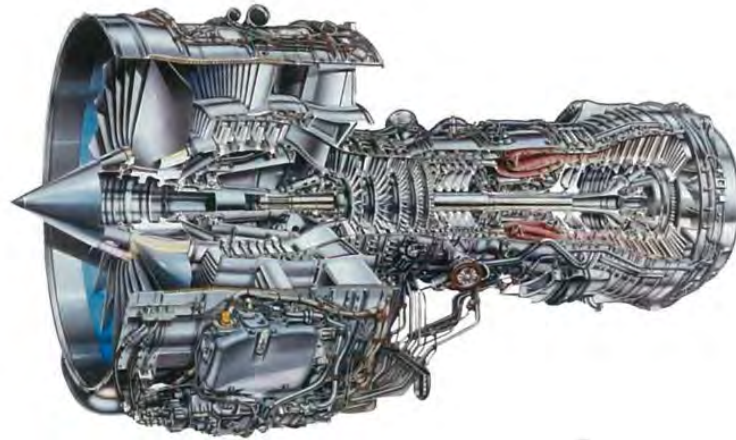




Engine Model: V2533-A5

ESN V17032 Technical Specification

Catalogue

No.	Description	Page No.
01	General Information	3
02	LLP list	4
03	Authorized Release Certificate	6
04	BSI Report	13
05	Last Cell Test	26
06	AD list	31
07	ACC list	43
08	Condition Monitoring Report	52

01 General Information

Engine Manufacturer	<i>International Aero Engines</i>	Status as of	<i>May 2025</i>
Engine Part Number	<i>V2533-A5</i>	TSN	<i>24,228.72</i>
Engine Serial Number	<i>V17032</i>	CSN	<i>8,349</i>
Authorized Thrust	<i>33,000</i>	TSLPR	<i>0</i>
Date of LSV	<i>28 Feb 2025</i>	CSLPR	<i>0</i>
Current Condition	<i>Preservation</i>	LLP Limiter	<i>11,651</i>
Expected delivery date	<i>Jun, 2025</i>	QEC configuration	<i>Full</i>

02 LLP list

Engine: V17032
 Engine Type: V2533-A5 SQ03
 Engine TSN : 24228,72

Customer: FTAI
 CSN: 8349

Signed: _____
 Project: M-0000323950

Date: February 24, 2025

Reference Document: Time Limits Manual (TLM) T-V2500-11A Rev.67. Issue 014, 01 May 2024

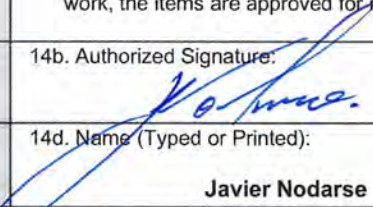
PART	PART NUMBER	S/N	TSN	CSN	CYCLES LIMIT	CYCLES REMAINING	MIN. REMAINING
FAN MODULE							
STAGE 1 FAN DISK	5A1948	RRXP011306	24228,72	8349	20000	11651	11651
FRONT BEARING COMPARTMENT							
STUB SHAFT	5A0895	JJJ539	24228,72	8349	20000	11651	11651
LP COMPRESSOR MODULE							
STAGE 1.5 TO 2.5 DISK	5R0159	JKBM2571	24228,72	8349	20000	11651	11651
HP COMPRESSOR MODULE							
STAGE 3 TO 8 DISK	6B1404	RRD12496	6704	3268	20000	16732	11651
STAGE 9 TO 12 DISK	6A7546	RRD7601	24228,72	8349	20000	11651	
SHAFT, ASSY REAR-HPC	6B1419	200581CQYX8163	24228,72	8349	20000	11651	
REAR ROTATING SEAL	6A5869	RFRQDY3450	24228,72	8349	20000	11651	
HP TURBINE MODULE							
STAGE 2 HP TURBINE AIR SEAL	2A4157	PKLBL26102	24228,72	8349	20000	11651	11651
STAGE 1 TURBINE HUB	2A5001	PKLBRB9243	6704	3268	20000	16732	
STAGE 1 HP TURBINE OUTER AIR SEAL	2A3423	PKLBL73496	24228,72	8349	20000	11651	
SEAL, AIR HPT 1STG	2A3923	PKLBL25622	24228,72	8349	20000	11651	
STAGE 2 TURBINE HUB	2A4802	PKLBTC9974	0	0	20000	20000	
STAGE 2 HP TURBINE BLADE RETAINING PLATE	2A3437	PKLBMC9299	24228,72	8349	20000	11651	
LP TURBINE ROTOR							
STAGE 3 TURBINE DISK	3A2513	JM5754D	24228,72	8349	20000	11651	11651
STAGE 4 TURBINE DISK	3A1984	MAL20671	24228,72	8349	20000	11651	
STAGE 5 TURBINE DISK	3A2430	MAL06241	24228,72	8349	20000	11651	
STAGE 6 TURBINE DISK	3A2996	MAK97375	24228,72	8349	20000	11651	
STAGE 7 TURBINE DISK	3A2514	JM5995D	24228,72	8349	20000	11651	
STAGE 3 TURBINE AIR SEAL	3A0963	MAA79907	24228,72	8349	20000	11651	
STAGE 4 TURBINE AIR SEAL	3A2422	MAL15573	24228,72	8349	20000	11651	
STAGE 5 TURBINE AIR SEAL	3A2423	MAA78276	24228,72	8349	20000	11651	
STAGE 6 TURBINE INNER AIR SEAL	3A3047	MAA78655	24228,72	8349	20000	11651	
STAGE 6 TURBINE AIR SEAL	3A1988	MAK98920	24228,72	8349	20000	11651	
STAGE 7 TURBINE AIR SEAL	3A2522	MAA80595	24228,72	8349	20000	11651	
LP TURBINE SHAFT	5A1762	JJDR892	24228,72	8349	20000	11651	

ENGINE MINIMUM REMAINING **11651**

Revision	Date	Reason
R00	February 24, 2025	Original document

03Authorized Release Certificate

1. Approving Competent Authority / Country AESA / SPAIN		2. AUTHORISED RELEASE CERTIFICATE EASA FORM 1			3. Form Tracking Number. MT/25855181		
4. Organization Name and Address: IBERIA MAINTENANCE		Adolfo Suarez Madrid - Barajas Intl. Airport - Zona Industrial No. 2 www.iberiamaintenance.com 28042 Madrid - Spain			5. Work Order / Contract / Invoice M-0000323950		
6. Item	7. Description	8. Part No.	9. Qty.	10. Serial No.	11. Status / Work		
1	ENGINE V2500	V2533-A5	1	V17032	REPAIRED		
12. Remarks Engine repaired iaw Engine Manual E-V2500-1IA Rev.137 (01/NOV/2024) and Tested iaw Engine Manual E-V2500-1IA Rev.137 (01/NOV/2024). // TSN=24228,72 and CSN=8349. EPR MOD=07-02 // N2K at 1.4 EPR=13586rpm // Variant=06 // EEC Software=SCN23/AC // FMU P/N 8061-640 switched HI (IAN SQ03). Main line bearings preservation and fuel system inhibition iaw Engine Manual E-V2500-1IA Rev.137 (01/NOV/2024). Up to 12 months preserved in an uncontrolled environment iaw EM Task 72-00-00-500-000 (Starting from the certified date specified in block 14e.). Works performed iaw Workscope V17032-20241212 Rev.01 (29/JAN/2025). *** See AD's Status, LLP's status, MPL Status, Deviation List and Carry Forward Sheet. *** The work identified in Block 11 and described herein has been accomplished in accordance with 14 CFR part 43 and in respect to that work, the items are approved for return to service under certificate No. ILAY037F. Records of works performed are on file at this maintenance organization under Work Order stated in block 5.							
13a. Certifies that the items identified above were manufactured in conformity to: ☐ approved design data and are in condition for safe operation ☐ non-approved design data specified in block 12		14a. ☒ Part 145.A.50 Release to Service ☒ Other regulation specified in block 12 Certifies that unless otherwise specified in Block 12, the work identified in block 11 and described in Block 12 was accomplished in accordance with part 145 and in respect to that work the items are considered ready for release to service.					
13b. Authorised Signature 		13d. Approval/Authorisation Number 		14b. Authorised Signature 		14c. Certificate/Approval Ref. No. ES.145.011	
13c. Name: 		13e. Date (dd/mmm/yyyy): 		14d. Name: J. A. LEONOR		14e. Date (dd/mmm/yyyy): 28/FEB/2025	
USER/INSTALLER RESPONSIBILITIES ☒ This certificate does not automatically constitute authority to install the item(s). ☒ Where the user/installer work in accordance with regulations of an airworthiness authority different than the airworthiness authority specified in block 1, it is essential that the user/installer ensures that this his/her airworthiness authority accepts items from the airworthiness authority specified in block 1. ☒ Statements in blocks 13a and 14a do not constitute installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.							

1. Approving Civil Aviation Authority/Country. FAA/United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: V1490	
4. Organization Name and Address: Aviation Precision Technology, Inc. 10820 NW 138TH ST BAY 2 HIALEAH GARDENS, FL 33018 8APR427B / 145.6536					5. Work Order/Contract/Invoice Number: 5884	
						
6. Item	7. Description	8. Part Number	9. Quantity	10. Serial number	11. Status/Work	
1	Turbofan Engine	V2533-A5	1	V17032	REPAIRED	
12. Remarks: Engine S/N V17032 Was Disassembled To The Extent Necessary To Correct LPC, HPC, Hot Section and HPT Module Discrepancies. Inspected, Repaired, and Reassembled IAW IAE E-V2500-11A Engine Manual Issue: 043 Dated Aug 01, 2024. Engine Released Pending Final Performance Test To Be Accomplished By Third-Party. Refere To Aviation Precision Technology, Inc AD and SB Summary List For History. ETT: 24,228:72 ETC: 8,349 TSLV: 00:00 CSOVH: 00:00 Time And Cycles Supplied By Customer						
						
All Pertinent details Of Work Performed Are On File At Aviation PrecisionTechnology, Inc Under W.O 5884						
Aviation Precision Technology, Inc., certifies that the work specified in Block 11/12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 acceptance certificate No. EASA. 145.6536						
13a. Certifies the items identified above were manufactured in conformity to: N/A Approved design data and are in a condition for safe operation. N/A Non-approved design data specified in Block 12.			14a. ☒ 14 CFR 43.9 Return to Service ☒ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service			
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:
N/A		N/A		 		8APR427B
13d. Name (Typed or Printed):		13e. Date (dd/mm/yyyy)		14d. Name (Typed or Printed):		14e. Date (dd/mm/yyyy):
N/A		N/A		Javier Nodarse		17/AUG/2024
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						



US Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 07/31/2026

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark	Serial No.	
	Make	Model	Series
2. Owner	Name (As shown on registration certificate)	Address (As shown on registration certificate)	
		Address _____ City _____ State _____ Zip _____ Country _____	

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☐	AIRFRAME	_____	(As described in Item 1 above)	_____
☒	☐	POWERPLANT	I.A.E TURBOFAN ENGINE	V2533-A5	V17032
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type _____ Manufacturer _____		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency	
Name	Aviation Precision Technology, Inc.	☐ U. S. Certified Mechanic	☐ Manufacturer
Address	10820 NW 138th Street, Bay 2	☐ Foreign Certified Mechanic	C. Certificate No.
City	Hialeah Gardens State Florida	☒ Certified Repair Station	8APR427B
Zip	33018 Country USA	☐ Certified Maintenance Organization	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐	Signature/Date of Authorized Individual <i>J. K. [Signature]</i> 08-17-2024.
--	---

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ Approved ☐ Rejected

BY	FAA Fit. Standards Inspector	Manufacturer	Maintenance Organization	Persons Approved by Canadian Department of Transport
	FAA Designee ☒	Repair Station	Inspection Authorization	Other (Specify)

Certificate or Designation No. 8APR427B	Signature/Date of Authorized Individual <i>J. K. [Signature]</i> 08-17-2024.
---	---

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

_____	08/17/2024
Nationality and Registration Mark	Date

Work Order:

Engine Model: V2533-A5

Engine Serial Number: V17032

Engine Total Time: 24,228:72

Engine Total Cycle: 8,349

Engine Data Plate:

Take-Off Rating-Bump: 31600 IDEAL 2-00

Variant: 06

EPR Mod: 08-00

Customer: Chromalloy Material Solutions.

Engine S/N V17032 was disassembled to the extent necessary to repair LPC, HPC, Hot Section, NGV's, HPT modules discrepancies.

Engine was disassembled, inspected, repaired and reassembled per IAE E-V2500-11A Engine Manual P/N 2A4406-01 Issue # 043 Dated Aug. 1, 2024.

Engine performance test 10 awaiting on accomplishment.

Refer to Aviation Precision technology, Inc. for engine AD Summary.

Refer to Aviation Precision technology, Inc. for engine SB Summary.

The following was accomplished:

1. Remove fan disk to facilitate repair on fan speed sensors and the electrical harness. Removed and replaced fan blades with OH condition fan blades. Inspect exposed areas and removed parts. Performed repairs VRS1154 and VRS1164.
Reference # S210993.
2. Remove LPC module to facilitate repair on fan speed sensors and the electrical harness. Inspect LPC module accordingly on exposed areas as an assembled module. Remove and replace LPC STG 1.5 stator vane assembly with O/H condition stator.
3. Remove FBC module to facilitate repair on fan speed sensors and the electrical harness. Inspect FBC module accordingly on exposed areas as an assemble module. Remove and replace engine fan speed balance probes Qty 3ea and the speed electrical harness with serviceable condition probes and harness.

☒ Additional Sheets Are Attached

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

_____	_____
Nationality and Registration Mark	08/17/2024 Date

4. Inspect Fan Case in-situ. Remove and replace 2ea rubber panel seal damage with panels in serviceable condition. Perform repairs VRS1822 acoustic lining filler sealant and VRS1819 attrition lining filler sealant.
5. Removed and inspect FEGV, front panel liners, rear fairing, splitter fairing to facilitate repair on fan speed Sensor and electrical harness. Removed and replace 2ea rubber panels seal with panels in serviceable condition. perform repair VRS1822 acoustic lining filler and VRS1819 attrition lining filler. Replaced 1ea bleed valve solenoid stg 10 with 1ea in SV condition solenoid that failed func. test-069.
6. High pressure comp. module was removed and disassembled to replaced drums 3-8 to comply with AD 2015-10-04 silver free nut modification. Replace 3-8 and 9-12 drums with OH condition drums. Front split cases were replaced with OH condition cases. Replace rear cases rotor path 7-12 with OH condition cases. Performed re-lubed and torque check on VSV's System. Performed 3-8 and 9-12 drum assembly, module grind and dynamic balance by third-party.
Replaced 3ea bleed valve STG 7 that failed func. test-066 with 2ea SV and 1ea OH condition valve. Replaced 1ea bleed valve STG 10 that failed func. test-068 with 1ea OH condition valve.
7. Diffuser was removed to facilitate HPC repair. Visual inspection performed as exposed. No discrepancies were noted. Performed diffuser M flange FPI per SB 72-A0706. Fuel nozzles were removed and replaced with OH condition fuel nozzles.
Reference: SO # 16655475.
8. Combustor module was removed accordingly. Inner comb. liners segment were removed and replace with OH condition and new condition liners per (VRS3173). Outer combustor chamber was replace with new outer combustor in overhaul condition.
9. No. 4 BRG compartment was removed to facilitate HPC repair. Visual & dimensional along with NDT inspection performed as exposed. Remove and replaced # 4 BRG seal face with OH condition seal. Vacuum Check Performed for Leaks (Pass Test).
10. NGV's Was Removed and Disassembled to Perform OH. Replaced 40ea Vanes With OH Condition Vanes. Replace Crack HPT STG 1 Support with SV Condition Support. Performed Visual & Dimensional Along With NDT Inspection to the Remaining Parts.
11. HPT was removed and disassembled to performed gas path restoration. Replaced STG 1 hub with OH condition and STG 2 hub with new condition hubs.
Replaced 64ea Stage 1 Blades With OH Condition Blades. Replaced 38ea Stage 1 Duct Segment With OH Condition Ducts Segment.
Replaced 72ea Stage 2 Blades With OH Condition Blades. Replaced 38ea STG 2 Duct Segment With OH Condition Ducts Segment.
Performed Static Balance on Stage 1 and Stage 2 Hub in Assembly Condition.
Replaced 19ea STG 2 Vanes Clusters with OH Condition Vanes. Performed Visual & Dimensional Along With NDT Inspection to the Remaining Parts.

☒ Additional Sheets Are Attached

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

_____	_____
Nationality and Registration Mark	08/17/2024 Date

12. LPT module was removed to facilitate discrepancies repairs on the HPC. Visual Inspection Performed as Exposed. No Discrepancies Were Noted.
13. TEC module was Visually Inspected In-situ as Exposed. No Discrepancies Were Noted. Vacuum Check Performed for Leaks on No.5 BRG. (Pass Test). Removed and Replaced 4ea Thermocouple Probes With OH Condition Probes.
14. EGB module was Visually Inspected In-situ as Exposed. No Discrepancies Were Noted.
15. Engine mounts were removed, disassembled, inspected and NDT to perform overhaul on FWD and RWD mounts.
16. Oil tank shipped empty.

☐ Additional Sheets Are Attached

04 BSI Report

S/N: V17032

CUSTOMER: FTAI

O/T: 41288721

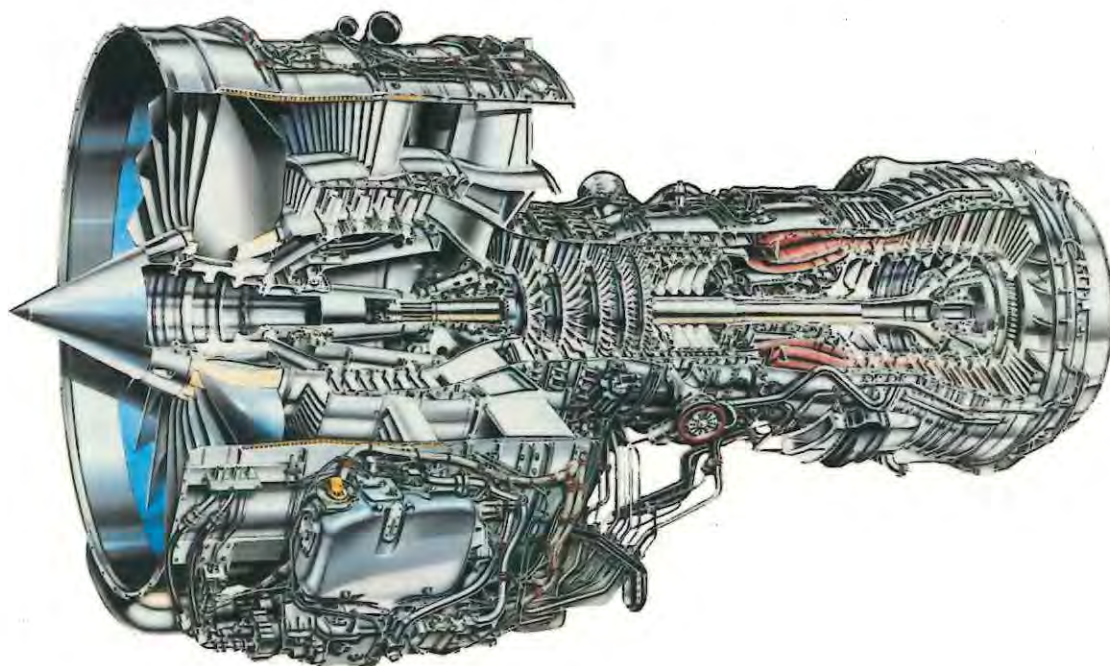
INSPECTED IN ACCORDANCE WITH A.M.M. LEASE, REV 15, Feb 01/24

INFORMATION OF THE BORESCOPE USED

Mentor Visual iQ Control Unit S/N: 1913A5002

Probe S/N: 1906A1096

Borescope Inspection Report Post Test Cell



Engine Type IAE V2500A5
SelectOne

BORESCOPE INSPECTION CONDITION

SERVICEABLE

X

NO SERVICEABLE

DECISION PENDING

ADDITIONAL REQUIREMENTS:

STAMP



S/N: V17032

CUSTOMER: FTAI

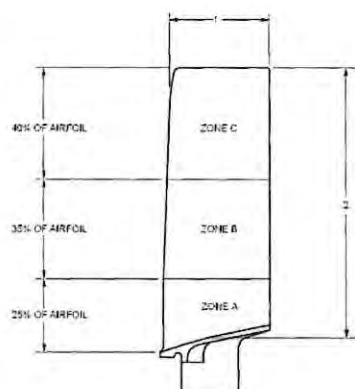
O/T: 41288721

INSPECTED IN ACCORDANCE WITH A.M.M. LEASE, REV 15, Feb 01/24

TASK 72-00-00-200-017-B Inspection of the LP Compressor

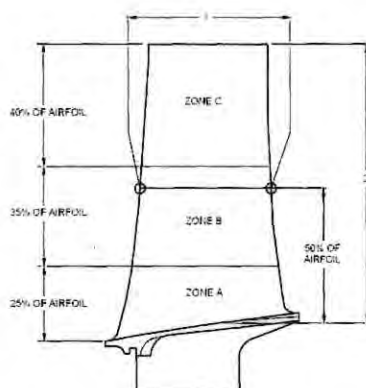
INSPECTION RESULTS

AREA	ACCESS PORT	STAGE	Nº OF BLADES	STAGE CONDITION	INSP. #
LP COMP RESSOR	INLET GUIDE VANES	1.5 L.E.	58	No damage found.	51173
	BLEED VALVE	2.5 L.E.	72	No damage found.	51173



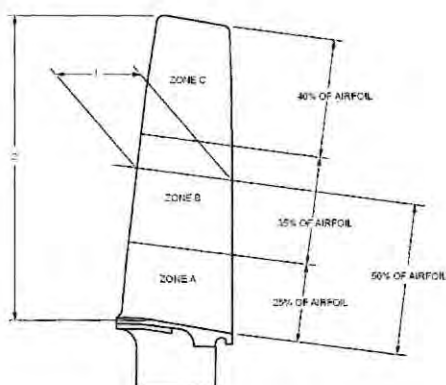
STAGE	APPROXIMATE BLADE HEIGHT (MEASURED IN A STRAIGHT LINE AT VERTICAL CENTER LINE)		APPROXIMATE BLADE WIDTH (MEASURED IN A STRAIGHT LINE AT AIRFOIL MIDDLE HEIGHT POSITION)			
	REF 2		REF 1			
	in	mm	TRUE WIDTH *		PROJECTION *	
1.5	3.75	64.0	1.52	38.5	—	—
2.5	—	—	—	—	—	—

* THE TRUE WIDTH IS MEASURED ALONG THE CHORD LINE AT MIDDLE AIRFOIL HEIGHT. THE PROJECTION IS THE APPARENT BLADE WIDTH WHICH WE CAN SEE FROM THE BORESCOPE POSITION AND SHOWS THE PROJECTION OF THE BLADE AIRFOIL.



STAGE	APPROXIMATE BLADE HEIGHT (MEASURED IN A STRAIGHT LINE AT VERTICAL CENTER LINE)		APPROXIMATE BLADE WIDTH (MEASURED IN A STRAIGHT LINE AT AIRFOIL MIDDLE HEIGHT POSITION)			
	REF 2		REF 1			
	in	mm	TRUE WIDTH *		PROJECTION *	
2.0	3.05	76.3	1.20	36.5	—	—

* THE TRUE WIDTH IS MEASURED ALONG THE CHORD LINE AT MIDDLE AIRFOIL HEIGHT. THE PROJECTION IS THE APPARENT BLADE WIDTH WHICH WE CAN SEE FROM THE BORESCOPE POSITION AND SHOWS THE PROJECTION OF THE BLADE AIRFOIL.



STAGE	APPROXIMATE BLADE HEIGHT (MEASURED IN A STRAIGHT LINE AT VERTICAL CENTER LINE)		APPROXIMATE BLADE WIDTH (MEASURED IN A STRAIGHT LINE AT AIRFOIL MIDDLE HEIGHT POSITION)			
	REF 2		REF 1			
	in	mm	TRUE WIDTH *		PROJECTION *	
2.3	2.95	74.9	1.15	29.0	—	—

* THE TRUE WIDTH IS MEASURED ALONG THE CHORD LINE AT MIDDLE AIRFOIL HEIGHT. THE PROJECTION IS THE APPARENT BLADE WIDTH WHICH WE CAN SEE FROM THE BORESCOPE POSITION AND SHOWS THE PROJECTION OF THE BLADE AIRFOIL.



STAGE	APPROXIMATE BLADE HEIGHT (MEASURED IN A STRAIGHT LINE AT VERTICAL CENTER LINE)		APPROXIMATE BLADE WIDTH (MEASURED IN A STRAIGHT LINE AT AIRFOIL MIDDLE HEIGHT POSITION)			
	REF 2		REF 1			
	in	mm	TRUE WIDTH *		PROJECTION *	
2.5	3.04	77.3	1.45	36.8	—	—

* THE TRUE WIDTH IS MEASURED ALONG THE CHORD LINE AT MIDDLE AIRFOIL HEIGHT. THE PROJECTION IS THE APPARENT BLADE WIDTH WHICH WE CAN SEE FROM THE BORESCOPE POSITION AND SHOWS THE PROJECTION OF THE BLADE AIRFOIL.

S/N: V17032

CUSTOMER: FTAI

O/T: 41288721

INSPECTED IN ACCORDANCE WITH A.M.M. LEASE, REV 15, Feb 01/24

TASK 72-00-00-200-016-B01 Inspection of the HP Compressor

INSPECTION RESULTS

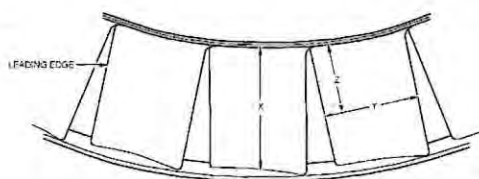
AREA	ACCESS PORT	STAGE	Nº OF BLADES	STAGE CONDITION		INSPEC.
H.P.C.	A & B	3 LE & TE	31	L.E: Some blades with small dents to small to be measured within limits. T.E: No damage found.		689146 -1
	A & B	Stage 3 Rotor Clapper TASK 72-00-00-210-012-A.	31	No damage found.		689146 -1
	B	4 LE	38	Some blades with a very small dents too small to be measured within limits.		689146 -1
	C	5 TE	64	No damage found.		689146 -1
	C	6 LE	79	Some blades with a small dents all within limits.		689146 -1
	C	STG 6 Rotor Path Coating		No damage found.		689146 -1
	D	7 TE	93	Some blades with very small dent too small to be measured within limits.		689146 -1
	D & E	8 LE & TE	84	L.E: Multiples blades with small dents all within limits. T..E: No damage found.		689146 -1
	D & E	STG 6; 7 & 8 Front and rear damper wire.		No damage found.		689146 -1
	E & F	9 LE & TE	89	L.E: Some blades with small dents too small to be measured all within limits. T.E: No significant damage found.		689146 -1
	F	10 LE	85	No significant damage found.		689146 -1
	G	11 TE	78	Some blades with small dents too small to be measures all within limits.		689146 -1
	G	12 LE	71	No damage found.		689146 -1
	Compressor Rotor Drum Stage 6 to 11 Stator Paths			Individual	Total	INSPE C.
	C	6 STATOR PATH CERAMIC COATING		No damage found.	No damage found.	689146 -1
	D	7 STATOR PATH CERAMIC COATING		5% Rubs and no damage found.		
	E	8 STATOR PATH CERAMIC COATING		No damage found.		
	F	9 STATOR PATH CERAMIC COATING		1% Rubs and no damage found.		
	F & G	10 STATOR PATH CERAMIC COATING		No damage found.		
	G	11 STATOR PATH CERAMIC COATING		10% Rubs and no damage found.		

S/N: V17032

CUSTOMER: FTAI

O/T: 41288721

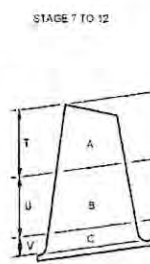
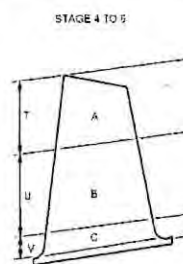
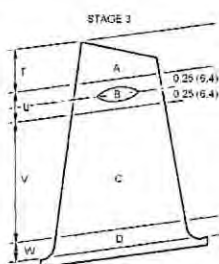
INSPECTED IN ACCORDANCE WITH A.M.M. LEASE, REV 15, Feb 01/24



VIEW ON THE
HP COMPRESSOR BLADES
WHEN YOU LOOK
FROM THE REAR OF THE ENGINE

STAGE	QUANTITY	DIMENSION X		DIMENSION Y		DIMENSION Z	
		in	mm	in	mm	in	mm
3	31	5.04	128.2	2.39	60.7	2.54	64.6
4	38	3.66	93.1	2.20	55.8	1.73	44.0
5	64	2.62	66.5	1.57	39.8	1.32	33.6
6	79	1.90	48.3	1.15	29.3	0.97	24.6
7	93	1.45	36.9	0.92	23.4	0.74	18.8
8	84	1.14	29.0	0.91	23.2	0.59	14.9
9	89	0.94	23.8	0.86	21.8	0.48	12.3
10	85	0.85	22.0	0.85	21.6	0.45	11.4
11	76	0.83	21.0	0.89	22.5	0.43	10.9
12	71	0.84	21.2	0.86	21.8	0.44	11.1

NOTE: THESE DIMENSIONS ARE FOR REFERENCE PURPOSES ONLY THE CHORD DIMENSION CHANGES FROM THE BLADE ROOT TO THE BLADE TIP THE CHORD DATUM (DIMENSION Y) IS AT DIMENSION Z MEASURED FROM THE BLADE ROOT DIMENSION X IS MEASURED AT THE MID CHORD POSITION



BORESCOPE DAMAGE ZONES

DIMENSION	STAGE										
	3	4	5	6	7	8	9	10	11	12	
T	R	S	S	S	S	S	S	S	S	S	
U	0.5 (13)	R	R	R	R	R	R	R	R	R	
V	R	0.5 (13)	0.5 (13)	0.5 (13)	0.25 (6.4)	0.25 (6.4)	0.25 (6.4)	0.25 (6.4)	0.25 (6.4)	0.25 (6.4)	
W	0.5 (13)	—	—	—	—	—	—	—	—	—	

NOTE: 1 ALL DIMENSIONS ARE IN INCHES (mm)
2 REMAINING HEIGHT = R
3 30% OF AIRFOIL HEIGHT = S

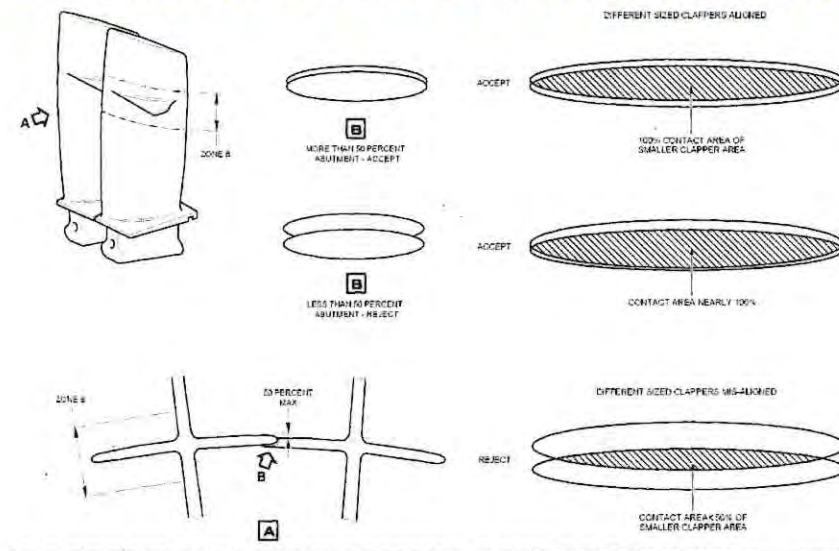
S/N: V17032

CUSTOMER: FTAI

O/T: 41288721

INSPECTED IN ACCORDANCE WITH A.M.M. LEASE, REV 15, Feb 01/24

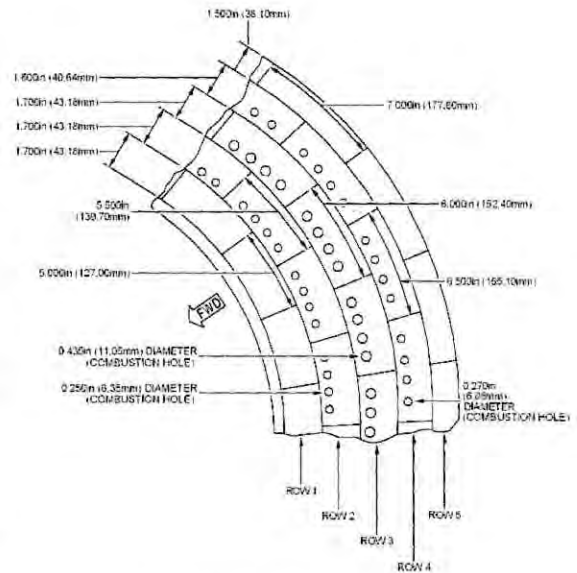
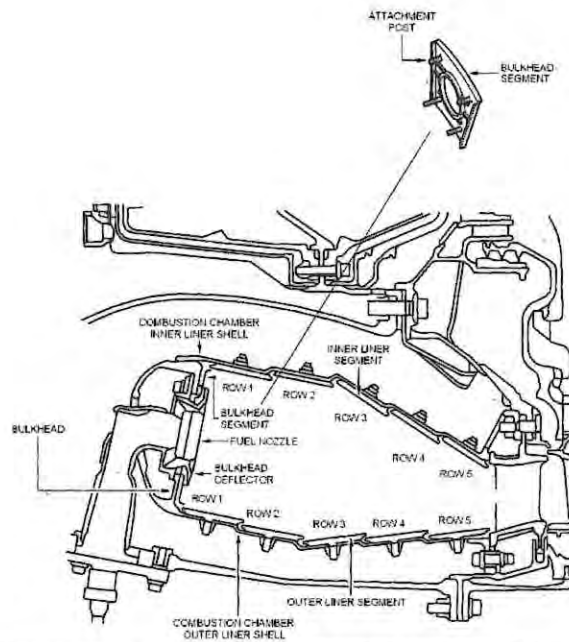
HP COMPRESSOR STAGE 3 ROTOR CLAPPER 72-00-00-210-012-A



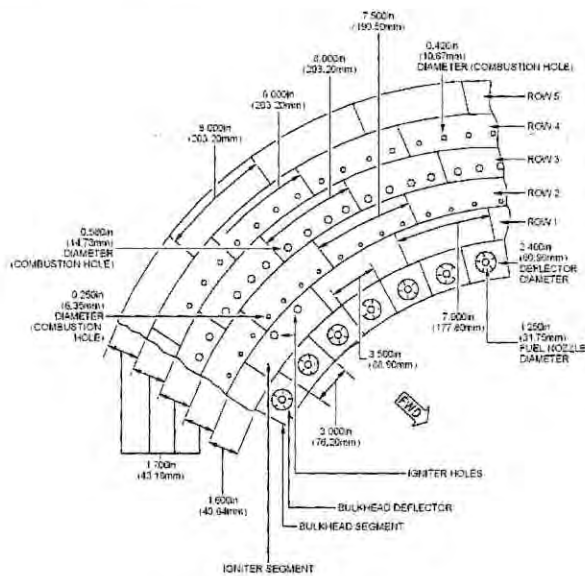
TASK 72-00-00-200-013-B Borescope Inspection of inside of Combustion Chamber and HP turbine stage 1 Vanes
INSPECTION RESULTS

AREA	ACCESS PORT	STAGE CONDITION			INSPE C.
<u>COMBUSTION CHAMBER</u>	B1 to B6 And IP1 - IP2	Liner Segment	Three row 5 with missing coating within limits.		
		Bulkhead Segment	No damage found.		
		Fuel Nozzles	No damage found.		
AREA	ACCESS PORT	STAGE	Nº OF VANES	STAGE CONDITION	INSPE C.
<u>NGV' s</u>	B1 to B6 And IP1 - IP2	NGV's	42	No damage found.	

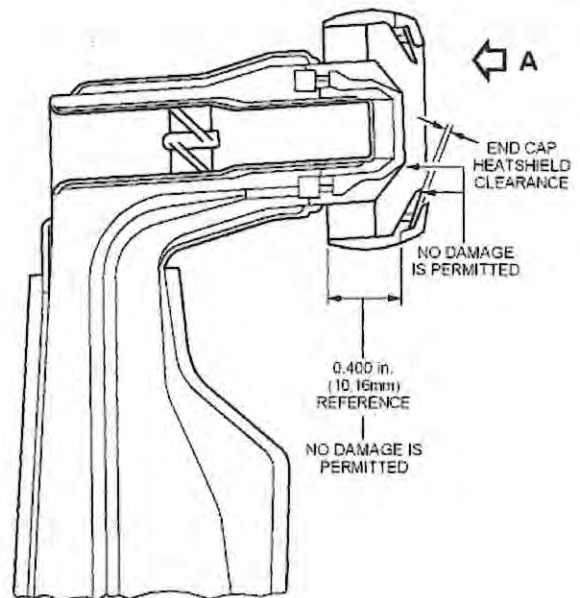
INSPECTED IN ACCORDANCE WITH A.M.M. LEASE, REV 15, Feb 01/24



VIEW OF THE INNER COMBUSTION CHAMBER SEGMENTS



VIEW OF THE OUTER COMBUSTION CHAMBER SEGMENTS



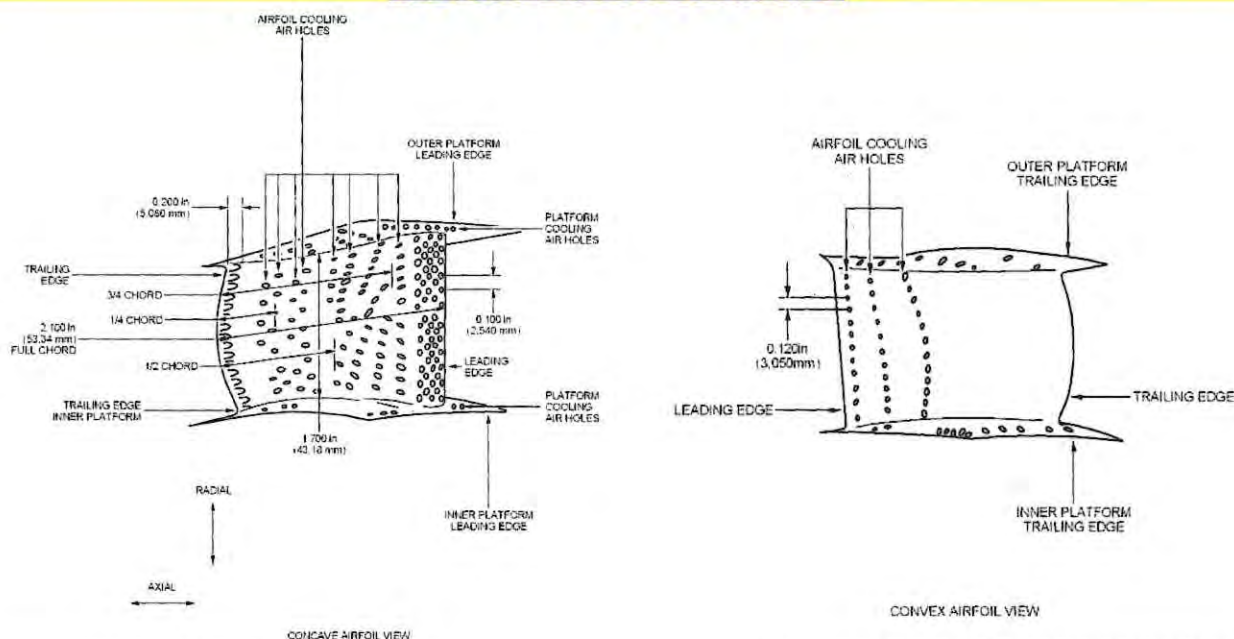
S/N: V17032

CUSTOMER: FTAI

O/T: 41288721

INSPECTED IN ACCORDANCE WITH A.M.M. LEASE, REV 15, Feb 01/24

HPT 1 NOZZLE 72-00-00-013-B



TASK 72-00-00-200-014-B01 Borescope Inspection of the Stage 1 HPT Blades and Duct Segments

INSPECTION RESULTS

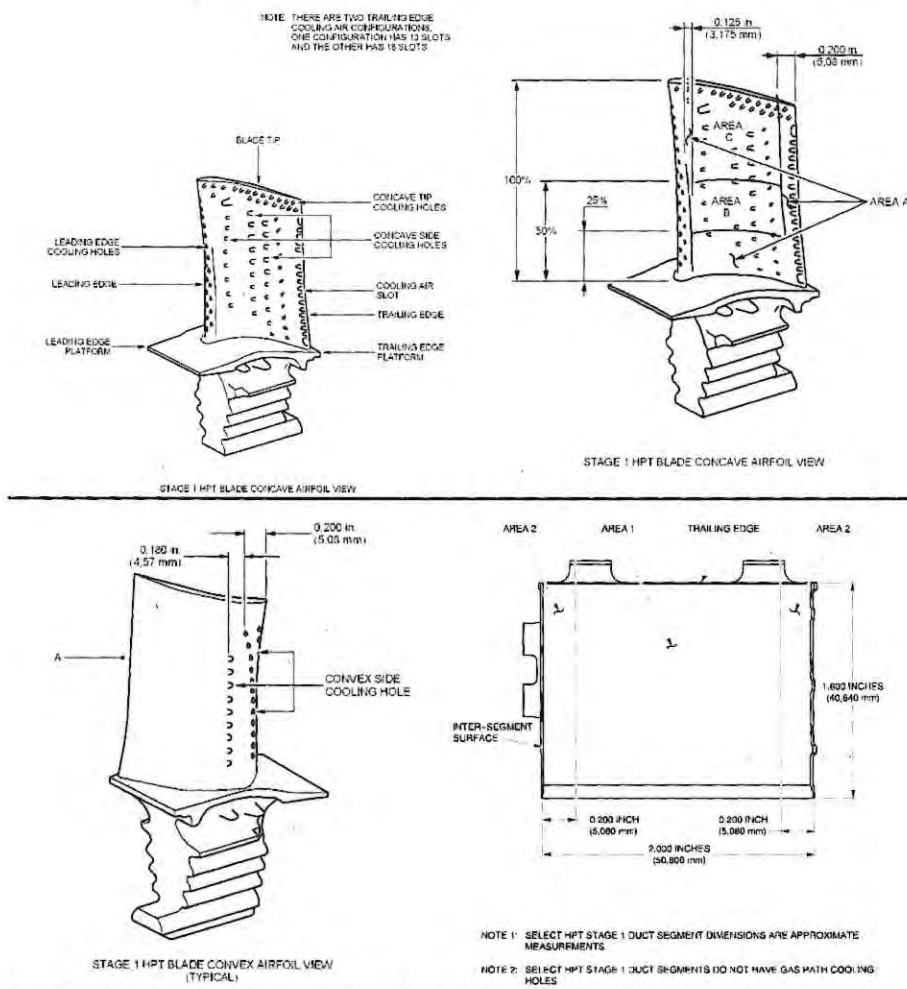
AREA	ACCESS PORT	STAGE	Nº OF BLADES	STAGE CONDITION	INSPE C.
H.P.T.	IP1 IGNITER PORT & T1/2R	1 LE & TE	64	No damage found.	
	IP1 IGNITER PORT & T1/2R	Stage 1 HPT Duct Segments	38	Light rubs within limits.	

S/N: V17032

CUSTOMER: FTAI

O/T: 41288721

INSPECTED IN ACCORDANCE WITH A.M.M. LEASE, REV 15, Feb 01/24



TASK 72-00-00-200-015-B Borescope Inspection of the Stage 2 HPT Blades and Duct Segments

INSPECTION RESULTS

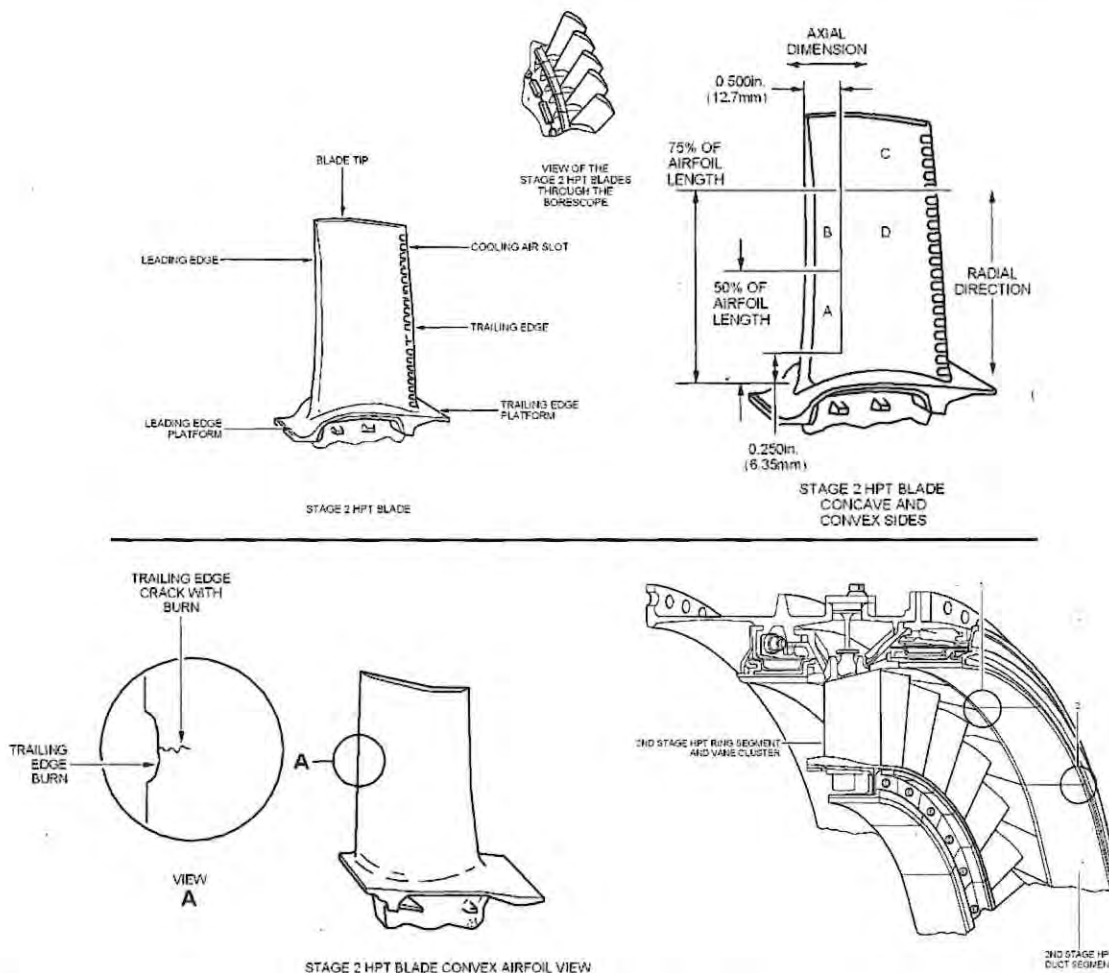
AREA	ACCESS PORT	STAGE	Nº OF BLADE S	STAGE CONDITION	INSPE C.
HPT	T1/2R & T2/3R	2 LE & TE	72	L.E: One blade with a smooth round bottom dent in platform not covered by TASK 72-00-00-200-015-B, but within limits IAW 72-45-32-220-053 D.(1). T.E: No damage found.	689146 -1
	T1/2R & T2/3R	Stage 2 HPT Duct Segments	19/38	NOTE: When you do a borescope inspection of the Stage 2 HPT blades, some areas of the Stage 2 HPT duct segments can be seen. No damage found.	689146 -1

S/N: V17032

CUSTOMER: FTAI

O/T: 41288721

INSPECTED IN ACCORDANCE WITH A.M.M. LEASE, REV 15, Feb 01/24



TASK 72-00-00-200-012-A Borescope Inspection of the Low Pressure (LP) Turbine
INSPECTION RESULTS

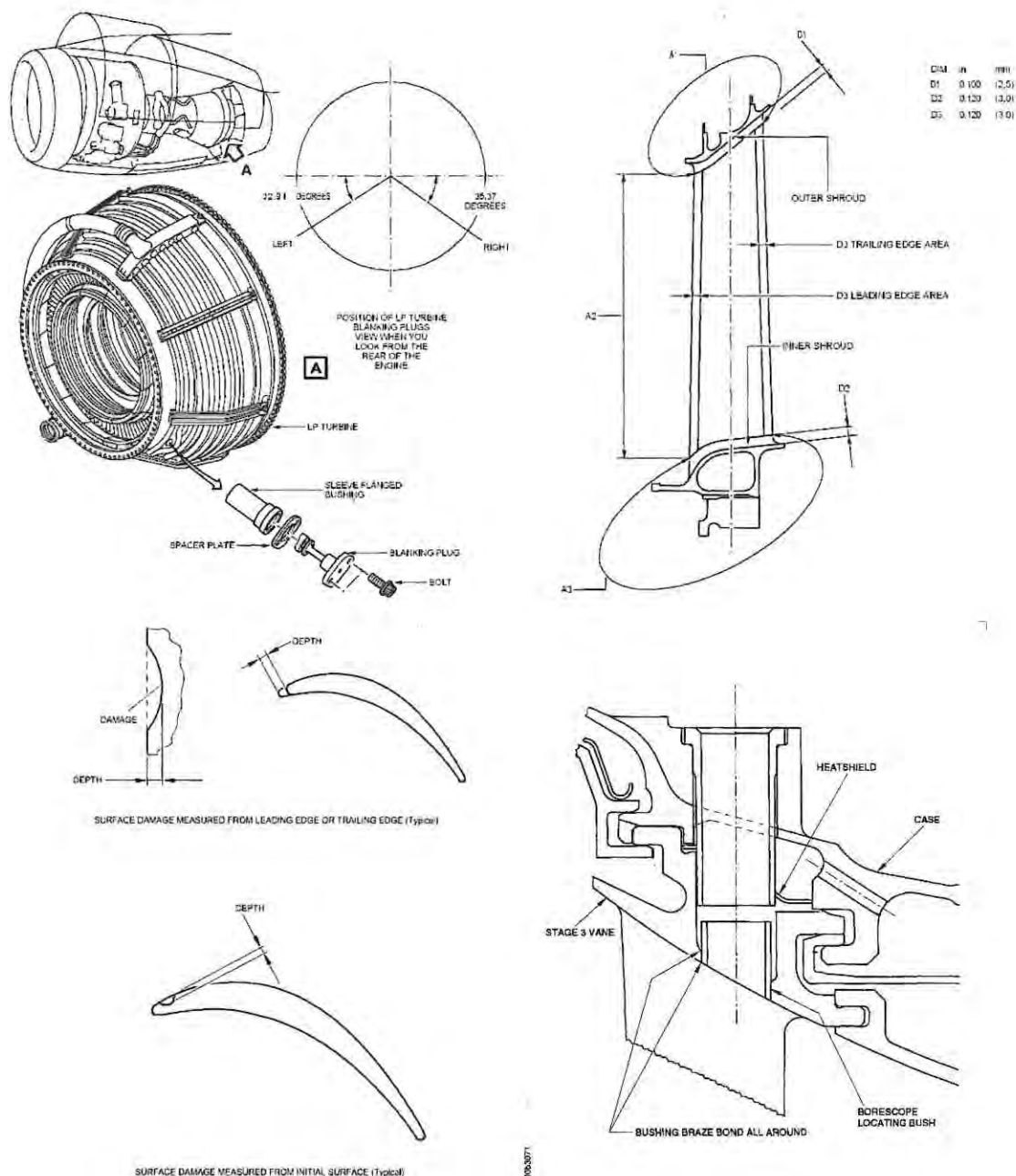
AREA	ACCESS PORT	STAGE	Nº OF BLADE S	STAGE CONDITION	INSPE C.
L.P.T.	LPT 3	3	97	Some Blades with missing/damaged coating and small dents within limits.	9811/3 -1
	LPT 3	Subtask72-00-00-220-171-A	2	This inspection is only applicable for brazed turbine vane clusters. Inspection over the whole circumference (360 degree all around) is NOT required One vane (right) with a possible small crack at borescope bushing within limits.	9811/3 -1

S/N: V17032

CUSTOMER: FTAI

O/T: 41288721

INSPECTED IN ACCORDANCE WITH A.M.M. LEASE, REV 15, Feb 01/24



S/N: V17032

CUSTOMER: FTAI

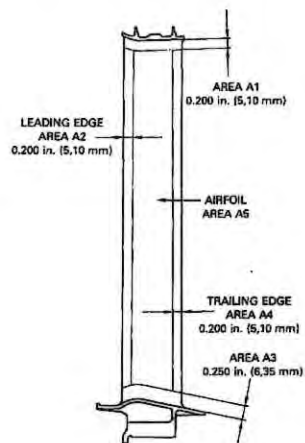
O/T: 41288721

INSPECTED IN ACCORDANCE WITH A.M.M. LEASE, REV 15, Feb 01/24

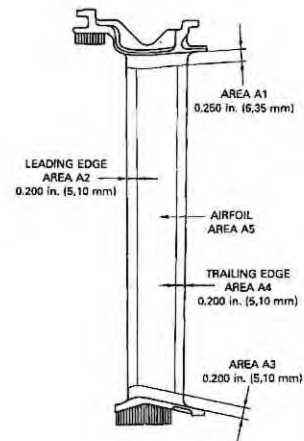
TASK 72-00-00-210-010-A Visually Examine the LP Turbine Stage 7

INSPECTION RESULTS

AREA	ACCESS PORT	STAGE	N° OF BLADES	STAGE CONDITION	INSPE C.
L.P.T.	EXHAUST	7 TE	89	Visual inspection	
				No damage found.	
	EXHAUST	7 TE Vanes	96	Visual inspection	
				No damage found.	



18545



18546

S/N: V17032	CUSTOMER: FTAI	O/T: 41288721
-------------	----------------	---------------

INSPECTED IN ACCORDANCE WITH A.M.M. LEASE, REV 15, Feb 01/24



HPT_Stage_2_LE_002.BMP

Section	HPT
Stage	Stage 2 LE
Length	1.12 mm
Comments	One blade with a smooth round bottom dent in platform not covered by TASK 72-00-00-200-015-B, but within limits IAW 72-45-32-220-053 D.(1).

NOTE: DIMENSION SPECIFIED IN ALL PICTURES ARE A GUIDE TO ASSESING DAMAGE

SIGNATURE/STAMP QUALITY CONTROL:	
----------------------------------	--

05 Last Cell Test

ENGINE TEST REPORT

Customer: **FTAI**
Work Order: **M-0000323950**
Test Date: **26-02-2025**

Engine Manual E-V2500-1IA Rev: **137** Dated: **Nov 01/24**

ENGINE DATA

Internal Ref.: **PO-ES-2.8-017**

ENGINE S/N: **V17032** OUTBOUND Rate: **V2533-A5** TESTING Rate: **V2533S0** Fuel LHV (BTU/lb): **18578**

Tested at 33K: ☒ Yes ☐ No HSI Refurbish: ☐ Yes ☒ No Select One/Two: ☒ Yes ☐ No EEC Soft. Version: ☐ SCN 19 or Lower ☒ SCN 20 or Higher

1.- EPR-0 vs Thrust Check

	TEST POINT			
	Band A	Band B	Band C	Band D
FNC (lbs)	30986	26545	22064	16116
EPR-0 Max (ratio)	1,6330	1,4957	1,3743	1,2397
EPR-0 (ratio)	1,6153	1,4758	1,3621	1,2269
EPR-0 Min (ratio)	1,6130	1,4757	1,3543	1,2197

2.- EPR Modifier Code Calculation

	TEST POINT			
	Band A	Band B	Band C	Band D
EPR-0 (ratio)	1,6153	1,4758	1,3621	1,2269
FNC (lbs)	30986	26545	22064	16116

EPR Mod N°: **7** EPR Class Bias: **2**

EPR MOD: **07-02**

3.- Acceptance Data

Engine Curves after EPR MOD Calculations

Power Point	FNK (lbs)	N1K (rpm)	N2K (rpm)	EGTK (°C)	TSFCK (lb/h/lbf)	EPR (ratio)
Band A	30986	5404	14072	558,46	0,3796	1,6243
Band B	26545	5010	13796	506,1	0,3543	1,4788
Band C	22064	4693	13481	465,63	0,3424	1,3651
Band D	16116	4162	12940	440,2	0,3359	1,2299

Standard Day Performances Interpolated against Check FNK

Power Point	FNK (lbs)	N1K (rpm)	N2K (rpm)	EGTK (°C)	TSFCK (lb/h/lbf)	EPR (ratio)
TakeOff 33K	30650	5368	14051	554	0,3771	1,612
Max Cont	26220	5011	13782	502	0,3544	1,474

Standard Day Performances Interpolated against Check EPR

Power Point	EPR (ratio)	N1K (rpm)	N2K (rpm)	EGTK (°C)
33K T/O EPR	1,61	5367	14052	553
M/C EPR	1,40	4788	13586	478

Engine Margins

EGT Margin TO at worst case (Sea Level, 55°C OAT)	(°C)	28
N1 margin at TO EPR (1,61)	(rpm)	43
	(%)	0,80
N2 margin at MC EPR (1,40)	(rpm)	139
	(%)	1,01

(*) N2 Margin has been calculated using 13725 rpm Limit

Approval:
A. Castilla
 **721350**  **721350**
Test Cell Engineer

ENGINE TEST REPORT

Engine Manual E-V2500-1IA

Rev: 137

Dated: Nov 01/24

Customer:

FTAI

Work Order:

M-0000323950

Test Date:

26-02-2025

4.- System Performance

	Actual	Limit
Accel. Time (sec)	6,0	5,3 - 6,6
Decel. Time (sec)	4,2	7
Oil Consumption (lt/h)	0,080	0,285
Oil Press (DMOPC) at Idle (psid)	137,4	115 - 170
Oil Press (DMOP) at T/O (psid)	222,7	160 - 295
Oil Temp (MOT) at T/O (°C)	102	155
Vib Max BB 1 (ips)	0,34	1,0
Vib Max N1Trk 1 (ips)	0,30	0,6
Vib Max N2Trk 1 (ips)	0,13	0,6
Vib Max BB 2 (ips)	0,32	1,0
Vib Max N1Trk 2 (ips)	0,28	0,6
Vib Max N2Trk 2 (ips)	0,13	0,6

5.- Observed Data

TEST POINT			
Band A	Band B	Band C	Band D

*** Main Parameters ***

EPR (ratio)	1,6243	1,4788	1,3651	1,2299
THRUST (lb)	27567	23698	19615	14074
N1 (rpm)	5254	4861	4532	3994
N2 (rpm)	13822	13441	13116	12607
WF (pph)	10582	8516	6819	4794
EGT (°C)	514,2	471,4	433,8	403,4

*** Pressures ***

PAMB (psia)	13,939	13,939	13,938	13,939
P2 (psia)	13,898	13,896	13,905	13,914
P12.5 (psia)	23,768	22,548	21,004	18,973
P2.5 (psia)	39,1	35,8	33,0	25,5
PS3 (psia)	446,4	383,6	328,7	247,9
P4.9 (psia)	22,5	20,5	18,9	17,1
MOP (psig)	328,9	320,2	303,5	271,2
#4 Brg Scav. (psig)	106,1	99,6	91,9	7,1
Diff Oil Press (psid)	222,7	220,6	211,6	264,1

*** Temperatures ***

AIT (deg C)	2,3	2,1	2,1	2,0
T25 (deg C)	105,1	92,5	82,8	58,2
T3 (deg C)	533,5	495,7	461,4	407,2
Fuel Temp (deg C)	13,3	13,3	13,3	13,3
Oil Tank Temp (deg C)	101,5	95,9	97,9	96,0

*** Humidity ***

Rel. Humidity (%)	71,88	73,15	74,22	73,97
Specific Humidity (grains)	0,0032	0,0032	0,0033	0,0032

*** VSV Schedule ***

EECVSV (pstr)	74,1	68,5	63,3	59,8
EECN2C26 (rpm)	11978	11833	11707	11623

*** Vibration Data (Steady) ***

BB Vib Channel 1 (ips)	0,18	0,20	0,19	0,34
N1 Trk Vib Channel 1 (ips)	0,14	0,09	0,16	0,30
N2 Trk Vib Channel 1 (ips)	0,07	0,04	0,07	0,13

BB Vib Channel 2 (ips)	0,17	0,18	0,19	0,32
N1 Trk Vib Channel 2 (ips)	0,14	0,08	0,16	0,28
N2 Trk Vib Channel 2 (ips)	0,06	0,04	0,08	0,13

6.- Corrections

TEST POINT			
Band A	Band B	Band C	Band D

*** Standard Corr. Factors ***

δt2	0,9457	0,9455	0,9462	0,9468
θt2	0,9559	0,9551	0,9553	0,9550
Kc	0,9720	0,9715	0,9716	0,9714
Kh	1,0059	1,0052	1,0045	1,0032
KDN	1,0006	1,0006	1,0006	1,0006
N1 θ Exp	0,4533	0,4737	0,4891	0,4995
T3 θ P3	1,9300	1,8100	1,7500	1,5600
T4.9 θ P49	1,2100	1,0800	0,9800	0,8800
T2.5 θ P25	0,3300	0,2900	0,2700	0,1900

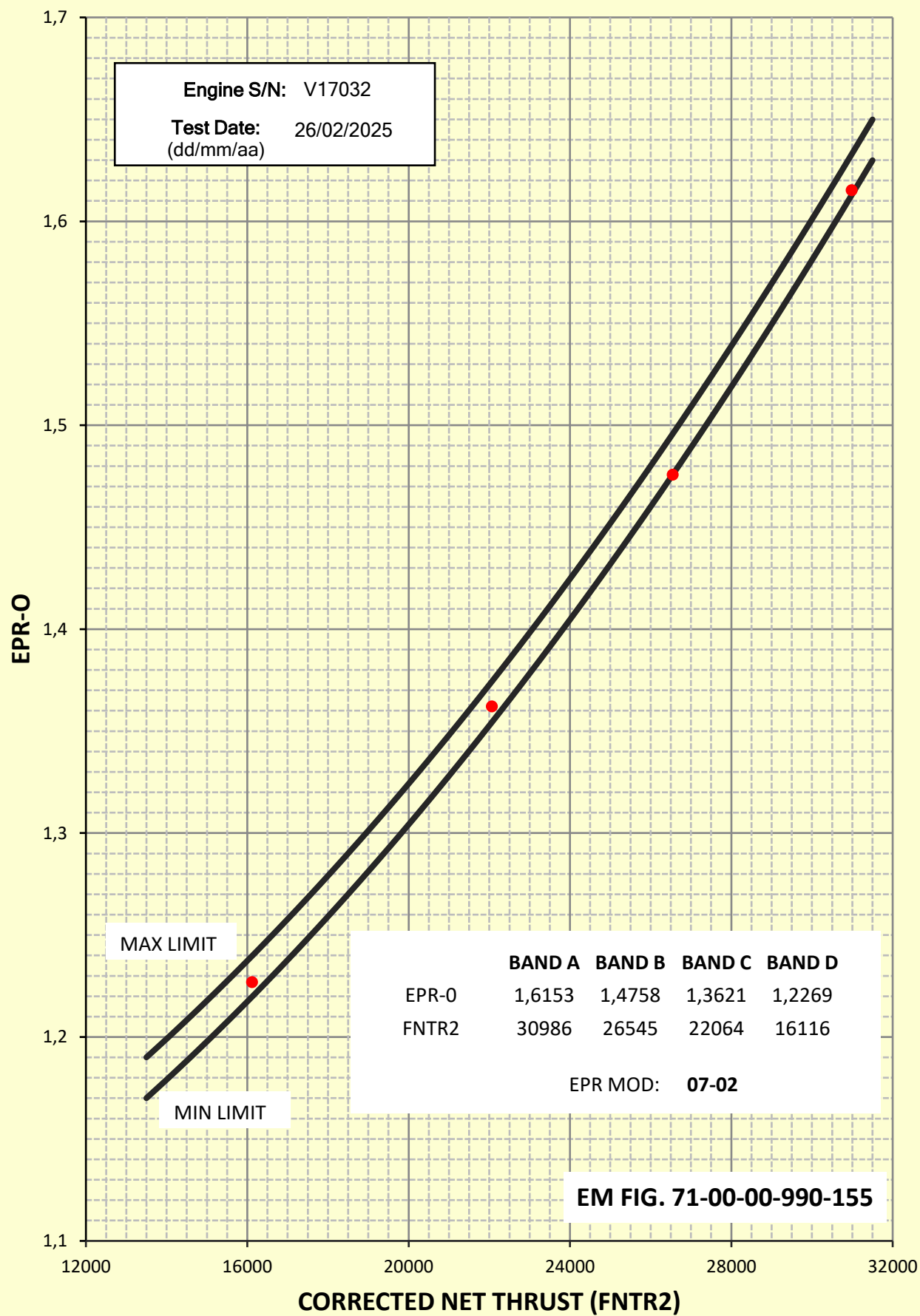
*** Test Cell Corr. Factors ***

ΔFN	1835,9	1483,1	1333,7	1251,2
ΔN1	44,45	44,86	60,77	77,01
ΔN2	-57,04	50,96	69,71	46,49
ΔWF	210,44	96,35	101,53	167,60
ΔEGT	4,8167	-3,1267	-3,8748	2,4418
ΔT2.5	0,0796	0,1941	-0,4493	-0,1758
ΔT3	0,1475	-0,1154	0,1917	0,6098
ΔP12.5/P2	0,0078	-0,0041	-0,0105	-0,0046
ΔP2.5/P2	-0,0164	-0,0044	-0,0072	-0,0052
ΔPS3/P2	0,0252	0,0401	0,0264	0,0392

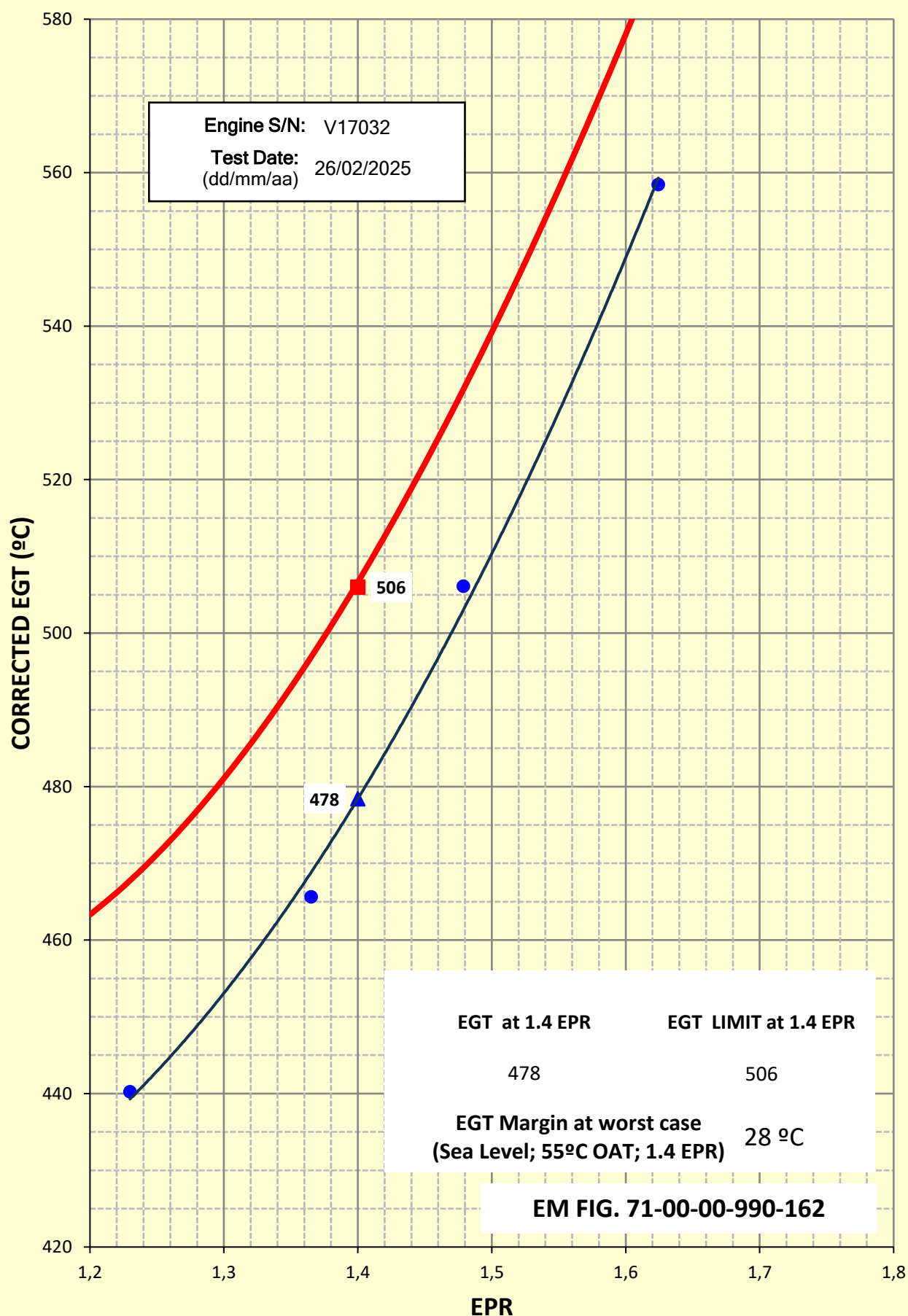
*** Std / Test Cell Corr. Values ***

EPR-0 (ratio)	1,6153	1,4758	1,3621	1,2269
FNC (lbs)	30986	26545	22064	16116
EGTC (deg C)	558,5	506,1	465,6	440,2
N1C (rpm)	5404	5010	4693	4162
N2C (rpm)	14072	13796	13481	12940
WFC (pph)	11763	9406	7556	5412
TSFC (lb/h/lb)	0,3796	0,3543	0,3424	0,3359
T2.5C (deg C)	122,8	110,0	99,1	73,6
T3C (deg C)	569,9	530,7	494,9	439,0
P12.5/P2C (ratio)	1,7179	1,6186	1,5001	1,3590
P2.5/P2C (ratio)	2,7993	2,5695	2,3651	1,8280
PS3/P2C (ratio)	32,15	27,64	23,67	17,86
PS3/P26 (ratio)	11,69	10,93	10,14	9,85

Gas Generator Check Curve (EPR vs Thrust)



EGT LIMIT vs EPR



06 AD List



Engine Shop - Engineering Department



V2500-A5 Airworthiness Directives Status Report

V2533-A5 SQ03		ENGINE:	V17032	TT:	24228,72	TC:	8349	W.O.: M-0000323950	DATE: February 25, 2025
COMPLIANCE CODE: CW=COMPLIED WITH AT THIS SHOP VISIT. PCW=COMPLIED PREVIOUSLY. N/A= NOT APLICABLE. OPEN= TASKS TO BE PERFORMED AT INTERVAL 'S REQUIREMENTS.									
N _o	CAA/EASA AD NUMBER	FAA EQUIVALENT	SB NUMBER	DESCRIPTION	COMPLIANCE STATUS	REMARKS	STAMP		
1	*	AD 99-04-03	V2500-ENG-72-0293	HPC 9-12 drum reduced life limits. (P/N affected 6A4156)	N/A	DRUM 9-12 P/N 6A7546 INSTALLED NOT AFFECTED			
2	*	AD 99-08-11		Enhanced Inspection of Fan Hub.	N/A	SUPERSEDED BY AD 2000-12-05	N/A		
3	*	AD 99-13-01	AOW 1046 Issue 2 V2500-ENG-72-0349	Borescope inspect certain eng. S/Ns, following IFSD and windmilling for evidence of oil or heat damage in the HPT hardware.	N/A	ESN NOT AFFECTED			
4	*	AD 99-13-01	V2500-ENG-72-0190	Borescope inspect certain eng. S/Ns, following IFSD and windmilling for evidence of oil or heat damage in the HPT hardware. - Terminating action.	N/A	ESN NOT AFFECTED			
5	AD 1999-050-126-B CN 99-050-126 (B)	AD 99-21-19	SB A320-71-1020 V2500-NAC-71-0210	Forward Mount. Replacement of all titanium thrust links with steel thrust links.	N/A	P/N 745-2010-503 INSTALLED NOT AFFECTED			
6	AD 1998-293-118-B R1 CN 98-293-118 (B)	AD 99-25-10	SB A320-71-1021 V2500-NAC-71-0135	Forward Mount. Replacement of mount bolts and monoball stop.	N/A	P/N 745-2010-503 INSTALLED NOT AFFECTED			

V2500 ENGINEERING				SHOP INSPECTION			
DATE: February 25, 2025		STAMP:		DATE: Feb. 28, 2025		STAMP:	

* DENOTES NO EQUIVALENT.



Engine Shop - Engineering Department



V2500-A5 Airworthiness Directives Status Report

V2533-A5 SQ03		ENGINE:	V17032	TT:	24228,72	TC:	8349	W.O.: M-0000323950		DATE: February 25, 2025	
COMPLIANCE CODE: CW=COMPLIED WITH AT THIS SHOP VISIT. PCW=COMPLIED PREVIOUSLY. N/A= NOT APPLICABLE. OPEN= TASKS TO BE PERFORMED AT INTERVAL'S REQUIREMENTS.											
N _o	CAA/EASA AD NUMBER	FAA EQUIVALENT	SB NUMBER	DESCRIPTION		COMPLIANCE STATUS	REMARKS		STAMP		
7	AD 2000-005-143-B	AD 2000-11-25	SB A320-73-1067	AD 2000-005-143(B). Requires replacement of FMU with alternate P/N unit before March 31, 2002. AD 2000-11-25. Requires replacement of FMU with alternate P/N unit within 18 months.		N/A	FMU P/N 8061-640 INSTALLED NOT AFFECTED				
8	*	AD 2000-12-05		Adds HPT 1st and 2nd disks to the FAA enhanced inspection at piece part opportunity. SUPERSEDES AD 99-08-11.		N/A	SUPERSEDED BY AD 2004-12-08		N/A		
9	*	AD 2000-22-07	(AOW 1053 Issue 2)	Removal from service of certain engines with improper HPT configuration prior to accumulating 5100 or 7600 cycles or at the next shop visit depending on the type of improper HPT V2500-A5/D5		N/A	ESN NOT AFFECTED				
10	*	AD 2003-11-23		Master magnetic chip detector (MCD), No. 3 Bearing.		N/A	SUPERSEDED BY AD 2006-25-01		N/A		
11	AD 2003-320-B	*	(AOW 1066 Issue 2)	Inspection of dual datum Goodrich Fuel Metering Units (FMU) (ATA 73).		N/A	CANCELLED 24 JULY 2010		N/A		
12	AD 2003-355-B	*	(AOW 1048) V2500-ENG-79-0085 V2500-ENG-79-0086	Inspect the area of the Fuel Filter cover of the Fuel Cooled Oil Cooler (FCOC).		N/A	SUPERSEDED BY EASA AD 2005-0012		N/A		

V2500 ENGINEERING		SHOP INSPECTION	
DATE: February 25, 2025	STAMP:	DATE: Feb. 28, 2025	STAMP:

* DENOTES NO EQUIVALENT.



Engine Shop - Engineering Department



V2500-A5 Airworthiness Directives Status Report

V2533-A5 SQ03	ENGINE: V17032	TT: 24228,72	TC: 8349	W.O.: M-0000323950	DATE: February 25, 2025
----------------------	-----------------------	---------------------	-----------------	---------------------------	--------------------------------

COMPLIANCE CODE: CW=COMPLIED WITH AT THIS SHOP VISIT. PCW=COMPLIED PREVIOUSLY. N/A= NOT APPLICABLE. OPEN= TASKS TO BE PERFORMED AT INTERVAL 'S REQUIREMENTS.

Nº	CAA/EASA AD NUMBER	FAA EQUIVALENT	SB NUMBER	DESCRIPTION	COMPLIANCE STATUS	REMARKS	STAMP
13	*	AD 2004-12-08		Enhanced inspection on LLPs. SUPERSEDES AD 2000-12-05. CRACK INSPECTION OF THE FAN DISC IAW (E.M.) SUBTASK 72-31-12-230-054	OPEN	PART NOT DISASSEMBLED INSPECT AT NEXT PIECE-PART OPPORTUNITY P/N: 5A1948 S/N: RFXP011306	
14	*	AD 2004-12-08		Enhanced inspection on LLPs. SUPERSEDES AD 2000-12-05. CRACK INSPECTION OF THE STG 3-8 DRUM IAW (E.M.) SUBTASK 72-41-11-230-104	OPEN	PART NOT DISASSEMBLED INSPECT AT NEXT PIECE-PART OPPORTUNITY P/N: 6B1404 S/N: RRD12496	
15	*	AD 2004-12-08		Enhanced inspection on LLPs. SUPERSEDES AD 2000-12-05. CRACK INSPECTION OF THE STG 9-12 DRUM IAW (E.M.) SUBTASK 72-41-12-230-079	OPEN	PART NOT DISASSEMBLED INSPECT AT NEXT PIECE-PART OPPORTUNITY P/N: 6A7546 S/N: RRD7601	
16	*	AD 2004-12-08		Enhanced inspection on LLPs. SUPERSEDES AD 2000-12-05. CRACK INSPECTION OF THE REAR SHAFT IAW (E.M.) SUBTASKS 72-41-13-230-097 AND 72-41-13-230-098.	OPEN	PART NOT DISASSEMBLED INSPECT AT NEXT PIECE-PART OPPORTUNITY P/N: 6B1419 S/N: 200581CQYX8163	
17	*	AD 2004-12-08		Enhanced inspection on LLPs. SUPERSEDES AD 2000-12-05. CRACK INSPECTION OF THE REAR ROTATING SEAL IAW (E.M.) SUBTASKS 72-41-14-230-064 AND 72-41-14-230-065.	OPEN	PART NOT DISASSEMBLED INSPECT AT NEXT PIECE-PART OPPORTUNITY P/N: 6A5869 S/N: RFRQDY3450	
18	*	AD 2004-12-08		Enhanced inspection on LLPs. SUPERSEDES AD 2000-12-05. CRACK INSPECTION OF THE STG 1 HUB HPT IAW (E.M.) SUBTASK 72-45-11-200-002.	OPEN	PART NOT DISASSEMBLED INSPECT AT NEXT PIECE-PART OPPORTUNITY P/N: 2A5001 S/N: PKLBRB9243	

V2500 ENGINEERING				SHOP INSPECTION			
DATE: February 25, 2025				DATE: Feb. 28, 2025			
STAMP:				STAMP:			

* DENOTES NO EQUIVALENT.



Engine Shop - Engineering Department



V2500-A5 Airworthiness Directives Status Report

V2533-A5 SQ03	ENGINE: V17032	TT: 24228,72	TC: 8349	W.O.: M-0000323950	DATE: February 25, 2025
----------------------	-----------------------	---------------------	-----------------	---------------------------	--------------------------------

COMPLIANCE CODE: CW=COMPLIED WITH AT THIS SHOP VISIT. PCW=COMPLIED PREVIOUSLY. N/A= NOT APPLICABLE. OPEN= TASKS TO BE PERFORMED AT INTERVAL 'S REQUIREMENTS.

Nº	CAA/EASA AD NUMBER	FAA EQUIVALENT	SB NUMBER	DESCRIPTION	COMPLIANCE STATUS	REMARKS	STAMP
19	*	AD 2004-12-08		Enhanced inspection on LLPs. SUPERSEDES AD 2000-12-05. CRACK INSPECTION OF THE STG 2 HUB HPT IAW (E.M.) SUBTASK 72-45-31-200-004	OPEN	PART NOT DISASSEMBLED INSPECT AT NEXT PIECE-PART OPPORTUNITY P/N: 2A4802 S/N: PKLBTC9974	
20	*	AD 2004-12-08		Enhanced inspection on LLPs. SUPERSEDES AD 2000-12-05. CRACK INSPECTION OF THE STG 3 DISC LPT IAW (E.M.) SUBTASK 72-50-31-200-006.	OPEN	PART NOT DISASSEMBLED INSPECT AT NEXT PIECE-PART OPPORTUNITY P/N: 3A2513 S/N: JM5754D	
21	*	AD 2004-12-08		Enhanced inspection on LLPs. SUPERSEDES AD 2000-12-05. CRACK INSPECTION OF THE STG 4 DISC LPT IAW (E.M.) SUBTASK 72-50-31-200-006.	OPEN	PART NOT DISASSEMBLED INSPECT AT NEXT PIECE-PART OPPORTUNITY P/N: 3A1984 S/N: MAL20671	
22	*	AD 2004-12-08		Enhanced inspection on LLPs. SUPERSEDES AD 2000-12-05. CRACK INSPECTION OF THE STG 5 DISC LPT IAW (E.M.) SUBTASK 72-50-31-200-006.	OPEN	PART NOT DISASSEMBLED INSPECT AT NEXT PIECE-PART OPPORTUNITY P/N: 3A2430 S/N: MAL06241	
23	*	AD 2004-12-08		Enhanced inspection on LLPs. SUPERSEDES AD 2000-12-05. CRACK INSPECTION OF THE STG 6 DISC LPT. IAW (E.M.) SUBTASK 72-50-31-200-006.	OPEN	PART NOT DISASSEMBLED INSPECT AT NEXT PIECE-PART OPPORTUNITY P/N: 3A2996 S/N: MAK97375	
24	*	AD 2004-12-08		Enhanced inspection on LLPs. SUPERSEDES AD 2000-12-05. CRACK INSPECTION OF THE STG 7 DISC LPT IAW (E.M.) SUBTASK 72-50-31-200-006.	OPEN	PART NOT DISASSEMBLED INSPECT AT NEXT PIECE-PART OPPORTUNITY P/N: 3A2514 S/N: JM5995D	

V2500 ENGINEERING				SHOP INSPECTION			
DATE: February 25, 2025				DATE: Feb. 28, 2025			
STAMP:				STAMP:			

* DENOTES NO EQUIVALENT.



Engine Shop - Engineering Department



V2500-A5 Airworthiness Directives Status Report

V2533-A5 SQ03		ENGINE:	V17032	TT:	24228,72	TC:	8349	W.O.: M-0000323950	DATE: February 25, 2025
COMPLIANCE CODE: CW=COMPLIED WITH AT THIS SHOP VISIT. PCW=COMPLIED PREVIOUSLY. N/A= NOT APPLICABLE. OPEN= TASKS TO BE PERFORMED AT INTERVAL'S REQUIREMENTS.									
N _o	CAA/EASA AD NUMBER	FAA EQUIVALENT	SB NUMBER	DESCRIPTION	COMPLIANCE STATUS	REMARKS	STAMP		
25	F-2004-074	AD 2004-15-14	SB A320-71-1030 V2500-NAC-70-0736	Arcing of the electrical harness of the Integrated Drive Generator (IDG) at the pin-to-socket connector at the firewall.	N/A	SUPERSEDED BY AD 2011-19-01	N/A		
26	AD 2005-0012	*	(AOW 1048) V2500-ENG-79-0086	Fuel filter cover – Inspection / modification.	PCW	SB 79-0086 ACCOMPLISHED FUEL COOLED OIL COOLER INSTALLED P/N: 55022001-11			
27	*	AD 2006-07-05	AOT A320-71A1036 V2500-NAC-71-0135	Inspecting for cracks or failure of the primary load path components of the engine forward mount	N/A	ESN NOT AFFECTED			
28	*	AD 2006-25-01	V2500-ENG-72-0452 V2500-ENG-72-0459	Master magnetic chip detector (MCD), No.3 Bearing. SUPERSEDES AD 2003-11-23.	N/A	ESN NOT AFFECTED			
29	*	AD 2007-21-05	V2500-ENG-72-0500 V2500-ENG-72-0501 V2500-ENG-72-0502	HPT Stage 2 Air Seals - Repetitive monitoring of N2 vibration	N/A	HPT STAGE 2 AIR SEAL INSTALLED P/N: 2A4157			
30	*	AD 2008-14-15	V2500-ENG-72-0541	No. 4 Bearing Oil System Components - HPT	N/A	ESN NOT AFFECTED			

V2500 ENGINEERING		SHOP INSPECTION	
DATE: February 25, 2025	STAMP:	DATE: Feb. 28, 2025	STAMP:

* DENOTES NO EQUIVALENT.



Engine Shop - Engineering Department



V2500-A5 Airworthiness Directives Status Report

V2533-A5 SQ03	ENGINE: V17032	TT: 24228,72	TC: 8349	W.O.: M-0000323950	DATE: February 25, 2025
----------------------	------------------------------	----------------------------	------------------------	----------------------------------	---------------------------------------

COMPLIANCE CODE: **CW**=COMPLIED WITH AT THIS SHOP VISIT. **PCW**=COMPLIED PREVIOUSLY. **N/A**= NOT APPLICABLE. **OPEN**= TASKS TO BE PERFORMED AT INTERVAL 'S REQUIREMENTS.

Nº	CAA/EASA AD NUMBER	FAA EQUIVALENT	SB NUMBER	DESCRIPTION	COMPLIANCE STATUS	REMARKS	STAMP
31	AD 2008-0191-E	AD 2008-25-06	AOT A320-71A1045	Power Plant – Rear Engine Mount Barrel Nuts – Inspection / Replacement. SUPERSEDES AD 90-079-009(B) R1 [A1 only].	OPEN	PART MISSING TO BE CHECKED ON INSTALLATION	
32	*	AD 2009-22-06	V2500 EO 72-00004 V2500-ENG-72-A0554	HPC Stage 9-12 Disc Assembly Life Limit Reduction and Drawdown	N/A	DRUM 9-12 P/N 6A7546 INSTALLED NOT AFFECTED	
33	*	AD 2010-06-18	V2500 EO 72-00001	Cracks in Vortex Reducer (HPC Stage 3-8 Drum) - V2500 Onetime fluorescent penetrant inspection of certain vortex reducers for cracks	N/A	INCOMING HPC 3-8 DRUM PN 6B1404 OUTGOING HPC 3-8 DRUM PN 6B1404	
34	*	AD 2010-20-07	V2500-ENG-72-0594 V2500-ENG-72-0601 V2500-ENG-72-0603	Ultrasonic inspections of the HPC 3-8 Drum	N/A	SUPERSEDED BY AD 2012-09-09	N/A
35	*	AD 2011-19-01 R1	SB A320-71-1030 V2500-NAC-70-0736	Replacement of the IDG harnesses and connectors. (The R0 of this AD supersedes AD2004-15-14)	N/A	IDG HARNESS POST SB NAC 70-0736	
36	*	AD 2011-25-08	V2500-ENG-72-0580 V2500-ENG-72-0483 V2500-ENG-72-0542	HPT Stage 1 Blade Outer Air Seal Inspections	N/A	ESN NOT AFFECTED	

V2500 ENGINEERING				SHOP INSPECTION			
DATE:	February 25, 2025	STAMP:		DATE:	Feb. 28, 2025	STAMP:	

* DENOTES NO EQUIVALENT.



Engine Shop - Engineering Department



V2500-A5 Airworthiness Directives Status Report

V2533-A5 SQ03		ENGINE:	V17032	TT:	24228,72	TC:	8349	W.O.: M-0000323950	DATE: February 25, 2025
COMPLIANCE CODE: CW=COMPLIED WITH AT THIS SHOP VISIT. PCW=COMPLIED PREVIOUSLY. N/A= NOT APPLICABLE. OPEN= TASKS TO BE PERFORMED AT INTERVAL'S REQUIREMENTS.									
N _o	CAA/EASA AD NUMBER	FAA EQUIVALENT	SB NUMBER	DESCRIPTION	COMPLIANCE STATUS	REMARKS	STAMP		
37	*	AD 2012-09-09	V2500-ENG-72-0615 R4	ULTRASONIC INSPECTIONS OF THE HPC 3-8 DRUM FOR STRESS CORROSION CRACKING. SUPERSEDES AD 2010-20-07.	N/A	SUPERSEDED BY AD 2015-10-04	N/A		
38	AD 2015-0004	AD 2016-06-03	SB A320-71-1064 R1 NAC-71-0323 R1	(V2500) Power Plant - Forward Engine Mount Pin - Identification / Replacement.	N/A	NOT APPLICABLE TO PARTS INSTALLED			
39	*	AD 2015-10-04	V2500-ENG-72-0615 R6	ULTRASONIC INSPECTIONS OF THE HPC 3-8 DRUM FOR STRESS CORROSION CRACKING. (INITIAL AND REPETITIVE USI) SUPERSEDES AD 2012-09-09. Set Grace Period in accordance with all applicable actions taken.	N/A	PART NOT DISASSEMBLED			
40	*	AD 2015-10-04	V2500-ENG-72-0632	Do NOT install silver-plated nuts (SB72-0632). TERMINATING ACTION: Within 9450 CY after eff.date (July/01/2015) but not later than Dec/31/2025, install an HPC 3-8 Drum that never operated with silver-plated nuts. SUPERSEDES AD 2012-09-09.	N/A	THE INSTALLED DRUM NEVER OPERATED WITH SILVER-PLATED NUTS.			
41	*	AD 2016-18-10	V2500-ENG-72-0661	OVER INSPECTION FOR MANUFACTURING DEFECTS STAGE 1 HPT HUB	N/A	NOT APPLICABLE TO PARTS INSTALLED			
42	*	AD 2016-18-10	V2500-ENG-72-0661	OVER INSPECTION FOR MANUFACTURING DEFECTS STAGE 2 HPT HUB	N/A	NOT APPLICABLE TO PARTS INSTALLED			

V2500 ENGINEERING				SHOP INSPECTION			
DATE: February 25, 2025				DATE: Feb. 28, 2025			
STAMP:				STAMP:			

* DENOTES NO EQUIVALENT.



Engine Shop - Engineering Department



V2500-A5 Airworthiness Directives Status Report

V2533-A5 SQ03		ENGINE:	V17032	TT:	24228,72	TC:	8349	W.O.: M-0000323950	DATE: February 25, 2025
COMPLIANCE CODE: CW=COMPLIED WITH AT THIS SHOP VISIT. PCW=COMPLIED PREVIOUSLY. N/A= NOT APPLICABLE. OPEN= TASKS TO BE PERFORMED AT INTERVAL 'S REQUIREMENTS.									
Nº	CAA/EASA AD NUMBER	FAA EQUIVALENT	SB NUMBER	DESCRIPTION	COMPLIANCE STATUS	REMARKS	STAMP		
43	*	AD2016-25-11	ENG72-0671 ENG72-0673	Nº3 BEARING.REDUCED MMCD INSPECTION INTERVAL FOR SUSPECT POPULATION (LOT A) IAW SB 72-0671. SB 72-0673 REPLACES POPULATION AFFECTED	N/A	ESN NOT AFFECTED			
44	*	AD2017-14-07	V2500-ENG-72-0677 V2500-ENG-72-0678	HPT STAGE 1 AIR SEAL P/N 2A3423. REPLACEMENT OF SUSPECT POPULATION.	N/A	NOT APPLICABLE TO PARTS INSTALLED OUTGOING HPT STAGE 1 AIR SEAL: P/N: 2A3423 S/N: PKLBL73496			
45	*	AD2017-14-07	V2500-ENG-72-0676	HPT STAGE 2 AIR SEAL P/N 2A4157. REPLACEMENT OF SUSPECT POPULATION.	N/A	NOT APPLICABLE TO PARTS INSTALLED OUTGOING HPT STAGE 2 AIR SEAL: P/N: 2A4157 S/N: PKLBL26102			
46	*	AD2017-14-07	V2500-ENG-72-0681 V2500-ENG-72-0682	HPT STAGE 2 RING PLATE P/N 2A3437. REPLACEMENT OF SUSPECT POPULATION.	N/A	NOT APPLICABLE TO PARTS INSTALLED OUTGOING HPT STAGE 2 RING PLATE: P/N: 2A3437 S/N: PKLBM9299			
47	*	AD2019-06-06	V2500-ENG-72-A0706 V2500-ENG-72-0709	DIFFUSER CASE M-FLANGE REPETITIVE VISUAL INSPECTION	N/A	SUPERSEDED BY AD 2021-05-18	N/A		
48	*	AD2019-18-01	V2500-ENG-72-A0697	DISK STAGE 6 LOW PRESSURE TURBINE REPLACEMENT OF SUSPECT POPULATION	N/A	NOT APPLICABLE TO PARTS INSTALLED OUTGOING LPT DISK STAGE 6: P/N: 3A2996 S/N: MAK97375			

V2500 ENGINEERING		SHOP INSPECTION	
DATE: February 25, 2025	STAMP:	DATE: Feb. 28, 2025	STAMP:

* DENOTES NO EQUIVALENT.



Engine Shop - Engineering Department



V2500-A5 Airworthiness Directives Status Report

V2533-A5 SQ03		ENGINE:	V17032	TT:	24228,72	TC:	8349	W.O.: M-0000323950	DATE: February 25, 2025
COMPLIANCE CODE: CW=COMPLIED WITH AT THIS SHOP VISIT. PCW=COMPLIED PREVIOUSLY. N/A= NOT APLICABLE. OPEN= TASKS TO BE PERFORMED AT INTERVAL'S REQUIREMENTS.									
N _o	CAA/EASA AD NUMBER	FAA EQUIVALENT	SB NUMBER	DESCRIPTION	COMPLIANCE STATUS	REMARKS		STAMP	
49	*	AD2020-07-51		REMOVAL FROM SERVICE OF AFFECTED HPT STAGE 1 HUB	N/A	NOT APPLICABLE TO PARTS INSTALLED OUTGOING HPT STAGE 1 HUB: P/N: 2A5001 S/N: PKLBRB9243			
50	*	AD2020-09-03	V2500-ENG-72-0707	REMOVAL FROM SERVICE OF AFFECTED DIFFUSER CASE	N/A	NOT APPLICABLE TO PARTS INSTALLED OUTGOING DIFFUSER CASE: P/N: 2A2891-01 S/N: PGGUA49050			
51	*	AD2021-01-03		REMOVAL FROM SERVICE OF CERTAIN HPT STAGE 1 HUBs	N/A	NOT APPLICABLE TO PARTS INSTALLED OUTGOING HPT STAGE 1 HUB: P/N: 2A5001 S/N: PKLBRB9243			
52	*	AD2021-01-03		REMOVAL FROM SERVICE OF CERTAIN HPT STAGE 2 HUBs	N/A	NOT APPLICABLE TO PARTS INSTALLED OUTGOING HPT STAGE 2 HUB: P/N: 2A4802 S/N: PKLBTC9974			
53	*	AD2021-05-18	V2500-ENG-72-A0706 V2500-ENG-72-0709	DIFFUSER CASE M-FLANGE REPETITIVE BORESCOPE INSPECTION	OPEN	M-FLANGE NOT INSPECTED AT THIS SHOP VISIT. P/N: 2A2891-01 S/N: PGGUA49050			
54	*	AD2021-11-51	V2500-ENG-72-0713	ULTRASONIC INSPECTION OF CERTAIN HPT STAGE 1 HUBS	N/A	NOT APPLICABLE TO PARTS INSTALLED OUTGOING HPT STAGE 1 HUB: P/N: 2A5001 S/N: PKLBRB9243			

V2500 ENGINEERING				SHOP INSPECTION			
DATE: February 25, 2025		STAMP:		DATE: Feb. 28, 2025		STAMP:	

* DENOTES NO EQUIVALENT.



Engine Shop - Engineering Department



V2500-A5 Airworthiness Directives Status Report

V2533-A5 SQ03		ENGINE: V17032	TT: 24228,72	TC: 8349	W.O.: M-0000323950	DATE: February 25, 2025	
COMPLIANCE CODE: CW =COMPLIED WITH AT THIS SHOP VISIT. PCW =COMPLIED PREVIOUSLY. N/A = NOT APPLICABLE. OPEN = TASKS TO BE PERFORMED AT INTERVAL 'S REQUIREMENTS.							
N _o	CAA/EASA AD NUMBER	FAA EQUIVALENT	SB NUMBER	DESCRIPTION	COMPLIANCE STATUS	REMARKS	STAMP
55	*	AD2021-11-51	V2500-ENG-72-0713	ULTRASONIC INSPECTION OF CERTAIN HPT STAGE 2 HUBS	N/A	NOT APPLICABLE TO PARTS INSTALLED OUTGOING HPT STAGE 2 HUB: P/N: 2A4802 S/N: PKLBTC9974	
56	*	AD2021-11-15	V2500-ENG-72-0713	ULTRASONIC INSPECTION OF CERTAIN HPT STAGE 1 HUBS	N/A	SUPERSEDED BY AD 2022-02-09	N/A
57	*	AD2021-11-15	V2500-ENG-72-0713	ULTRASONIC INSPECTION OF CERTAIN HPT STAGE 2 HUBS	N/A	SUPERSEDED BY AD 2022-02-09	N/A
58	*	AD2021-14-19	V2500-ENG-72-0713	ULTRASONIC INSPECTION OF CERTAIN HPT STAGE 1 HUBS	PCW	NOT APPLICABLE TO PARTS INSTALLED OUTGOING HPT STAGE 1 HUB: P/N: 2A5001 S/N: PKLBRB9243	
59	*	AD2021-14-19	V2500-ENG-72-0713	ULTRASONIC INSPECTION OF CERTAIN HPT STAGE 2 HUBS	N/A	NOT APPLICABLE TO PARTS INSTALLED OUTGOING HPT STAGE 2 HUB: P/N: 2A4802 S/N: PKLBTC9974	
60	*	AD2022-02-09	V2500-ENG-72-0713	ULTRASONIC INSPECTION OF CERTAIN HPT STAGE 1 HUBS	PCW	NOT APPLICABLE TO PARTS INSTALLED OUTGOING HPT STAGE 1 HUB: P/N: 2A5001 S/N: PKLBRB9243	
V2500 ENGINEERING				SHOP INSPECTION			
DATE: February 25, 2025				DATE: Feb. 28, 2025			
STAMP:				STAMP:			

* DENOTES NO EQUIVALENT.



Engine Shop - Engineering Department



V2500-A5 Airworthiness Directives Status Report

V2533-A5 SQ03	ENGINE: V17032	TT: 24228,72	TC: 8349	W.O.: M-0000323950	DATE: February 25, 2025
----------------------	------------------------------	----------------------------	------------------------	----------------------------------	---------------------------------------

COMPLIANCE CODE: **CW**=COMPLIED WITH AT THIS SHOP VISIT. **PCW**=COMPLIED PREVIOUSLY. **N/A**= NOT APLICABLE. **OPEN**= TASKS TO BE PERFORMED AT INTERVAL 'S REQUIREMENTS.

Nº	CAA/EASA AD NUMBER	FAA EQUIVALENT	SB NUMBER	DESCRIPTION	COMPLIANCE STATUS	REMARKS	STAMP
61	*	AD2022-02-09	V2500-ENG-72-0713	ULTRASONIC INSPECTION OF CERTAIN HPT STAGE 2 HUBS	N/A	NOT APPLICABLE TO PARTS INSTALLED OUTGOING HPT STAGE 2 HUB: P/N: 2A4802 S/N: PKLBTC9974	
62	*	AD2024-11-03	V2500-ENG-72-0720	ULTRASONIC INSPECTION AND REPLACEMENT OF CERTAIN HPT STAGE 1 HUBS	N/A	NOT APPLICABLE TO PARTS INSTALLED OUTGOING HPT STAGE 1 HUB: P/N: 2A5001 S/N: PKLBRB9243	
63	*	AD2024-11-03	V2500-ENG-72-0720	ULTRASONIC INSPECTION AND REPLACEMENT OF CERTAIN HPT STAGE 2 HUBS	N/A	NOT APPLICABLE TO PARTS INSTALLED OUTGOING HPT STAGE 2 HUB: P/N: 2A4802 S/N: PKLBTC9974	

V2500 ENGINEERING				SHOP INSPECTION			
DATE: February 25, 2025				DATE: Feb. 28, 2025			
STAMP:				STAMP:			

* DENOTES NO EQUIVALENT.

Revision	Date	Reason
REV.00	25 February 2025	Original document

V17032_ADs_R00A

07 ACC List

ACCESSORIES STATUS

Engine Type: **V2533-A5 SQ03**

TSN: **24228,72**

Work Order: **M-0000323950**

Operator:



Engine SN: **V17032**

CSN: **8349**

Rev. // Issue Date: **00 // 28/FEB/2025**

Issued by: **OMH 983124**

N°	Ref. I.P.C.	Denomination	INCOMING		OUTGOING		
			P/N received	S/N received	P/N installed	S/N installed	Work Acc.
1	24-21-15-01-100	IDG OIL TEMPERATURE THERMOCOUPLE	73984300	12700	73984300	12700	VISUALLY CHECKED IN ASSEMBLED CONDITION
2	24-21-42-01-100	IDG OIL COOLER FUEL COOLED	16844-000	5709	16844-000	5709	VISUALLY CHECKED IN ASSEMBLED CONDITION
3	26-12-15-01-100	DETECTOR ASSY FIRE ACCESSORY	7702-04	3640	7702-04	3640	VISUALLY CHECKED IN ASSEMBLED CONDITION
4	26-12-17-01-100	DETECTOR ASSY FIRE	7827-03	3001	7827-03	3001	VISUALLY CHECKED
5	29-11-17-01-010	HYDRAULIC PRESSURE SWITCH	450-1-3100-00	10328	450-1-3100-00	10328	VISUALLY CHECKED IN ASSEMBLED CONDITION
6	30-21-51-01-010	VALVE, INTAKE ANTI-ICE	MISSING	MISSING	MISSING	MISSING	MISSING
7	36-11-41-01-050	CHECK VALVE	2293B020000	19579	2293B020000	19579	VISUALLY CHECKED
8	36-11-51-01-050	PNEUMATIC HP BLEED VALVE	6773E010000	02492	6773E010000	02492	VISUALLY CHECKED
9	36-11-52-01-050	BLEED PRESSURE REGULATOR VALVE	6774G010000	6774-09522	6774G010000	6774-09522	VISUALLY CHECKED
10	36-11-57-01-100	CABIN AIR STAGE 10 AIR SOLENOID VALVE	AC69576	EGL396	AC69576	EGL396	VISUALLY CHECKED IN ASSEMBLED CONDITION
11	71-51-49-01-100	BOX RELAY	005RL05	6675	005RL05	6675	VISUALLY CHECKED IN ASSEMBLED CONDITION
12	73-12-41-01-100	FUEL PUMP	5009913G	8938	5009913G	8938	VISUALLY CHECKED IN ASSEMBLED CONDITION
13	73-13-42-01-100	FUEL DIVERTER AND RETURN VALVE	17300G09	6698	17300G09	6698	VISUALLY CHECKED IN ASSEMBLED CONDITION
14	73-22-34-01-280	ELECTRONIC ENGINE CONTROL	824972-7-022	2540-1086	824972-7-026	2540-1086	VISUALLY CHECKED IN ASSEMBLED CONDITION SB 73-0248
15	73-22-35-01-100	DATA ENTRY PLUG	2A3106	V17032	2A4378	V17032	VISUALLY CHECKED IN ASSEMBLED CONDITION
16	73-22-38-01-100	DEDICATED ALTERNATOR STATOR	430201	N13340105	430201	N13340105	VISUALLY CHECKED IN ASSEMBLED CONDITION

ACCESSORIES STATUS

Engine Type: **V2533-A5 SQ03**

TSN: **24228,72**

Work Order: **M-0000323950**

Operator:



Engine SN: **V17032**

CSN: **8349**

Rev. // Issue Date: **00 // 28/FEB/2025**

Issued by: **OMH 983124**

N°	Ref. I.P.C.	Denomination	INCOMING		OUTGOING		
			P/N received	S/N received	P/N installed	S/N installed	Work Acc.
17	73-22-52-01-100	FUEL METERING UNIT	8061-640	14346670	8061-640	14346670	VISUALLY CHECKED IN ASSEMBLED CONDITION
18	73-31-17-01-100	FUEL FLOW TRANSMITTER	9-217-94	291714-154	9-217-94	291714-154	VISUALLY CHECKED IN ASSEMBLED CONDITION
19	73-34-15-01-010	FUEL FILTER DIFFERENTIAL PRESSURE SWITCH	21SN04-300B	P029014	21SN04-300B	P029014	VISUALLY CHECKED IN ASSEMBLED CONDITION
20	73-35-15-01-100	THERMOCOUPLE FUEL TEMPERATURE	22912-200	1128123	22912-200	1128123	VISUALLY CHECKED IN ASSEMBLED CONDITION
21	74-11-38-01-120	IGNITION EXCITER (BOTTOM)	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
22	74-11-38-01-120	IGNITION EXCITER (TOP)	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
23	74-21-43-02-100	IGNITION LEAD (LOWER)	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
24	74-21-43-01-100	IGNITION LEAD (UPPER)	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
25	75-22-41-01-100	NO.4 BEARING COMPARTMENT AIR COOLER (ACAC)	D1876-5000A	0024385909	D1876-5000A	24385909	VISUALLY CHECKED
26	75-23-51-01-100	10 STG TO HPT AIR VALVE	MISSING	MISSING	MISSING CONFIG	MISSING CONFIG	MISSING CONFIG
27	75-23-52-01-100	10 STG TO HPT SOLENOID VALVE	MISSING	MISSING	MISSING CONFIG	MISSING CONFIG	MISSING CONFIG
28	75-24-51-01-100	ACC VALVE	5860016-139	5961	5860016-139	5961	VISUALLY CHECKED
29	75-24-52-01-100	ACC ACTUATOR VALVE	5860017-144	3419	5860017-144	3419	VISUALLY CHECKED IN ASSEMBLED CONDITION
30	75-31-42-01-100	ACTUATOR MASTER LPC BLEED	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
31	75-31-43-01-100	LPC BLEED SLAVE ACTUATOR	1778MK1	N.V.	1778MK1	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
32	75-32-41-01-100	STATOR VANE ACTUATOR	G40000VSVA02	AAG18-874	G40000VSVA02	AAG18-874	VISUALLY CHECKED IN ASSEMBLED CONDITION

ACCESSORIES STATUS

Engine Type: **V2533-A5 SQ03**

TSN: **24228,72**

Work Order: **M-0000323950**

Operator:



Engine SN: **V17032**

CSN: **8349**

Rev. // Issue Date: **00 // 28/FEB/2025**

Issued by: **OMH 983124**

N°	Ref. I.P.C.	Denomination	INCOMING		OUTGOING		
			P/N received	S/N received	P/N installed	S/N installed	Work Acc.
33	75-32-51-01-100	7TH STG HPC BLEED VALVE SOLENOID AFT	AA1051-00	NKB612	AA1051-00	NKB612	VISUALLY CHECKED IN ASSEMBLED CONDITION
34	75-32-51-01-100	7TH HPC BLEED VALVE SOLENOID FWD	AA1051-00	NKB530	AA1051-00	NKB530	VISUALLY CHECKED IN ASSEMBLED CONDITION
35	75-32-51-01-100	7TH STG HPC BLEED VALVE SOLENOID MID	AA1051-00	NKB610	AA1051-00	NKB610	VISUALLY CHECKED IN ASSEMBLED CONDITION
36	75-32-52-01-100	7TH STG HPC BLEED VALVE 7H	AC69924	EFA204	AC69924	EFA204	VISUALLY CHECKED IN ASSEMBLED CONDITION
37	75-32-52-01-100	7TH STG HPC BLEED VALVE 5H	AC69924	N.V.	AC69924	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
38	75-32-52-01-100	7TH STG HPC BLEED VALVE 1H	AC69924	EHK318	AC69924	EHK318	VISUALLY CHECKED IN ASSEMBLED CONDITION
39	75-32-53-01-400	10TH HPC BLEED VALVE SOLENOID	AA1060-00	MN578	AA1060-00	MN578	VISUALLY CHECKED IN ASSEMBLED CONDITION
40	75-32-54-01-400	10TH HPC STAGE BLEED VALVE	5950041-108	5470	5950041-108	5470	VISUALLY CHECKED IN ASSEMBLED CONDITION
41	75-41-15-01-100	NACELLE TEMPERATURE SENSOR	RP162-02	11465	RP162-02	11465	VISUALLY CHECKED
42	77-32-15-01-100	VIBRATION TRANSDUCER	144-171-000-032	AJ74901	144-171-000-032	AJ74901	VISUALLY CHECKED IN ASSEMBLED CONDITION
43	79-21-42-01-100	AIR COOLED OIL COOLER (ACOC)	50069001-1	HEX35-4529	50069001-1	HEX35-4529	VISUALLY CHECKED IN ASSEMBLED CONDITION
44	79-21-43-01-100	FUEL COOLED OIL COOLER (FCOC)	55022001-11	63086	55022001-11	63086	VISUALLY CHECKED IN ASSEMBLED CONDITION
45	79-21-51-01-100	ACOC AIR MODULATING VALVE	8910-620	18910581	8910-620	18910581	VISUALLY CHECKED IN ASSEMBLED CONDITION
46	79-22-43-01-001	FILTER, ASSY OIL SCAVENGE	4A7033B	FAG78329	4A7033B	FAG78329	VISUALLY CHECKED IN ASSEMBLED CONDITION
47	79-22-45-01-050	MAGNETIC - CHIP DETECTOR ASSY #1 - MASTER	1A6794	42231	1A6794	42231	VISUALLY CHECKED
48	79-22-45-01-059	MAGNETIC - CHIP DETECTOR ASSY #2 -No. 1, 2 & 3 BRG SCAVENGE	1A6794	42330	1A6794	42330	VISUALLY CHECKED

ACCESSORIES STATUS

Engine Type: **V2533-A5 SQ03**

TSN: **24228,72**

Work Order: **M-0000323950**

Operator:



Engine SN: **V17032**

CSN: **8349**

Rev. // Issue Date: **00 // 28/FEB/2025**

Issued by: **OMH 983124**

N°	Ref. I.P.C.	Denomination	INCOMING		OUTGOING		
			P/N received	S/N received	P/N installed	S/N installed	Work Acc.
49	79-22-45-01-103	MAGNETIC - CHIP DETECTOR ASSY #3 - NO. 5 BRG SCAVENGE	1A6794	42329	1A6794	42329	VISUALLY CHECKED
50	79-23-51-01-100	NO.4 BEARING SCAVENGE VALVE	32F0003	TE023328	32F0003	TE023328	VISUALLY CHECKED IN ASSEMBLED CONDITION
51	79-32-15-01-200	THERMOCOUPLE OIL ACOC	73984300	12698	73984300	12698	VISUALLY CHECKED IN ASSEMBLED CONDITION
52	79-32-16-01-010	THERMCPL, OIL SCAV TEMP	56B98	DD159	56B98	DD159	VISUALLY CHECKED IN ASSEMBLED CONDITION
53	79-33-15-01-010	OIL PRESSURE TRANSMITTER	41SG240-4	P028136	41SG240-4	P028136	VISUALLY CHECKED IN ASSEMBLED CONDITION
54	79-33-16-01-100	TRANSDUCER PRESS. BRG. NO.4	41SG272-6	P003423	41SG272-6	P003423	VISUALLY CHECKED IN ASSEMBLED CONDITION
55	79-34-15-01-010	LOW OIL PRESSURE SWITCH	21SN04-275A	P027248	21SN04-275A	P027248	VISUALLY CHECKED IN ASSEMBLED CONDITION
56	79-35-16-01-010	SCAVENGE FILTER PRESSURE SWITCH	21SN04-298A	P007339	21SN04-298A	P007339	VISUALLY CHECKED IN ASSEMBLED CONDITION
57	80-13-41-01-200	STARTER PNEUMATIC	790425A9	A20040	790425A9	A20040	VISUALLY CHECKED IN ASSEMBLED CONDITION
58	80-13-41-01-260	STARTER HOUSING-ADAPTER	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
59	80-13-51-01-200	STARTER AIR VALVE	790424-4	A6003	790424-4	A6003	VISUALLY CHECKED IN ASSEMBLED CONDITION
60	24-21-41-01-010	QAD ADAPTER KIT	767669B	4674	767669B	4674	VISUALLY CHECKED IN ASSEMBLED CONDITION
61	24-21-51-01-010	INTEGRATED DRIVE GENERATOR	772292	AAJ4003732	772292	AAJ4003732	VISUALLY CHECKED
62	29-11-51-01-010	HYDRAULIC PUMP	3031863-001	MX728521	3031863-001	MX728521	VISUALLY CHECKED
63	73-22-11-01-010	P2/T2 SENSOR	MISSING	MISSING	MISSING	MISSING	MISSING
64	24-21-43-01-010	IDG CABLE ASSEMBLY	740-5869-507	M5958	740-5869-507	M5958	REPAIRED

ACCESSORIES STATUS

Engine Type: **V2533-A5 SQ03**

TSN: **24228,72**

Work Order: **M-0000323950**

Operator:



Engine SN: **V17032**

CSN: **8349**

Rev. // Issue Date: **00 // 28/FEB/2025**

Issued by: **OMH 983124**

N°	Ref. I.P.C.	Denomination	INCOMING		OUTGOING		
			P/N received	S/N received	P/N installed	S/N installed	Work Acc.
65	71-51-41-01-005	EEC FAN HARNESS	6A8281	X03663	6A8281	X03663	MINOR REPAIRED IN ASSEMBLED CONDITION
66	71-51-42-01-005	EEC FAN HARNESS AND IGNITION SUPPLY HARNESS	745-5870-503	M5127	745-5870-503	M5127	MINOR REPAIRED IN ASSEMBLED CONDITION
67	71-51-43-01-005	GENERAL SERVICE HARNESS	740-5966-529	M5535	740-5966-529	M5535	MINOR REPAIRED IN ASSEMBLED CONDITION
68	71-51-44-01-005	IGNITION SUPPLY HARNESS	MISSING	MISSING	MISSING CONFIG	MISSING CONFIG	MISSING CONFIG
69	71-52-41-01-005	HARNESS - CORE SERVICES	745-5967-517	M2926	745-5967-517	M2926	MINOR REPAIRED IN ASSEMBLED CONDITION
70	71-52-42-01-005	FIRE DETECTION (B) HARNESS	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
71	71-52-43-01-005	EEC HARNESS CORE	6A8172	X04379	6A8172	X04379	VISUALLY CHECKED IN ASSEMBLED CONDITION
72	71-52-44-01-005	EEC LINK HARNESS CORE	6A7569	4535	6A7569	4535	VISUALLY CHECKED IN ASSEMBLED CONDITION
73	71-52-45-01-005	EGT HARNESS	6A5878	6193	6A5878	6193	VISUALLY CHECKED IN ASSEMBLED CONDITION
74	72-32-37-01-100	NF HARNESS, ELECTRICAL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
75	77-32-43-01-100	VIBRATION HARNESS	922-167-000-031	062813	922-167-000-031	062813	VISUALLY CHECKED IN ASSEMBLED CONDITION
76	77-32-43-02-100	VIBRATION HARNESS	922-168-000-031	112016	922-168-000-031	112016	VISUALLY CHECKED IN ASSEMBLED CONDITION
77	72-32-18-01-100	PROBE, FAN SPEED	788-1-13	4341W			VISUALLY CHECKED (RESISTANCE CHECK PASSED)
78	72-32-18-01-200	PROBE, FAN SPEED	788-1-13	4359W			VISUALLY CHECKED (RESISTANCE CHECK PASSED)
79	72-32-18-01-400	PROBE, FAN SPEED	788-1-13	4399W			VISUALLY CHECKED (RESISTANCE CHECK PASSED)
80	72-32-18-01-300	PROBE, TRIM BALANCE	1021-3-13	AE103946			VISUALLY CHECKED (RESISTANCE CHECK PASSED)

ACCESSORIES STATUS

Engine Type: **V2533-A5 SQ03**

TSN: **24228,72**

Work Order: **M-0000323950**

Operator:



Engine SN: **V17032**

CSN: **8349**

Rev. // Issue Date: **00 // 28/FEB/2025**

Issued by: **OMH 983124**

N°	Ref. I.P.C.	Denomination	INCOMING		OUTGOING		
			P/N received	S/N received	P/N installed	S/N installed	Work Acc.
81	73-22-15-01-030	PROBE, TEMP&PRESS DIFSR CASE	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
82	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
83	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
84	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
85	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
86	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
87	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
88	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
89	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
90	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
91	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
92	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
93	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
94	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
95	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
96	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION

ACCESSORIES STATUS

Engine Type: **V2533-A5 SQ03**

TSN: **24228,72**

Work Order: **M-0000323950**

Operator:



Engine SN: **V17032**

CSN: **8349**

Rev. // Issue Date: **00 // 28/FEB/2025**

Issued by: **OMH 983124**

N°	Ref. I.P.C.	Denomination	INCOMING		OUTGOING		
			P/N received	S/N received	P/N installed	S/N installed	Work Acc.
97	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
98	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
99	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
100	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
101	73-13-41-01-020	NOZZLE & SUPPORT ASSY, FUEL	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
102	73-13-43-01-020	FUEL DISTRIBUTION VALVE (FDV)	N.V.	PSGCAM4091	N.V.	PSGCAM4091	VISUALLY CHECKED IN ASSEMBLED CONDITION
103	74-21-41-01-100	PLUG, IGNITER	MISSING	MISSING	CH31991	N.V.	NEW PART
104	74-21-41-01-100	PLUG, IGNITER	N.V.	N.V.	CH31991	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
105	77-21-15-20-100	EGT THERMOCOUPLE	30404-000	1147971	30404-000	1147971	VISUALLY CHECKED (RESISTANCE CHECK PASSED)
106	77-21-15-20-100	EGT THERMOCOUPLE	30404-000	2121279	30404-000	2121279	VISUALLY CHECKED (RESISTANCE CHECK PASSED)
107	77-21-15-20-100	EGT THERMOCOUPLE	30404-000	2021492	30404-000	2021492	VISUALLY CHECKED (RESISTANCE CHECK PASSED)
108	77-21-15-20-100	EGT THERMOCOUPLE	30404-000	2020839	30404-000	2020839	VISUALLY CHECKED (RESISTANCE CHECK PASSED)
109	77-21-43-01-010	EGT HARNESS AND JUNCTION BOX	25975-000	1012974	25975-000	1012974	VISUALLY CHECKED IN ASSEMBLED CONDITION
110	79-11-41-01-001	OIL TANK ASSEMBLY	4A7004-01F	FAG97088	4A7004-01F	FAG97088	VISUALLY CHECKED IN ASSEMBLED CONDITION
111	79-11-51-01-001	OIL TANK PRESSURIZATION VALVE	N.V.	N.V.	N.V.	N.V.	VISUALLY CHECKED IN ASSEMBLED CONDITION
112	79-21-41-01-001	OIL PUMP & FILTER	4A7110B	FAG95982	4A7110B	FAG95982	VISUALLY CHECKED IN ASSEMBLED CONDITION

ACCESSORIES STATUS

Engine Type: **V2533-A5 SQ03**

TSN: **24228,72**

Work Order: **M-0000323950**

Operator:



Engine SN: **V17032**

CSN: **8349**

Rev. // Issue Date: **00 // 28/FEB/2025**

Issued by: **OMH 983124**

N°	Ref. I.P.C.	Denomination	INCOMING		OUTGOING		
			P/N received	S/N received	P/N installed	S/N installed	Work Acc.
113	79-22-41-01-001	OIL SCAVENGE PUMP	4A7124A	FAG95966	4A7124A	FAG95966	VISUALLY CHECKED IN ASSEMBLED CONDITION
114	79-22-45-01-115	MAGNETIC - CHIP DETECTOR ASSY #4 - GEARBOX RH.	1A6794	42238	1A6794	42238	VISUALLY CHECKED IN ASSEMBLED CONDITION
115	79-22-45-01-115	MAGNETIC - CHIP DETECTOR ASSY #5 - ANGLE GEARBOX SCAVENGE	1A6794	42318	1A6794	42318	VISUALLY CHECKED IN ASSEMBLED CONDITION
116	79-22-45-01-115	MAGNETIC - CHIP DETECTOR ASSY #6 - GEARBOX LH	1A6794	42251	1A6794	42251	VISUALLY CHECKED IN ASSEMBLED CONDITION
117	79-22-45-01-115	MAGNETIC - CHIP DETECTOR ASSY #7 - DE-OILER	1A6794	42244	1A6794	42244	VISUALLY CHECKED IN ASSEMBLED CONDITION
118	79-31-15-01-010	OIL QUANTITY TRANSMITTER	76-167-10	AAE0559	76-167-10	AAE0559	VISUALLY CHECKED IN ASSEMBLED CONDITION

08 Engine Condition Monitoring

ETM ^{Plus} Fleet Performance Report:

Avianca

V2500 / A320

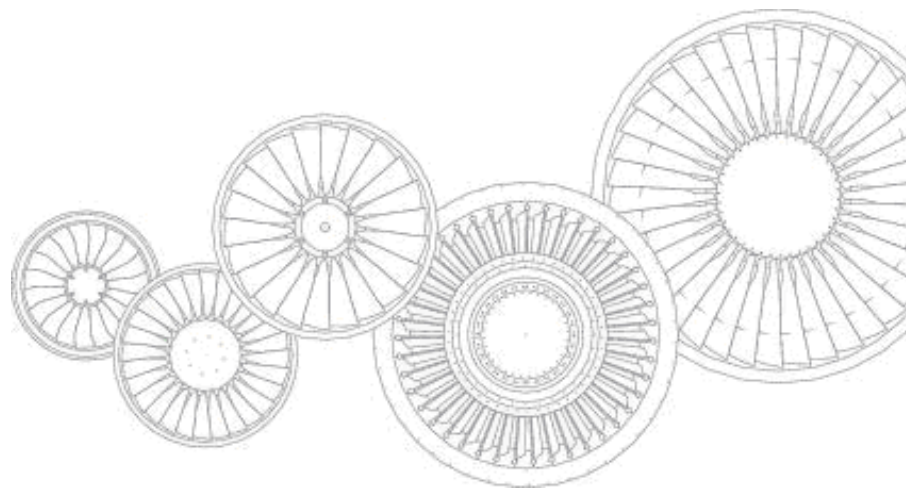


Prepared by

KELLER, Christian

Report creation date (UTC):

27.05.2022 06:56





Diagrams explanation notes

Summary table / charts for the following aircraft / engines:

N692AV: 17036-2

ESTOCK: 17032-0

Table parameter list:

EGTM_sm [°C] (TAKEOFF) -- THRDRT_m_sm [%] (TAKEOFF)

Chart parameter list:

d_EGT [°C] -- d_EGT_sm [°C] (CRUISE)

d_FF [%] -- d_FF_sm [%] (CRUISE)

d_N2 [%] -- d_N2_sm [%] (CRUISE)

d_N1 [%] -- d_N1_sm [%] (CRUISE)

VB1 [A/C units] -- N1_Vib_sm [A/C units] (CRUISE)

VB2 [A/C units] -- N2_Vib_sm [A/C units] (CRUISE)

OIP [psia] -- OIP_sm [psia] (CRUISE)

OIT [°C] -- OIT_sm [°C] (CRUISE)

EGTMar [°C] -- EGTM_sm [°C] (TAKEOFF)

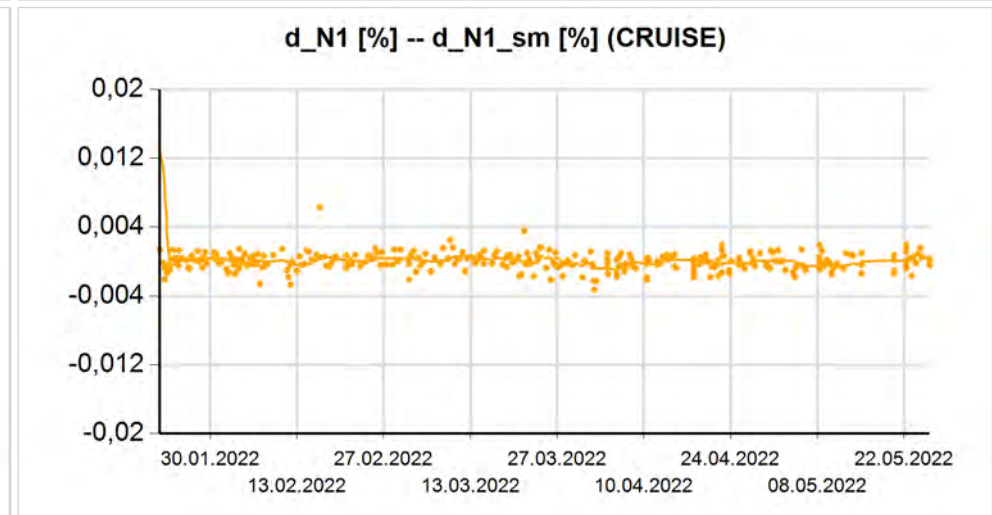
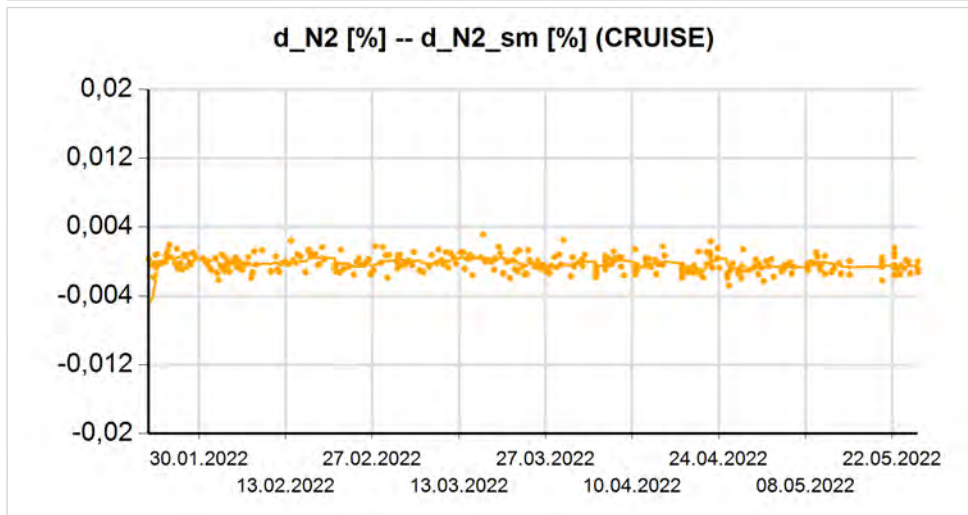
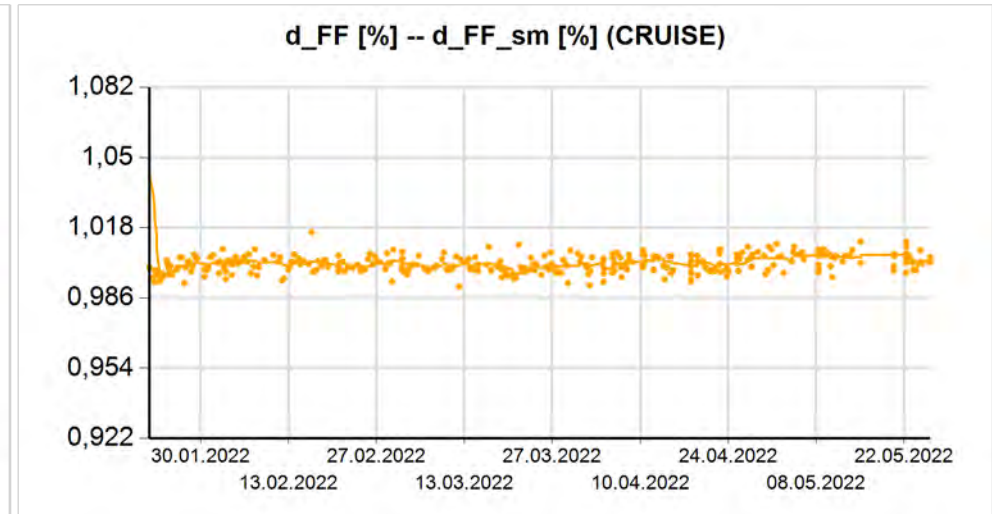
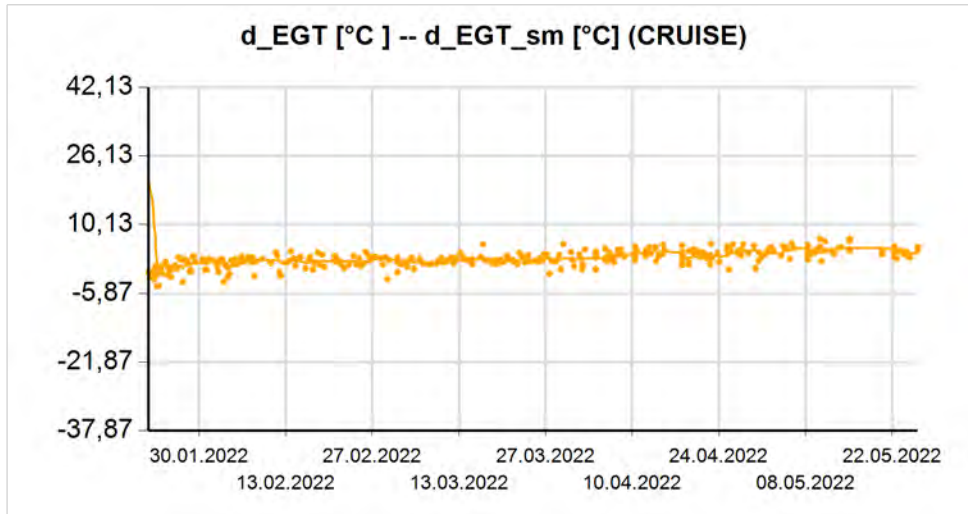
THRDRT_m [%] -- THRDRT_m_sm [%] (TAKEOFF)

Fleet Performance Report: 15.01.2022 - 27.05.2022

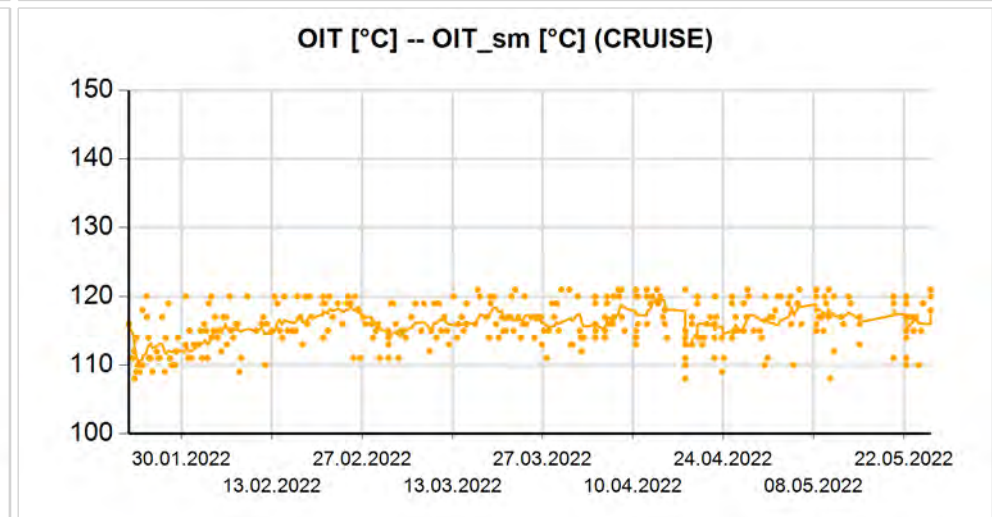
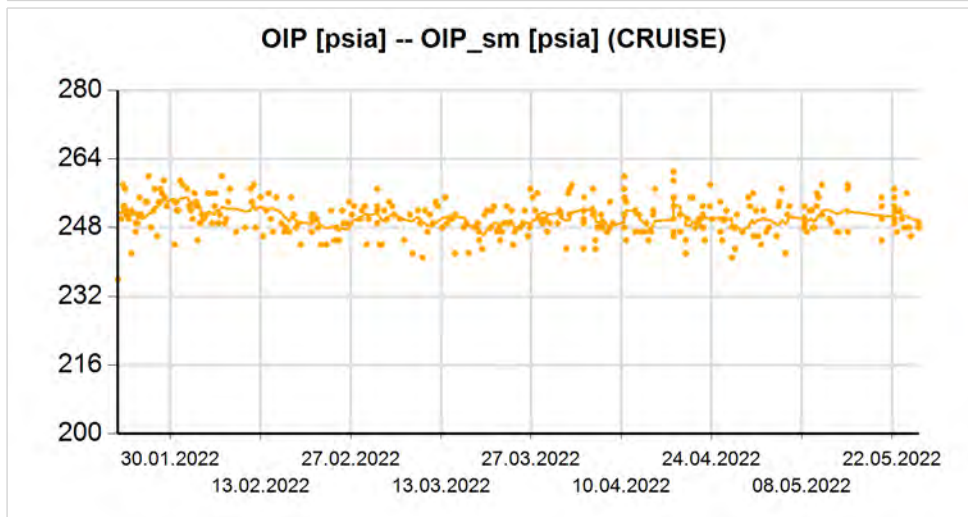
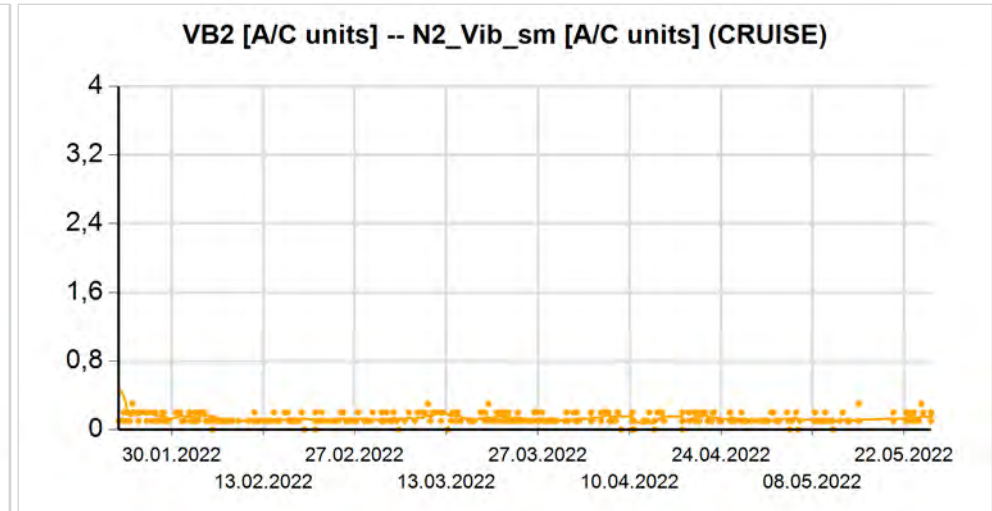
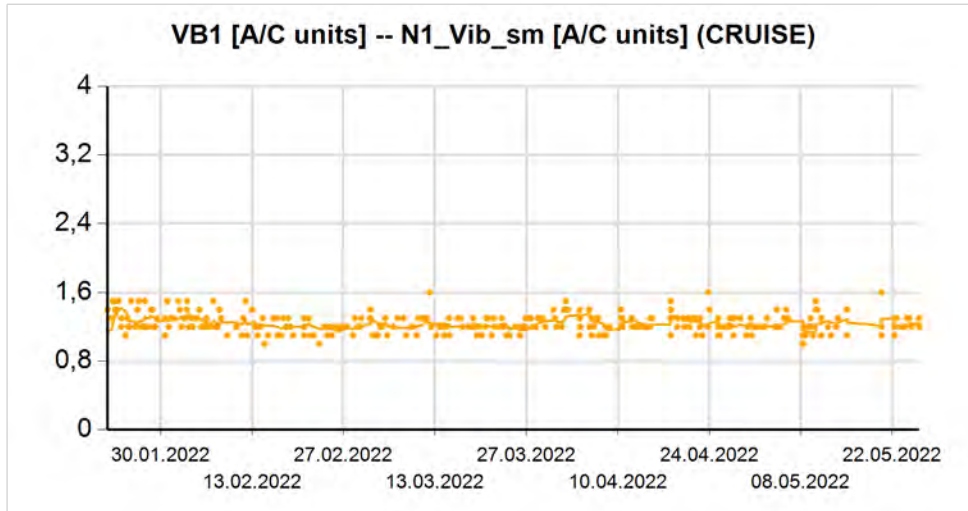
Tail ID	Pos	ESN	EGTM_sm [°C]	THRDRT_m_ sm [%]	NotifReport (date/reason)
N692AV	2	17036	42,2	18,15736	-
ESTOCK	0	17032	12,2	18,95277	-

Fleet Performance Report: 15.01.2022 - 27.05.2022

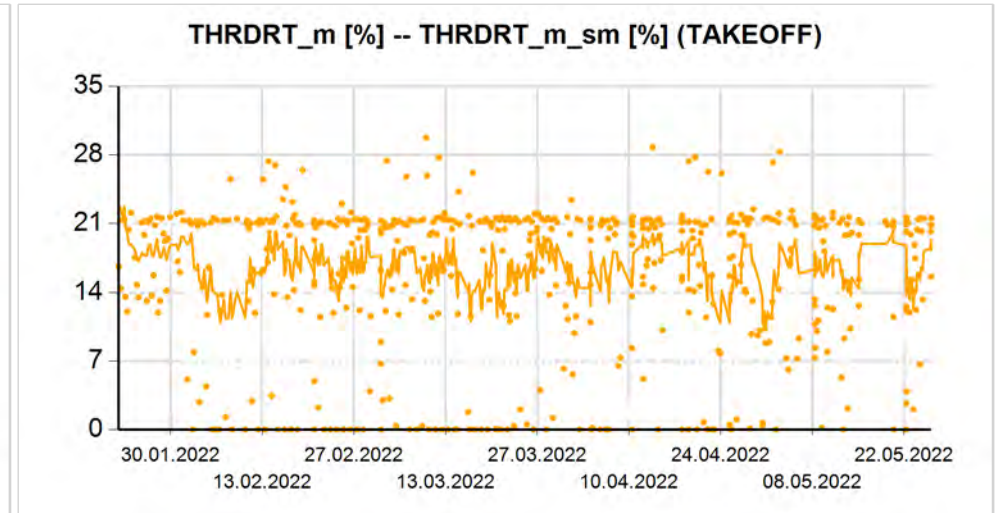
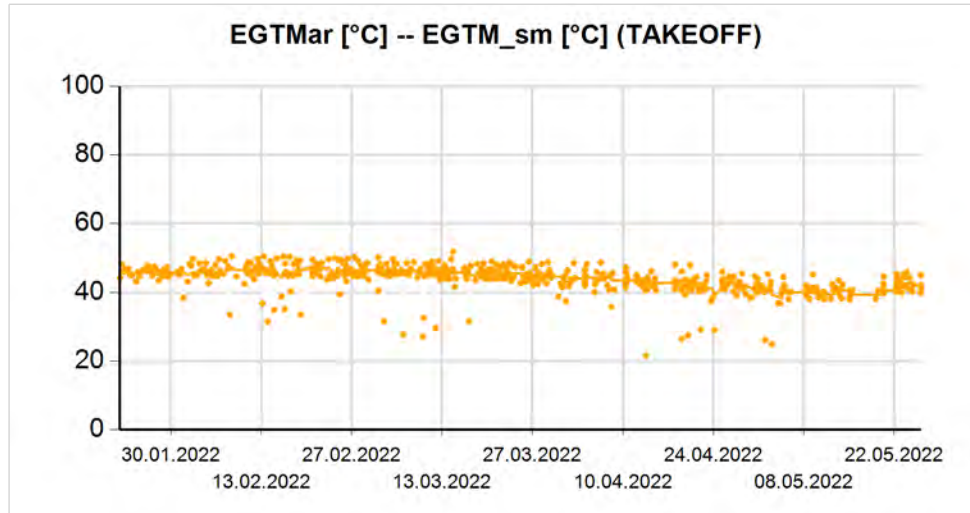
N692AV: 17036-2: filled Circle (Orange) / Solid (Orange) [S]



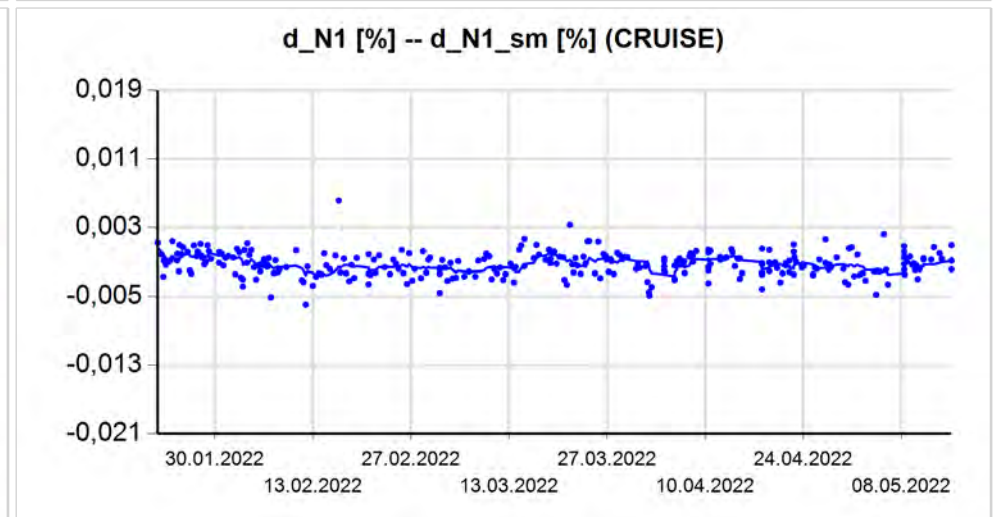
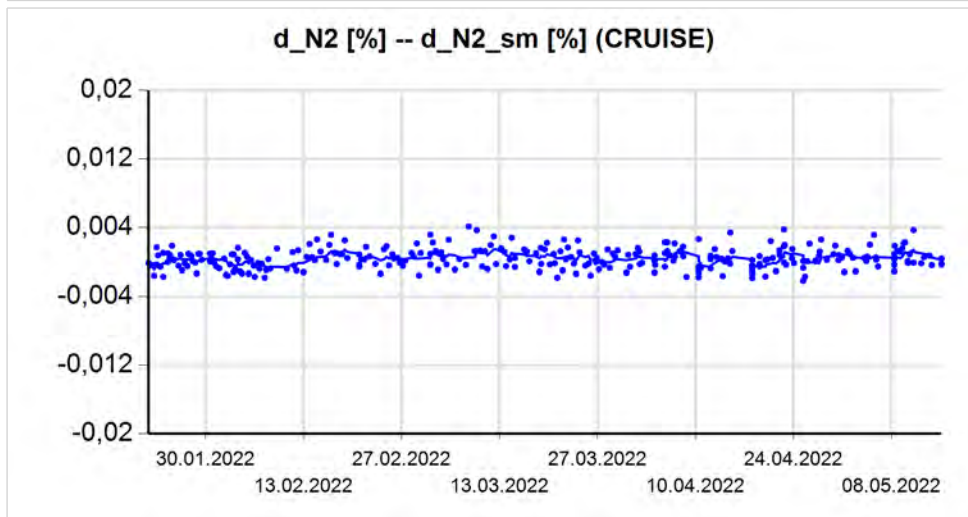
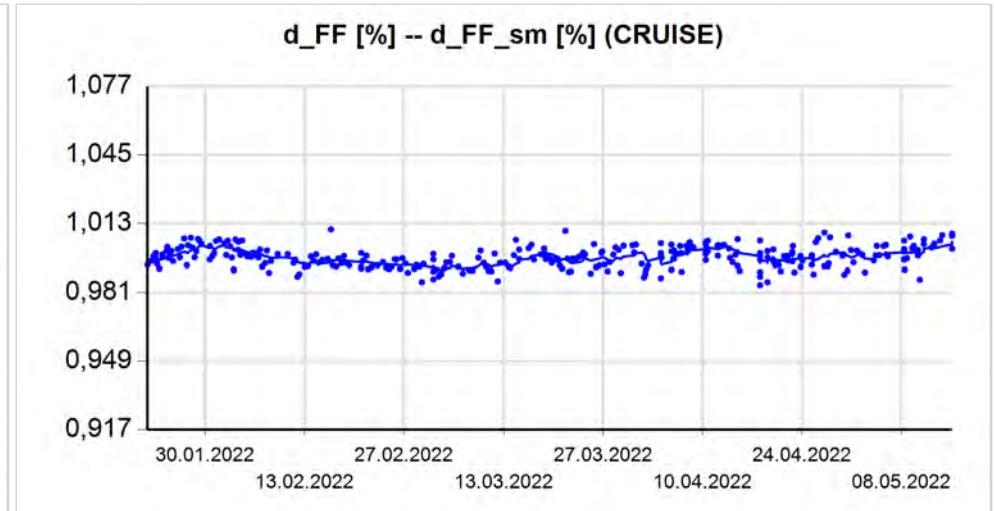
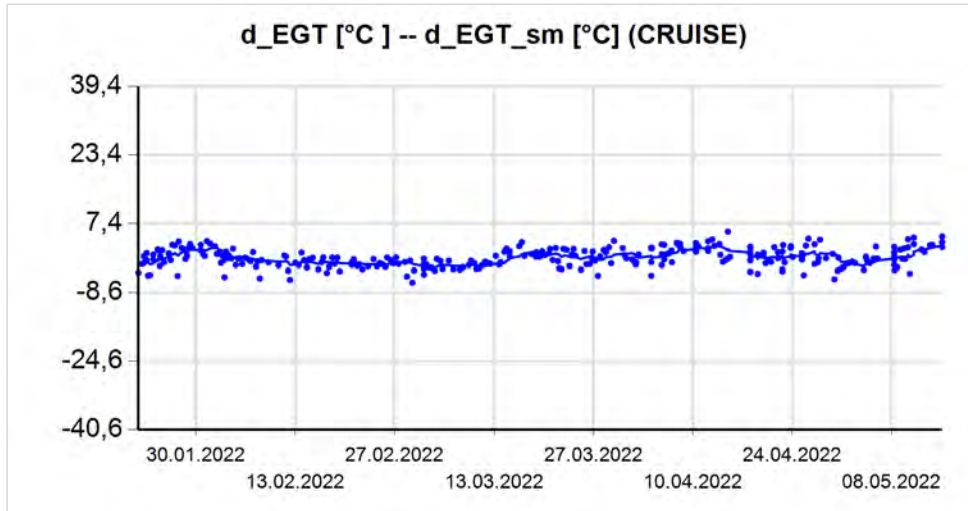
N692AV: 17036-2: filled Circle (Orange) / Solid (Orange) [S]



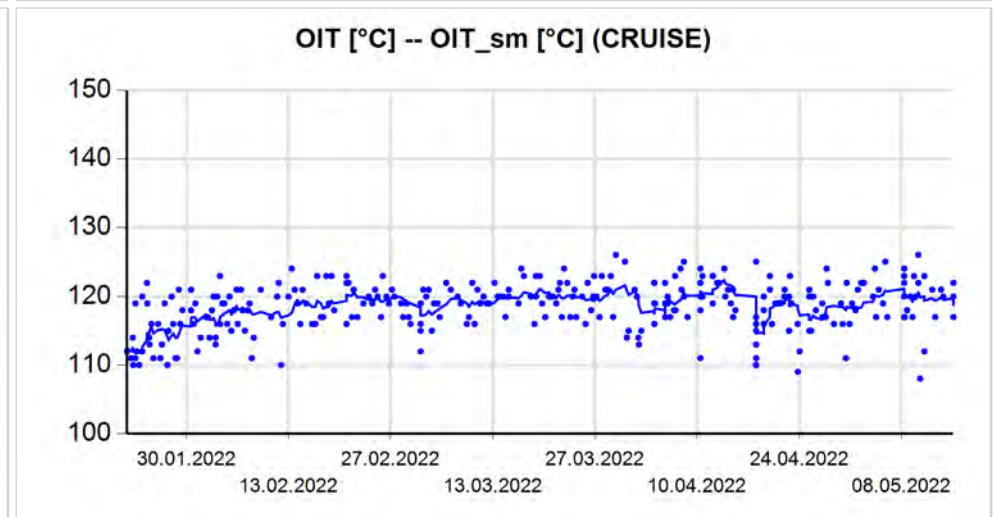
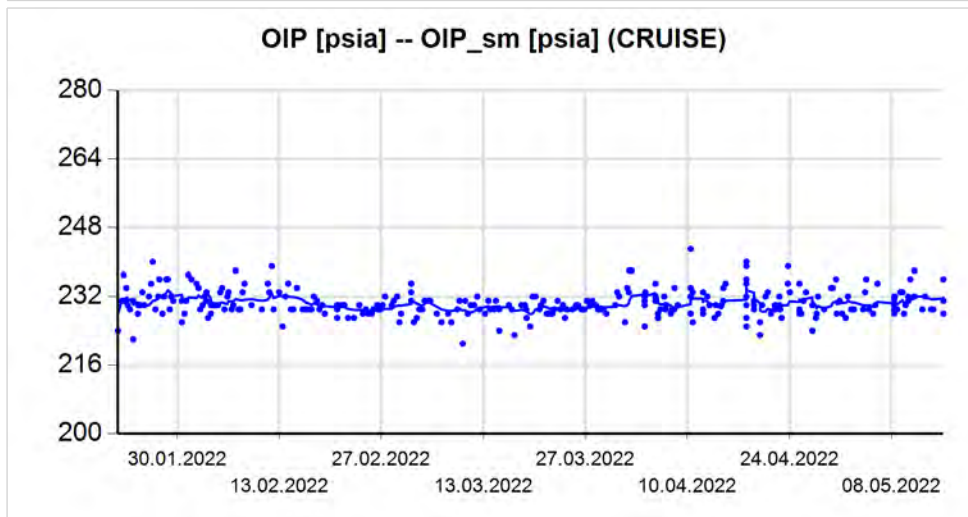
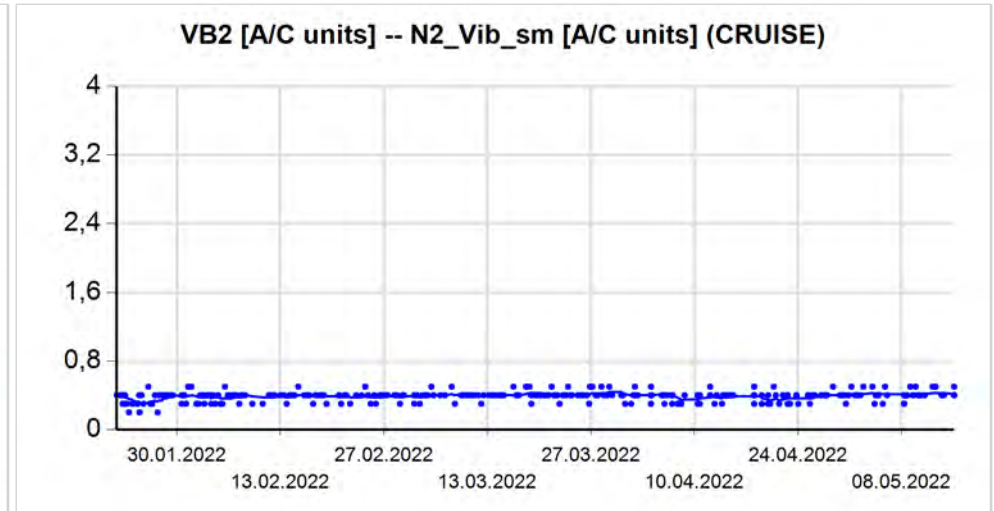
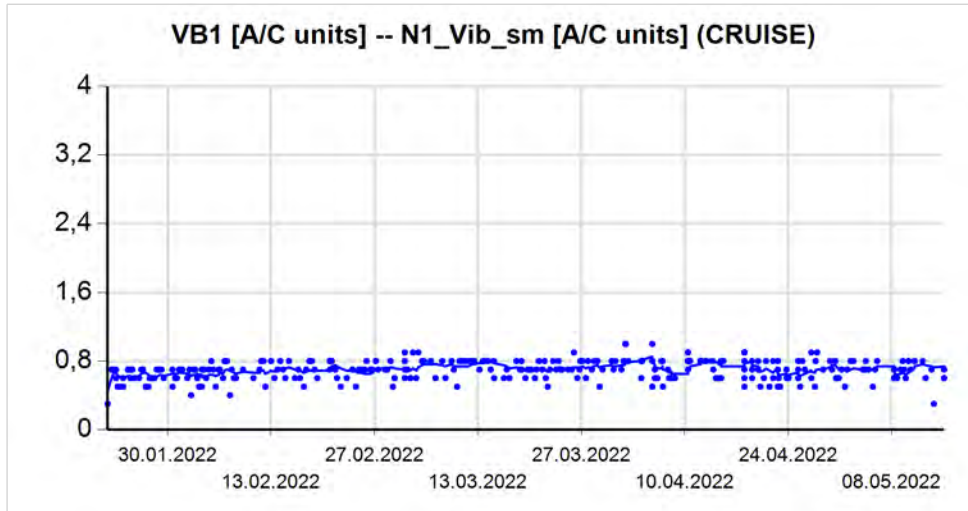
N692AV: 17036-2: filled Circle (Orange) / Solid (Orange) [S]



ESTOCK: 17032-0: filled Circle (Blue) / Solid (Blue) [S]



ESTOCK: 17032-0: filled Circle (Blue) / Solid (Blue) [S]



ESTOCK: 17032-0: filled Circle (Blue) / Solid (Blue) [S]

