de ASIANA AIRLINES	X				X	-	-	-	18179,26	26447	13-abr-06	-	-	-	-	-	-	-	-
		Tanque RH - Cumplido mediante MEO SB 737-300766-28-2.002.06 de ASIANA AIRLINES.	X				X	-	-	-	18194,48	26468	15-abr-06	-	-	-	-	-	-	-	-
		Tanque LH - Cumplido mediante hoja de libro de vuelo 594239						-	-	-	17897,1	26031	28-feb-06	-	-	-	-	-	-	-	-
2005-07-12	12-may-05	FUSELAGE - FRAMES AND BULKHEADS - STA 277 TO STA 328 FRAME INSPECTIONS, STRINGER 7R TO 17R - FORWARD GALLEY DOOR CUTOUT. supercedida por la AD 2014-22-06						-	-	-	-	-	-	-	-	-	-	-	-	-	-
2005-07-19 SB 53-A1228	12-may-05	FUSELAGE - MAIN CARGO DOOR CUTOUT, FORWARD CARGO DOOR CUTOUT, AFT CARGO DOOR CUTOUT - SKIN, BEARSTRAP AND FRAME INSPECTIONS.	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
		Inspeccion inicial - Cumplida mediante EO 5313-01033 de ASIANA AIRLINES.	X				X	-	-	-	14602	21326	7-ago-04	-	-	-	-	-	-	-	-
		FORWARD CARGO DOORWAY FWD FRAME INSPECTION. Cumplida mediante documento TASK NO. FAA 2005-07-19.000.04 de ASIANA AIRLINES.					X	X	-	4000	-	37470,53	54583	7-mar-15	-	58583	-	-	-	920	-
		FORWARD CARGO DOORWAY AFT FRAME INSPECTION - MEO SB 737-53A1228.101.00					X	X	-	4000	-	32173,96	46947	9-may-12	-	66947	-	-	-	9284	-
		AFT CARGO DOORWAY AFT FRAME INSPECTION. Cumplido mediante documento MEO SB 737-53-A1228.103.00 de ASIANA AIRLINES					X	X	-	4000	-	31121,45	45418	19-nov-11	-	65418	-	-	-	7755	-





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<b>APPLICABILITY</b>		ENGINE LH		ENGINE RH	
<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
2006-07-12	-	FUSELAGE - SKIN - INSPECTIONS OF THE FUSELAGE SKIN FOR SCRIBE LINES. supercedida por la AD 2010-05-13								-	-	-	-	-	-	-	-	-	-	-	-
2006-12-06 SB 25-1438 R4	12-jul-06	<b>EQUIPMENT/FURNISHINGS - BUFFET/GALLEY - GALLEY WIRE BUNDLE PROTECTIVE WRAP REPLACEMENT.</b> N/A por V/N en efectividad del SB 25-1438 R4 (JUN 18, 2009), solo aplican los V/N PV271-PV-272.								-	-	-	-	-	-	-	-	-	-	-	-
2006-12-23 SB 55A-1070 R4	align="center">3/07/2006	<b>STABILIZERS - ELEVATOR - TAB ASSEMBLY - INSPECTION OF ELEVATOR TAB HINGE AND ELEVATOR TAB CONTROL MECHANISM</b> Configuration, free-play, damage or discrepancy inspection. Completada accion inicial establecida en la AD 2002-01-01 mediante EO 5521-01002 (ABR 6, 2000). Accion inicial complementaria de la AD 2006-12-13 cumplida mediante documento MEO SB 737-55A1070.000.00.	X							-	-	-	19691,37	28665	2-dic-06	-	-	-	-	-	-
		Inspection of Elevator Tab Hinge and Free play. Ultimo cumplimiento mediante WO 18027-057057 de CEMA.								2000	1500	-	38937,2	56679	21-jun-18	40937,2	58179	-	697,12	516	-
		Inspection of Elevator Tab Control Mechanism. Ultimo cumplimiento mediante WO 18027-057057 de CEMA.								6000	4500	-	38937,2	56679	21-jun-18	44937,2	61179	-	4697,12	3516	-
2006-13-13	7-jul-06	<b>AFM REVISION - IMPROVED PROCEDURES FOR PRE-FLIGHT SETUP OF THE CABIN PRESSURIZATION SYSTEM AND FOR INTERPRETING AND RESPONDING TO THE CABIN ALTITUDE/CONFIGURATION WARNING HORN.</b> Cumplida mediante incorporacion por revision al AFM								-	-	-	-	-	-	-	-	-	-	-	-
2006-15-05 SB 28-A1192 R2	24-ago-06	<b>FUEL - PRESSURE FUELING SYSTEM - AUXILIARY TANKS - IMPROVED FUELING FLOAT SWITCH AND WIRING PROVISION INSTALLATION.</b> N/A por V/N en efectividad el SB 28-A1192 R2 (ABR 27, 2006), no aplican V/N iniciados en PV.								-	-	-	-	-	-	-	-	-	-	-	-
2006-15-18 SB 33-1132	5-sep-06	<b>LIGHTS - MASTER DIM AND TEST - SUPPLY TEST GROUND SIGNAL WIRING SEPARATION.</b> Cumplida mediante EO 3315-01003	X							-	-	-	12734,32	18593	18-sep-03	-	-	-	-	-	-
2006-16-18 R1	18-jul-08	<b>INSTALLING A WARNING PLACARD ON THE TAWS/RMI.</b> N/A aplica por equipo TAWS/RMI en la efectividad de la AD, esta aeronave no cuenta con TAWS/RMI.								-	-	-	-	-	-	-	-	-	-	-	-

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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	<b>APU</b>
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<b>TSN</b>	40240,1	<b>CSN</b>	57663
BOEING				<b>TSO</b>	N/A
				<b>CSO</b>	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS
2006-18-11	27-sep-06	<b>AUXILIARY FUEL SYSTEM - ONE-TIME DEACTIVATION - REPETITIVE VENTING OF THE AUXILIARY FUEL TANKS.</b> N/A por MSN en efectividad de la AD.					-	-	-	-	-	-	-	-	-	-	-	-
2006-21-01 SB 27-1272	14-nov-06	<b>FLIGHT CONTROLS - AILERON AND AILERON TRIM CONTROL SYSTEM - AILERON BALANCE TAB FREEPLAY CHECK AND LUBRICATION</b>					-	-	-	-	-	-	-	-	-	-	-	-
		Cumplida Inspeccion inicial PARTE I (Aileron Balance Tab Freeplay) mediante MEO SB 737-27-1272.000.00.	X		X		-	-	-	20897,2	30432	31-may-07	-	-	-	-	-	-
		Lubrication - Aileron Balance Tab/Rod End (LO QUE OCURRA PRIMERO). Cumplida IAW WO 0617 de GCA		X	X		4000	-	360	40238,2	57661	31-jul-21	44238,2	-	26-jul-22	3998,12	-	327
		Aileron Balance Tab Freeplay Inspection. Ultimo cumplimiento mediante WO 0158 de GCA. (LO QUE OCURRA PRIMERO)		X	X		8000	-	720	39230,42	56932	3-mar-20	47230,42	-	21-feb-22	6990,33	-	172
2006-26-09 SB 53-1216 R1	8/02/2007	<b>FUSELAGE - MAIN FRAME - AIR-CONDITIONING BRACKET ATTACHMENTS, STATION 360 THROUGH STATION 907 AT STRINGER 8L AND STRINGER 8R - INSPECTION, PREVENTIVE MODIFICATION, AND REPAIR.</b>					-	-	-	-	-	-	-	-	-	-	-	-
		Modification and Repair - Cumplida accion terminante mediante MEO SB 737-53-1216.000.00.	X		X		-	-	-	20897,28	30432	18-jun-07	-	-	-	-	-	-
		Post-modificacion Inspection. cumplida mediante MEO SB 737-53-1216 (AGO 19, 2014)	X		X		-	-	-	36414,36	53162	19-ago-14	-	-	-	-	-	-
2006-26-13 SB 25-1412 R3	12-feb-07	<b>EQUIPMENT/FURNISHINGS - FLIGHT/PASSENGER COMPARTMENT - OBSERVER/ATTENDANT SEAT - CHANGE ATTACHMENT OF SHOULDER RESTRAINT SYSTEM.</b> N/A por V/N en efectividad del SB 25-1412 R3 (DIC 2, 2004), solo aplican los V/N PV231-PV237.					-	-	-	-	-	-	-	-	-	-	-	-
2007-03-03 SB 53-A1225	7-mar-07	<b>FUSELAGE - FRAMES AND BULKHEADS 178 PRESSURE BULKHEAD.</b> Inspeccion inicial cumplida mediante la EO 5311-01043 (MAR 16, 2002). Accion terminante cumplida mediante MEO SB 737-53A1225.000.01.	X		X		-	-	-	37470,53	54583	26-mar-15	-	-	-	-	-	-



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<b>APPLICABILITY</b>		ENGINE LH		ENGINE RH	
<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
2007-03-07 SB 27-1255	16-feb-07	<b>FLIGHT CONTROLS - RUDDER AND RUDDER TRIM CONTROL SYSTEM - INSTALLATION OF AN ENHANCED RUDDER CONTROL SYSTEM (RSEP).</b> Cumplida mediante MEO SB 737-27-1255.002.03	X		X		-	-	-	20897,2	30432	16-jun-07	-	-	-	-	-	-			
2007-07-02 SB 25-1548	2-may-07	<b>EQUIPMENT AND FURNISHINGS - PASSENGER COMPARTMENT - MODIFICATION OF THE OXYGEN ACTIVATION MECHANISM WITHIN THE PASSENGER SERVICE UNITS.</b> Cumplido mediante MEO SB 737-25-1548.000.00	X		X		-	-	-	28424,57	41620	4-oct-10	-	-	-	-	-	-			
2007-10-12	27/06/2007	<b>REINFORCED FLIGHT DECK DOOR MODIFICATION AND INSPECTION</b>	X		X		-	-	-	-	-	-	-	-	-	-	-	-			
		Inspeccion Visual al cableado cumplida mediante MEO SB B221001-52A02.000.00 Chequeo Funcional. Ultimo Cumplimiento mediante WO 18027-051167			X	X		3000	-	-	38937,2	56679	21-jun-18	41937,2	-	-	-	3000,00	-	-	
2007-11-07	-	<b>FUEL TANK BOOST PUMP - INSPECTION OF THE WIRING IN THE FUEL TANKS.</b> supercedida por la AD 2007-24-02					-	-	-	-	-	-	-	-	-	-	-	-	-		
2007-16-05 SB 27-A1266	13-sep-07	<b>FLIGHT CONTROLS - ELEVATOR AND TAB CONTROL SYSTEM - INSPECTION OF THE TAB CONTROL RODS AND CONTROL ROD JAM NUTS.</b> Inspección inicial cumplida mediante MEO SB 737-27A1266.000.01 (MAR 19, 2008). Ultima inspección cumplida mediante TASK NO 2007-16-05.000.01. (LO QUE OCURRA PRIMERO)		X	X		6000	4500	-	37470,53	54583	6-mar-15	43470,53	59083	-	3230,45	1420	-			
2007-18-03 SB 25-A1363 R2	12-sep-07	<b>EQUIPMENT/FURNISHINGS - FLIGHT COMPARTMENT - CAPTAIN AND FIRST OFFICER CREW SEATS - INSPECTION AND INSTALLATION OF THE REAR TRACK LOCK BRACKET THROUGH BOLTS.</b> Cumplida mediante EO 2511-01022	X		X		-	-	-	10000,1	14642	7-may-02	-	-	-	-	-	-			
2007-24-02	6/12/2007	<b>FUEL - ENGINE FUEL FEED SYSTEM - FUEL BOOST PUMP ELECTRICAL WIRE AND SLEEVE IN FUEL TANK CONDUIT - INSPECTION / REPLACEMENT.</b> Inspeccion inicial cumplida mediante MEO SB 737-28A1263.000.01 (JUN 18, 2007). Ultima inspección cumplida mediante TASK NO 2007-24-02.000.00.		X	X		15000	-	-	31277,1	49747	9-may-12	46277,1	-	-	6037,02	-	-			







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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	
<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>APU</b>	<b>CSO</b>
				N/A	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
2008-22-01	26-nov-08	<b>DEACTIVATION OF PATS AIRCRAFT, LLC, AUXILIARY FUEL TANKS.</b> N/A por STC en la aplicabilidad de la AD, solo aplican las aeronaves 737-400 que tienen incorporado los STC SA553NE / SA725NE / SA725NE-D (Tanques auxiliares de combustible).					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
2008-23-07	25-nov-08	<b>REVISING THE AIRPLANE FLIGHT MANUAL TO INCLUDE A NEW FLIGHTCREW BRIEFING TO PREVENT FAILURE OF THE FLIGHTCREW TO RECOGNIZE AND REACT PROPERLY TO A VALID CABIN ALTITUDE WARNING HORN.</b> Cumplida mediante revisión al AFM.					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
2008-23-09 SB 25-1572	15-dic-08	<b>EQUIPMENT/FURNISHINGS - FLIGHT COMPARTMENT/PASSENGER COMPARTMENT/CARGO COMPARTMENT/MAIN EQUIPMENT CENTER - INSULATION BLANKET REPLACEMENT.</b> N/A por S/N en efectividad de la AD. N/A por V/N en la aplicabilidad del SB 25-1572 (ABR 30, 2008), solo aplican los V/N PV001-PV010.					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
2008-24-05 SB 30-A1059 R2	30-dic-08	<b>ICE AND RAIN PROTECTION - WINDSHIELD WIPER SYSTEM - CHANGE THE WINDSHIELD WIPER MOTOR.</b> N/A por V/N en efectividad del SB 30-A1059 R2 (OCT 31, 2011). No aplican V/N iniciados en PV.					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
2009-02-06 R1	23-sep-09	<b>FUSELAGE - FRAMES AND BULKHEADS - FORWARD CARGO BAY STRINGER 19 TO STRINGER 22 - FRAMES AND FRAME REINFORCEMENTS - INSPECTION, REPAIR AND PREVENTATIVE MODIFICATION AT WIRE PENETRATION HOLE, 0.50 INCH DIAMETER HOLE, STANDOFF/TOOLING HOLES AND PRODUCTION FASTENER HOLES.</b> supercedida por la AD 2011-23-05					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
2009-04-15 SB 53-1268	2-abr-09	<b>FUSELAGE - STRINGER TIE CLIPS - STA 360 TO STA 907, S-10L TO S-10R, INSPECTION AND REPLACEMENT.</b> Inspección inicial cumplida mediante MEO SB 737-53-1268.000.00.	X		X		-	-	-	28424,4	41620	5-oct-10	-	-	-	-	-	-	-	-	-
		Internal Eddy Current and Detailed Inspection of stringer tie clips. Ultimo cumplimiento mediante MEO SB 737-53-1268.000.00.		X		X		-	20000	-	28424,4	41620	5-oct-10	-	61620	-	-	-	3957	-	-

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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>APU</b>	<b>CSO</b>
					N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
2009-07-11 SB 53A1266	14-may-09	<b>FUSELAGE - SKIN - CROWN SKIN INSPECTION AND REPAIR (BS857 TO BS1006) AND DORSAL FIN NOSE CAP MODIFICATION (BS 860 TO BS 887).</b> Cumplida accion terminante mediante implementación del SB 53-A1266, mediante MEO SB 737-53A1266.000.00. Reverificado mediante WO 18027-057062 de CEMA.	X		X		-	-	-	24689,86	36011	24-ene-09	-	-	-	-	-	-			
2009-09-06 SB 53-A1269 R1	11-jun-09	<b>FUSELAGE - STRUCTURE AROUND OPENINGS - FORWARD AIRSTAIR DOOR CUTOUT - UPPER SILL WEB AND INTERCOSTALS INSPECTION AND MODIFICATION.</b> N/A por V/N en aplicabilidad del SB 53-A1269 R1 (OCT 26, 2009), solo aplica los V/N PV296, PV301 - PV302, PV356 - PV357.					-	-	-	-	-	-	-	-	-	-	-				
2009-09-07 SB 32-A1355	11-jun-09	<b>LANDING GEAR - MAIN GEAR EXTENSION AND RETRACTION - RETRACT ACTUATOR BEAM INSPECTION AND REWORK.</b> Cumplida durante el último Overhaul del MLG LH realizado por SUNCOAST, mediante WO 2783. Cumplida durante el último Overhaul del MLG RH realizado por SUNCOAST, mediante WO 2938.		X	X		-	21000	120	-	0	15-mar-18	-	21000	15-mar-28	-	20016	2386			
2009-09-07 SB 32-A1355	11-jun-09	<b>LANDING GEAR - MAIN GEAR EXTENSION AND RETRACTION - RETRACT ACTUATOR BEAM INSPECTION AND REWORK.</b> Cumplida durante el último Overhaul del MLG RH realizado por SUNCOAST, mediante WO 2938.		X	X		-	21000	120	-	0	7-may-18	-	21000	07-may-28	-	20016	2439			
2009-12-05 SB 21-A1156 R1	16-jul-09	<b>AIR CONDITIONING- EQUIPMENT COOLING SYSTEM - ELECTRONIC FLIGHT INSTRUMENT SYSTEM COOLING SUPPLY OFF LIGHT WIRING CHANGE.</b> Cumplido mediante MEO SB 737-21A1156.000.01.	X		X		-	-	-	24690,9	36011	25-ene-09	-	-	-	-	-				
2009-12-06 SB 24-1145 R1	16-jul-09	<b>ELECTRICAL POWER - ELECTRICAL LOAD DISTRIBUTION - INSTALLATION OF THE INFLIGHT ENTERTAINMENT SYSTEM CIRCUIT BREAKER, RELAYS AND WIRING.</b> N/A por V/N en efectividad del SB 24-1145 R1 (ENE 19, 2015), no aplican V/Ns iniciados en PV.					-	-	-	-	-	-	-	-	-	-	-				
2009-14-04 SB 28-1241	12-ago-09	<b>FUEL - ENGINE FUEL SHUTOFF VALVE - WIRE BUNDLE REPLACEMENT W802 AND W804 AND INSPECTION OF GROMMETS AND FAIRLEADS IN THE ELECTRICAL JUNCTION BOXES IN THE MAIN WHEEL WELL.</b> Cumplido mediante MEO SB 737-28-1241.000.00.	X		X		-	-	-	20897,28	30432	7-jun-07	-	-	-	-	-				



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<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>CSO</b>	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE	INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O		FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS
2009-26-03 SB 24-A1166 R4	1-feb-10	<b>ELECTRICAL POWER - STANDBY POWER GENERATION - REMOVAL AND REPLACEMENT OF THE STATIC INVERTER MODULE.</b> N/A por V/N en efectividad del SB 24-A1166 R4 (MAY 21, 2009), solo aplican los V/N PV231-PV237.																	
2010-01-09	-	FUSELAGE - SKIN - CRACKS AT CHEM-MILLED STEPS IN CROWN SKINS AT SPECIFIED LOCATIONS - INSPECTION. supercedida por la AD 2011-24-12																	
2010-05-13 SB 53-A1262 R3	13-abr-10	<b>FUSELAGE - SKIN - INSPECTIONS OF THE FUSELAGE SKIN FOR SCRIBE LINES.</b>  Inspección de la Zona 1 cumplida mediante EO 5331-01049. Inspección de la Zona 2 cumplida mediante EO 5331-01050. Inspección de la zona 3 cumplida mediante EO 5331-01051. Acciones Futuras No Requeridas.	X		X						16835,15	24479	6-sep-05						
		Inspección de las zonas 4 y 5 cumplida mediante WO 18027-051178 de CEMA.	X		X						38937,2	56679	4-abr-18						
2010-14-06 SB 35-1099 R1	5-ago-10	<b>OXYGEN - PASSENGER OXYGEN SYSTEM - OXYGEN MASK ASSEMBLY INSPECTION AND MODIFICATION.</b> Cumplida mediante implementación del SB 35-1099 R1 (ABR 23, 2009) mediante MEO SB 737-35-1099.002.01.	X		X						28424,57	41620	4-oct-10						
2010-15-08	31-ago-10	CARRIAGE SPINDLE OF THE OUTBOARD MID-FLAPS - REPETITIVE INSPECTIONS, GAP CHECKS, OVERHAUL, REPLACEMENT. supercedida por la AD 2014-05-12.																	
2010-16-06 SB 35-A1053 R1	9-sep-10	<b>OXYGEN - CREW OXYGEN - CREW OXYGEN SYSTEM LOW PRESSURE FLEX HOSE REPLACEMENT.</b> Cumplido mediante EO 3510-01012.	X		X						6274,67	9253	4-jun-00						
2010-25-06 SB 53-A1254 R2	1-feb-11	<b>FUSELAGE - FRAME AND OVERWING STUB BEAM AT BODY STATIONS 616, 639 AND 597 OR 601 - BUTTOCK LINE 45.5 FLOOR BEAM WEB AT BODY STATION 639 - INSPECTION AND REPAIR.</b> Supercedida por la AD 2018-19-20																	
2011-01-09	9-feb-11	<b>PBE UNITS REPLACEMENT.</b> N/A por S/N de PBE en efectividad de la AD, de acuerdo a la tabla "Hard Time Component Status", solo aplican los PBE de P/N 119003-11 con S/N entre 003-50730M y 003-51329M.																	



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<b>APPLICABILITY</b>		ENGINE LH	ENGINE RH	APU	
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>CSO</b>	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
2011-01-12 SB 54-A1044 R2	9-feb-11	<b>NACELLES/PYLONS - NACELLE STRUT - MIDSPAR FUSE PIN - INSPECTION, REPAIR AND REPLACEMENT.</b> Acción terminante realizada en el cumplimiento de la AD 2008-21-03, de acuerdo al parrafo (i) de esta AD.					-	-	-	-	-	-	-	-	-	-	-	-	-	-	
2011-03-14 SB 31-A1325	14-mar-11	<b>INDICATING / RECORDING SYSTEMS - AURAL WARNING SYSTEM - INSTALLATION OF THE CABIN ALTITUDE WARNING SYSTEM AND TAKEOFF CONFIGURATION WARNING SYSTEM LIGHTS.</b> Cumplido mediante MEO SB 737-31A1325.000.00	X				-	-	-	28424,57	41620	4-oct-10	-	-	-	-	-	-	-		
2011-04-09	14-mar-11	<b>OXYGEN GENERATORS - SUPPLEMENTAL OXYGEN SYSTEM INSTALLATION.</b> supercedida por la AD 2012-11-09					-	-	-	-	-	-	-	-	-	-	-	-	-		
2011-04-10 SB 57-A1304 R1	25-mar-11	<b>AILERON BALANCE PANEL HINGE SUPPORT MODIFICATION - WING OUTBOARD REAR SPAR.</b> Inspeccion cumplida mediante MEO SB 737-57A1304.001.00, verificacion pag libro de vuelo 777783, N/A de acuerdo a Tabla 1 del SB 57-A1319 R2 (MAR 16, 2015).	X				-	-	-	23406,32	34162	1-jul-08	-	-	-	-	-	-			
2011-08-51	upon receipt	<b>FUSELAGE - SKIN - CROWN SKIN LAP JOINT - INSPECTION FOR LOWER ROW CRACKS AT STRINGER S-4, BETWEEN BODY STATION 360 AND BODY STATION 1016.</b> supercedida por la AD 2015-16-08					-	-	-	-	-	-	-	-	-	-	-	-			
2011-12-01	1-ago-11	<b>KOITO INDUSTRIES, LTD., SEATS AND SEATING SYSTEMS REMOTION</b> N/A por P/N de silla y/o conjunto de silla instalado en aplicabilidad de la AD, de acuerdo al IPC 25-22.					-	-	-	-	-	-	-	-	-	-	-	-			
2011-12-09 SB 28-A1212 R1	15-jul-11	<b>FUEL - ENGINE FUEL FEED SYSTEM - FUEL PUMP CONTROL GROUND FAULT INTERRUPTER (GFI) RELAY INSTALLATION</b>  Part 1: Wiring Changes, Relay Replacements, and Certain Bonding Resistance Measurements for Certain Airplanes. Cumplida mediante MEO SB 737-28A1212.000.01.	X				-	-	-	28424,57	41620	4-oct-10	-	-	-	-	-	-			

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<b>APPLICABILITY</b>		ENGINE LH		ENGINE RH	
<b>MANUFACTURER</b>	☑	TSN	CSN	TSO	CSO
BOEING		40240,1	57663	N/A	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING			
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS				
		Maintenance Program Revisions. Cumplida con la adición del documento D6-38278-CMR, Revisión Agosto de 2010, al MPD al momento de dar cumplimiento a la AD 2008-10-09 R1.					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	
2011-21-08	23-nov-11	<b>CRACKS ON CERTAIN SICMA AERO SEAT PASSENGER SEAT ASSEMBLIES - CORRECTIVE ACTIONS.</b> N/A por P/N de "backrest link" de las sillas instalado de acuerdo a aplicabilidad de la AD, según IPC 25-22.					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	
2011-23-05	16-nov-11	FUSELAGE - FRAMES AND BULKHEADS - FORWARD CARGO BAY STRINGER 19 TO STRINGER 22 - FRAMES AND FRAME REINFORCEMENTS - INSPECTION, REPAIR AND PREVENTATIVE MODIFICATION AT WIRE PENETRATION HOLE, 0.50 INCH DIAMETER HOLE, STANDOFF/TOOLING HOLES AND PRODUCTION FASTENER HOLES. supercedida por la AD 2016-11-04.					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	
2011-24-12 SB 53-A1301 R1	5-ene-12	<b>FUSELAGE - SKIN - CRACKS AT CHEM-MILLED STEPS IN CROWN SKINS AT SPECIFIED LOCATIONS - INSPECTION.</b> Ultimo cumplimiento mediante WO 18027-051183 de CEMA.		X	X		-	500	-	40238,2	57661	23-jul-21	-	58161	-	-	498	-	-	-	-	
2011-27-03 SB 27-A1278 R1	10-feb-12	<b>FLIGHT CONTROLS - HORIZONTAL STABILIZER TRIM CONTROL SYSTEM - BALLSCREW AND BALL NUT INSPECTION AND LUBRICATION</b>					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	
		WP 1. INSPECTION. Inspección inicial cumplida mediante MEO SB 737-27A1278.000.00 (ABR 3, 2008). Ultimo cumplimiento mediante WO 0260 de GCA. (LO QUE OCURRA PRIMERO).		X	X		4000	-	720	39273,3	56968	2-oct-20	43273,3	-	22-sep-22	3033,22	-	385	-	-	-	-
		WP 1. LUBRICATION. Inspección inicial cumplida mediante MEO SB 737-27A1278.001.00 (ABR 3, 2008). Ultimo cumplimiento mediante WO 0618 de GCA. (LO QUE OCURRA PRIMERO)		X	X		2000	-	360	40238,2	57661	30-jul-21	42238,2	-	25-jul-22	1998,12	-	326	-	-	-	-
		WP 5. OVERHAUL . Cumplimiento inicial mediante TRP card GTL 28851.		X	X		24000	-	-	24384	-	14-may-08	48384	-	-	7276,80	-	-	-	-	-	-

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<b>APPLICABILITY</b>		ENGINE LH	ENGINE RH	APU	
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>	<b>CSO</b>
BOEING		40240,1	57663	N/A	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
2012-08-17	31-may-12	<p><b>REVISING THE MAINTENANCE PROGRAM TO INCORPORATE CERTAIN LIMITATIONS ON ANALOG TRANSIENT SUPPRESSION DEVICES INSTALLED ACCORDING TO STC ST00146BO.</b>                      N/A por STC instalado de acuerdo a efectividad de la AD, según tabla "STC STATUS", solo aplican aeronaves con el STC ST00146BO ( analog transient suppression devices ) incorporado.</p>								-	-	-	-	-	-	-	-	-	-	-	-
2012-11-09 R1	10-ago-12	<p><b>OXYGEN GENERATORS - SUPPLEMENTAL OXYGEN SYSTEM INSTALLATION.</b>                      Cumplida mediante MEO SB 737-35-1147.000.04.</p>	X		X					-	-	-	37801,5	54980	18-jun-15	-	-	-	-	-	-
2012-12-04	23-jul-12	<p>FUSELAGE - SKIN - STRINGER S-1 AND STRINGER S-2R, STATION 400 TO STATION 460 INSPECTION, REPAIR AND MODIFICATION. supercedida por la AD 2016-16-13.</p>								-	-	-	-	-	-	-	-	-	-	-	-
2012-12-05	1-nov-05	<p>FUSELAGE - MAIN FRAME - FRAMES AND BULKHEADS - BODY STATION BS 360 - BS 348.2 AND BS 303.9 - BS 294.5 STRINGERS S-7L - S-16L LEFT INTERCOSTAL INSPECTION / REPAIR / MODIFICATION. supercedida por la AD 2018-04-06</p>								-	-	-	-	-	-	-	-	-	-	-	-
2012-13-07 SB 57-A1319		<p><b>WINGS - TRAILING EDGE FLAPS - OUTBOARD FLAP, CARRIAGE SPINDLE - INSPECTION FOR CORROSION.</b>                      N/A por P/N de Flap Carriage instalado, de acuerdo a Tabla 1 del SB 57-A1319 R2 (MAR 16, 2015). Verificado mediante MEO SB 737-57A1304.001.00.</p>								-	-	-	-	-	-	-	-	-	-	-	-
2012-18-13 R1		<p><b>FUSELAGE - BODY STATION 1016 AFT PRESSURE BULKHEAD WEB CRACKS - INSPECTION AND REPAIR</b></p>								-	-	-	-	-	-	-	-	-	-	-	-
		<p>Inspections of the Web at the "Y" Chord Upper Bulkhead From S-15L to S-15R. Ultimo cumplimiento mediante WO 18027-057060 de CEMA.</p>		X	X					-	6000	-	38937,2	56679	4-abr-18	-	62679	-	-	5016	-
		<p>Inspections of the Web at the "Y" Chord Upper Bulkhead From FROM S-5L TO S-7L and S-5R TO S-9R. Ultimo cumplimiento mediante WO 18027-057060 de CEMA.</p>		X	X					-	1200	-	38937,2	56679	4-abr-18	-	57879	-	-	216	-
		<p>Inspections of the Web at the "Y" Chord in the Lower Bulkhead From S-15L to S15R. Ultimo cumplimiento mediante WO 18027-057060 de CEMA.</p>		X	X					-	6000	-	38937,2	56679	4-abr-18	-	62679	-	-	5016	-

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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	
<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>CSO</b>	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
SB 53-A1214	17-jul-12	One-Time Inspection Under the Tear Strap. Ultimo cumplimiento mediante WO 18027-057060 de CEMA.	X			X	-	-	-	38937,2	56679	4-abr-18	-	-	-	-	-	-	-	-	
		Inspection for Oil-Canning. Ultimo cumplimiento mediante WO 18027-057060 de CEMA.		X	X		-	6000	-	38937,2	56679	4-abr-18	-	62679	-	-	-	5016	-	-	
		Inspection of the Dome Cap at the Center of the Bulkhead. Ultimo cumplimiento mediante WO 18027-057060 de CEMA.		X	X		-	6000	-	38937,2	56679	4-abr-18	-	62679	-	-	-	5016	-	-	
		Inspection of the Forward Flange of the "Z" Stiffeners at the Dome Cap. Ultimo cumplimiento mediante WO 18027-057060 de CEMA.		X	X		-	12000	-	38937,2	56679	4-abr-18	-	68679	-	-	-	11016	-	-	
		Inspection for Existing Repairs on the Bulkhead. Inspección de Reparacion en el Dome Cap cumplida de acuerdo a WO 18027-067096. Cumplir de acuerdo al SRM 50-80-08 Pg 201 - 207 Rep. 9.		X			X	-	6000	-	PENDIENTE PRIMER CUMPLIMIENTO	PENDIENTE PRIMER CUMPLIMIENTO	PENDIENTE PRIMER CUMPLIMIENTO	-	86679	-	-	-	29016	-	-
2012-19-11	7-nov-12	IMPROVE THE RELIABILITY OF THE CABIN ALTITUDE WARNING SYSTEM. supercedida por la AD 2015-21-11					-	-	-	-	-	-	-	-	-	-	-	-	-	-	
2012-23-04 SB 53-A1166 R2	26-dic-12	<b>INSPECTIONS FOR CRACKS IN THE BS 727 FRAME OUTBOARD CHORDS AND THE RADIUS OF THE AUXILIARY CHORD</b>					-	-	-	-	-	-	-	-	-	-	-	-	-	-	
		Detailed and Eddy Current Inspections for cracks BS 727 Auxiliary Chord (L/H). Ultimo cumplimiento mediante documento TASK NO. FAA 2012-23-04.000.00 de ASIANA AIRLINES.		X	X		-	15000	-	32034,26	46733	1-abr-12	-	61733	-	-	-	4070	-	-	
		Detailed and Eddy Current Inspections for cracks BS 727 Auxiliary Chord (R/H). Ultimo cumplimiento mediante documento TASK NO. FAA 2012-23-04.001.00 de ASIANA AIRLINES.		X	X		-	-	-	32034,26	46733	1-abr-12	-	61733	-	-	-	4070	-	-	
		Ultrasonic and surface HFEC Inspections BS727 Frame Outbd Chord FWD Flange(L/H). Ultimo cumplimiento mediante documento TASK NO. FAA 2012-23-04.002.00 de ASIANA AIRLINES.		X	X		-	5000	-	37470,53	54583	13-mar-15	-	59583	-	-	-	1920	-	-	
		Ultrasonic and surface HFEC Inspections BS727 Frame Outbd Chord FWD Flange(R/H). Ultimo cumplimiento mediante documento TASK NO. FAA 2012-23-04.003.00 de ASIANA AIRLINES.		X	X		-	-	-	37470,53	54583	26-mar-15	-	59583	-	-	-	1920	-	-	
		Open hole Eddy Current Inspection BS727 Frame Outbd Chord (L/H). Ultimo cumplimiento mediante documento TASK NO. FAA 2012-23-04.004.00 de ASIANA AIRLINES.		X	X		-	-	-	34715,6	50618	28-ago-13	-	65618	-	-	-	7955	-	-	

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<b>APPLICABILITY</b>		ENGINE LH	ENGINE RH	APU	
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
		Open hole Eddy Current Inspection BS727 Frame Outbd Chord (R/H). Ultimo cumplimiento mediante documento TASK NO. FAA 2012-23-04.005.00 de ASIANA AIRLINES.		X	X			-		-		34715,6	50724	28-ago-13	-	65724	-	-	8061	-	
2012-23-08 SB 54-1043 R2	31-dic-12	<b>NACELLES/PYLONS - STRUT DRAINS AND SEALS - AFT FAIRING DRAIN TUBE ASSEMBLY REPLACEMENT.</b> N/A por modelo de aeronave en efectividad de la AD, solo aplican los modelos 737-600 en adelante.						-	-	-		-		-		-		-	-	-	
2012-23-14 SB 53-A1318	7-ene-13	<b>FUSELAGE - SKIN - SKIN CRACKING AT THE AFT WASTE WATER DRAIN MAST - INSPECTION AND REPAIR.</b> Cumplimiento soportado en la Página 036410 del FLIGHT AND MAINTENANCE LOG BOOK de ASIANA AIRLINES.	X		X			-	-	-		34024,15	49756	2-abr-13	-	-	-	-	-	-	
2012-26-02 SB 53-A1234 R2	6-feb-13	<b>FUSELAGE - SKIN - CRACKS AT CHEM-MILLED STEPS - INSPECTION AND REPAIR.</b>						-	-	-		-		-		-		-	-	-	
		(m) Fuselage Skin Inspections at Chem-Mill Steps Common to Lap Joints. Ultimo cumplimiento mediante WO 18027-051195 de CEMA.		X	X			-	500	-		40238,2	57661	23-jul-21	-	58161	-	-	498	-	
		(O) Fuselage Skin Inspections at Chem-Mill Steps Common to Shear Wrinkle Areas. Ultimo cumplimiento mediante WO 18027-051195 de CEMA.		X	X			-	1000	-		40238,2	57661	23-jul-21	-	58661	-	-	998	-	
		Fuselage Skin Inspections at Specified Vertical Chem-Mill Step Locations (LO QUE OCURRA PRIMERO). Ultimo cumplimiento mediante WO 18027-051195 de CEMA.		X	X			1800	1800	-		38937,2	56679	27-mar-18	40737,2	58479	-	497,12	816	-	
		(q) Fuselage Skin Inspections at Chem-Mill Steps in General Pocket-to-Pocket Areas (Accomplishment Instruction 3.B.1.d. of SB 53-A1234 R2). Ultimo cumplimiento mediante WO 18027-051195 de CEMA.		X	X			-	500	-		40238,2	57661	24-jul-21	-	58161	-	-	498	-	
		Fuselage Skin Inspections at Chem-Mill Steps in General Pocket-to-Pocket Areas (Accomplishment Instruction 3.B.1.e. of SB 53-A1234 R2). Aun cuenta con tiempo remanente para programar el primer cumplimiento cumplimiento inicial a los 66000 ciclos y en adelante cada 500 ciclos.		X		X		-	500	-		POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	66000	-	-	8337	-	

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<b>APPLICABILITY</b>		ENGINE LH		ENGINE RH	
<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>APU</b>	<b>CSO</b>
				N/A	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE	INTERVAL			LAST DONE			NEXT DUE			REMAINING					
			U	R	C	O		FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
		Fuselaje Skin Inspections at Chem-Mill Steps in General Pocket-to-Pocket Areas (Accomplishment Instruction 3.B.1.f. of SB 53-A1234 R2). Aun cuenta con tiempo remanente para programar el primer cumplimiento. Primer cumplimiento a los 81000 ciclos y despues cada 500 ciclos		X			X	-	500	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	81000	-	-	23337	-			
2013-01-03 SB 737-23-1302	22-feb-13	<b>COMMUNICATIONS - PASSENGER ENTERTAINMENT SYSTEM - AUDIO ENTERTAINMENT MULTIPLEXER POWER RELAY WIRING CHANGE.</b> N/A por V/N en efectividad del SB 23-1302 (AGO 24, 2009). No aplican V/N iniciados en PV.						-	-	-	-	-	-	-	-	-	-	-	-			
2013-04-05 SB 53-1287 R1	28-mar-13	<b>FUSELAGE - SKIN - SKIN CRACKING UNDER THE NUMBER 3 VHF ANTENNA - INSPECTION, REPAIR AND PREVENTIVE MODIFICATION.</b> N/A por MSN en aplicabilidad del SB 53-1287 R1 (NOV 15, 2010). Solo aplican los MSN del 292 al 2790, para aeronaves B 737-400.						-	-	-	-	-	-	-	-	-	-	-	-			
2013-09-01 SB 53-A1255 R3	13-jun-13	<b>FUSELAGE - SKIN - CROWN SKIN LAP JOINT - INSPECTION FOR LOWER ROW CRACKING AT STRINGERS S-4, S-10, S-14 BETWEEN STA 360 AND STA 908.</b> N/A por V/N en efectividad el SB 53-A1255 R3 (AGO 7, 2015), solo aplican los V/N PV 231 al PV 237.						-	-	-	-	-	-	-	-	-	-	-	-			
2013-09-02 SB 57-A1271 R3	13-jun-13	<b>WINGS - TRAILING EDGE AND TRAILING EDGE DEVICES - TRAILING EDGE FLAPS - OUTBOARD FLAP, INBOARD TRACK INSPECTION/OVERHAUL/REPLACEMENT AT WBL 254</b>						-	-	-	-	-	-	-	-	-	-	-	-			
		ITEM (S) Inspections of Flap-to-Wing Attachment if Repairs Are Done or if No Damage Is Found in Flap Track Web and Flanges: DET, HFEC, Ultrasonic Insp. (LO QUE OCURRA PRIMERO) WO 18027-053043 de CEMA.						X	X	-	-	-	38937,2	56679	22-jun-18	-	-	-	-	-		
		ITEM (T) Overhaul/Replace Outboard TE Flap Inboard Track # 2 y #7. SEE HT CONTROL								-	-	-	-	-	-	-	-	-	-	-		
		ITEM (U) Post-Overhaul Inspections: DET, Eddy Current, Visual Insp						X		X	-	20000	1440	38937,2	56679	22-jun-18	-	76679	01-jun-22	-	19016	272
2013-09-08 SB 28-A1307	18-jun-13	<b>FUEL - ENGINE FUEL FEED SYSTEM - ENGINE FUEL SUCTION FEED OPERATIONAL TEST.</b> Ultimo cumplimiento mediante WO 0264 de GCA. (LO QUE OCURRA PRIMERO)						X	X	7500	-	720	39273,25	56968	19-oct-20	46773,25	-	09-oct-22	6533,17	-	402	



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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	<b>APU</b>
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<b>TSN</b>	40240,1	<b>CSN</b>	57663
BOEING				<b>TSO</b>	N/A
				<b>CSO</b>	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
2014-01-05 SB 53-A1322	14-mar-14	STA 360 and STA 908 Zone A (S-5L to S-3L, S-3R to S-5R) Inspection. Ultimo cumplimiento mediante WO 18027-051229 de CEMA.		X		X	-	20000	-	38937,2	56679	5-jun-18	-	76679	-	-	19016	-			
		STA 360 and STA 908 Zone B (S-10L to S-5L, S-3L to S-3R, S-5R to S-10R) Inspection. Ultimo cumplimiento mediante WO 18027-051229 de CEMA.		X		X	-	20000	-	38937,2	56679	5-jun-18	-	76679	-	-	19016	-			
		Internal Detailed Inspection of the Stringer Splice according with paragraph 3.B. - PARTII of SB. Tabla 11 del SB		X		X	-	30000	-	PENDIENTE PRIMER CUMPLIMIENTO	PENDIENTE PRIMER CUMPLIMIENTO	PENDIENTE PRIMER CUMPLIMIENTO	-	86679	-	-	29016	-			
2014-03-06	9-abr-14	WINGS - LANDING GEAR SUPPORT BEAM - REAR SPAR AND LANDING GEAR SUPPORT FITTING INSPECTION. Fecha de efectividad 9 abril de 2014 supercedida por las AD's 2014-12-13 y 2015-21-08.					-	-	-	-	-	-	-	-	-	-	-				
2014-04-05 SB 21-1186	8-abr-14	AIR CONDITIONING - CONTROL CABIN CONDITIONED AIR DISTRIBUTION SYSTEM - INSPECTION OF THE METAL CLAMP INSTALLED ON HOSES TO THE AIR CONDITIONING TEMPERATURE SENSOR, GASPER AIR OUTLET AND DIFFUSER. Cumplido mediante WO 18027-051228 de CEMA.	X		X		-	-	-	38937,2	56679	16-mar-18	-	-	-	-	-				
2014-05-02	-	FUSELAGE - AFT PRESSURE BULKHEAD INSPECTION, MODIFICATION, AND REPAIR. supercedida por la AD 2014-18-02					-	-	-	-	-	-	-	-	-	-	-				
2014-05-12 SB 57-A1277R1		CARRIAGE SPINDLE OF THE OUTBOARD MID-FLAPS - REPETITIVE INSPECTIONS, GAP CHECKS, OVERHAUL, REPLACEMENT.					-	-	-	-	-	-	-	-	-	-	-				
		WORK PACKAGE 1 - On Wing NDT Ultrasonic & General Visual Inspection (LO QUE OCURRA DESPUES). Ultimo cumplimiento bajo WO 18027-053044 de CEMA.		X		X	-	1500	180	38937,2	56679	5-jun-18	-	58179	02-dic-18	-	516	-1005			
		Parag. (H). WORK PACKAGE 2 - Midflap/Spindle Carriage Gap Check (LO QUE OCURRA DESPUES). El cumplimiento del Gap Check se realiza después de la inspección ultrasónica. Ultimo cumplimiento bajo WO 0584 by GCA		X		X	-	140	14	40180,2	57625	10-jul-21	-	57765	24-jul-21	-	102	-40			
		OVH - OUTBOARD TE MIDFLAP CARRIAGE #1. Ultimo cumplimiento bajo TASK NO. FAA 2014-05-12.002.00.		X	X		-	12000	2880	36414,36	53162	30-ago-14	-	65162	19-jul-22	-	7499	320			



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<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
					<b>CSO</b>
					N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
SB 57-A1218R5		OVH - OUTBOARD TE MIDFLAP CARRIAGE #2. Ultimo cumplimiento bajo TASK NO. FAA 2014-15-12.003.00.		X	X			-	12000	2880	36414,36	53162	30-ago-14	-	65162	19-jul-22	-	7499	320		
		OVH - OUTBOARD TE MIDFLAP CARRIAGE #7. Ultimo cumplimiento bajo TASK NO. FAA 2014-15-12.004.00.		X	X			-	12000	2880	36414,36	53162	30-ago-14	-	65162	19-jul-22	-	7499	320		
		OVH - OUTBOARD TE MIDFLAP CARRIAGE #8. Ultimo cumplimiento bajo TASK NO. FAA 2014-15-12.005.00.		X	X			-	12000	2880	36414,36	53162	30-ago-14	-	65162	19-jul-22	-	7499	320		
		REPLACEMENT - OUTBOARD TE MIDFLAP CARRIAGE #1		X		X		-	40000	-	36414,36	53162	30-ago-14	-	83602	-	-	25939	-		
		REPLACEMENT - OUTBOARD TE MIDFLAP CARRIAGE #2		X		X		-	40000	-	36414,36	53162	30-ago-14	-	83602	-	-	30440	-		
		REPLACEMENT- OUTBOARD TE MIDFLAP CARRIAGE #7		X		X		-	40000	-	36414,36	53162	30-ago-14	-	83602	-	-	30440	-		
		REPLACEMENT- OUTBOARD TE MIDFLAP CARRIAGE #8		X		X		-	40000	-	36414,36	53162	30-ago-14	-	83602	-	-	30440	-		
2014-05-21 SB 53-A1200	22-abr-14	<b>FUSELAGE - SKIN - FORWARD GALLEY SERVICE DOORWAY UPPER AND LOWER CORNERS - INSPECTION, REPAIR AND PREVENTIVE MODIFICATION.</b>						-	-	-	-	-	-	-	-	-	-	-			
		Forward Entry Doorway Cutout Inspection ( Upper and Lower Hinge). Cumplimiento inicial mediante MEO SB 737-53A1200.000.00 ( FEB 08, 2007). Ultimo cumplimiento soportado en la pagina 009981 del MAINTENANCE FLIGHT LOG BOOK de ASIANA AIRLINES.		X	X			-	18000	-	32171,96	46947	4-may-12	-	64947	-	-	7284	-		
		Forward Galley Service Doorway Cutout Inspection ( Upper and Lower Hinge). Cumplimiento inicial mediante MEO SB 737-53A1200.001.01 (ENE 26, 2009). Ultimo cumplimiento soportado en la pagina 048360 del MAINTENANCE FLIGHT LOG BOOK de ASIANA AIRLINES.		X	X			-	18000	-	36417,02	53162	5-ago-14	-	71162	-	-	13499	-		
2014-05-27	2-may-14	<b>ROCKWELL COLINS ATC TRANSPONDER TEST AND CALIBRATION.</b> Cumplida mediante WO 18027-051202 de CEMA.		X		X		-	-	1440	38937,2	56679	4-abr-18	-	-	14-mar-22	-	-	193		
2014-08-08 53-A1329	3-jun-14	<b>FUSELAGE - SKIN - FORWARD ENTRY DOORWAY - LOWER CORNERS - INSPECTION AND REPAIR.</b> Ultimo cumplimiento mediante WO 18027-051203 de CEMA.		X		X		-	4500	-	38937,2	56679	4-abr-18	-	61179	-	-	3516	-		







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<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<b>TSN</b>	<input type="checkbox"/>	<b>CSN</b>	<b>TSO</b>	<b>CSO</b>
BOEING		40240,1		57663	N/A	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
2015-21-08 SB 57-A1326	12-nov-15	<b>WING - WING SPARS - INSPECTION OF UPPER WING SKIN AT REAR SPAR, WING BUTTOCK LINE 157.</b>					-	-	-	-	-	-	-	-	-	-	-	-	-	-	
		Eddy Current Inspection (LH). Cumplimiento inicial mediante MEO SB 737-57A1326.000.00 (OCT 07, 2015). Ultimo cumplimiento mediante WO 18027-053046 de CEMA.		X	X		-	-	-	40238,2	57661	25-jul-21	-	58161	-	-	498	-	-	-	
		Eddy Current Inspection (RH). Cumplimiento inicial mediante MEO SB 737-57A1326.000.00 (OCT 07, 2015). Ultimo cumplimiento mediante Ultimo cumplimiento mediante WO 18027-053046 de CEMA.		X	X		-	500	-	40238,2	57661	25-jul-21	-	58161	-	-	498	-	-	-	
2015-21-11 SB 21-1164 R2	12-nov-15	<b>IMPROVE THE RELIABILITY OF THE CABIN ALTITUDE WARNING SYSTEM.</b>					-	-	-	-	-	-	-	-	-	-	-	-	-	-	
		(G). Install A Redundant Cabin Altitude Pressure Switch, Replace The Auralwarning Module (AWM). Cumplido mediante WO 18027-05-1205 de CEMA, soportado con forma FAA 337.	X		X		-	-	-	38937,2	56679	26-jun-18	-	-	-	-	-	-	-	-	
		(H). Modify the instrument panels, install light assemblies, modify the wire bundles, and install a new circuit breaker. Cumplida mediante implementacion del SB correspondiente bajo documento MEO SB 737-31A1325.000.00 registrada en hoja 002010 del "heavy maintenance log"	X		X		-	-	-	38383,68	55913	5-oct-10	-	-	-	-	-	-	-	-	
2015-23-08 SB 53-A1323	4-ene-16	<b>FUSELAGE - FRAMES - CRACKING OF FRAMES AT STA 328 BETWEEN STRINGERS S-20R TO S-21R, STA 360 BETWEEN STRINGERS S-19R TO S-21R, AND STA 380 BETWEEN STRINGERS S-18R TO S-20R, AND MISSING STRAP ON FRAME AT STA 312 - INSPECTION, MODIFICATION, AND REPAIR.</b>					-	-	-	-	-	-	-	-	-	-	-	-	-	-	
		Inspection of the Frame at STA 312 to Make Sure Strap is Installed. Cumplimiento soportado en la página 054847 del FLIGHT MAINTENANCE LOG BOOK de ASIANA AIRLINES.	X		X		-	-	-	37470,53	54583	24-mar-15	-	-	-	-	-	-	-	-	
		Inspections of the Frame at STA 328 for Cracking. Cumplimiento soportado en la página 054847 del FLIGHT MAINTENANCE LOG BOOK de ASIANA AIRLINES.		X	X		-	19000	-	37470,53	54583	24-mar-15	-	73583	-	-	15920	-	-	-	

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<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>APU</b>	<b>CSO</b>
					N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING				
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS					
2015-23-09	28-dic-05	<b>FLIGHTCREW OXYGEN MASK REGULATORS IDENTIFICATION AND REPLACEMENT.</b>	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-		
		Inspect each flightcrew oxygen mask regulator to identify the part number and serial number. Cumplido mediante WO 18027-051226 de CEMA.	X				X			-	-	-	38937,2	56679	14-mar-18	-	-	-	-	-	-	-	-
		AFM Revision. Cumplido mediante WO 18027-051226 de CEMA.	X				X			-	-	-	38937,2	56679	14-mar-18	-	-	-	-	-	-	-	-
		Placard Installation. Cumplido mediante WO 18027-051226 de CEMA.	X				X			-	-	-	38937,2	56679	14-mar-18	-	-	-	-	-	-	-	-
2016-01-05	17-feb-16	<b>DETAILED INSPECTION OF CERTAIN CARGO PLACARDS AND HARNESES.</b> N/A por S/N en efectividad de la AD, solo aplican los S/N s 23865, 24231, 24706, 24474, 25417, 27003, 27149, 25375, 26281, 28661, y 28881, los cuales tienen implementado el STC ST0114WI.																					
		<b>WINGS - FRONT SPAR - WING FRONT SPAR LOWER CHORD BETWEEN WBL 114 AND WBL 183 AND WING SKIN COMMON TO THE NACELLE OUTBOARD SIDE LOAD FITTING FASTENERS AT WBL 198.6 - INSPECTION</b>																					
2016-03-01 SB 57A1323	10-mar-16	Detailed Inspection (DI) of the Left And Right Wing Front Spar Lower Chord and Inspar Skin between (WBL) 114 and (WBL) 183. Ultimo cumplimiento mediante WO 18027-053047 de CEMA.										38937,2	56679	4-jun-18	-	61679	-	-	-	-	4016	-	
		PAR(i); Do PART 2, HFEC Insp through mouse hole of the left and right wing at (WBL) 198.6 for cracks. Accomplished IAW WO 0157 & EWO 0121											39251,1	56950	12-mar-20	-	66950	-	-	-	-	9287	-
2016-06-02 SB 55-A1100	29-mar-06	<b>HORIZONTAL STABILIZER - CENTER SECTION REAR SPAR UPPER CHORD INSPECTION, REPAIR, AND ALTERATION.</b>																					
		Horizontal Stabilizer Center Section Rear Spar Upper Chord Horizontal Flange - Inspection.															60000	-	-	-	-	2337	-
		Horizontal Stabilizer Center Section Rear Spar Upper Chord Vertical Flange - Inspection														60000	-	-	-	-	-	2337	-



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<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>APU</b>	
				<b>CSO</b>	
				N/A	

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE	INTERVAL			LAST DONE			NEXT DUE			REMAINING			
			U	R	C	O		FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS	
2016-11-04 SB 53-A1279 R2	6-jul-16	Part 4 – Repeat Detail and HFEC Inspections. Cumplimiento inicial mediante MEO SB 737-53A1279.001.00 (ENE 26, 2009). Último cumplimiento mediante TASK NO. 2011-23-05.000.01. Se cancela acción repetitiva por cumplimiento de la acción terminante.		X	X			-	4500	-	37464,14	54583	4-mar-15	-	-	-	-	-	-	
		Acción terminante cumplida mediante cumplimiento del parrafo (p) Preventative Modification of the Frames Between S-19 and S-22. Mediante WO 18027-052164 de CEMA.	X			X			-	-	-	38937,2	56679	6-jun-18	-	-	-	-	-	-
2016-11-20	-	<b>RISK OF FIRE AND/OR FUNCTIONAL IMPAIRMENT ON UNSERVICEABLE B/E AEROSPACE PROTECTIVE BREATHING EQUIPMENT (PBE).</b> supercedida por la AD 2017-18-12.						-	-	-	-	-	-	-	-	-	-	-	-	
2016-13-10	9-ago-16	FUSELAGE - SKIN - STRINGER S-1 AND STRINGER S-2R, STATION 400 TO STATION 460 INSPECTION, REPAIR AND MODIFICATION. supercedida por la AD 2016-16-13.						-	-	-	-	-	-	-	-	-	-	-	-	
2016-13-11 SB 57A129 R2	9-ago-16	<b>WINGS - CENTER WING BOX - LONGITUDINAL FLOOR BEAMS BETWEEN BODY STATION 656 AND BODY STATION 727 - INSPECTION, REPAIR AND PREVENTIVE MODIFICATION.</b> Longitudinal Floor Beam Inspection and Repair - Detailed Visual and Open Hole HFEC in web, upper and lower chord (LH & RH). Cumplimiento inicial mediante documento MEO SB 737-57-1296.000.00 (ENE 19, 2009). Último cumplimiento mediante documento TASK NO. FAA 2008-05-06.001.02 (GTL 37520) de Asiana Airlines.		X	X			-	7000	-	36414,36	53162	24-ago-14	-	60162	-	-	2499	-	
2016-16-13 SB 53-A1293 R3	15-ago-16	<b>FUSELAGE - SKIN - STRINGER S-1 AND STRINGER S-2R, STATION 400 TO STATION 460 INSPECTION, REPAIR AND MODIFICATION</b>						-	-	-	-	-	-	-	-	-	-	-	-	
		External Detailed Inspection (DI) and external Non Destructive Inspection (NDI) of the fuselage skin chem-mill steps. Cumplida mediante documento MEO SB 737-53A1293.000.00 (SEP 20, 2008).	X			X			-	-	-	23945	34958	20-sep-08	-	-	-	-	-	-
		External NDI of the fuselage skin chem-mill steps. Último cumplimiento mediante documento WO 18027-051207 de CEMA.		X		X			-	500	-	40238,2	57661	25-jul-21	-	58161	-	-	498	-



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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	
<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>APU</b>	<b>CSO</b>
				N/A	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE	INTERVAL			LAST DONE			NEXT DUE			REMAINING			
			U	R	C	O		FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS	
2016-18-16 SB 28-A1210 R2 SB 28-A1216 R3	31-oct-16	<b>FUEL - ENGINE FUEL FEED SYSTEM - INSTALLATION OF THE AUTO SHUTOFF PROTECTION FOR THE CENTER AND AUXILIARY FUEL TANK BOOST PUMPS</b>																		
		Installation of Automatic Shutoff System for the Center and Auxiliary Tank Fuel Boost Pumps. Cumplida bajo documento MEO SB 737-28A1216.	X			X						32717,96	46947	10-may-12						
		Installation of P5-2 Fuel System Module Assembly. Cumplida mediante documento MEO SB 737-28A1210.	X			X						32717,96	46947	9-may-12						
		AFM REVISION - Cumplida mediante incorporacion por revision al AFM																		
		Airworthiness Limitations (AWLs) Revision for Automatic Shutoff System. Cumplido mediante incorporación del documento D6-38278-CMR al MPD.																		
		Installation of Un-commanded ON (UCO) Protection System	X				X										31-oct-21			59
		AWLs Revision for UCO Protection System	X				X										31-oct-21			59
2016-20-02 SB 53-A1214 R5	25-nov-17	<b>FUSELAGE - BODY STATION 1016 AFT PRESSURE BULKHEAD WEB CRACKS - INSPECTION AND REPAIR.</b>																		
		Inspections of the Aft Pressure Bulkhead Web at the "Y"-Chord Upper Bulkhead -Tabla 10 del SB.					X		X										18337	
		Repetitive Inspections of the Aft Pressure Bulkhead Web at the "Y"-Chord Below S-15.					X		X										18337	
2016-23-08 SB 53-A1347	30-dic-16	<b>FUSELAGE - FRAMES AND BULKHEADS - BODY STATIONS 559, 578 AND 601 - BODY FRAME AND OVERWING STUB BEAM INSPECTION AND REPAIR</b>																		
		Overwing Stub Beam Upper Chord at STA 559, STA 578 And STA 601 - HFEC Inspection For Any Crack.					X		X										514	
		Post-Replacement Inspections and Corrective Action (N/A)																		
2017-02-05 SB 53-A1350,	14-mar-17	<b>FUSELAGE - SKIN - UPPER AND LOWER CORNERS OF AFT DOORWAYS INSPECTION AND REPAIR</b>																		
		Low Frequency Eddy Current (LFEC) and Detailed Inspections of Aft Entry & Aft Galley Doorway Corners.					X		X										3514	
2017-02-12 SB 55-A1092	28-mar-17	<b>STABILIZERS - HORIZONTAL STABILIZER - FRONT SPAR CHORD LUGS - INSPECTIONS</b>																		
		Inspect for Corrosion and any Crack on The Horizontal Stabilizer Front Spar Chord Lugs. Cumplido WO 0024 de GCA					X		X								13-sep-21			11

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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	<b>APU</b>
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<b>TSN</b>	<input type="checkbox"/>	<b>CSN</b>	<b>TSO</b>
BOEING		40240,1		57663	N/A
					<b>CSO</b>
					N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
		Inspect for Corrosion on The Horizontal Stabilizer Front Spar Chord Lug Bores. Cumplido WO 0024 de GCA		X		X	-	-	2880	38945,6	56686	14/09/2019	-	-	03-ago-27	-	-	2161			
2017-04-08 SB 53-A1261 R1	3-abr-17	<b>FUSELAGE - FRAMES AND BULKHEADS - UPPER FRAME TO SIDE FRAME SPLICE - INSPECTION, REPAIR AND MODIFICATION.</b> N/A por V/N en efectividad del SB 53-A1261 R1 (ENE 30, 2015), Solo aplican los V/N PV001 - PV040.					-	-	-	-	-	-	-	-	-	-	-	-			
2017-06-14 SB 55-1059 R1	2-may-17	<b>STABILIZERS - HORIZONTAL STABILIZER - LOWER SKIN INSPECTION AND REPAIR</b>					-	-	-	-	-	-	-	-	-	-	-	-			
		Detailed Inspection of the Horizontal Stabilizer Lower Skin. Post-Repair Inspection of the Horizontal Stabilizer Lower Skin for any Crack and for any Loose or any Missing Fastener. (N/A)		X		X	-	4500	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	63679	-	-	6016	-			
2017-10-07 SB 53-1187 R3	5-jul-17	<b>FUSELAGE - AFT SKIN - INSPECTION, REPAIR AND SKIN PANEL REPLACEMENT.</b> N/A por V/N en efectividad del SB 53-1187 R3 (JUL 10, 2015), solo aplican los V/N PV231-PV237.					-	-	-	-	-	-	-	-	-	-	-				
2017-10-21 SB 53-A1343	29-jun-17	<b>FUSELAGE - SKIN - LAP JOINT SKIN MODIFICATION AT STRINGER S-4R AND S-4L</b>					-	-	-	-	-	-	-	-	-	-	-				
		Lap Joint Skin Modification.	X			X	-	-	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	59677	-	-	2014	-			
		Inspection of the Critical Fastener Row.		X		X	-	-	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	97677	-	-	40014	-			
2017-12-05 SB 53-A1267 R1	25-jul-17	<b>FUSELAGE - SKIN - STA 259.5 CIRCUMFERENTIAL BUTT SPLICE - INSPECTION AND REPAIR.</b> N/A por V/N en efectividad del SB 53-A1267 R1 (MAR 8, 2016), solo aplican los V/N del PV231-PV237.					-	-	-	-	-	-	-	-	-	-	-				
2017-13-01 SB 53-A1366	27-jul-17	<b>FUSELAGE - FRAMES AND BULKHEADS - EMERGENCY EXIT FRAME INBOARD CHORD - INSPECTIONS.</b> Detailed Inspection and HFEC Inspection for Any Crack at the Frame Inboard Chord, Forward of the Emergency Exit Opening, Left and Right Side		X		X	-	6000	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	61177	-	-	3514	-			
2017-13-13 SB 53-A1354	9-ago-17	<b>FUSELAGE - FRAMES AND BULKHEADS - SECTION 41 LOWER LOBE FRAMES - INSPECTIONS.</b> N/A por V/N en efectividad el SB 53-A1354 (DIC 02, 2016), solo aplican los V/N PV231 - PV237.					-	-	-	-	-	-	-	-	-	-	-				
2017-15-12 SB 53-A1365	5-sep-17	<b>FUSELAGE - SKIN - S-14 LAP SPLICE LOWER FASTENER ROW INSPECTION.</b> Eddy Current Inspection For Any Crack At S-14 Skin Lap Splice.		X		X	-	6000	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	61177	-	-	3514	-			

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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	<b>APU</b>
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<b>TSN</b>	40240,1	<b>CSN</b>	57663
BOEING				<b>TSO</b>	N/A
				<b>CSO</b>	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
2017-17-04 SB 57-A1332	25-sep-17	<b>WINGS - PLATES/SKIN - UPPER AFT SKIN AT REAR SPAR FROM WING BUTTOCK LINE 80 TO 155 - INSPECTION.</b> Detailed inspection of the upper aft skin on the left and right wing from Wing Buttock Line (WBL) 80 to WBL 155		X		X	-	4500	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	61177	-	-	-	3514	-		
2017-17-06 SB 53-A1357	25-sep-17	<b>FUSELAGE - FRAMES AND BULKHEADS - STATION 947.5 FRAME BETWEEN STRINGER S-24R AND STRINGER S-25R INSPECTION AND REPAIR.</b> High frequency eddy current (HFEC) inspection for any cracking in the fuselage frame at station (STA) 947.5 common to the water tank 6 support intercostal clip located between stringers S-24R and S-25R.		X		X	-	12000	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	62677	-	-	-	5014	-		
2017-17-09 SB 57-1320	29-sep-17	<b>WINGS - TRAILING EDGE - TANK ACCESS DOOR - REPLACEMENT</b> New Door Assembly, Electrical Bond Check, and Related Corrective Actions. Cumplido mediante WO 0285 GCA.	X			X	-	-	-	39273,3	56968	18/10/2020	-	-	-	-	-	-	-		
		Revise the Maintenance or Inspection Program. Cumplido mediante incorporación por revisión al documento D6-38278-CMR (CERTIFICATION MAINTENANCE REQUIREMENTS). Incorporado en la Revisión Original (30 de Enero 2018) del MPM capítulo 8.	X			X	-	-	-	-	-	-	30-ene-18	-	-	-	-	-	-	-	
2017-18-12	16-oct-17	<b>RISK OF FIRE AND/OR FUNCTIONAL IMPAIRMENT ON UNSERVICEABLE B/E AEROSPACE PROTECTIVE BREATHING EQUIPMENT (PBE).</b> Cumplida mediante documento WO 18027-051227.	X			X	-	-	-	38937,2	56679	5-jun-18	-	-	-	-	-	-	-		
2017-19-05	31-oct-17	<b>INSPECTION AND REPLACEMENT OF THE AFFECTED SMOKE DETECTORS.</b> N/A por P/N de los "Smoke Detectors" en efectividad de la AD, solo aplican los "Smokes Detectors" fabricados por Siemens S.A.S. con P/N's PMC1102-02, PMC3100-00, GMC1102-02.					-	-	-	-	-	-	-	-	-	-	-	-	-		
2017-20-14	16-nov-17	<b>REVISING THE MAINTENANCE PROGRAM TO ADD SSID.</b> Revision of the Maintenance or Inspection Program <b>(LO QUE OCURRA DESPUES)</b> .	X			X	-	-	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	66000	16-nov-18	-	8337	-1021			
		Inspection for detect cracks on all structure identified in Boeing Document D6-82669. <b>(LO QUE OCURRA DESPUES)</b> .	X			X	-	-	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	66000	16-nov-18	-	8337	-1021			

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<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE		STATE		INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS
2017-22-04 SB 53-A1349	30-nov-17	<b>FUSELAGE - SKIN - INSPECTION, REPAIR, AND REPLACEMENT OF DISBONDED FUSELAGE SKIN PANELS</b>					-	-	-	-	-	-	-	-	-	-	-	-
		External General Visual and Detailed Inspections for any Crack, Corrosion or Existing Disbond Repair.		X		X	-	4500	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	61179	-	-	3516	-
		External High Frequency Bond Test Inspection for any Disbond: S-4L and S-4R lap splice between BS 259,50 and BS 1016 / S-10L and S-10R lap splice between BS 360 and BS 1016.		X		X	-	5000	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	66179	-	-	8516	-
		External High Frequency Bond Test Inspection for any Disbond: BS 540 butt joint between S-10L and S-10R / BS 727 butt joint between S-10L and S-10R / BS 500C butt joint between S-10L and S-10R / BS 727G butt joint between S-10L and S-10R.		X		X	-	22000	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	83179	-	-	25516	-
		External High Frequency Bond Test Inspection for any Disbond: Butt joint FWD of BS 360 from S-4L to S-4R and AFT of BS 360 from S-10L to S-10R / BS 908 butt joint between S-10L and S-10R.		X		X	-	47000	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	108179	-	-	50516	-
		Internal and External Inspection of Fuselage Skin Panels for Disbond		X		X	-	35000	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	65432	-	-	7769	-
2017-23-02 SB 53-A1358	20-dic-17	<b>FUSELAGE - SKIN - 0.036 INCH CROWN SKIN PANELS BETWEEN S-10L AND S-10R, STA 360 TO STA 540 AND STA 727 TO STA 908 - INSPECTION, REPAIR AND REPLACEMENT.</b> N/A por V/N en efectividad del SB 53-A1358 (ABR 27, 2017), solo aplican los V/N PV231 - PV237.					-	-	-	-	-	-	-	-	-	-	-	-
2017-23-05 SB 57-A1335	20-dic-17	<b>WINGS - PLATES/SKIN - UPPER AFT SKIN AT AND FORWARD OF THE REAR SPAR FROM WING BUTTOCK LINE 159 TO 220 - INSPECTION.</b> Detailed Inspection of the Upper Wing Aft Skin From WBL 159 To 220 For Any Crack.		X		X	-	4500	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	61179	-	-	3516	-
2017-23-07 SB 57-A1333	26-dic-17	<b>WINGS - PLATES/SKIN - LOWER CHORD OF REAR SPAR AND LOWER AFT SKIN AT WING BUTTOCK LINE 157 - INSPECTION.</b> Low Frequency Eddy Current (LFEC) Inspection of the lower chord of the rear spar and detailed inspection of the lower aft skin on the left and right wing at Wing Buttock Line (WBL) 157.		X		X	-	4500	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	66000	-	-	8337	-



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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	
<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>CSO</b>	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
2017-25-12 SB 53-A1364	22/01/2018 Replaced by AD 2019-14-15	High frequency eddy current (HFEC) and detailed inspections of the STA 685 and STA 706 stub beam webs for any crack.		X		X	-	-	-	-	-	-	-	-	-	-	-	-	-	-	
		General Visual Inspection for Any Existing Repair and High Frequency Eddy Current (HFEC) Inspection for Any Crack in the STA 685 and STA 706 Stub Beam Webs In Repaired Areas. N/A debido a que no hay reparaciones entre las estaciones 685 y 706 de acuerdo al "DENT & BUCKLE CHART".					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
		High Frequency Eddy Current (HFEC) and Detailed Inspections for Any Crack in the STA 695 Stub Beam Web		X		X	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
2018-01-08 53- A1166 R2	16-feb-18	<b>FUSELAGE - FRAMES - BODY STATION 727 OUTBOARD CHORD AT STRINGER 18A, INSPECTION AND PREVENTIVE MODIFICATION</b>					-	-	-	-	-	-	-	-	-	-	-	-	-	-	
		PAR (G). Internal detailed and High Frequency Eddy Current (HFEC) inspections to detect cracks in the auxiliary chord radius. Cumplida con WO 0142.		X	X		-	15000	-	39166,6	56877	11/02/2020	-	71877	-	-	71877	-	-	-	
		Inspection for Interim Modification and Repair. N/A debido a que la aeronave no tiene la modificación preventiva ni se ha efectuado el reemplazo del frame outboard chord en la estación 727 según el "DENT & BUCKLE CHART".					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
2018-04-06 SB 53-A1240 R2	30-mar-18	<b>DETECT AND CORRECT FATIGUE CRACKING OF THE INTERCOSTALS ON THE FORWARD AND AFT SIDES OF THE FORWARD ENTRY DOOR CUTOUT</b>					-	-	-	-	-	-	-	-	-	-	-	-	-	-	
		Inspection of S-16L door stop intercostals from BS 348.2 to BS 360 & Inspection of S-7L to S-16L intercostals from BS 294.5 to BS 303.9 and S-7L to S-15L intercostals from BS 348.2 to BS 360. Cumplimiento inicial mediante EO 5311-01054 (AGO 31, 2005). Ultimo cumplimiento mediante TASK NO. FAA 2012-12-05.000.00.		X	X		-	6000	-	35808,9	52428	29-abr-14	-	58428	-	-	765	-	-	-	









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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>APU</b>	<b>CSO</b>
				N/A	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING			
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS				
		<b>INSPECTION AND REPETITIVE OVERHAUL TO WING OUTBOARD FLAP TRACK ASSEMBLY</b>	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	
		<b>Wing Outboard Flap Track Pos 1 &amp; 8</b>	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	
		"Do a detailed inspection of the flap track flanges and webs for any crack, any nick, any corrosion pit, any galling, any piece broken off, and any stop-drill repair. Do a surface High Frequency Eddy Current (HFEC) inspection of the flap track flanges and webs for any crack. Do a surface HFEC inspection and an ultrasonic inspection of the upper flap track flange for any crack. Do a detailed inspection of the anti-fret strip and the tapered shim for any damage and any corrosion. Do a detailed inspection of the flap track at the rear spar attachment for any crack, any corrosion and any loose fastener." Cumplimiento Inicial mediante WO 0013 & 0014 de GCA.					-	10000	1460	38945,6	56686	06/sep/2019	-	66686	05-sep-23	-	9023	733				
		Overhaul Pos 1 & 8 Wichever occurs first SEE HT STATUS	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	
		Parag. (h).1: POS 2 & 7 "Do a detailed inspection of the flap track flanges and webs for any crack, any nick, any corrosion pit, any galling, any piece broken off, and any stop-drill repair. Do a surface High Frequency Eddy Current (HFEC) inspection of the flap track flanges and webs for any crack. Do a surface HFEC inspection and an ultrasonic inspection of the upper flap track flange for any crack. Do a detailed inspection of the anti-fret strip and the tapered shim for any damage and any corrosion. Do a detailed inspection of the flap track at the rear spar attachment for any crack, any corrosion and any loose fastener." (Lo que ocurra primero)WO 18027-053043 de CEMA.					-	10000	1460	38937,2	56679	22-jun-18	-	66679	21-jun-22	-	9016	292				
		Overhaul Pos 2 & 7 Wichever occurs first SEE HT STATUS	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	
		Detailed Inspection of BS 616 and BS 639 frames from three fastener rows above S-16 down to the lower end of the frames and stub beam for cracks iaw 3.B.1.b.1. del SB. Cumplimiento previo de acuerdo a la AD 2010-25-06 mediante WO 18027-051184 de CEMA (ABR 04, 2018).					-	9000	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	65679	-	-	8016	-				



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<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<b>TSN</b>	40240,1	<b>CSN</b>	57663
BOEING				<b>TSO</b>	N/A
				<b>CSO</b>	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
2019-03-06 SB 25-1728	29-mar-19	<b>PASSENGER SERVICE UNIT (PSU) AND LIFE VEST PANEL MODIFICATION -</b> At each PSU, install lanyard assemblies IAW SB. <b>Superseded by AD 2020-17-04</b>	X		X		-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
2019-03-24 737-53A-1380 RB	8-abr-19	<b>STRINGERS AND OTHER LONGITUDINAL MEMBERS</b>		X		X	-	4500	-	-	-	-	-	61186	-	-	-	-	3523	-	
		<b>REPETITIVE INSPECTION:</b> Overwing emergency exit doors lower sill splice plate inspection		X		X	-	9000	-	After completing the initial inspection	After completing the initial inspection	After completing the initial inspection	-	-	-	-	-	-	-	-	-
2019-08-02 737-53A-1360	13-jun-19	<b>STATION 277 TO STATION 520 AND STATION 727A TO STATION 1006, BETWEEN STRINGER S-10 AND STRINGER S-17 INSPECTIONS</b> N/A per SB effectivity.					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
2019-08-03 737-53A-1361	17-jun-19	<b>SECTION 43 LOWER LOBE FRAMES FROM STA 380 TO STA 520 INSPECTIONS.</b> N/A per SB effectivity.					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
2019-09-01 737-30A-1064	26-jun-19	<b>ICE AND RAIN PROTECTION - P5 AND P18-3 PANELS CHANGE TO ACTIVATE AUTOMATED PROBE HEAT SYSTEM.</b> Perform the replacement of P5-9 Window and Pitot Heat Module. Global AMOC 790-21-8902	X			X	-	-	730	-	-	-	-	-	-	25-jun-22	-	-	-	-	296
		Do Pitot Static and Temperature Probe Anti-Icing System Test. Applicable to Group 3.		X		X	11000	-	-	-	After completing the replacement of the paragraph (h)	After completing the replacement of the paragraph (h)	After completing the replacement of the paragraph (h)	-	-	-	-	-	-	-	-
2019-10-03 SB-737-531246	11-jul-19	<b>Left and right side-strut support fitting lugs at BS 685</b>					-	-	-	-	-	-	-	-	-	-	-	-	-	-	-
		Detail inspection of the right side-strut support fitting lugs at BS 685	X			X	-	-	365	39273,3	56968	18/oct/2020	-	-	-	-	-	-	-	-	-
		Repeat Detailed inspection of the right side-strut support fitting lugs at BS 685. Cumplido mediante WO 0268 GCA.		X		X	-	4500	730	39273,3	56968	18/oct/2020	-	61468	18-oct-22	-	3805	411			
		Part 3F/11F: Detailed And High Frequency Eddy Current (Hfec) Inspection Of The Left Side-Strut Support Fitting Lugs At Bs 685 With Bushings Removed For Any Corrosion Or Any Crack. (WOF)		X		X	-	30000	3650		After completing the initial inspection of the paragraph (h)	After completing the initial inspection of the paragraph (h)	After completing the initial inspection of the paragraph (h)	-	71.686	11-jul-29	71.686	-	47310		

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<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
Rev. 01		Part 3B/11B: High Frequency Eddy Current (Hfec) Inspection Of The Lugs At Bs 685 On The Side-Strut Support Fitting Of The Left Main Landing Gear For Any Crack.		X		X	-	-	-	-	-	-	-	71.686	11-jul-29	71.686	-	-			
		Part 2B/10B: Determine The Lug Bore Configuration.	X				-	-	-	-	-	-	-	71.686	11-jul-29	71.686	-	-			
		Part 7A/15A: Determine If The Lugs Are Within Acceptable Limits.	X				-	-	-	-	-	-	-	71.686	11-jul-29	71.686	-	-			
		Part 4A/12A: Preventive Modification Of The Lugs At Bs 685 On The Side-Strut Support Fitting Of The Left Main Landing Gear.	X				-	-	-	-	-	-	-	71.686	12-jul-29	71.686	-	-			
2019-12-09	23-jul-19	<b>Conflicting TCAS information on Flight Display System Application FDSA-6500</b> Not applicable due to component installed					-	-	-	-	-	-	-	-	-	-	-	-			
2019-13-02 SB-737-53A1379 RB	23-ago-19	<b>Lower Lobe Section 43, S-14 To S-19 Skin Panel Assembly Replacement</b> N/A per SB effectivity					-	-	-	-	-	-	-	-	-	-	-	-			
2019-13-03	27-ago-19	<b>ATC Transponder System</b> N/A per Part Number Installed					-	-	-	-	-	-	-	-	-	-	-	-			
2019-14-02 SB- 737-29-1127 RB	27-ago-19	<b>Main Landing Gear Wheel Well - Pressure Module Check Valve Replacement</b>	X			X	5360	-	-	-	38945,6	-	-	44305,6	-	-	4065,52	-	-		
2019-14-15 737-53A-1364 R1	19-sep-19	<b>Fuselage - Cracks in the webs of the stub beams</b>	X			X	-	-	-	-	-	-	-	-	-	-	-	-			
		<b>Initial Inspection:</b> High Frequency Eddy Current (HFEC) Inspection for Any Crack in the STA 685 and STA 706 Stub Beams	X			X	-	4500	-	-	-	-	-	-	61186	-	-	3523	-		
		<b>Repetitive Inspection:</b> High Frequency Eddy Current (HFEC) Inspection for Any Crack in the STA 685 and STA 706 Stub Beams (NO CRACK FOUND IN ANY STUB BEAM)		X		X	-	4500	-	After completing the initial inspection	After completing the initial inspection	After completing the initial inspection	-	-	-	-	-	-	-		
		<b>Initial Inspection:</b> General Visual Inspection for Any Existing Repair and High Frequency Eddy Current (HFEC) Inspection for Any Crack in the STA 685 and STA 706 Stub Beam Webs In Repaired Areas	X			X	-	4500	-	-	-	-	-	-	61186	-	-	3523	-		
		<b>Repetitive Inspection:</b> General Visual Inspection for Any Existing Repair and High Frequency Eddy Current (HFEC) Inspection for Any Crack in the STA 685 and STA 706 Stub Beam Webs In Repaired Areas (NO EXISTING REPAIR FOUND)		X		X	-	9000	-	After completing the initial inspection	After completing the initial inspection	After completing the initial inspection	-	-	-	-	-	-	-		
		<b>Initial Inspection:</b> High Frequency Eddy Current (HFEC) and Detailed Inspections for Any Crack in the STA 695 Stub Beam Web	X			X	-	4500	-	-	-	-	-	61186	-	-	3523	-			

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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	
<b>MANUFACTURER</b>	☑	☐	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>APU</b>	<b>CSO</b>
					N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
		<b>Repetitive Inspection:</b> High Frequency Eddy Current (HFEC) and Detailed Inspections for Any Crack in the STA 695 Stub Beam Web		X		X	-	4500	-	After completing the initial inspection	After completing the initial inspection	After completing the initial inspection	-	-	-	-	-	-			
2019-15-07 BASB 737-53A-1362	19-sep-19	<b>Fuselage:</b> Inspection of section 46 Lower Lobe Frames, Station 727A to Station 1006, between S-17L and S-17R N/A per SB Effectivity					-	-	-	-	-	-	-	-	-	-	-	-			
2019-16-12 SB 737-57A-1269	9-oct-19	<b>Wings:</b> Wing Center Section Rear Spar – Replacement of Stiffeners at Left and Right BBL 6.15	X			X	-	4500	-	-	-	-	61190	-	-	-	3527	-			
2019-18-03	3/oct/2019	<b>Exhaust:</b> Alteration of Thrust Reverser upper locking actuators N/A per AD effectivity					-	-	-	-	-	-	-	-	-	-	-	-			
2019-19-10	16/ene/2020	<b>AFM revision for Crossfeed valve check within "Extender Range Opeations".</b> Notified by Engineering Alert A-28-001 and complied by AFM Revision 33 (29-APR-20), Document D6-8734.48E.	X			X	-	-	120	-	-	29/04/2020	-	-	-	-	-	-			
2019-20-03	19/nov/2019	<b>The modification of the USB Receptacles installed as Part of "Provisions for " EFB</b> N/A per the aircraft does not have the EFB system installed					-	-	-	-	-	-	-	-	-	-	-	-			
2019-21-06	13/dic/2019	<b>Ipeco pilot and co-pilot seats</b> N/A per the aircraft does not have Ipeco seats P/N 3A063-0099-01-1 or P/N 3A063-0100-01-1 installed					-	-	-	-	-	-	-	-	-	-	-	-			
2019-23-12 SB 737-57A-1321	13/ene/2020	<b>Fuel Tank Fastener sealant application, Wire bundle Clamp replacement and Teflon Sleeve Installation.</b>		X		X	-	-	1825	-	-	-	-	-	11/01/2025	-	-	1227			
2019-23-14	21-ene-20	<b>Revision del MPM para incorporar la informacion de las sub-secciones C.1, C.2, y C.3 de AWL y CMR</b> Previamente cumplida con la revision 2 del programa de mantenimiento 15-Oct-19)	X			X	-	-	-	-	-	15-oct-19	-	-	-	-	-	-			
2019-23-16 SB 737-57A-1344 RB	3-feb-20	<b>Wings-Plates/Skin - Inspection of Upper Skin at Wing Buttock Line 157 between Stringer 4 and 5</b>		X		X	-	4500	-	-	-	-	-	61369	-	-	3706	-			



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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A
				<b>APU</b>	<b>CSO</b>
				N/A	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
		Do an external surface High Frequency Eddy Current (HFEC) inspection and ultrasonic inspection of the center wing box, STA 663.75 rear spar, lower skin and lower chord for any crack.	X				X	-	9000	-	POR PROGRAMAR	POR PROGRAMAR	POR PROGRAMAR	-	60170	-	-	2507	-		
2021-08-04 SB MXP1/4-35-175 R2, May 10, 2011	24-may-21	Crew Oxygen Mask Stowage Box Units, Inspection/Replace.  AD is not applicable iaw Zodiac Aerospace SB MXP1/4-35-175 R2, May 10, 2011 / Appendix Tables 1 & 2.  Refer to GCA Work Order No. 0497	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	
2021-08-12 SB 737-57A1349	28-may-21	WINGS - LOWER OUTER WING SKIN, LH & RH SIDE - AFT EDGE INSPECTION.  AD 2021-08-12: Par. (g)(1).  SB 737-57A1349 (TABLE 1): Do Eddy Current HFEC inspection of the lower aft wing skin, aft edge at flap tracks numbers 1 to 8 attachment location for any crack.		X			-	4500			PROGRAMAR CONTADOR A PARTIR 28/05/2021	PROGRAMAR CONTADOR A PARTIR 28/05/2021	PROGRAMAR CONTADOR A PARTIR 28/05/2021	N/A	61988	N/A	N/A	4325	N/A		
2021-08-14 MOM-MOM-21-0145-01B(R2)	2-jun-21	A/T BITE TEST LRU interface for validate the airplanes' wiring integrity EO-737-22-001. Cumpido con GCA W.O. 0523.		X		X	-	2000	-	39983,1	57484	27/05/2021	41983,1	-	-	1743,0	-	-			
2021-10-11 SB-737-57A1347	29-jul-21	Parg(h) Initial Inspection: Surface High Frequency Eddy Current (HFEC) inspection in the radius of the Station (STA) 540 bulkhead splice angle for any crack.	X			X	-	5000	-	-	-	-	-	62661	-	-	4998	-			
		Parg(h) Repeat Inspection: Surface High Frequency Eddy Current (HFEC) inspection in the radius of the Station (STA) 540 bulkhead splice angle for any crack.		X		X	-	12000	-	-	-	-	-	69661	-	-	11998	-			



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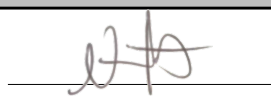
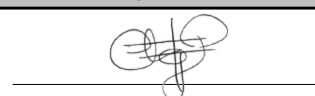
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<b>APPLICABILITY</b>		<b>ENGINE LH</b>		<b>ENGINE RH</b>	
<b>MANUFACTURER</b>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	<b>TSN</b>	<b>CSN</b>	<b>TSO</b>
BOEING			40240,1	57663	N/A

A.D./S.B.	EFFECTIVE DATE	DESCRIPTION & REMARKS	COMPLIANCE				STATE			INTERVAL			LAST DONE			NEXT DUE			REMAINING		
			U	R	C	O	FH	FC	DAYS	TSN	CSN	DATE	TSN	CSN	DATE	FH	FC	DAYS			
AD 2021-13-18 SB 737-71-1911	3-sep-21	Procedures for replacing each affected SDOS actuator with a new SDOS actuator and verifying that safety markers are installed. N/A, LEAP-1B engines applicable for this AD are not effective to GCA B737-400 Fleet.	X				N/A	N/A	N/A				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
AD 2021-14-20 MOM-MOM-21-0292-01B, dated June 23, 2021. AMM Task Card 31-026-01-01	20-jul-21	<b>CABIN ALTITUDE WARNING PRESSURE SWITCH INSPECTION</b>  Within 90 days after the effective date July 20, 2021 or 2,000 FH after the last functional test (or Entry Into Service) whichever occurs later. After, Repetitive interval not to exceed 2,000 FH.  <b>NOTE: Last Done 26/jun/18 "1C" Check.</b>					2000						40937,2				697,1				

FULLFILLMENT		
ENGINEER:  	ENGINEER CHIEF:  	QUALITY CONTROL DIRECTOR:  