MI-35P HELICOPTER



MAIN COMBAT TASKS

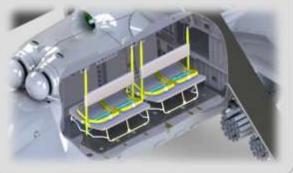


VERSIONS OF APPLICATION

TROOP CARRIER



Airlifting of 8 fully armed troopers in cargo compartment.





CARGO CARRIER

Airlifting of cargo up to 1500 kg inside the cargo compartment and up to 2400 kg on an external sling.

AMBULANCE VERSION



Airlifting of two stretcher cases and two sitting wounded men with one medical attendant.



MI-35P HELICOPTER

Provides for round-the-clock accomplishment of combat tasks using the installed armament in various physiographic regions.

MAIN CHARACTERISTICS

ARMAMENT COMPLEX

PROVISION
OF ROUND-THE-CLOCK
COMBAT APPLICATION

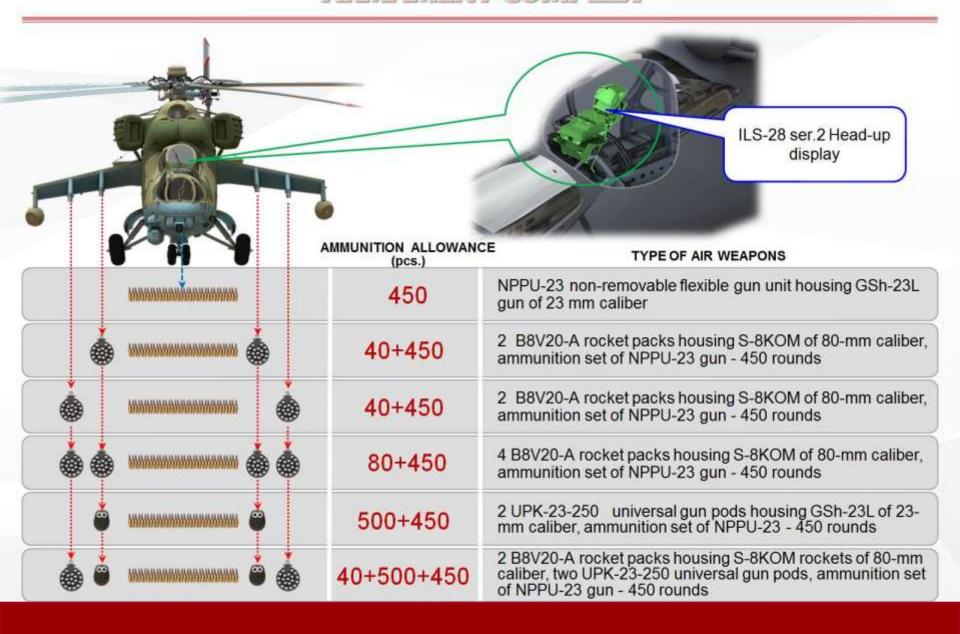
AIRCRAFT COMBAT SURVIVABILITY

AIRCRAFT AND RADIO-ELECTRONIC EQUIPMENT

OPTIONAL EQUIPMENT

MAIN CHARACTERISTICS

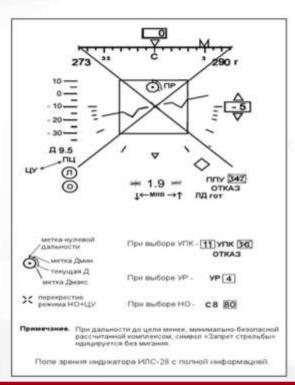
TV3-117VMA Normal take-off weight, kg 11200 Engine type Maximum take-off weight, kg 11500 Take-off power, H.P. 2x2200 2400 3 Cargo weight on the external sling, kg Crew Maximum speed, km/h 320 Cruise speed, km/h 280 17.51 m 1750 Hover ceiling, m Ø 3.9 m Service ceiling, m 4500 450 Operational range, km up to 1000 Ferry range, km Ø 17.3 m 4500 SERVICE CEILING 320 3.97 m MAXIMUM SPEED km/h 4.39 m 450 6.66 m FLIGHT RANGE



ILS-28 ser.2 HEAD-UP DISPLAY

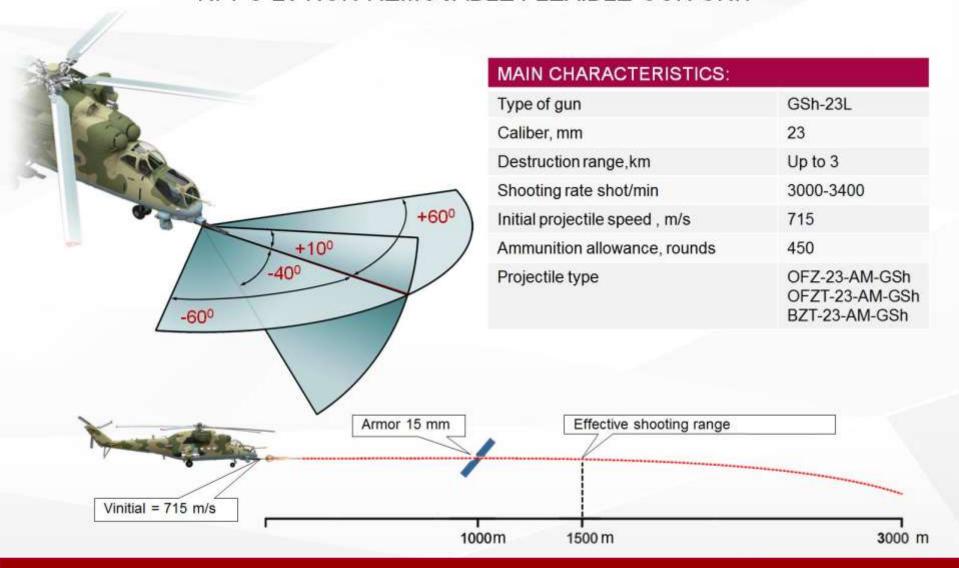
The helicopter is fitted out with ILS-28 ser. 2. head-up display which provides sighting by the pilot.

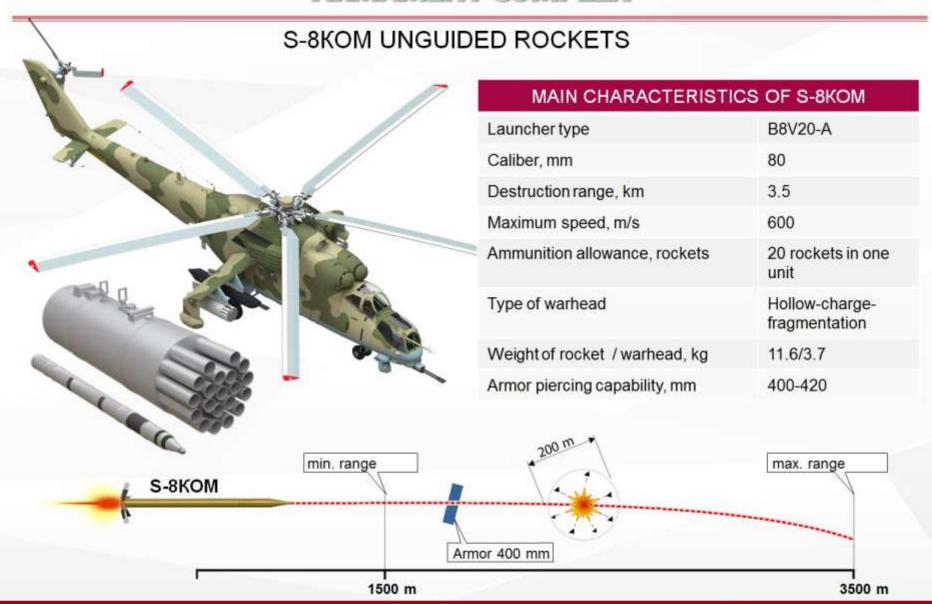
The display is intended for indication of flight and navigation data and sighting data in all modes of flight and combat application of the helicopter.

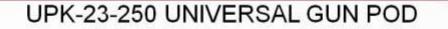




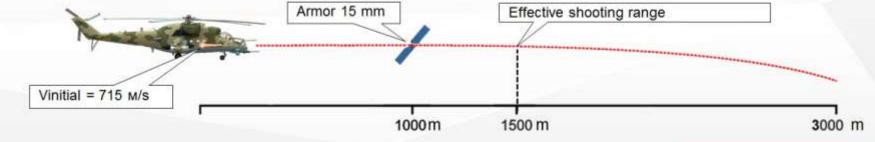
NPPU-23 NON-REMOVABLE FLEXIBLE GUN UNIT











KNEI-24E-1 NAVIGATION AND ELECTRONIC INDICATION COMPLEX

The complex is intended for processing of data coming from aircraft systems and equipment and for output of the following pages on MFD:







4. ADB page

Airborne	Date Rase
Navigation points	Date, time, airfields pressure
Airfields	
Radio beacons	
Memory	Settings
Rosto	Technological page
Write	Read OK
NVG PLT GOES ADD	HSS AW RES

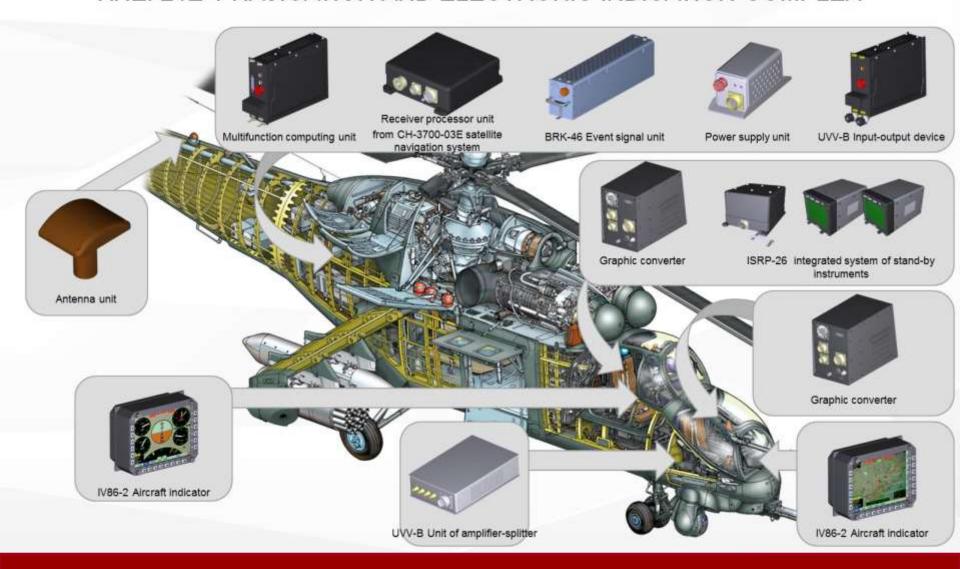
5. TSS page



6. AW page



KNEI-24E-1 NAVIGATION AND ELECTRONIC INDICATION COMPLEX



KNEI-24E-1 NAVIGATION AND ELECTRONIC INDICATION COMPLEX

MAIN CHARACTERISTICS:	
Accuracy of detection of the helicopter location: a) in autonomous reckoning mode, using data coming from Doppler system and attitude and heading reference system 6) in correction mode of GLONASS satellite navigation system (Russia) and/or NAVSTAR (USA) satellite system	not more than 6 % of the covered distance during the first hour of flight not more than 160 m
Time required to be ready to start operation Data programming and storage in nonvolatile memory of the aircraft data base of navigation objects and flight routes.	not more than 3 min provided, database can store up to 8 routes (up to 24 waypoints in each) and 192 navigation objects
Memorization and storage of coordinates of operative waypoints, date, time and waypoints fly-over time in non-volatile memory	provided, data of up to 20 operative waypoints can be stored in the database
Storage of electronic map of location in nonvolatile memory	provided, display module of MFD ensures storage of maps separately for MFD of pilot-in-command and for MFD of operator
Graphic representation of the covered route	provided, with the possibility of scale changing at the background of the electronic map of location or without the map
Indication of data about the failures of the components of the complex and aircraft systems relating to the complex as well as data about aircraft reaching the limits in pitch, roll and flight altitude	provided, data about failures of 8 types of equipment and reaching pitch angle of 30°, roll angle of 15° 45° (maximum value depends on flight altitude) is indicated, as well as aircraft descent to the alert altitude selected at the

indicator of radio altimeter

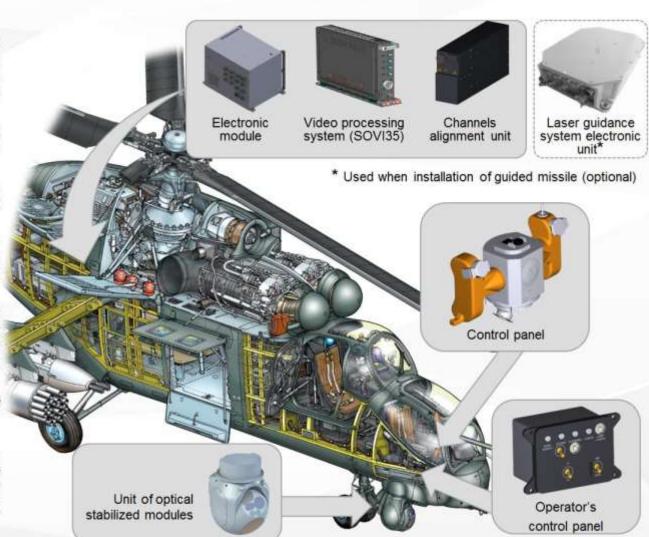
OPS-24N-1L SURVEILANCE SIGHTING SYSTEM

System provides:

- □ Underline surveillance, search, destroy, recognition of different targets and objects, including low speed areal targets round- the-clock in simple and adverse weather conditions:
- Measuring of slant range to the target by the Pilot, Operator and its display to sighting-computing system;
- Manual, automatic and programcorrected target tracking during combat application of armament systems
- Application of unguided rockets, UPK-23-250, NPPU-23 in OPS-0 mode as well as application of NPPU-23 by the Operator in movable mode;
- Output of video signal to MFD as a "OPS" information page containing pointing mark in coincidence with optical axis, IR/TV-channel, laser range finder.

In terms of optional equipment the system provides the following:

- Application of laser guidance system;
- Generation of launch permission signal in PrVK-24-2, output of signals of moving mark angular position on MFD to PrVK-24-2 and Pilot's sight during his operation with guided missile.



STRUCTURAL ELEMENTS OF OPS-24N-1L

STABILIZED OPTIC MODULES BLOCK

The following is used in OPS-24N-L structure: high-sensitive thermal imager operating (matrix 640x512) both night and day time in long-wave band (8+12 mk), fitted out with long-focus compact objective with three fields of view.

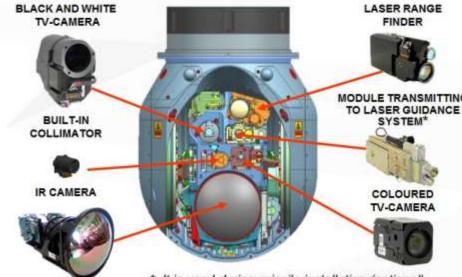
Except the thermal imager a high-definition black and white (matrix 1024x768) and coloured cameras (matrix 1024x768) being a part of the OPS-24N-1L are used for day operations.

A new laser range finder used in OPS-24N-1L allows to measure the distance to the chosen objects and targets with a pinpoint accuracy.

SOVI35 VIDEO PROCESSING SYSTEM

system to enhance black and white / coloured images.





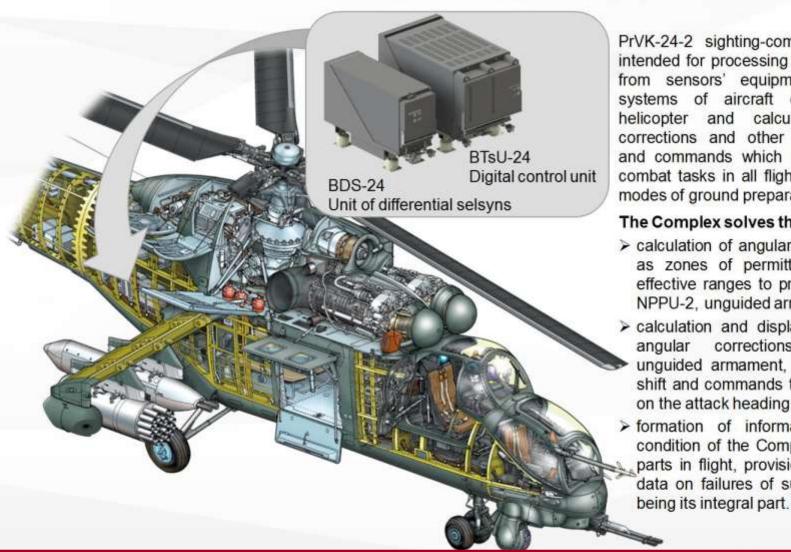
SOVI35 item Изделие as apart of the OPS-24N-11 provides the following:

- Image enhancement due to the signal conditioning in artificial and natural interferences (haze, fog, rain, snow, dust, whiteout, glare) - full-range histogram processing of brightness, maintaining of definite-level brightness (parrying of white,
- > Automation of objects' detection in the field of view with display of potential target markers on indicator's screen to take a decision by the Operator;
- Formation of character-graphic information on the display screen;
- two-contour combined tracking as per frame and object

OPS-24N-1L SURVEILANCE SIGHTING SYSTEM

MAIN CHARACTERISTICS:	
Maximum range of target detection of "T-80 tank in the open" type with the dimensions of (7012x3582x2355) mm in normal conditions, m: a) daytime (for TV channel) b) at night (for IR channel)	8000-10000 8000-10000
Maximum range of target recognition (identification) of «T-80 tank in the open» with the dimensions of(7012x3582x2355) mm in normal conditions, m: a) daytime (for TV channel) 6) at night (for IR channel)	6000-8000 6000-8000
Range of slant ranges to be measured by laser range finder, m	from 300 to 10000
Range of angles of line of sight, degree	In heading \pm 230, in pitch from minus 80 up to \pm 30
Maximum angular speed of control over line of sight, degree/s	from 2,7 to 10,0; minimum 0,02
Field of view of TV channel	wide 10.0°x7,5°, middle 6°x4.5°, narrow 1,4°x1,1°, in scale mode the image is zoomed twice, fourfold increase
Field of view of IR channel	wide 10,0°x8,0°, middle 4,8°x3,8°, narrow 1,5° x 1,2°, in scale mode the image is zoomed twice

PrVK-24-2 SIGHTING-COMPUTING COMPLEX



PrVK-24-2 sighting-computing complex is intended for processing of input information from sensors' equipment and matched systems of aircraft equipment of the helicopter and calculation of ballistic corrections and other output parameters and commands which provide decision of combat tasks in all flight modes as well as modes of ground preparation and control.

The Complex solves the following tasks:

- > calculation of angular corrections as well as zones of permitted, prohibited and effective ranges to provide application of NPPU-2, unguided armament.
- calculation and display for indication of corrections for NPPU-23. unguided armament, helicopter angle of shift and commands to put the helicopter on the attack heading;
- > formation of information on technical condition of the Complex and its integral parts in flight, provision of the crew with data on failures of subsystem and units

ADAPTATION OF INTERNAL AND EXTERNAL LIGHTS FOR APPLICATION OF NIGHT VISION GOGGLES

The helicopter Lights (external and internal lighting equipment) have been modified to exclude optical emission within the waves' band length of 630-900nm for NVG application by the Pilot and Operator.

Lights adaptation provides the helicopter flying day and night at different level of illumination.

The helicopter external Lights include: VPPF-1A landing and search light and FPP-9 taxi light. VPPF-1A light provides optical emission within the following wavelength band:

- Visible (light) "VISIBLE" operation mode;
- ➤ IR "STEALTH" operation mode

FPP-9 light is provided with removable IR light filter.



HELICOPTER COMBAT SURVIVABILITY

ELEMENTS OF ARMOR PROTECTION AND PROTECTED FUEL TANKS

Armor plates of MGB compartment (2 pcs.) and end compartment (cowlings' rear part) (3 pcs.)

Armor plates on engine compartment cowlings (2 pcs.)

Bullet-proof glass (2 pcs.)

Side armor plates of fuselage nose part (2 pcs.)

Suspended armor plates in the cargo compartment to protect troopers (4 pcs.)

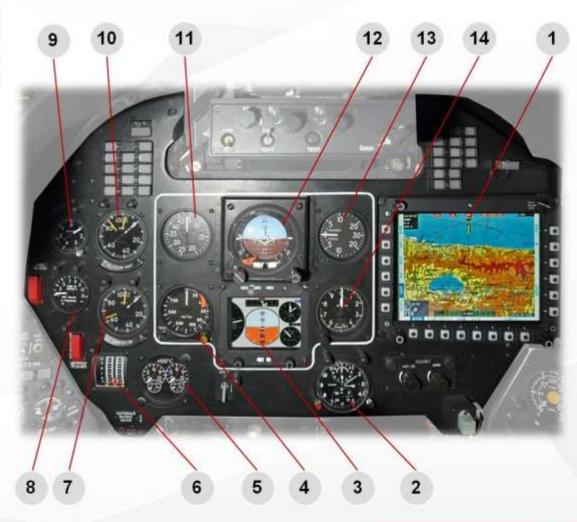
Protected down fuel tanks filled with polyurethane foam

7.62 mm Armor-piercing bullet

PILOT INSTRUMENT PANEL

In order to improve the Pilot's cockpit ergonomics the location of main piloting and navigation instruments, power plant operability instruments and annunciators on the Pilot's instrument panel have been relocated.

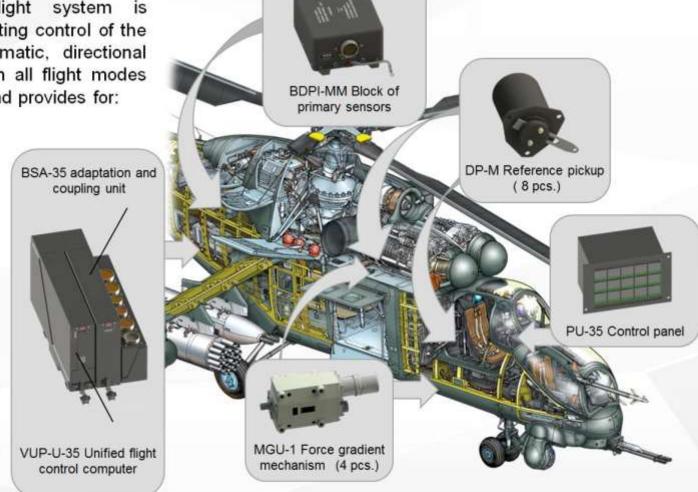
- IV86-2Multifunctional indicator
- AChS-1;K Clock
- Video module VM-26
- Radio altimeter
- 2UT-6K Engine gas temperature indicator
- 6. UR-117V Engines mode indicator
- 7. Main Rotor rpm indicator
- UShV-1K ser. 2 Rotor pitch indicator
- UP-4 Accelerometer
- ITE-2T Engines compressor RPM indicator
- 11. USV-350PZ Speed indicator
- 12. AGR-29RS-15 Gyro horizon
- 13. VR-30PZ-1 Rate-of-climb indicator
- 14. VM-15PZ-1 gyro horizon



PKV-8-35 INTEGRATED FLIGHT SYSTEM

PKV-8-35 integrated flight system is designed for efficient piloting control of the aircraft in manual, automatic, directional and combined controls in all flight modes from take-off to landing and provides for:

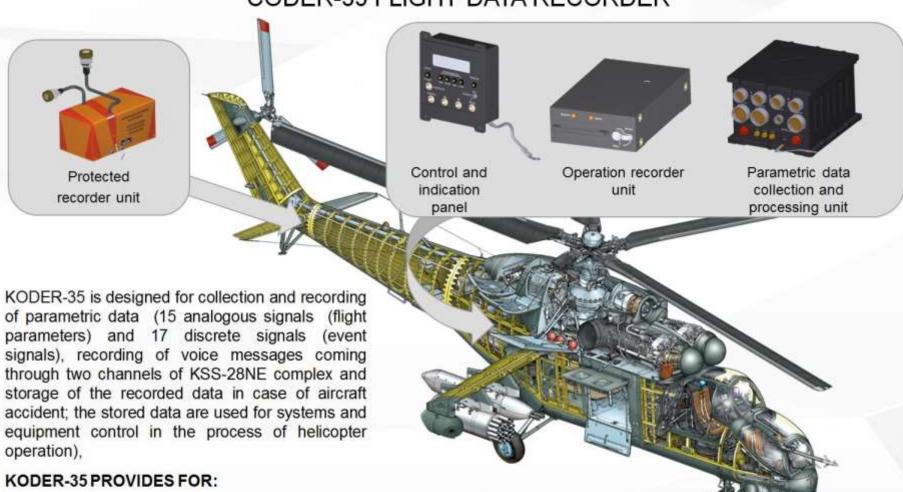
- Enhancement of helicopter controllability and stability;
- Automatization of control of helicopter angular position and attitude in all flight modes:
- Reduction and simplification of pilots actions in manual, automatic, directional and combined piloting controls



PKV-8-35 INTEGRATED FLIGHT SYSTEM

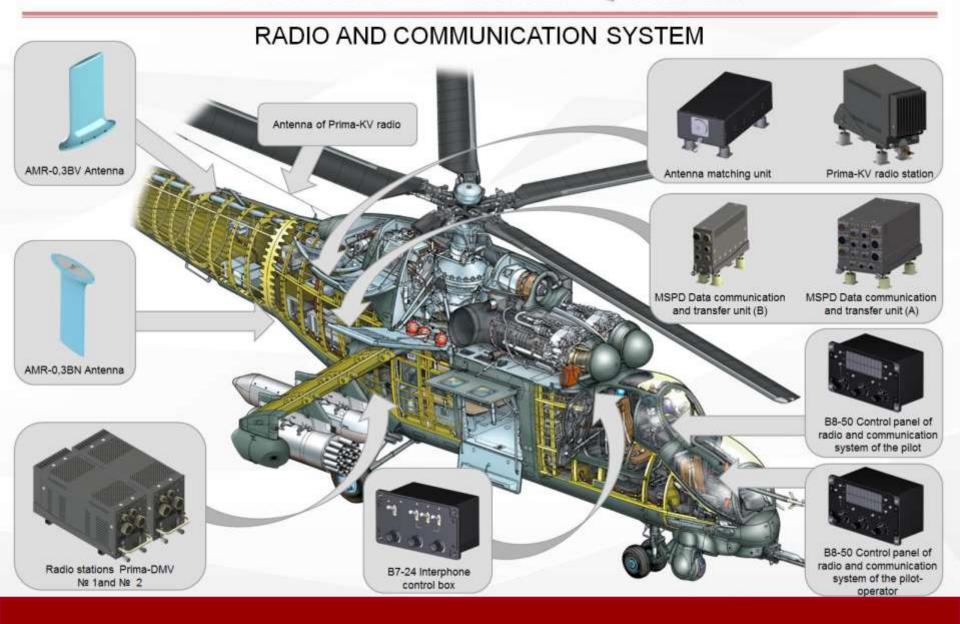
MAIN CHARACTERISTICS:	
Hold error of helicopter flight parameters:	
- yaw angle	±1,5°
- roll angle	±1°
- pitch angle	±1°
- pressure altitude	±20 m
- indicated speed	±10 km/h
- in hovering mode	
by altitude	±1,5 m
in longitudinal and lateral directions	2 m
- in terms of longitudinal and lateral ground speed components set by the pilot in "Low Speed Hold" mode	±0,5 m/s
- in "Route" mode	
·when helicopter is automatically held on the route	±20 m
·when stabilizing the course, manually set by pilot	±2°

CODER-35 FLIGHT DATA RECORDER



Recording of service parameters: current time, built-in-test features, and operation parameters: flight date, flight number helicopter tail number.

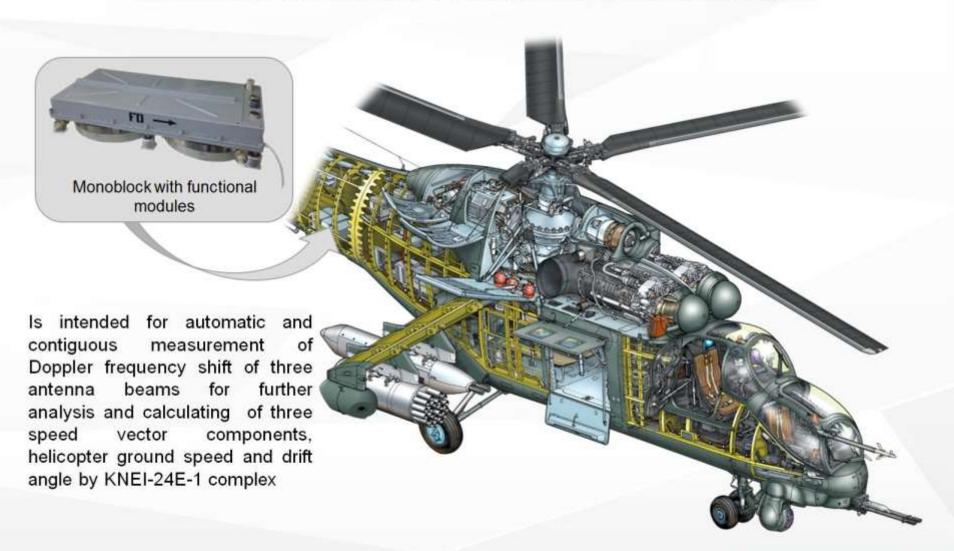
Number of system engagements every time the system was switched on;



RADIO AND COMMUNICATION SYSTEM

MAIN CHARACTERISTICS:	
VHF radio stations PRIMA-DMV provide:	
- receipt and transmission of amplitude-modulated signal within the frequency range of 118,000 - 136,975 MHz with spacing of operational frequency 8,33 kHz and 25 kHz	Emission class A3EJN
 receipt and transmission of frequency and phase modulated signal with spacing of operational frequency 25 kHz within frequency range: 	
• 100,000 - 149,975 MHz • 156,000 - 173,975 MHz • 220,000 - 399,975 MHz	Emission classes F1BBN, F3EJN Emission class G3EJN Emission classes F1BBN, F3EJN
Number of pre-programmed communication channels	40
Time of continuous operation for cycle transmission 1:3 (1 minute- transfer, 3 minutes- receipt), hr	24
Guard channel emergency frequency channel	121,5; 156,8 and 243 MHz
HF radio station PRIMA-KV provide:	
Receipt and transmission of signals within frequency range from 2,000 to 29,9999 MHz with spacing of operational frequency 100 Hz	Emission classes J3E, R3E, H3E/A3E, J3E, F1B, G1B, J7D
Number of pre-programmed communication channels	40
Time of continuous operation for cycle transmission 1:3 (1 minute- transfer, 3 minutes- receipt), hr	12
7 (*)	

DISS-32-28E DOPPLER SPEED AND DRIFT ANGLE SENSOR

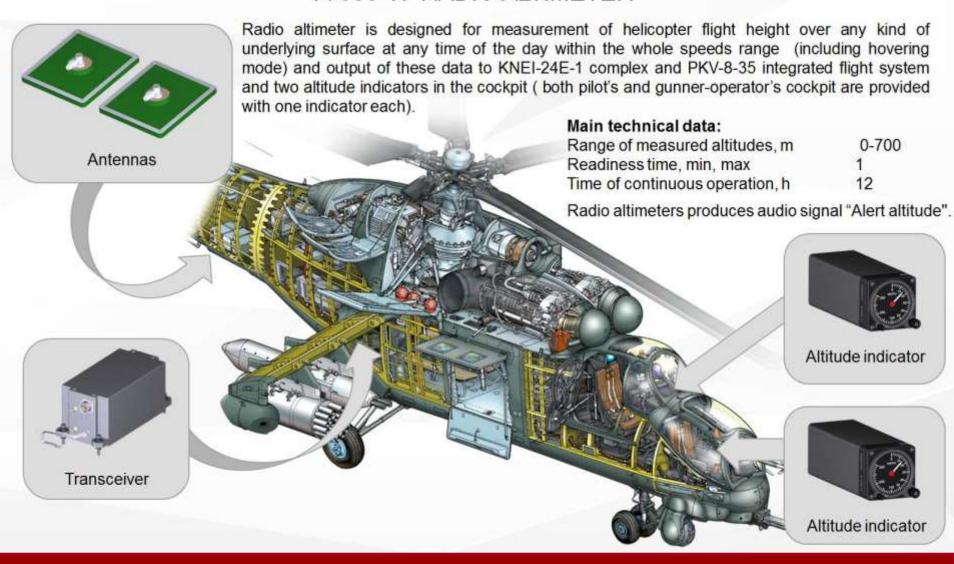


DISS-32-28 DOPPLER SPEED AND DRIFT ANGLE SENSOR

The sensor is operable when flying over any kind of underlying surface (land, sea, sands, ice) regardless of visual conditions and season of the year.

MAIN CHARACTERISTICS :	
Type of emission	Continuous electromagnetic oscillations
frequency of emitted UHF-energy, MHz	13320 ± 3
Operational frequency range, m	2 – 6500
Range of vector components based on measurement of Doppler frequency shift by body-fixed coordinates :	
• longitudinal, km/h • lateral, km/h • vertical, km/h	From minus 200 up to 500 ±120 ±50
Time of continuous operation, h, not more than	6

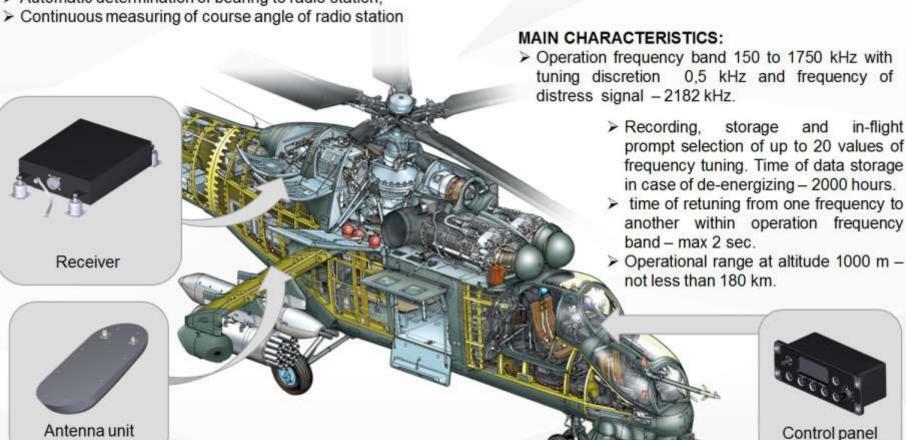
A-053-17 RADIO ALTIMETER



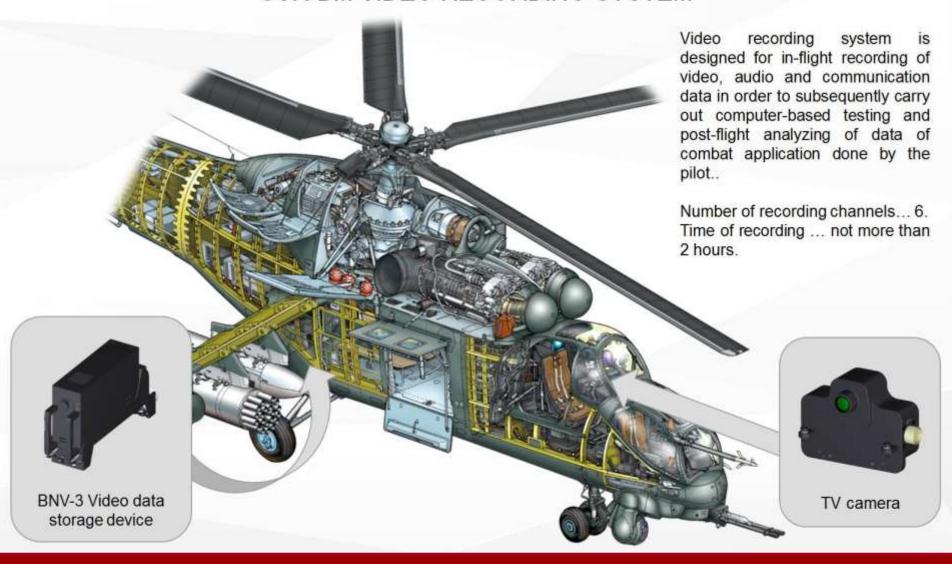
ARK-35-1 AUTOMATIC DIRECTION FINDER

ARK-35-1 automatic direction finder is designed for navigation purposes based broadcasting radio stations and enables solution of the following navigation tasks:

- > Flight to and from radio station and provision of visual indication of course angle;
- Automatic determination of bearing to radio station;



SVR-BM VIDEO RECORDING SYSTEM



GEO-ONV1-01M night vision goggles



UV-26M flare dispenser







Switch box

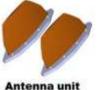


Dispense



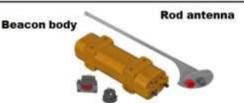
Control box

4280MSE-01 IFF transponder





S406-2NM emergency radio beacon



RT-600 SAR DF-517A (RT-600A NVG)
 Wide-range aircraft direction finder



GEO-ONV1-01M NIGHT VISION GOGGLES

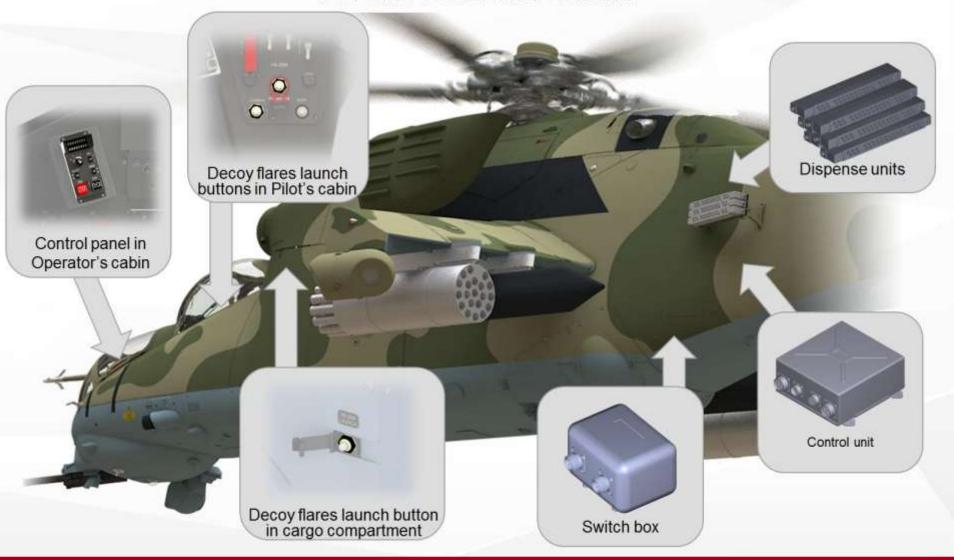
NVG is intended for observing the external environment on terrain when flying the helicopter at night in conditions of natural night illumination from 5x10⁻³ to 1 lx. NVG is a binocular passive optical-electronic device made on III-generation electronic-optical converters.

NVG PROVIDE THE FOLLOWING:



- Helicopter takeoff and landing at night from unequipped and unlit sites;
- Objects detection of power lines masts, tree, armoured vehicle, forest area outline etc.;
- 3 flights at extremely low altitude;
- search of targets and objects on ground and water surface;

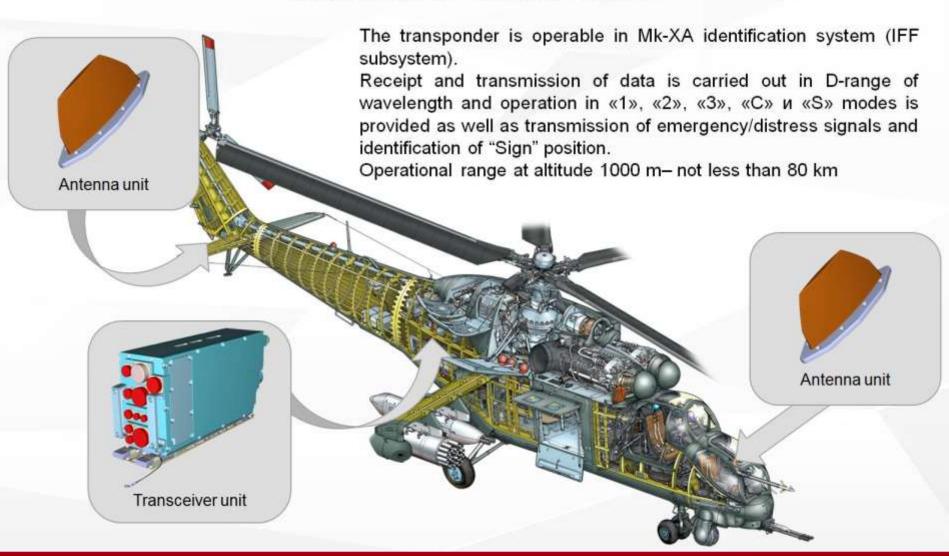
UV-26M FLARE DISPENSER



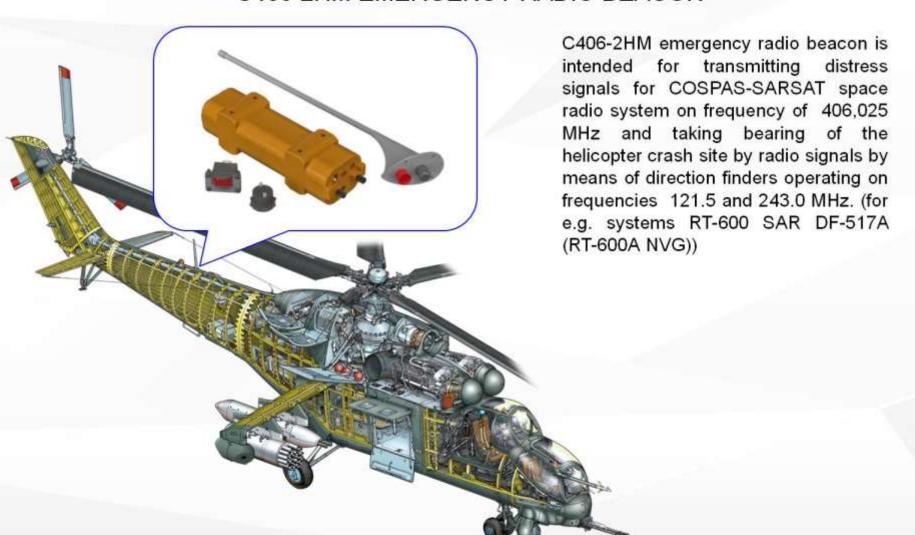
UV-26M FLARE DISPENSER

MAIN CHARACTERISTICS :	
Caliber of loaded flare-cartridges, mm	26
Maximum number of flare-cartridges in dispense unit, pc.	192
Indication of remaining flare-cartridges	Accuracy up to 1
Number of pre-programmed and in-flight selected programmes	8
Time of nonvolatile storage of data about current number of flare-cartridges, days	30
Indication of data of current number of flare-cartridges	On control panel
Time of speeded dispense of the complete set of flare- cartridges, s	Not more than 1,2
Operation mode	manually— selection of dispense programme, dispense from star board or port side is initiated by the operator manually. Selection of dispense programme is done from the control panel by pressing the number of a dispense programme from those which have been preliminary recorded in the memory unit.

4280MSE-01 IFF TRANSPONDER



C406-2HM EMERGENCY RADIO BEACON



RT-600 SAR DF-517A (RT-600A NVG) WIDE-RANGE AIRCRAFT DIRECTION FINDER



UPGRADED MI-35P HELICOPTER IS EQUIPPED WITH ALL THE NECESSARY STATE-OF-THE-ART TECHNICAL AIDS, EQUIPMENT AND SYSTEMS REQUIRED FOR ACCOMPLISHMENT OF ITS MISSIONS DAY AND NIGHT



- NIGHT APPLICATION
- GLASS COCKPIT
- OPTIMIZED WEAPONRY COMPOSITION
- DIGITAL AUTOPILOT SYSTEM
- DIGITAL COMMUNICATION SYSTEM
- > THE BEST CRITERIA "EFFECTIVENESS-COST»