

# DOCUMENTS & DRAWINGS FOR WORKING

PROJECT	SEMI-SUBMERSIBLE DRILLING RIG JBF14000 #1		MA/P245
OWNER			
SHIPYARD			
DESIGNER	HUISMAN,		
PREPARED BY	LX	MAIN GENERATOR SET 8xCAT 3616 4400eKW@900rpm	
CHECKED BY			
APPROVED BY			
DATE	SEP04, 2009		
VERSION	01		
HULL NO.	YRO 2007-242		

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## 1.0 TECHNICAL DATA

### 1.1 Diesel Engine

Model	CAT 3616
Type	V-16, 4 cycle
Aspiration	Turbocharger, Aftercooled
Rated power max	4600 bkW
Rated speed	900 rpm
Emission	IMO compliance
Overload 110% for 1 hr	5060 bkW
Rating and emission based on following ambient condition	
Air inlet of the turbocharger	45 °C
Barometric pressure	100 kPa
Relative humidity	60 %
Sea water temperature	32 °C
BSFC (w/ pump +5 %/-0%) @100% load	195.5 g/kW-hr
TBO approx.	40,000 hours
Bore / Stroke	280 / 300mm
Displacement	296 L
Compression ratio	13:1
Firing pressure, max.	17300 kPa
BMEP@ 100% load	2283 kPa
Mean piston speed	9.0 m/s
Engine dry weight (approx)	28,500 kg

### 1.2 Alternator

Type	Synchronous brushless revolving field w/ a direct connected rotating brushless exciter	
Code	Kato 8P12-4700	
Power rating	5500KVA, 4400eKW, 0.8ph+110% overload for 1 hr	
Duty	Continuous	
Rated speed	900rpm, 8poles	
Electrical rating	11.2KV, 60Hz, 3ph	
Connection	6 wire, wye	
Temperature rise	90°C/50°C amb.	
Insulation	Class F	
Enclosure	IP44	
Cooling	Totally Enclosed Water to Air	
Bearing design	Two Sleeve w/ext. lube	
Net weight (approx.)	25,946Kg	

### 1.3 Air Inlet System

Flow of air at 100% load	501.6 m3/min
Air temp. @ air cleaner, max.	45 °C
Air temp. after after-cooler, alarm	60 °C
Intake manifold press. at 100% load	277 kPa
Air inlet restriction, new/max.	1.25/3.8 kPa
AC press. diff at 100%load, clean state	3.4 kPa

**1.4 HT Cooling Water System – JW**

Pump capacity @ 900 rpm	2630	Lpm
Pump rise @ 900 rpm	240	kPa
Pump inlet pressure, min.	30	kPa
Inlet temperature, nom.	90	°C
Inlet temperature, max.	95	°C
Inlet temperature, min.	83	°C
Outlet temperature, alarm	100	°C
Outlet temperature, stop	104	°C

**1.5 LT Cooling Water System - AC/OC**

AC/OC water flow rate required	1560	Lpm
Inlet temperature, nom.	32	°C
Inlet temperature, max.	38	°C

**1.6 Exhaust Gas System**

Gas flow at 100% load	1045	m <sup>3</sup> /min
Stack temp. at 100% load	356	°C
Stack temp. alarm at 100% load	550	°C
Temp. to turbo at 100% load	523	°C
Temp. to turbo alarm at 100% load	650	°C
Back pressure, max	2.54	kPa

**1.7 Fuel Oil System**

Flow rate, supply @ 900rpm	72.0	Lpm
Flow rate, return @ 900rpm	52	Lpm
Pump suction restriction, max.	-20	kPa
Manifold pressure at 100% load	430-676	kPa
Return line back pressure, max.	350	kPa
BSFC (w/ pump +5 %/-0%) @100% load	195.5	g/kW-hr
Fuel oil recommendation	ISO8217 DMA, DMB, DMX (20 cSt max, 1.4 cSt min. @ 38 °C)	

\* Fuel consumption are based on 35API, 16°C fuel having a lower heating value of 42,780 kJ/kg used at 29°C with density of 838.9 g/L

**1.8 Lubricating Oil System**

Lube oil flow	1117	Lpm
Manifold pressure, nom.	380	kPa
Manifold press, alarm (650-900rpm)	320	kPa
Manifold pressure, alarm (0-655rpm)	120	kPa
Manifold press. stop (650-900rpm)	260	kPa
Manifold pressure, stop (0-655rpm)	105	kPa
Manifold temperature, alarm	92	°C
Manifold temperature, stop	98	°C
Manifold temperature, nom.	85	°C
Prelube pump cap. continuous	23	Lpm
Prelube pump cap. intermittent	76	Lpm
Lube oil filter diff. press. max.	104	kPa
BSOC at 100% load	0.7	g/ kw-hr



Lube oil viscosity grade:	SAE40 or SAE15W40
Lube oil API classification:	CF
Caterpillar lube oil name:	DEO (API CG-4)

### **1.9 Starting Air System**

Air pressure, nominal	1225	kPa
Air pressure, min.	620	kPa
Air pressure, max.	1575	kPa
Low air pressure, alarm	850	kPa

## 2.0 SCOPE OF SUPPLY

### 2.1 General

The scope of supply listed below are based on single engine unless otherwise specified

Engine rotation viewed fr. rear of flywheel	CCW
Engine service side viewed fr. rear of flywheel	Right Hand
Power supply on board	480V/60Hz/3ph
	230V/1ph
	24V DC

### 2.2 Air Intake System

Front mounted turbocharger, BASIC 297, engine oil lubricated  
 Aftercooler, fresh water, corrosion resistant coated (air side)  
 Air inlet adapter (90 deg.), mounted to turbocharger inlet  
 Air cleaner, vertical, standard duty, normal volume  
 Air vertical support bracket  
 Air inlet shutoff

### 2.3 Control System

Woodward 2301D electronic governor with load sharing interface, hydraulic actuator, 24VDC power, loose supplied  
 Load sharing control is not our scope of supply

### 2.4 Cooling Water System

Separate cooling circuit suitable for sea water 32°C max  
 Plate type heat exchanger for JW circuit, titanium 0.4 mm, accessory module mounted, based on 32°C SW and sized to supply JW  
 Plate type heat exchanger for AC/OC/fuel/alternator cooler, titanium 0.4 mm, accessory module mounted, based on 32°C SW and size to supply these components with 38°C freshwater  
 Plate-type oil cooler, stainless steel  
 Plate-type fuel cooler, stainless steel  
 Jacket water pump, gear driven  
 AC/OC pump, gear driven  
 Boost pump, gear driven, not self-priming  
 Expansion tank, accessory module mounted  
 JW thermostat, 90°C  
 AC/OC thermostat, 32°C  
 Heat recovery connection, including thermostat, loose supplied  
 JW/LO heater, 480V /3ph/60Hz, base mounted  
 Cooling connections will include a single seawater inlet and a single seawater outlet  
 ANSI connection incl. a single sea water inlet and a single seawater outlet  
 Flexible connection  
 Sea water pump not included

### 2.5 Exhaust System

Dry gas tight exhaust manifolds with soft shields  
 Exhaust adapters, rectangular to round, vertical orientation, loose supplied  
 Exhaust expander, 14"-18", loose supplied  
 Exhaust flexible fitting, 18", loose supplied

Exhaust weld flange, 18", loose supplied  
Exhaust silencer not included

## 2.6 Fuel Oil System

Fuel unit injectors at each cylinder  
Fuel transfer pump, gear driven  
Manual fuel priming pump  
Duplex fuel filter, engine mounted  
[Duplex fuel strainer, shipped loose, customer remote mounted](#)  
Flexible conn, At least 500mm long hose, incl. fuel pump inlet and return oil out let, loose supplied, with type approved or certificated by class

## 2.7 Alternator & Accessories

### Alternator & Accessories

- ◆ Basler DECS 200 digital voltage regulator (supplied loose). 0.25% regulation, field over volt & over current, Exc. Diode Monitor, Under/Overvolt, Loss of Sensing, Under/Over-exc. Limiter, Auto Tracking & Switching to Manual Control, VAR/PF Control, with reactive load sharing interface
- ◆ Reactive load sharing control in not our scope of supply
- ◆ PMG (300% SC 10 Sec) for Regulator Power
- ◆ Kato shall supply & mount coupling hub suitable for Caterpillar 3600 series engine
- ◆ Primary bearing lube oil pump for both exc. and drive end sleeve bearings. Oil. Oil Cooler, Pre-lube pump, and all piping by others
- ◆ Sleeve Bearing oil line connection fittings for external lubrication (per bearing). 60 Deg C maximum oil to be supplied by customer
- ◆ Drive End Sleeve Bearing -non-insulated - self lubricating
- ◆ Exciter End Sleeve bearing - insulated - self lubricating
- ◆ Stand-Off Terminal Connectors, mounted in outlet box.
- ◆ (6) Kato supplied CT's for differential protection. Three mounted in outlet box, three furnished loose for customer mounting.
- ◆ Oversized Outlet Box
- ◆ Double Tube copper-nickel air to water heat exchanger w/water leak detector. External Maximum 38°C water is required and to be supplied by others.
- ◆ Special frame for water cooled totally enclosed alternators.
- ◆ Six - 100 ohm RTD's embedded in stator windings for temperature monitoring, 2 per phase.
- ◆ Two - 100 ohm RTD's installed in bearing housing for temperature monitoring (1 per brg)
- ◆ Thermostat for commercial space heaters
- ◆ Commercial Space Heater, Single Phase 250V or less.
- ◆ ABS Test & Certification. Includes Air-gap, Over-speed Test, Overload Test, and Steady state short circuit Test. Heat Test on first unit may be required.
- ◆ Heat test per IEEE 115 for first unit only.
- ◆ Witness of Tests by Customer or Customers Agent per day for one unit only.
- ◆ Calculated  $X'D = 0.137$  saturated per request
- ◆ IP 55 Terminal Box located LHSFEE with bottom cable entry
- ◆ 2/3's Stator Winding Pitch
- ◆ Inrush current reduction module for AVR, mounted in switchboard, shipped loose

### Spares

- ◆ Spare Rectifier/Suppressors (2), one (1) set per shipset
- ◆ Spare Bearing Liners, one (1) set per shipset

## 2.8 Lube Oil System - Generator

Primary mechanical lube oil pump, generator driven

Lube oil module:

- ◆ Pre-lube/standby electrical lube oil pump, 480 V, 3 ph, 60 Hz
- ◆ Lube oil tank
- ◆ Heat exchanger, 38C fresh water
- ◆ Lube oil filter
- ◆ Lube flow divider
- ◆ Module incl. wiring of the unit, motor starter, 24VDC controls, NEMA 4 enclosure, mounting bracket for mounting the unit to the base, lines group for connecting to the generator bearings
- ◆ Redundant air pre-lube pump for black start

## 2.9 Lube Oil System - Engine

Breather, crankcase top-mounted

Custom dry sump base assembly with an integral sump in the base assembly for 15 deg. static and 22.5 deg dynamic tilt capability

Lube oil pump, gear driven

Duplex lube oil filter, paper element, accessory module mounted

Centrifugal oil filters with single shutoff, service side engine mounted

Oil filter and dipstick valve

Oil pressure regulating valve

Crankcase explosion relief

Prelube pump, continuous mode, electrical driven, incl. 24VDC motor starter, engine base mounted

Backup/black start prelube pump, intermittent mode, air driven, base mounted

Lube oil is not scope of supply

## 2.10 Monitoring System

### 2.10.1 Genset Monitoring System (GMS)

Providing protection, monitoring and control, housed in a NEMA 4 (IP66) enclosure, sensors are wired to GMS

Control panel on accessory module

10 inch color monitor, displays all engine parameters and alarm annunciation

Start / prelube control switch

Fuel control switch

Emergency stop button

Speed control switch

Contacts are available for customer use.

- ◆ Selection of local/remote control of engine
- ◆ Selection of idle/rated control of engine
- ◆ Equipped for remote communication
- ◆ Four 4-20 mA outputs (programmable)
- ◆ Relay contact signals to the remote monitoring system

Gauge

- ◆ Engine hour meter
- ◆ Digital tachometer
- ◆ Starting air pressure gauge

Light

- ◆ Pre-lube status light
- ◆ Summary alarm light

- ◆ Summary shutdown light
- ◆ PLC failure light

#### Engine sensors

##### Contactors:

- ◆ Lube oil pressure (hi/low speed)
- ◆ Jacket water pressure
- ◆ AC/OC pressure
- ◆ Start air pressure
- ◆ Crankcase pressure

##### 4-20 mA transducers:

- ◆ Lube oil pressure (to filter/to engine)
- ◆ Fuel pressure (to filter/to engine)
- ◆ Inlet air manifold pressure

##### RTD (PT100):

- ◆ Lubricating oil to engine temperature
- ◆ Inlet air manifold temperature
- ◆ Fuel to engine temperature
- ◆ AC/OC inlet temperature
- ◆ Jacket water outlet temperature (alarm)
- ◆ Jacket water outlet temperature (shutdown)
- ◆ Generator rear bearing temperature
- ◆ Generator front bearing temperature
- ◆ Generator stator A temperature
- ◆ Generator stator B temperature
- ◆ Generator stator C temperature

##### Switches:

- ◆ Jacket water detector
- ◆ Metal particle detector
- ◆ Starting oil pressure or detector

##### Thermocouples:

- \* Exhaust thermocouple temperatures (one per cylinder plus inlet to turbine and stack)

#### MODBUS communication

All monitored parameters and status available except engine start/stop control witch is connected by hardwire

#### Alarms

##### Pressures:

- ◆ Low oil pressure
- ◆ High oil filter differential
- ◆ Low fuel pressure
- ◆ High fuel filter differential
- ◆ High inlet air manifold pressure
- ◆ Low starting air pressure
- ◆ Low jacket water pressure
- ◆ Low AC/OC water pressure
- ◆ Low raw/sea water pressure (customer supplied contact)

##### Temperatures:

- ◆ High lube oil temperature
- ◆ High inlet air manifold temperature
- ◆ High fuel temperature
- ◆ High AC/OC inlet temperature
- ◆ High jacket water outlet temperature

- ◆ High generator rear bearing temperature
- ◆ High generator front bearing temperature
- ◆ High generator stator A temperature
- ◆ High generator stator B temperature
- ◆ High generator stator C temperature
- ◆ High individual exhaust port temperature
- ◆ High turbine inlet temperature
- ◆ High exhaust stack temperature
- ◆ High exhaust port deviation temperature
- Other:
  - ◆ Low battery voltage
  - ◆ Low oil level
  - ◆ Sensor failure
  - ◆ Jacket water detection
  - ◆ Low coolant level (switch is supplied with expansion tank)
  - ◆ Metal particle detection

#### Shutdowns

##### Pressure:

- ◆ Low oil pressure
- ◆ High crankcase pressure
- ◆ Temperature:
  - ◆ High jacket water temperature
  - ◆ High lube oil temperature
  - ◆ High generator bearing temperature
- ◆ Other:
  - ◆ Metal particle detector
  - ◆ Engine overspeed
  - ◆ Customer shutdown (normally open contact customer supplied)

#### Programmable Inputs

2 of customer supplied RTD's

2 of customer supplied 4-20mA (0-10 VDC) sensors

3 of discrete alarms, and 3 of discrete shutdowns

#### Beacon and horn

##### Oil mist detector incl. drain GP

#### Engine mounted gauges

Fuel pressure

Lube oil pressure

Inlet air restriction

#### Other engine protection

Cylinder pressure relief valve

Mechanical cylinder press GA valve

## 2.10.2 Remote Monitoring

The items of remote monitoring, alarm and protection are complied with ABS class notation of ACCU.

All monitored parameters, alarms, protection and engine status are available on communication port (RS232, RS422 or RS485 are available).

**2.11 Mounting and Base**

Skids mounted, three point mounting, 36" I-beam base  
Vibration isolators, shipped loose

**2.12 Starting Air System**

Air start motors, vane type, incl. silencer, lubricator, air strainer, shutoff valve and flexible hose, max. air pressure 1575 kPa  
Air pressure reducing valve containing relief valve, 3100-850 kPa, loose supplied, shipyard installed  
Flexible hoses  
Starting air vessel not included

**2.13 General Items**

High inertia flywheel  
Electric barring device, 480V/3ph/60Hz motor, equipped with a pendant switch on a 40 ft. cord. Incl. 50:1 manual barring device and a 24 VDC motor starter  
Standard damper guard  
Flywheel and coupling guard  
Torsional coupling for two bearing generator, ABS certificafied  
Engine lifting eyes  
Custom accessory module  
Paint, CAT yellow  
Shrink wrap and tarpaulin protection for transportation and storage  
Approved custom genset TVA, one per shipset  
Noise and vibration technical data  
Test/commissioning procedure  
Factory Acceptance Test for one gen-set per shipset

**2.14 Spare Parts & Tools**Spare parts kit

As per ABS recommened for unrestricted voyage zone and CAT standard, one (1) set per shipset

Service tool group

3600 specialized tooling, one (1) set per shipset  
Manuel service pump, one (1) per shipset  
Protection system calibration kit, one (1) set per shipset  
Oil mist detector tool kit, one (1) set per shipset

**2.15 Drawing & Document**

Drawing & Document for approval	7 paper+1CD per shipset
Drawing & Document for working	7 paper+1CD per shipset
Parts Book	7 paper+7CD per shipset
Service Manual	7 paper per shipset
Technical. Manual	7 paper per shipset
ABS Engine Certificates	1 per engine
ABS Generator Certificates	1 per generator
IMO EIAPP Certification	1 per engine

# LAFAYETTE POWER SYSTEMS

## Generator Specifications

Price List Code:

Spec. No: 8P12-4700

Printed: 10/16/2008

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### \*\*\*\*\* Specifications \*\*\*\*\*

Poles	8
Excitation:	PMG
Pitch:	0.667
Connection:	Wye
Max Overspeed (60 sec)	125%
Number of Bearings	Two
Number of Leads	Six
Number of Terminals	Four

### \*\*\*\*\* Ratings \*\*\*\*\*

Power	4400 EkW
K.V.A.	5500
P.f.	0.8
Voltage-L.L.	11,200 V
Voltage-L.N.	6467 V
Current-L.L.	284 A
Frequency	60 Hz
Speed	900 RPM

### \*\*\*\*\* Efficiency and Heat Dissipation \*\*\*\*\*

(As per NEMA and IEC at 95°C)

Load PU	Kilowatts	Efficiency	Heat Rejection
0.25	1100.0	92.0%	326461 BTU/hr
0.50	2200.0	95.3%	370309 BTU/hr
0.75	3300.0	96.2%	444896 BTU/hr
1.00	4400.0	96.5%	544665 BTU/hr

### \*\*\*\*\* Temperature & Insulation Data \*\*\*\*\*

Ambient Temperature	50 °C
Temperature Rise	90 °C
Insulation Class	F (155 °C)
Insulation Resistance	100 Megaohms
(as shipped)	(at 40 °C )

### \*\*\*\*\* Fault Currents \*\*\*\*\*

Instantaneous 3-Ø symmetrical fault current	2070 Amps
Instantaneous L-N symmetrical fault current	2709 Amps
Instantaneous L-L symmetrical fault current	1682 Amps

### \*\*\*\*\* Exciter Armature Data \*\*\*\*\*

(at full load, 0.8 p.f.)	
Voltage	274 V
Current	102 A

### \*\*\*\*\* Time Constants \*\*\*\*\*

OC Transient - Direct Axis	T <sup>DO</sup>	4.436 Sec
SC Transient - Direct Axis	T <sup>D</sup>	1.113 Sec
OC Subtransient - Direct Axis	T <sup>DO</sup>	0.044 Sec
SC Subtransient - Direct Axis	T <sup>D</sup>	0.031 Sec
OC Subtransient - Quadrature Axis	T <sup>QO</sup>	0.022 Sec
SC Subtransient - Quadrature Axis	T <sup>Q</sup>	0.006 Sec
Armature SC	TA	0.077 Sec

### \*\*\*\*\* Resistances \*\*\*\*\*

Base Impedence	22.807 ohms
Stator (at 25 °C)	0.086 ohms
Field (at 25 °C)	1.86 ohms
Zero Sequence R0	0.26 ohms
Positive Sequence R1	0.11 ohms
Short Circuit Ratio	1.21

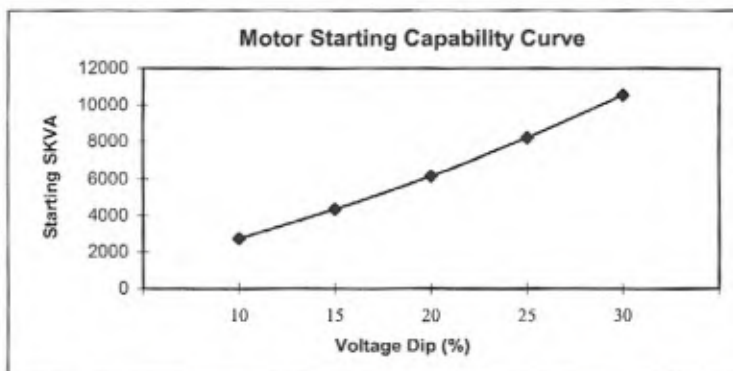
### \*\*\*\*\* Reactances \*\*\*\*\*

		Saturated		Unsaturated	
		Per Unit	Ohms	Per Unit	Ohms
Subtransient - Direct Axis	X <sup>''D</sup>	0.137	3.1	0.161	3.7
Subtransient - Quadrature Axis	X <sup>''Q</sup>	0.173	3.9	0.203	4.6
Transient - Direct Axis	X <sup>'D</sup>	0.197	4.5	0.223	5.1
Transient Quadrature Axis	X <sup>'Q</sup>	0.159	3.6	0.604	13.8
Synchronous - Direct Axis	X <sup>D</sup>	0.830	18.9	1.052	24.0
Synchronous - Quadrature Axis	X <sup>Q</sup>	0.459	10.5	0.604	13.8
Negative Sequence	X <sup>2</sup>	0.155	3.5	0.182	4.2
Zero Sequence	X <sup>0</sup>	0.022	0.5	0.025	0.6



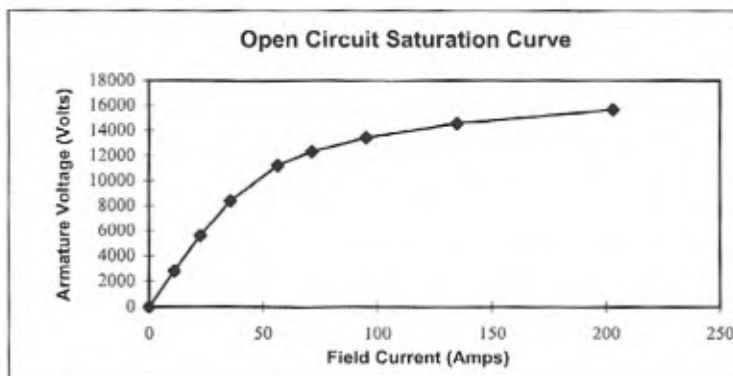
## \*\*\*\*\* Motor Starting Capability Data \*\*\*\*\*

Voltage Dip (%)	Starting SKVA
10	2737
15	4346
20	6157
25	8210
30	10555



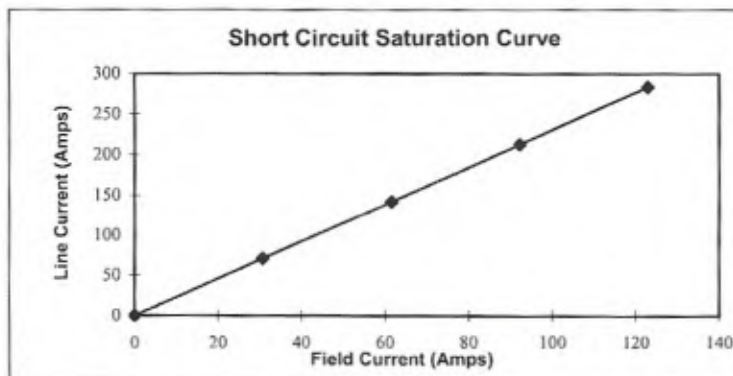
## \*\*\*\*\* Open Circuit Saturation Data \*\*\*\*\*

Field Current (Amps)	Armature Voltage (Volts)
0.0	0
11.1	2800
22.6	5600
35.7	8400
56.5	11200
71.5	12320
95.2	13440
134.8	14560
203.1	15680



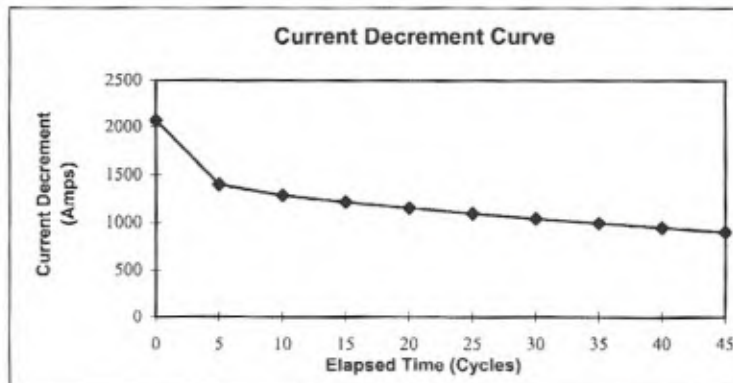
## \*\*\*\*\* Short Circuit Saturation Data \*\*\*\*\*

Field Current (Amps)	Line Current (Amps)
0.0	0.0
30.8	70.9
61.5	141.8
92.3	212.6
123.0	283.5



## \*\*\*\*\* Current Decrement Data \*\*\*\*\*

Elapsed Time (Cycles)	Current Decrement (Amps)
0	2070
5	1404
10	1289
15	1219
20	1155
25	1096
30	1042
35	992
40	945
45	901




**3.0 INSTALLATION DRAWING**

No	Drawing No	REV.	Drawing Name	Total Pages
1	332-2698-01	01	Genset Right Side View	1
2	-02	01	Genset Rear & Front View	1
3	-03	01	Genset Left Side View	1
4	-04	01	Genset Top View	1
5	-05	01	Genset Plan View (TBA)	1
6	-06	01	Detail Sheet	1
7	-07	01	Detail Sheet	1
8	-08	01	Detail Sheet	1
9	-09	01	Detail Sheet	1
10	-10	01	Detail Sheet	1
11	-11	01	Connection Data Sheet	1
12	-12	01	3616 Cooling Schematic	1
13	7W4451	01	Erection and Handling Procedure	1



CERTIFIED TO BUILD

THE  LINE IS USED AS A LINEAR DATUM FOR THE VIEW IN WHICH IT APPEARS. DIMENSIONS ARE PRINTED PARALLEL TO THE DATUM LINE TO WHICH THEY PERTAIN. DIMENSIONS WITH ARROW-HEADS DO NOT PERTAIN TO THE DATUM LINES EXCEPT IN AUXILIARY VIEWS.

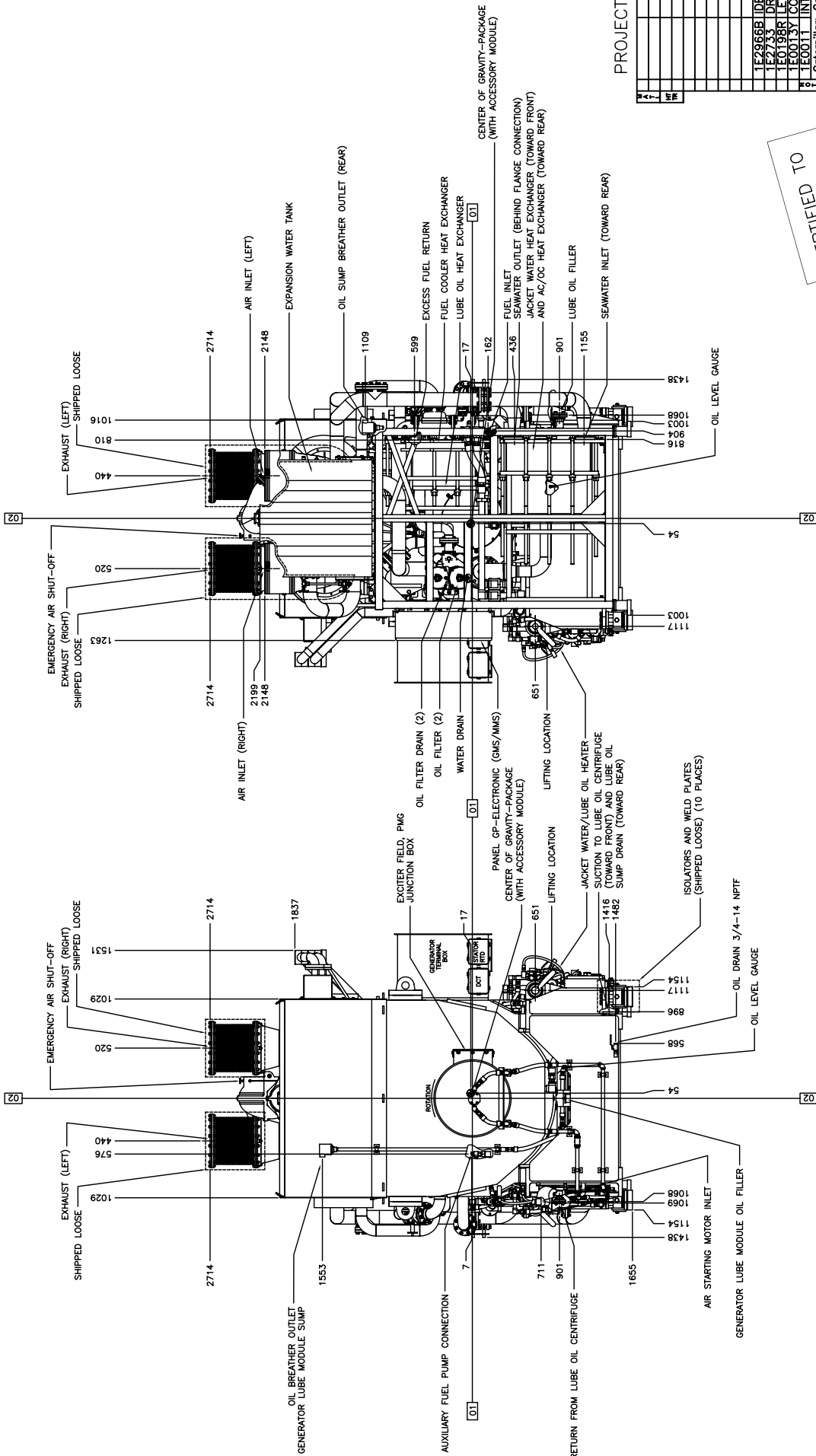
RIGHT HAND SERVICE  
RIGHT HAND COOLANT CONNECTIONS  
SCALE 1=20

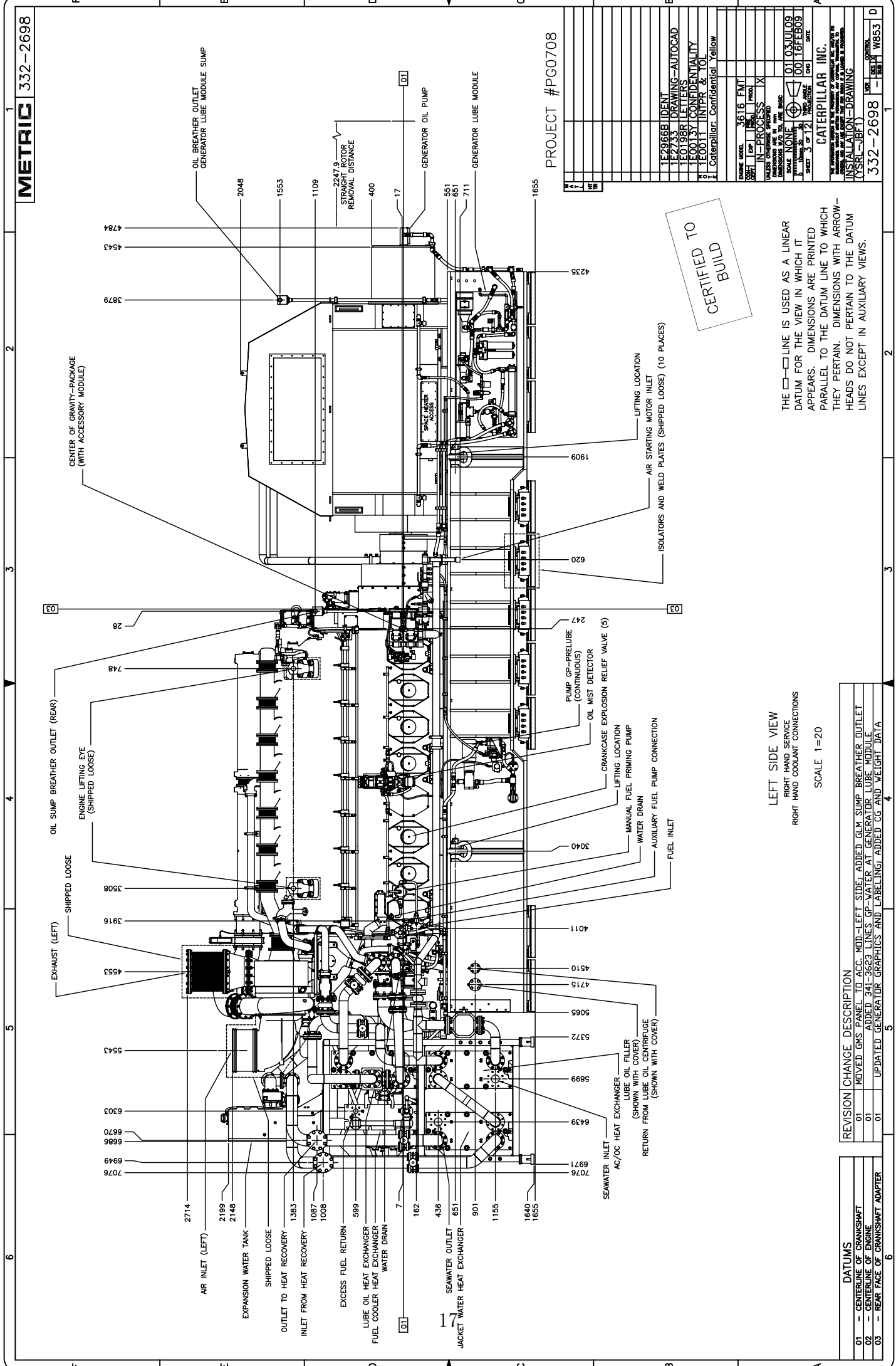
REVISION CHANGE DESCRIPTION

01	MOVED GMS PANEL TO ACC. MOD-LEFT SIDE, ADDED GLM SUMP BREATHER OUTLET
01	ADDED 341-3623 LINES GP-WATER AT GENERATOR LUBE MODULE
01	UPDATED GENERATOR GRAPHICS AND LABELING/ ADDED CG AND WEIGHT DATA

DATUMS

01	CENTERLINE OF CRANKSHAFT
02	CENTERLINE OF ENGINE
03	REAR FACE OF CRANKSHAFT ADAPTER





PROJECT #PG0708

CERTIFIED TO BUILD

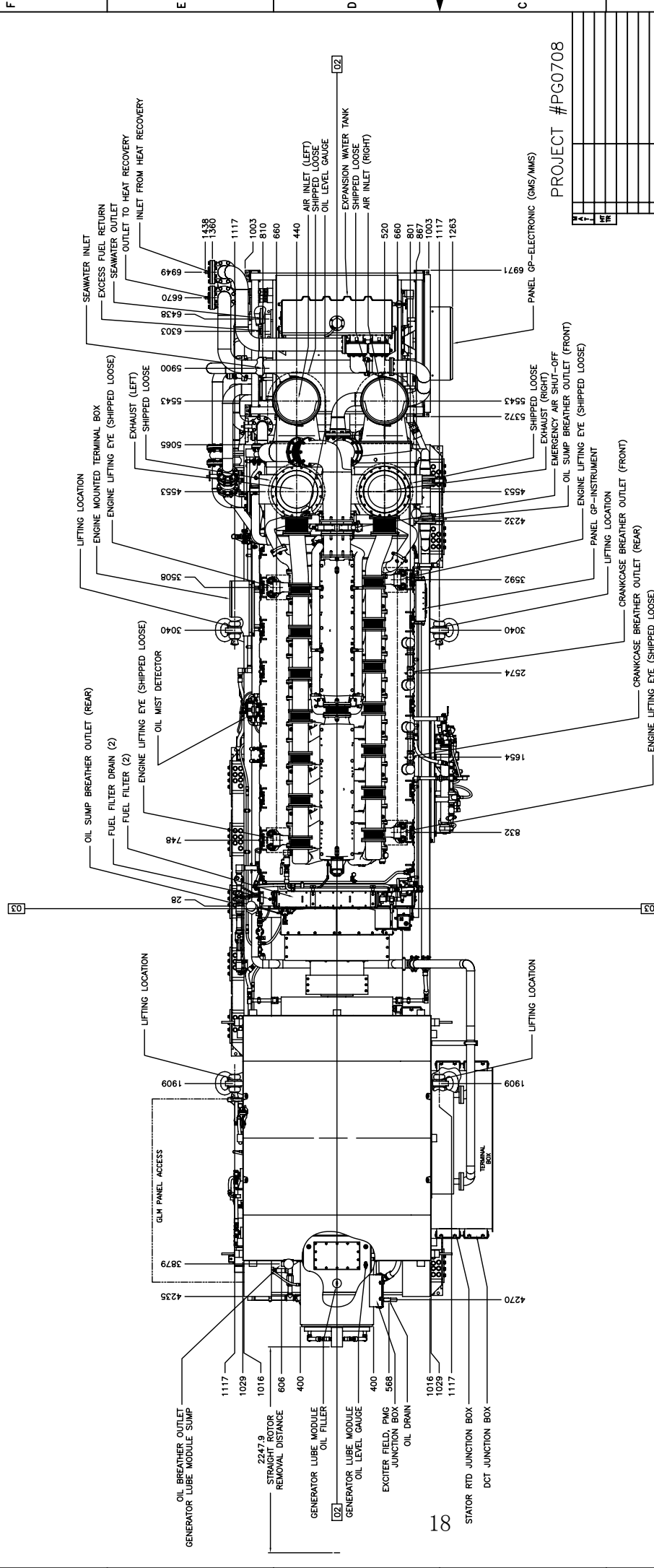
LEFT SIDE VIEW  
RIGHT HAND SERVICE  
RIGHT HAND COOLANT CONNECTIONS  
SCALE 1=20

THE LINE IS USED AS A LINEAR DATUM FOR THE VIEW IN WHICH IT APPEARS. DIMENSIONS ARE PRINTED PARALLEL TO THE DATUM LINE TO WHICH THEY PERTAIN. DIMENSIONS WITH ARROW-HEADS DO NOT PERTAIN TO THE DATUM LINES EXCEPT IN AUXILIARY VIEWS.

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1E2956B IDENT	1E2733 DRAWING-AUTOCAD	1E0188R LETTERS	1E0013Y CONFIDENTIALITY	1E0011 INTPR & TOL	Caterpillar: Confidential Yellow
ENGINE MODEL	3616 FMT	EXP	IN-PROCESS	PROB	
DATE	01/03/09	00116FEB09	00116FEB09	00116FEB09	00116FEB09
SCALE	NONE	1:1	1:1	1:1	1:1
SHEET	3 OF 12	PRODUCTION	DATE		
CATERPILLAR, INC.					
INSTALLATION-DRAWING					
332-2698					
W853					



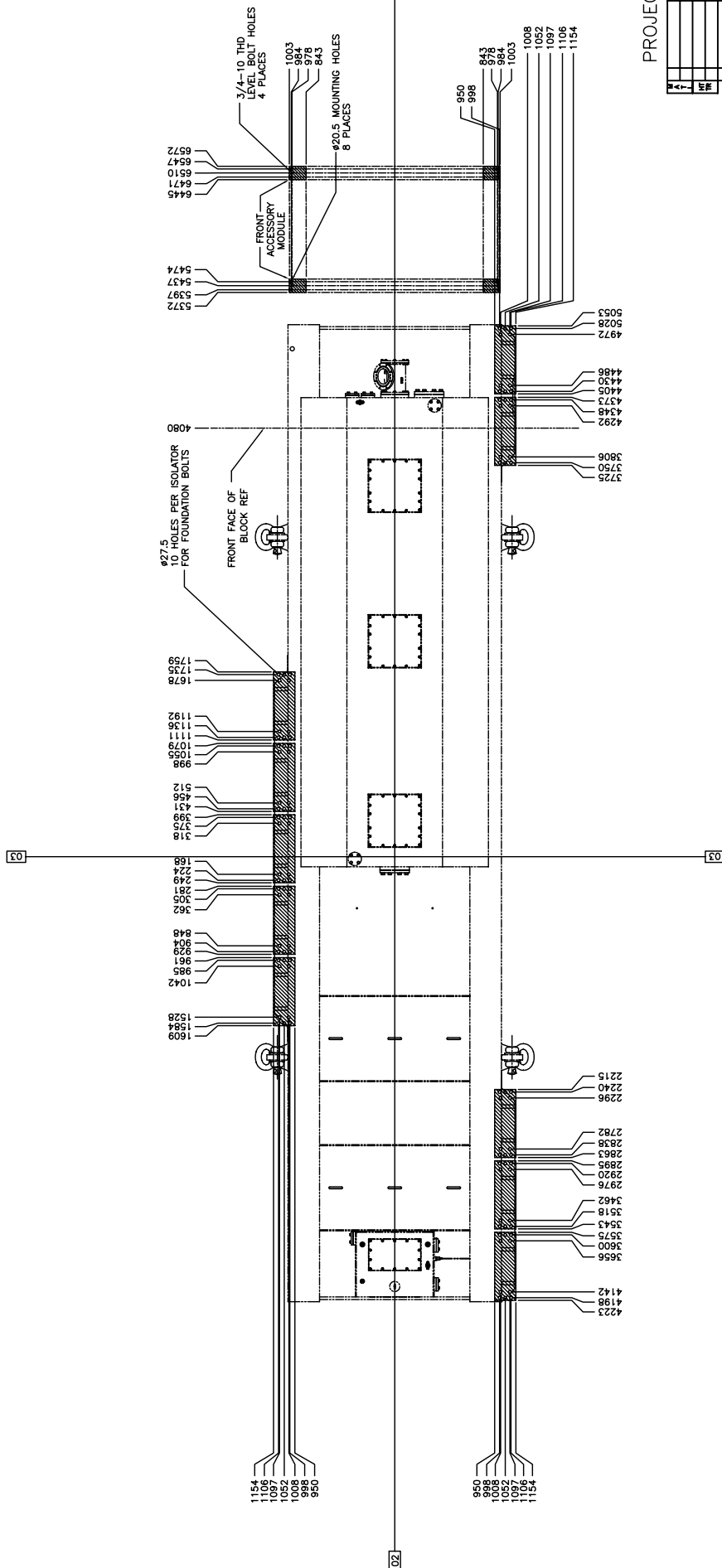
CERTIFIED TO  
BUILD

TOP VIEW  
RIGHT HAND SERVICE  
RIGHT HAND COOLANT CONNECTIONS  
SCALE 1=20

THE □-□ LINE IS USED AS A LINEAR DATUM FOR THE VIEW IN WHICH IT APPEARS. DIMENSIONS ARE PRINTED PARALLEL TO THE DATUM LINE TO WHICH THEY PERTAIN. DIMENSIONS WITH ARROWS DO NOT PERTAIN TO THE DATUM LINES EXCEPT IN AUXILIARY VIEWS.

DATUMS			REVISION	CHANGE DESCRIPTION
01	- CENTERLINE OF CRANKSHAFT		01	Moved GWS PANEL TO ACC. MOD.-LEFT SIDE; ADDED GLM SUMP BREATHER OUTLET
02	- CENTERLINE OF ENGINE		01	ADDED 341-3623 LINES GP-WATER AT GENERATOR LUBE MODULE
03	- REAR FACE OF CRANKSHAFT ADAPTER		01	UPDATED GENERATOR GRAPHICS AND LABELING; ADDED CG AND WEIGHT DATA
				5                  4
				6


[illegible]



PLAN VIEW  
SCALE 1"=20'

NOTE A: ALL DIMENSION TOLERANCES  $\pm 1.3\text{mm}$ .

NOTE B: IF MOUNTS ARE TO BE BOLTED TO THE FOUNDATION, THE PREFERRED METHOD IS TO DRILL ANCHOR BOLT HOLES AFTER THE ENGINE PACKAGE IS ON-SITE, USING THE ISOLATORS FOR DRILLING TEMPLATES. THE ANCHOR BOLTS ARE THEN INSTALLED, AND SECURED WITH EPOXY RESIN. CONCRETE IS THEN CAST TO THE ANCHORS, FORGED TO 54-81 N (40-60 lb-ft) (REFERENCE THE 3800 EPG APPLICATION AND INSTALLATION GUIDE FOR FURTHER DETAIL).

THE  LINE IS USED AS A LINEAR DATUM FOR THE VIEW IN WHICH IT APPEARS. DIMENSIONS ARE PRINTED PARALLEL TO THE DATUM LINE TO WHICH THEY PERTAIN. DIMENSIONS WITH ARROW HEADS DO NOT PERTAIN TO THE DATUM LINES EXCEPT IN AUXILIARY VIEWS.

[illegible]

DATUMS		REVISION		CHANGE DESCRIPTION	
01	- CENTERLINE OF CRANKSHAFT	01		MOVED GMS	ADDED GLM SUMP BREATHER OUTLET
02	- CENTERLINE OF ENGINE	01		ADDED 341-3623 LINES GP-WATER AT GENERATOR	LUBE MODULE
03	- REAR FACE OF CRANKSHAFT ADAPTER	01		UPDATED GENERATOR GRAPHICS AND LABELING	ADDED CG AND WEIGHT DATA
				5	4
				6	













CONNECTION DATA SHEET

RIGHT HAND SERVICE

RIGHT HAND COOLANT CONNECTIONS

	ITEM DESCRIPTION	CONNECTION SPECIFICATIONS	DIMENSIONS			OTHER DETAIL
			01	02	03	
WATER	CUSTOMER SEAWATER INLET	DETAIL ON DRAWING 332-2698 SHEET 10 4" ASME-150#	1155	816	5899	PARTNER FLANGE NOT PROVIDED
	CUSTOMER SEAWATER OUTLET	DETAIL ON DRAWING 332-2698 SHEET 10 4" ASME-150#	436	816	6439	PARTNER FLANGE NOT PROVIDED
FUEL	FUEL INLET	DETAIL ON DRAWING 332-2698 SHEET 8 PART #164-7576 (2" 150# ANSI)	705	980	4729	PARTNER FLANGE NOT PROVIDED
	EXCESS FUEL RETURN	DETAIL ON DRAWING 332-2698 SHEET 8 PART #164-7576 (2" 150# ANSI)	932	980	4729	PARTNER FLANGE NOT PROVIDED
	AUXILIARY FUEL PUMP CONNECTION	1 5/16-12 THD (37° FLARE)	12	573	3916	PARTNER FLANGE NOT NEEDED
OIL	LUBE OIL FILLER	DETAIL ON DRAWING 332-2698 SHEET 10 PART #153-7340 (2" 150# ANSI)	901	1068	4715	PARTNER FLANGE NOT PROVIDED
	LUBE OIL SUMP DRAIN	DETAIL ON DRAWING 332-2698 SHEET 10 PART #330-8558 (1 1/2" 150# ANSI)	1416	896	580	PARTNER FLANGE NOT PROVIDED
	SUCTION TO LUBE OIL CENTRIFUGE	DETAIL ON DRAWING 332-2698 SHEET 10 PART #330-8558 (1 1/2" 150# ANSI)	1416	896	730	PARTNER FLANGE NOT PROVIDED
	RETURN FROM LUBE OIL CENTRIFUGE	DETAIL ON DRAWING 332-2698 SHEET 10 PART #304-9847 (1 1/2" 150# ANSI)	901	1068	4510	PARTNER FLANGE NOT PROVIDED
	AIR INLET (RIGHT)	DETAIL ON DRAWING 332-2698 SHEET 9 PART #190-3600	2148	440	5543	PARTNER FLANGE NOT NEEDED
	AIR INLET (LEFT)	DETAIL ON DRAWING 332-2698 SHEET 9 PART #190-3600	2148	520	5543	PARTNER FLANGE NOT NEEDED
AIR	EXHAUST (RIGHT)	DETAIL ON DRAWING 332-2698 SHEET 6 PART #2W-2682	2714	440	4553	PARTNER FLANGE PROVIDED
	EXHAUST (LEFT)	DETAIL ON DRAWING 332-2698 SHEET 6 PART #2W-2682	2714	520	4553	PARTNER FLANGE PROVIDED
	BREATHER OUTLET" OIL SUMP (FRONT)	FOR 60.3 OD TUBING	785	867	4232	PARTNER FLANGE PROVIDED
	BREATHER OUTLET OIL SUMP (REAR)	FOR 60.3 OD TUBING	1109	1016	28	PARTNER FLANGE PROVIDED
	OIL BREATHER OUTLET GLM SUMP	FOR 60.3 OD TUBING	3879	1552	606	PARTNER FLANGE NOT NEEDED
	CRANKCASE BREATHER OUTLET (FRONT)	FOR 60.3 OD TUBING	1584	801	2574	PARTNER FLANGE NOT NEEDED
	CRANKCASE BREATHER OUTLET (REAR)	FOR 60.3 OD TUBING	1584	801	1654	PARTNER FLANGE NOT NEEDED
	AIR STARTING MOTOR INLET	DETAIL ON DRAWING 332-2698 SHEET 10 PART #4P-4113	711	1069	620	PARTNER FLANGE NOT NEEDED

CERTIFIED TO  
BUILD

PROJECT #PG0708

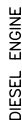
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1E01988 LETTERS	1E0013Y CONFIDENTIALITY
1E0011 INTNPR & TOL	1E0011 INTNPR & TOL
Caterpillar: Confidential Yellow	

ENGINE MODEL	3616 FMT
SIZE	1500
IN-PROCESS	TX
DATE	01/03/09
BY	0016FEB09
SHEET	1 OF 12

CATERPILLAR INC.	
INSTALLATION-DRAWING	
(YSRL-JBF1)	
332-2698	WB53

REVISION	CHANGE DESCRIPTION
01	MOVED GMS PANEL TO ACC. MID-LEFT SIDE, ADDED GLM SUMP BREATHER OUTLET
01	ADDED 341-3623 LINES GP-WATER AT GENERATOR LUBE MODULE
01	UPDATED GENERATOR GRAPHICS AND LABELING, ADDED CG AND WEIGHT DATA

## YRSL-JBF1



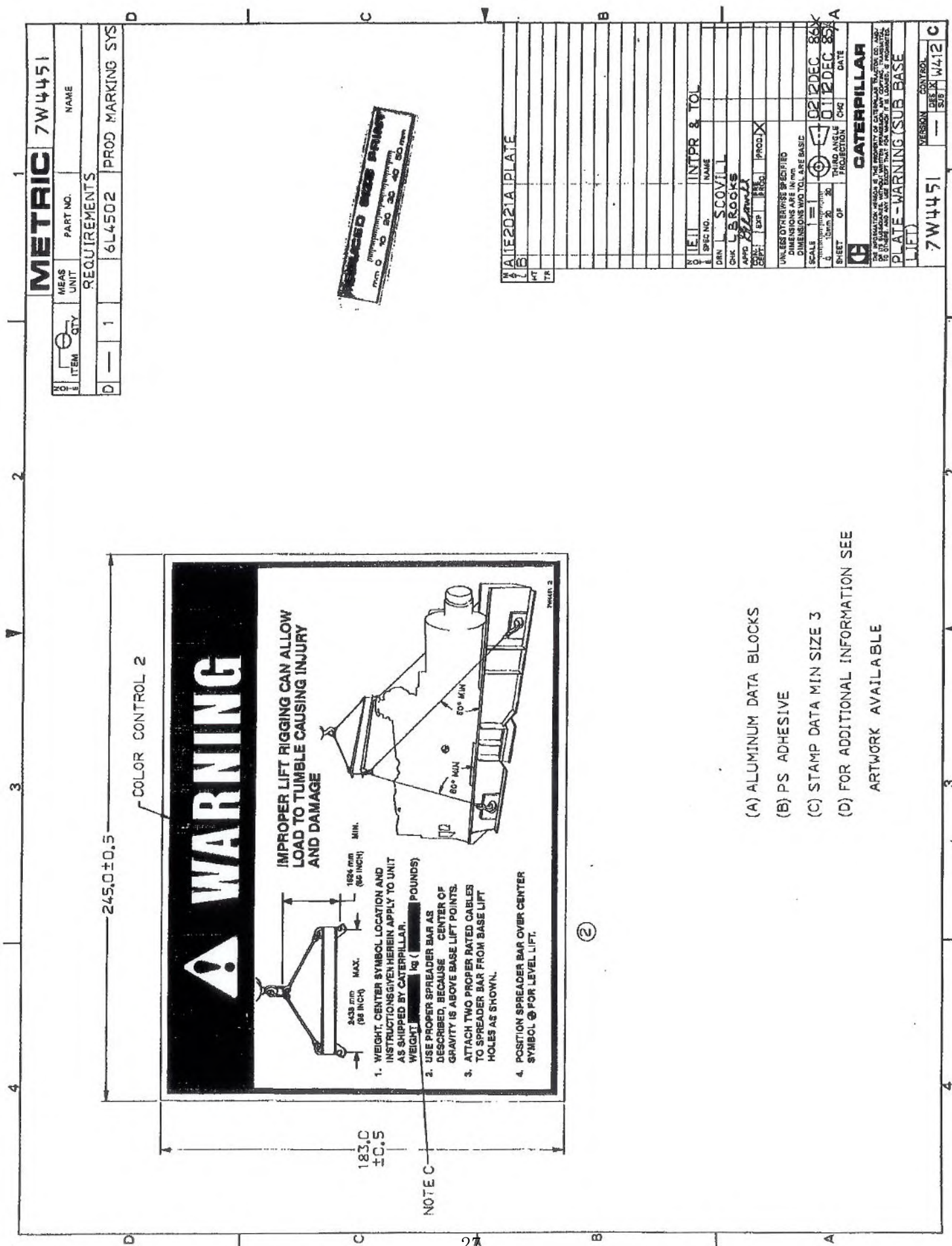
CUSTOMER SUPPLIES 2800 L/MIN SEAWATER • 27°C (32°C MAX)

THE INFORMATION HEREON IS THE PROPERTY OF CORPULIFE INC. AND/OR ITS  
 SUBSIDIARIES WITHOUT WRITTEN PERMISSION, ANY COPIING, TRANSMISSION, TO  
 OTHERS, AND ANY USE EXCEPT THAT FOR WHICH IT IS ISSUED IS PROHIBITED.

VER		CONTROL	
DES	XX	DES	XX
DATE		DATE	

332-2698  
 (YSRL-JBF1)  
 INSTALLATION-DRAWING  
 D W853

[illegible][illegible]



METRIC		7W4451	
MEAS	UNIT	PART NO.	NAME
REQUIREMENTS			
D	1	6L4502	PROD MARKING SYS

UNLESS OTHERWISE SPECIFIED			
DIMENSIONS ARE IN mm			
DIMENSIONS WHO TOL ARE BASIC			
SCALE	1	1	1
THIRD ANGLE	PROJECTION	CHG	DATE
CATERPILLAR			
THE INFORMATION HEREON IS THE PROPERTY OF CATERPILLAR. IT IS TO BE KEPT SECRET AND NOT REPRODUCED OR TRANSMITTED IN ANY FORM OR BY ANY MEANS, ELECTRONIC OR MECHANICAL, INCLUDING PHOTOCOPYING, RECORDING, OR BY ANY INFORMATION STORAGE AND RETRIEVAL SYSTEM, WITHOUT THE WRITTEN PERMISSION OF CATERPILLAR.			
PLATE-WARNING (SUB BASE)			
LIFT			
VERSION	CONTROL	DATE	BY
7W4451	—	DEC 85	W412

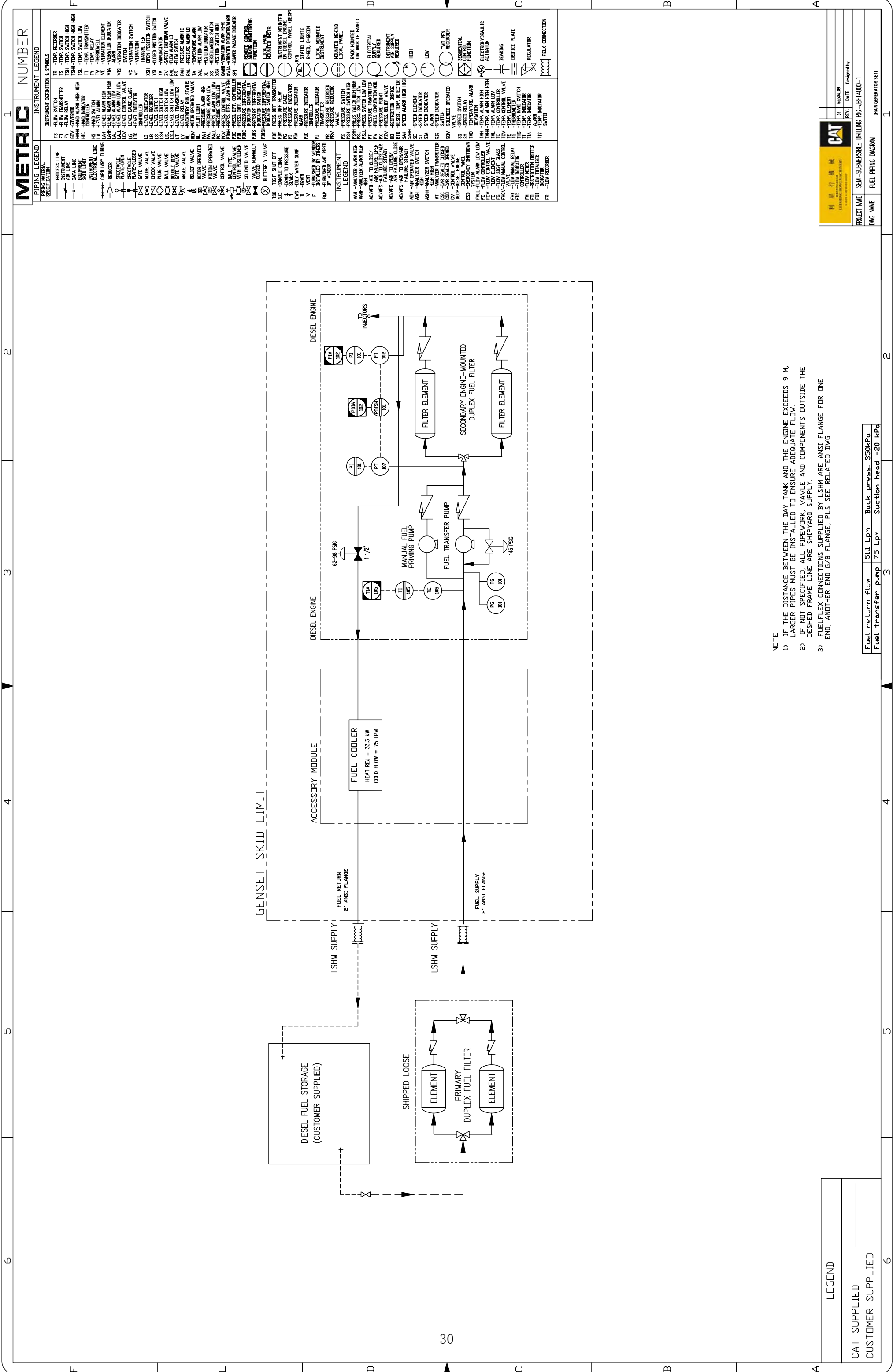
- (A) ALUMINUM DATA BLOCKS
- (B) PS ADHESIVE
- (C) STAMP DATA MIN SIZE 3
- (D) FOR ADDITIONAL INFORMATION SEE ARTWORK AVAILABLE

#### 4.0 PIPING & ELECTRICAL DIAGRAMS

No	Drawing No	REV.	Drawing Name	Total Pages
1	MA/P245-01-MG	01	Cooling Piping Diagram	1
2	MA/P245-02-MG	01	Fuel Piping Diagram	1
3	MA/P245-03-MG	01	Generator set Lube Oil Piping Diagram	1
4	MA/P245-04-MG	01	Air Inlet & Exhaust Piping Diagram	1
5	MA/P245-05-MG	01	Starting Air Piping Diagram	1
6	HC090907-001		Custom Fuel and Lube Oil Flexible Hose	1
7	MA/P245-06-MG	01	Electrical Interface Diagram	1
8	888-00218-25		Kato generator wiring diagram	1
9			2301D Outline Drawing & Plant Wiring Drawing	4
10			DVR DECS200 Typical Connections & Dimensions Drawing	3
11	334-3086	01	Wiring Diagram for GMS	25
12				



















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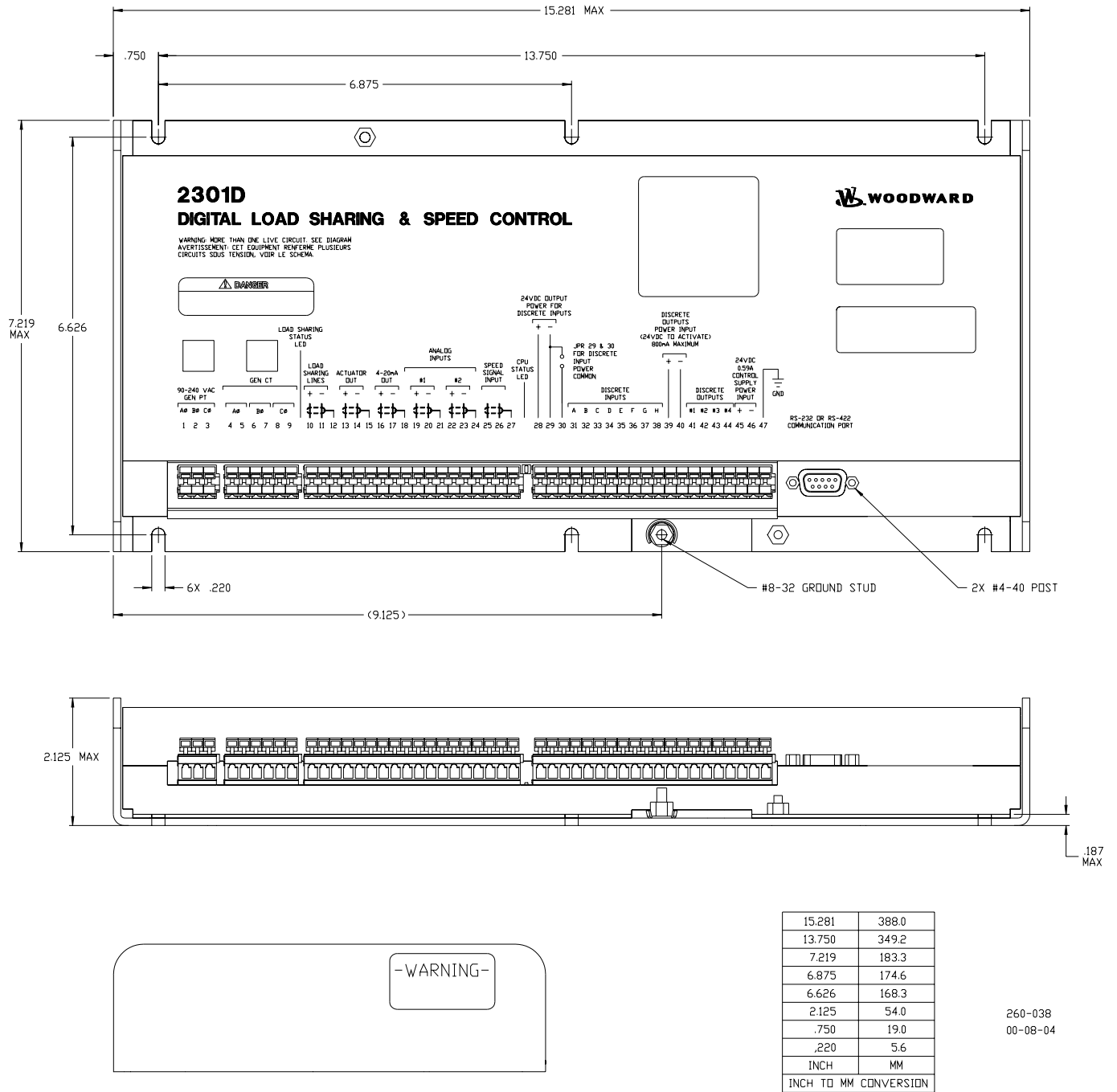


Figure 1-1a. 2301D Outline Drawing (Ordinary Locations)

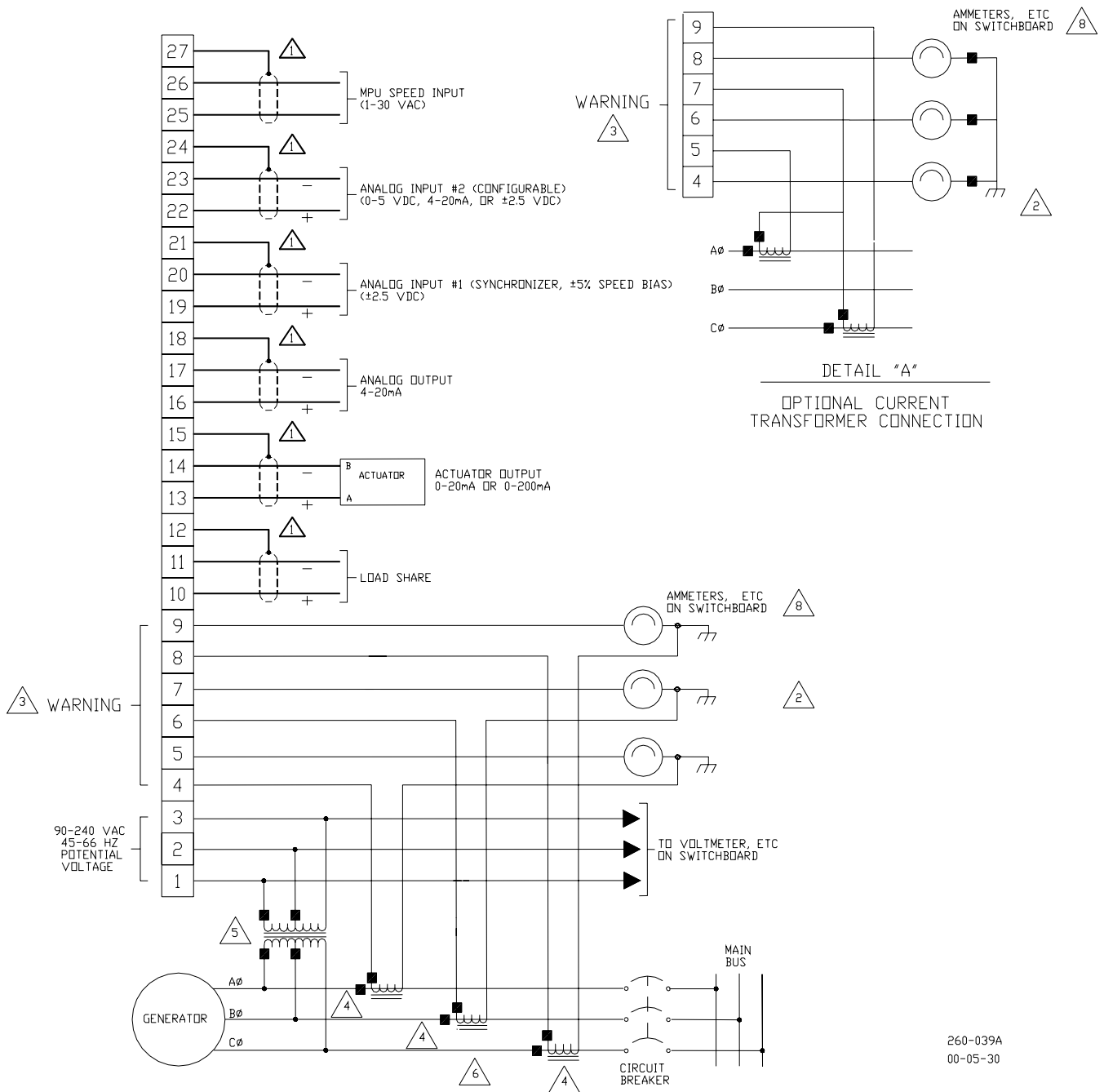
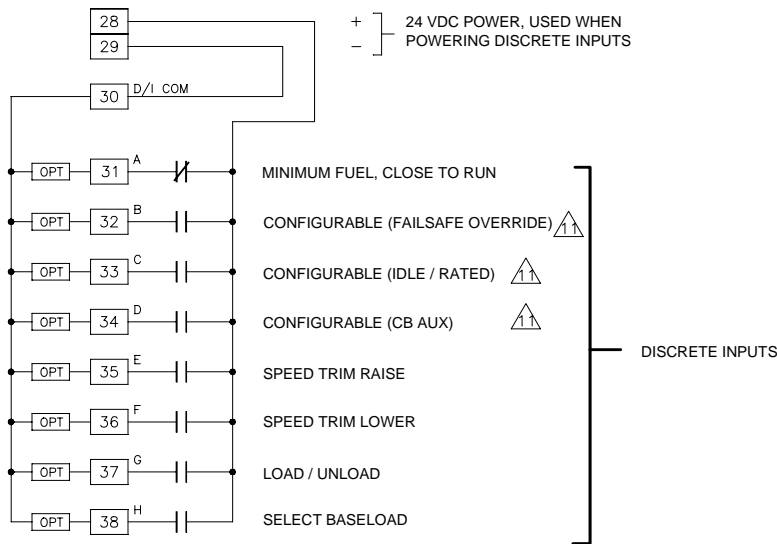
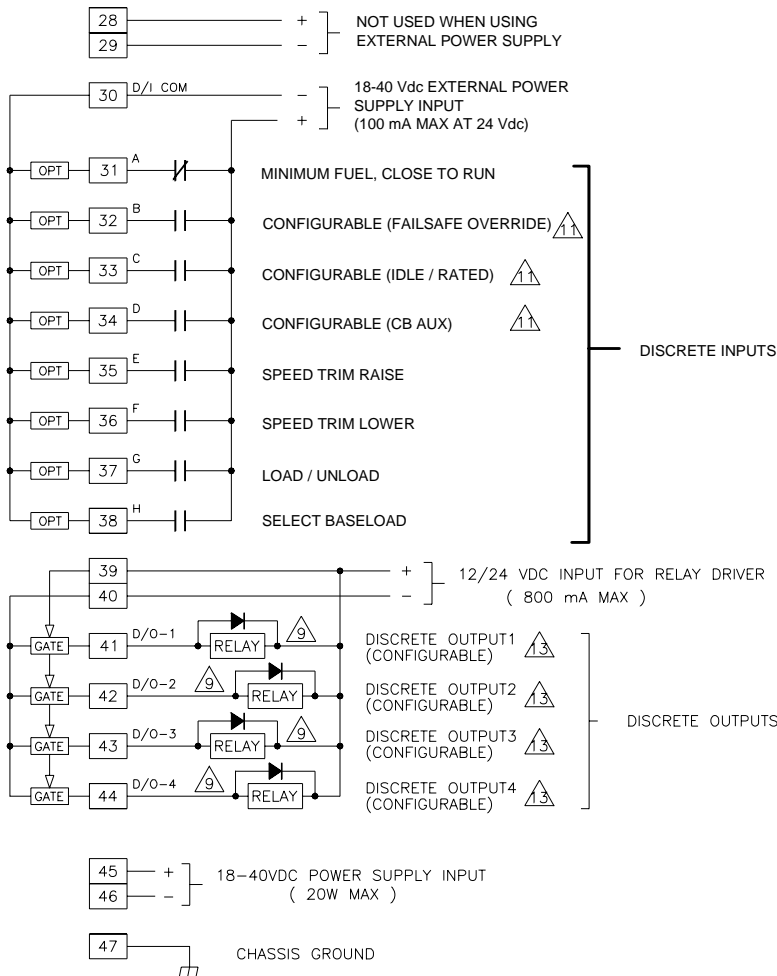


Figure 1-2a. 2301D Plant Wiring Diagram (sheet 1)

## DISCRETE INPUTS WITH INTERNAL POWER SUPPLY



## DISCRETE INPUTS WITH EXTERNAL POWER SUPPLY



RS-232 OR RS-422  
COMMUNICATION PORT  
PIN ASSIGNMENT  
(FOR SERVLINK/MODBUS)

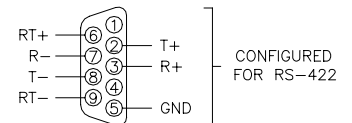
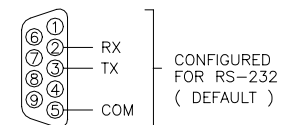


Figure 1-2b. 2301D Plant Wiring Diagram (sheet 2)

## NOTES:

- 1 SHIELDED WIRES TO BE TWISTED PAIRS, WITH SHIELD GROUNDED AT CONTROL END ONLY.
- 2 POINT OF GROUNDING IF REQUIRED BY WIRING CODE.
- 3 INTERNAL CURRENT TRANSFORMER BURDEN MUST BE CONNECTED ACROSS POWER SOURCE CURRENT TRANSFORMER AT ALL TIMES, TO PREVENT LETHAL HIGH VOLTAGES.
- 4 POWER SOURCE CURRENT TRANSFORMERS SHOULD BE SIZED TO PRODUCE 5 A SECONDARY CURRENT WITH MAXIMUM GENERATOR CURRENT, CURRENT TRANSFORMER BURDEN IS LESS THAN 0.1 VA PER PHASE.
- 5 WITH A BALANCED THREE PHASE LOAD AND UNITY POWER FACTOR, THE CURRENT TRANSFORMERS SHOULD BE WIRED IN THE CORRECT POTENTIAL LEG AND MUST BE PHASED AS FOLLOWS:  
 PHASE A: POTENTIAL TERMINAL 1, WITH RESPECT TO NEUTRAL, IN PHASE WITH CT TERMINALS 4 ( ) TO 5.  
 PHASE B: POTENTIAL TERMINAL 2, WITH RESPECT TO NEUTRAL, IN PHASE WITH CT TERMINALS 6 ( ) TO 7.  
 PHASE C: POTENTIAL TERMINAL 3, WITH RESPECT TO NEUTRAL, IN PHASE WITH CT TERMINALS 8 ( ) TO 9.
- 6 FOR OPTIONAL CURRENT TRANSFORMER CONNECTION, SEE DETAIL "A".
- 7 **WARNING:** DO NOT USE FOR EMERGENCY SHUTDOWN. THE PRIME MOVER SHOULD BE EQUIPPED WITH SEPARATE OVERSPEED, OVERTEMPERATURE OR OVERPRESSURE SHUTDOWN DEVICE(S) TO PROTECT AGAINST RUNAWAY OR DAMAGE TO THE PRIME MOVER WITH POSSIBLE PERSONAL INJURY OR LOSS OF LIFE.
- 8 IF METERS ARE NOT USED, JUMPERS MUST BE INSTALLED IN PLACE OF METERS SHOWN.
- 9 INDICATES RELAY COIL OR LAMP, 200 MA MAXIMUM PER CHANNEL.
- 10 OPTIONAL SEE BOM
- 11 DISCRETE INPUTS WITH CABLE LENGTHS GREATER THAN 30 METERS THAT ARE USED FOR CRITICAL FUNCTIONS, SUCH AS EMERGENCY STOP, SHOULD NOT BE FLOATED IN EITHER AN ON OR OFF STATE. THESE INPUTS SHOULD BE SWITCHED TO +24 VDC OR GROUND.

260-039C  
00-07-26

Figure 1-2c. 2301D Plant Wiring Diagram (notes)

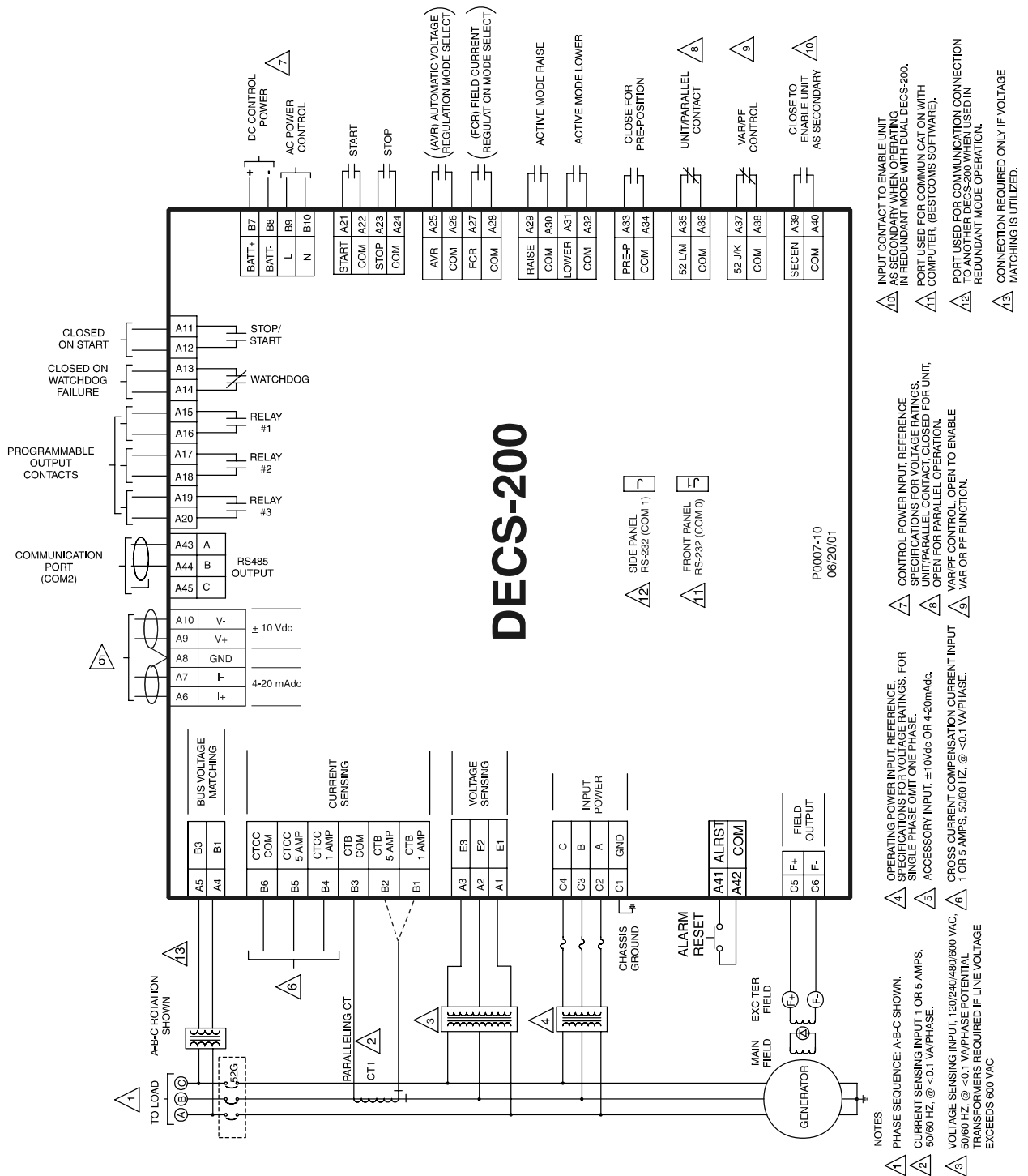


Figure 4-9. Typical Connections

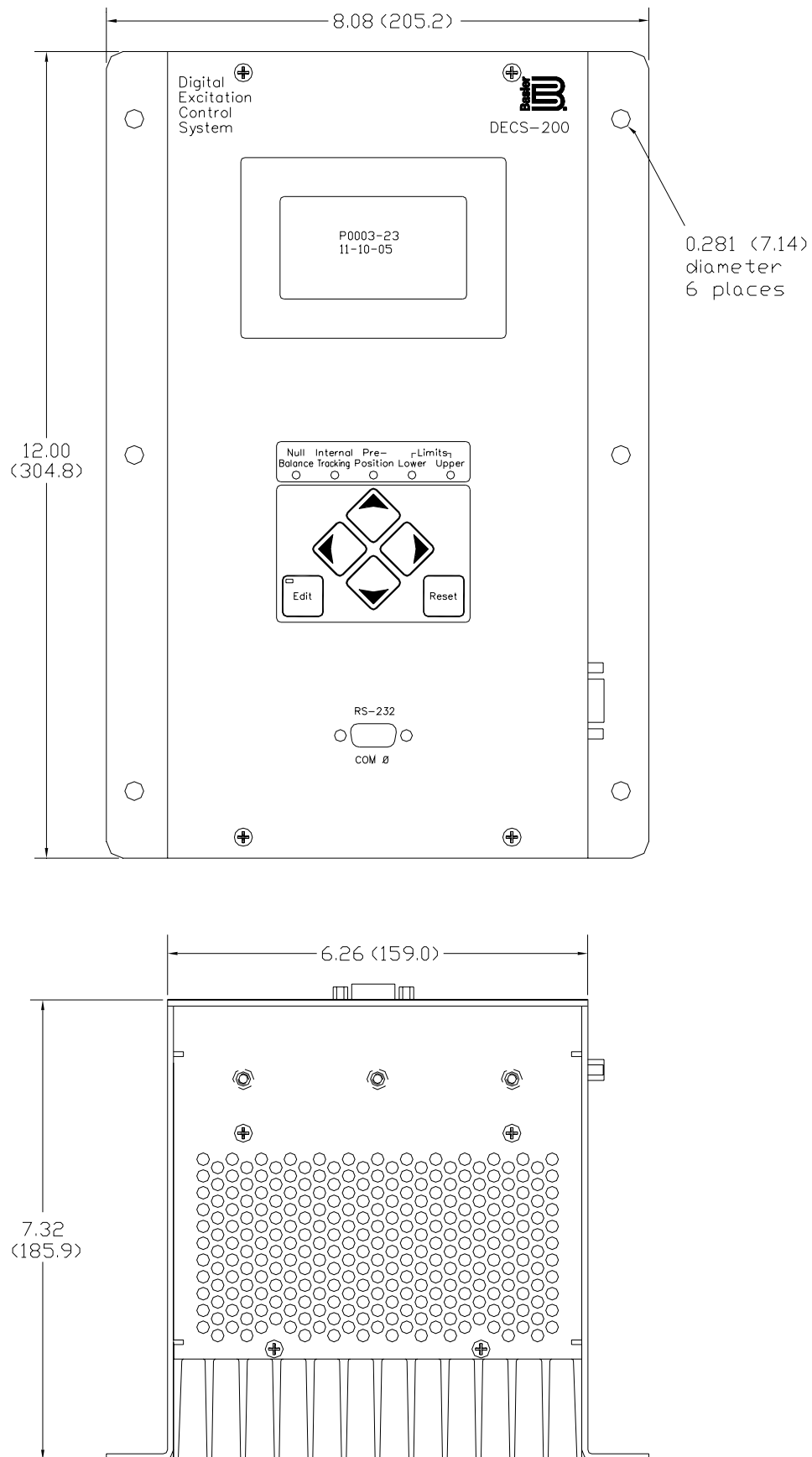


Figure 4-1. Overall Dimensions

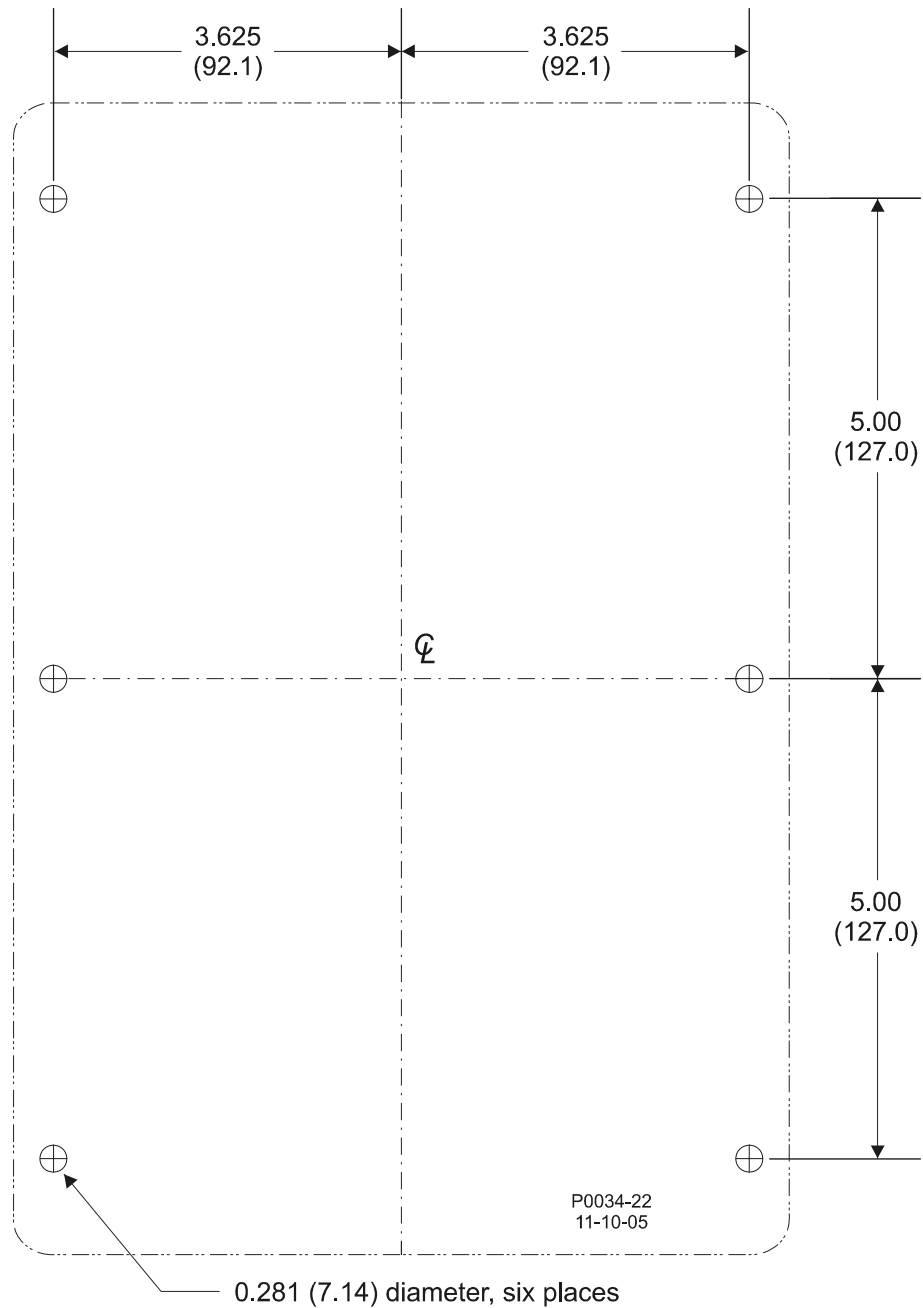


Figure 4-2. Panel Drilling Diagram, Projection Mount

TABLE OF CONTENTS	
SHEET	DESCRIPTION
1	CONFIGURATION/NOTES
2	LEGEND
3	SYSTEM BLOCK DIAGRAM
4	CIRCUIT SCHEMATIC
5	CIRCUIT SCHEMATIC
6	CIRCUIT SCHEMATIC
7	CIRCUIT SCHEMATIC
8	CIRCUIT SCHEMATIC
9	CIRCUIT SCHEMATIC
10	CIRCUIT SCHEMATIC
11	CIRCUIT SCHEMATIC
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13	CIRCUIT SCHEMATIC
14	CIRCUIT SCHEMATIC
15	CIRCUIT SCHEMATIC
16	CIRCUIT SCHEMATIC
17	CIRCUIT SCHEMATIC
18	INTERCONNECT
19	INTERCONNECT
20	INTERCONNECT
21	INTERCONNECT
22	INTERCONNECT
23	MODBUS INTERCONNECT
24	LOADSHARE
25	VOLTAGE REGULATOR

UNLESS OTHERWISE SPECIFIED, RECOMMENDED STANDARD WIRE SIZE IS 16 AWG ENGINE, 18 AWG MMS PANEL.

SHIELD TWISTED PAIR WIRE SIZE IS 18 AWG MINIMUM.

USE K-TYPE THERMOCOUPLE WIRE FOR THERMOCOUPLE CONNECTIONS.

DRY CONTACTS ARE RATED FOR 5 AMPS AT 30 VDC, 3 AMPS AT 120 VAC

ALL PRESSURE SWITCHES SHOWN WITHOUT PRESSURE PRESENT.

ALL LEVEL SWITCHES SHOWN WITHOUT FLUID PRESENT.

ALL RELAY CONTACTS SHOWN IN THE DE-ENERGIZED MODE.

INTERCONNECT WIRING, UNLESS OTHERWISE SPECIFIED, RECOMMENDED STANDARD WIRE SIZE IS 16 AWG, FACTORY PROVIDED FOR ACCESSORY MODULE MOUNTED GMS OR MMS II, CUSTOMER PROVIDED IF SHIP LOOSE.

ALL COMMON TERMINALS FOR ENGINE WIRING ARE JUMPERED IN ENGINE TERMINAL BOX FOR INTERCONNECT WIRING TO GMS OR MMS II PANELS.

TB #	LINE #	TB #	LINE #	TB #	LINE #	TB #	LINE #
G	313	101	508	164B	1172	251	817
GND	300	102	508	164C	1170	253	810
SH	344	102A	1161	165	648	257A	1175
1	300	102B	1165	167	723	257B	1179
2	300	102C	1163	168	729	257C	1177
3	320	103	514	169	723	258A	1168
1A	306	104	520	173	430	258B	1172
1B	310	105	520	174	436	258C	1170
1C	341	105A	1161	175	430	452	746
1D	323	105B	1165	200	400	500	329
1E	315	105C	1163	201	400	501	328
14A	1188	106	507	202	602	502	326
14B	1192	107	529	203	741	511	1040
14C	1190	107A	1168	204	708	512	1036
30A	1151	107B	1172	205	711	513	1039
30C	1151	107C	1170	205A	1194	514	1041
55A	1181	108	529	205B	1199	515	1044
55B	1186	109	538	205C	1196		
55C	1183	109A	1154	208	980		
56A	1175	109B	1158	209	983		
56B	1179	109C	1156	210	952		
56C	1177	110	538	210A	1194		
57A	1181	111	543	210B	1199		
57B	1186	113	449	210C	1196		
57C	1183	114	944	211	955		
67	617	115	845	211A	1154		
68	620	116	846	211B	1158		
70	635	140	1013	211C	1156		
71	632	141	1016	212	957		
72	632	142	1019	212A	1161		
73	718	146	607	212B	1165		
74	620	147	611	212C	1163		
75	620	148	607	213	960		
80	511	149	614	214	963		
81	438	150	609	215	966		
82	723	150A	1188	216	969		
82A	720	150B	1192	217	971		
82B	729	150C	1190	218	974		
83	441	151	620	219	977		
84	444	151A	824	220	838		
85	438	151B	827	221	840		
86	444	151C	824	222	1222		
86A	1175	152	626	223	1223		
86B	1179	153	626	230	1214		
86C	1177	153A	1154	231	1216		
87	547	153B	1158	233	1205		
88	549	153C	1156	234	1207		
90	504	161	638	235	991		
91	517	162	642	236	994		
92	501	163	642	248	807		
93	505	164	645	249	818		
		164A	1168	250	807		

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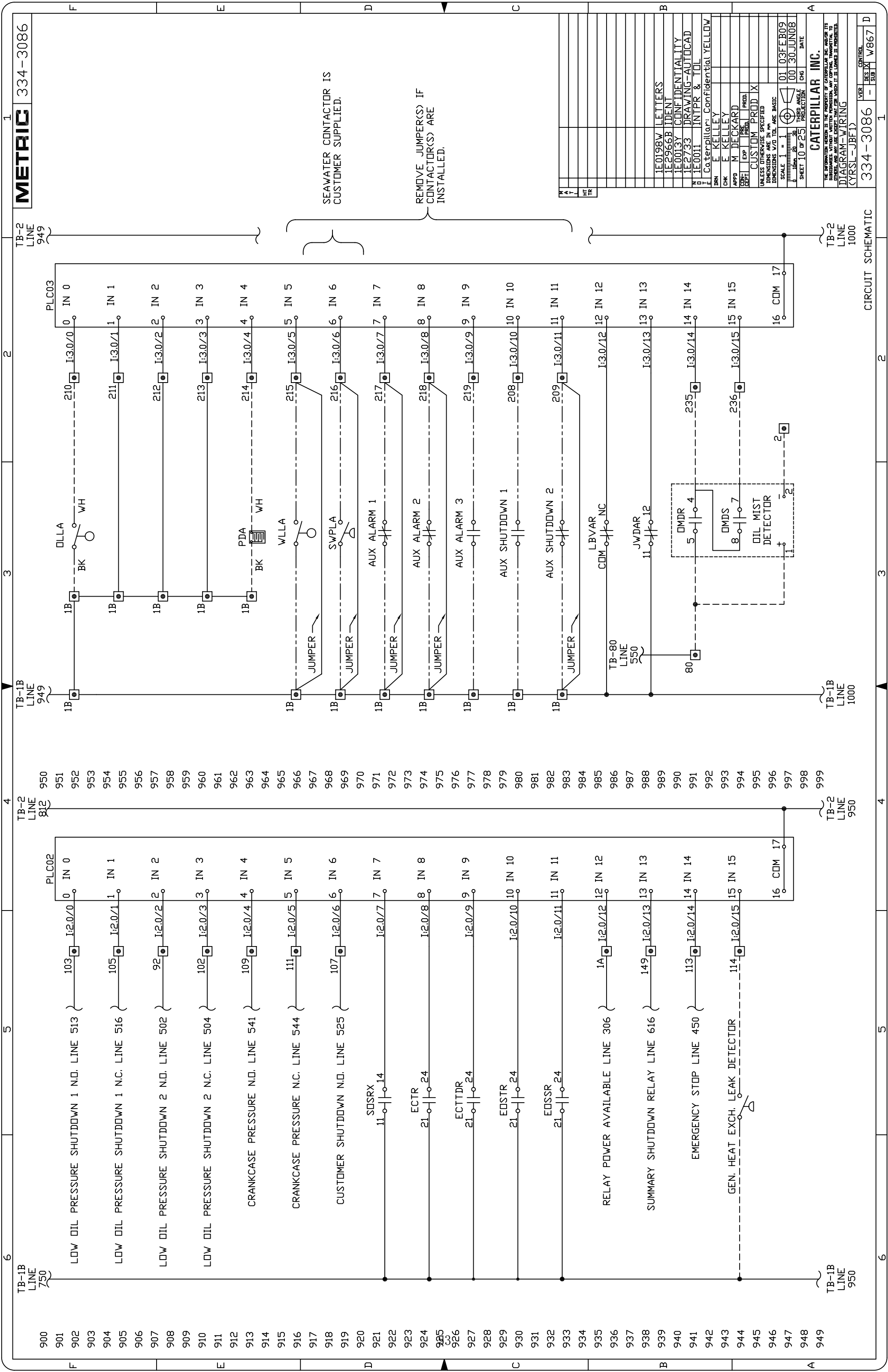








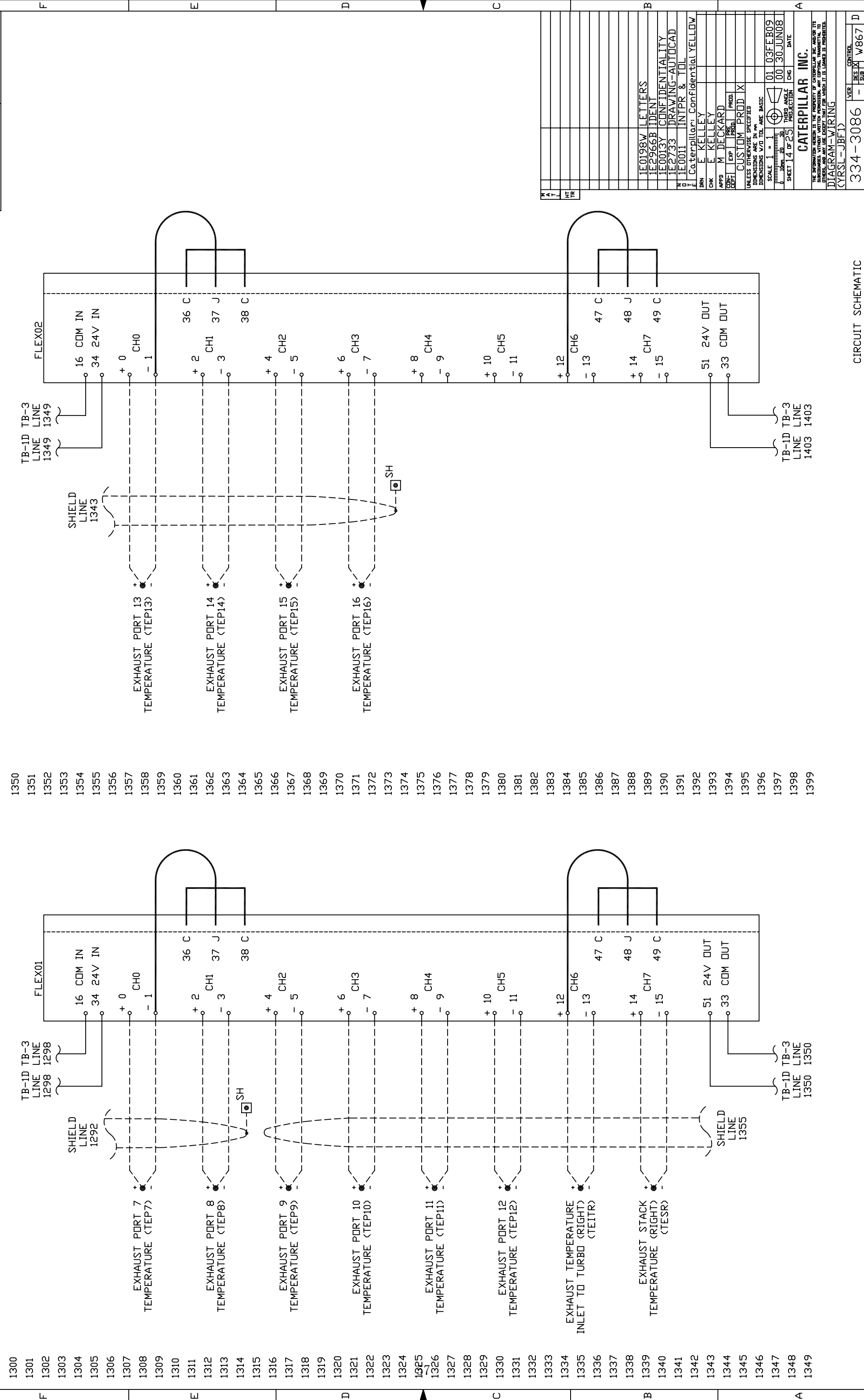


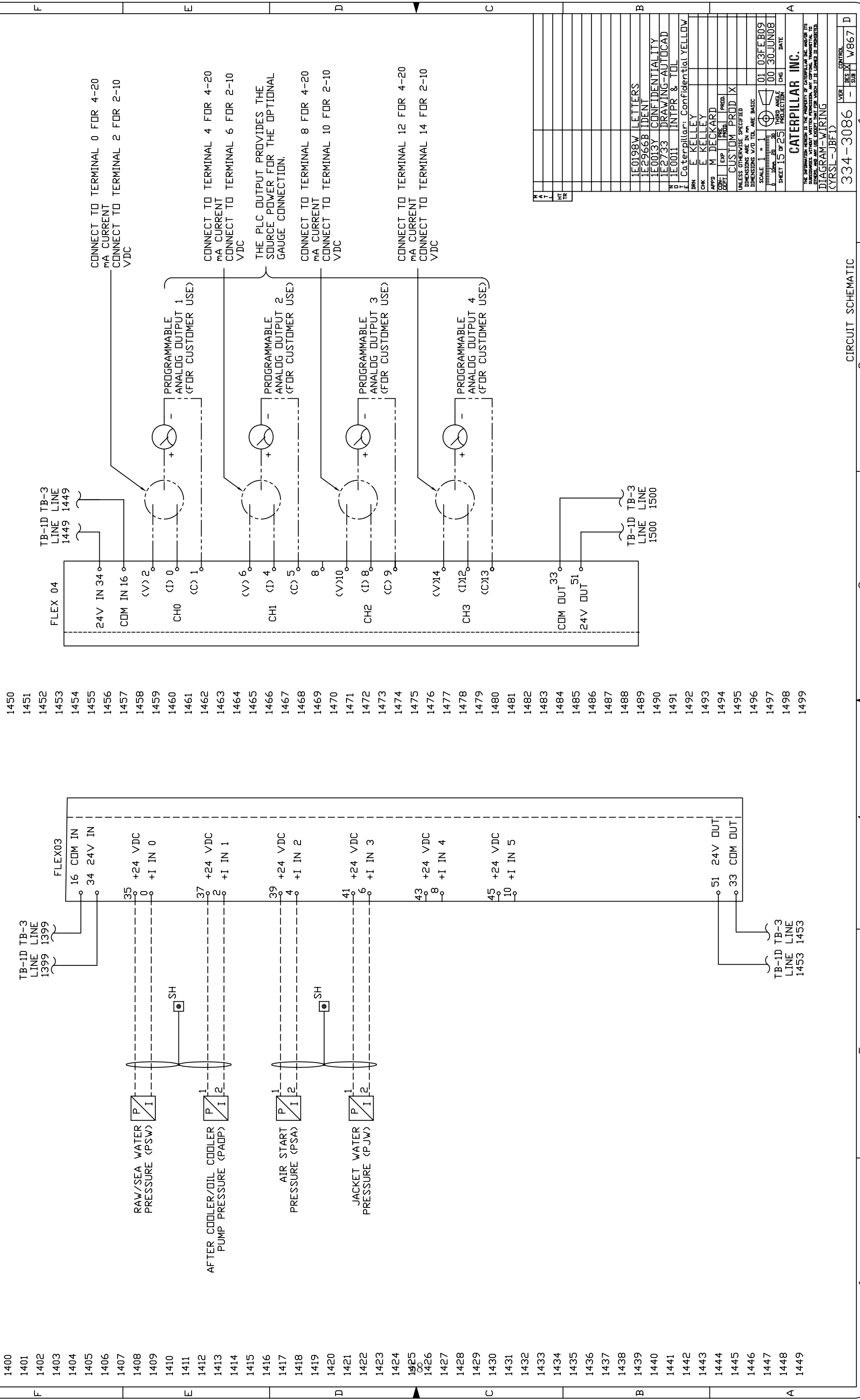








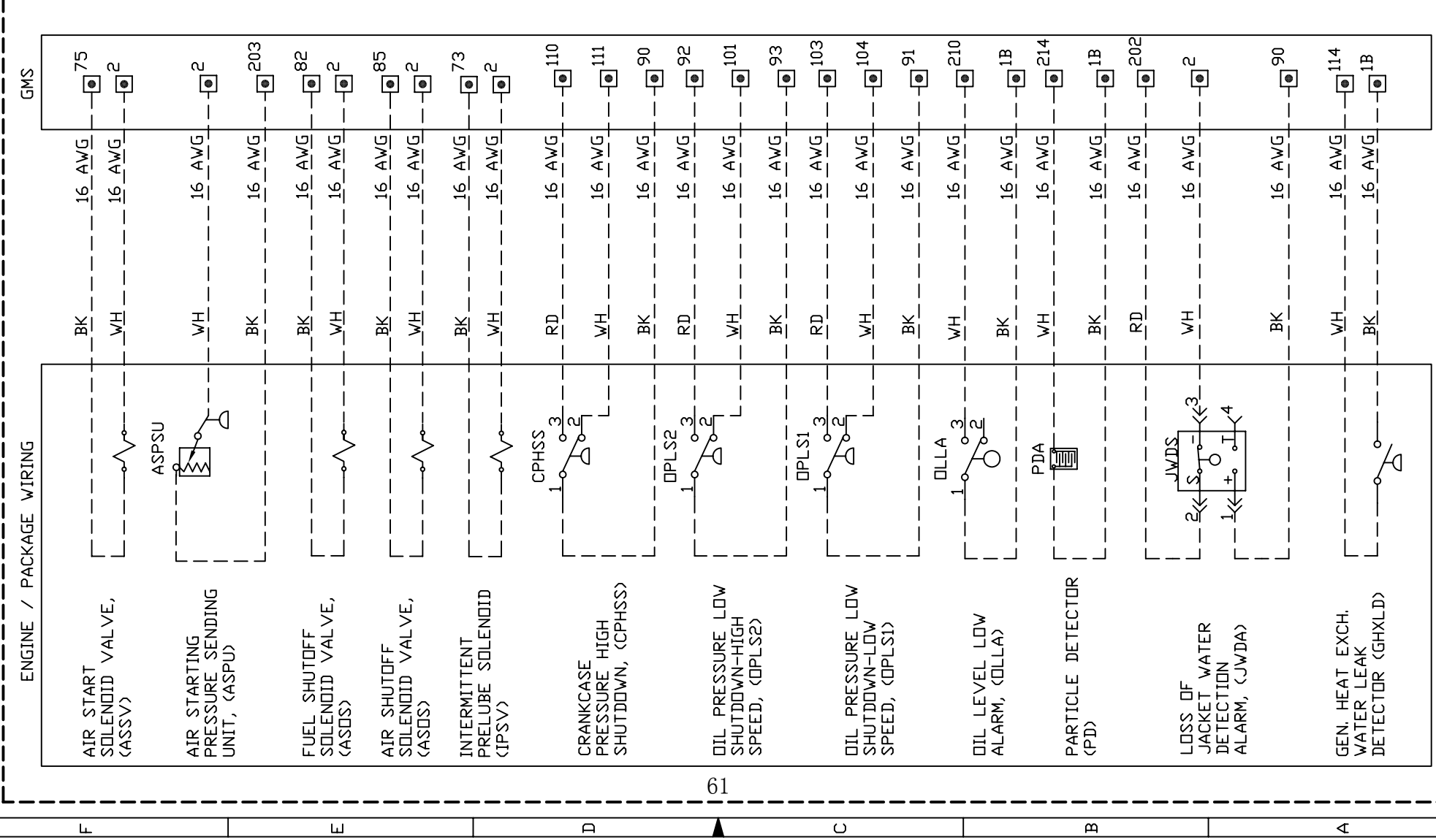
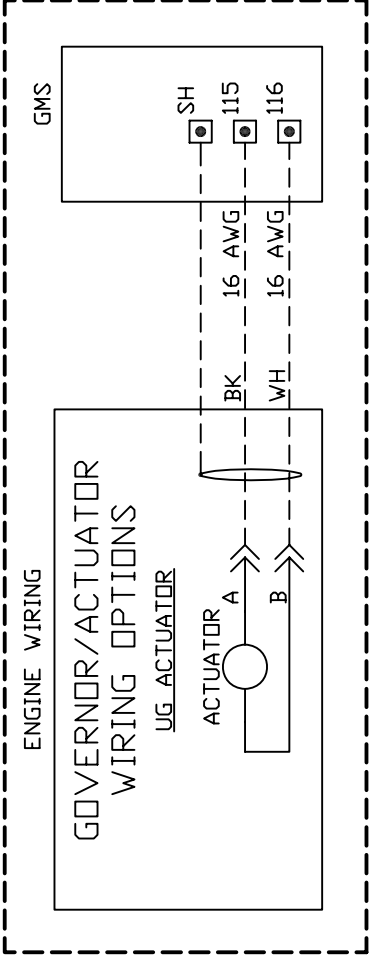
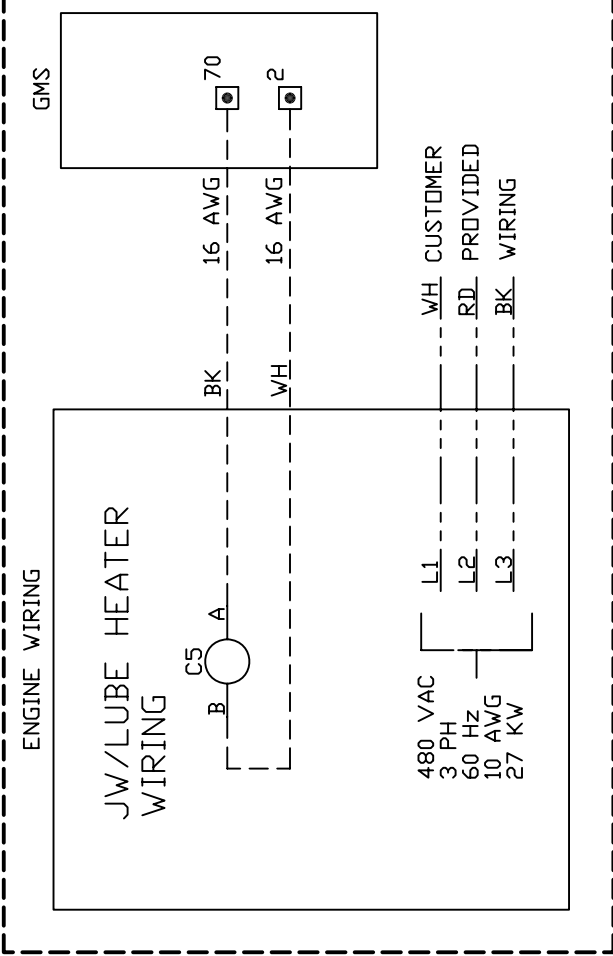
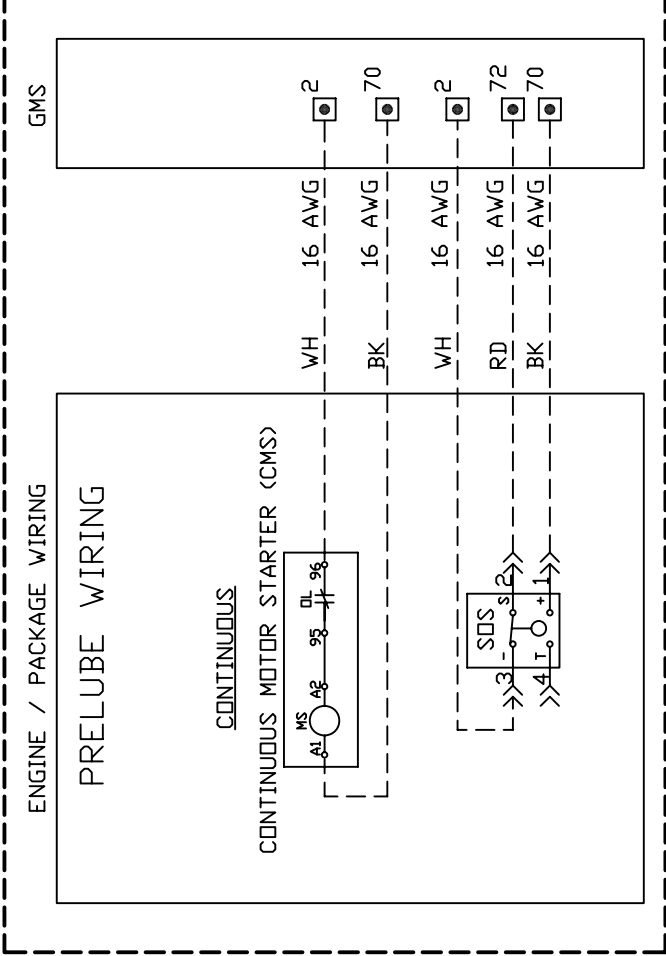
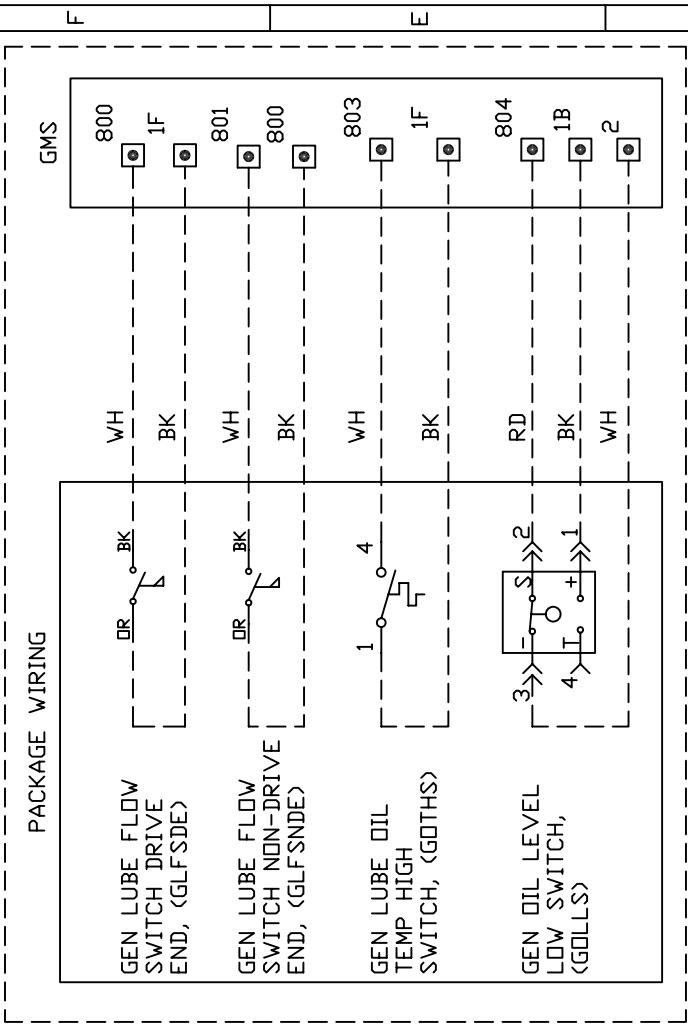


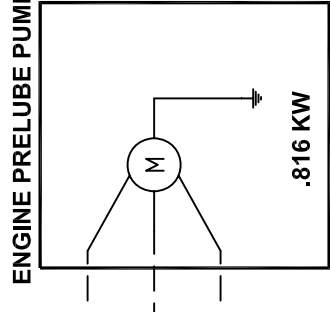
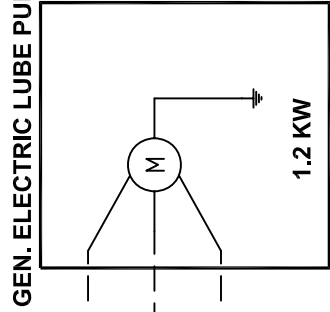
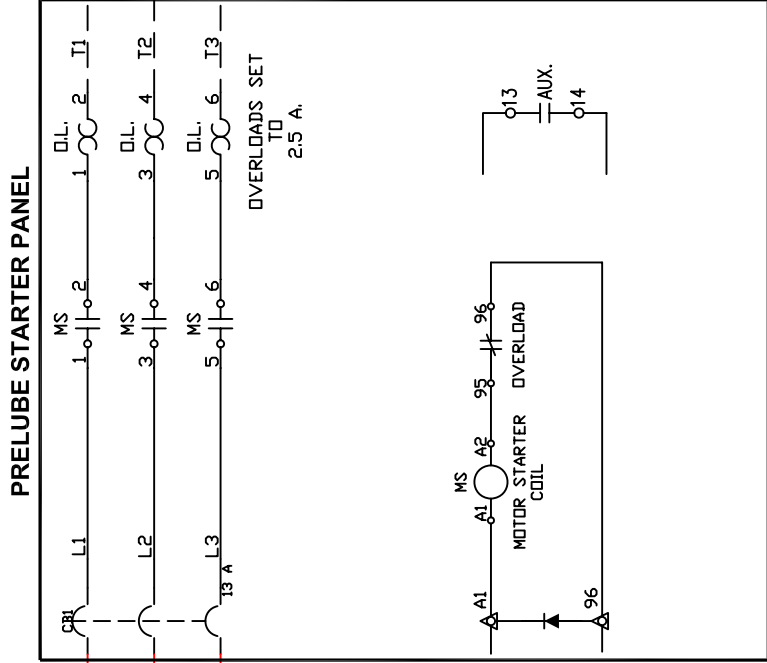
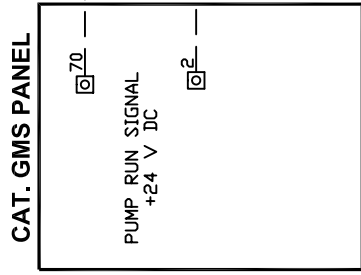
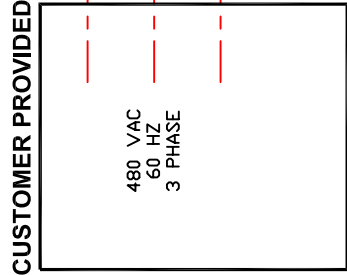
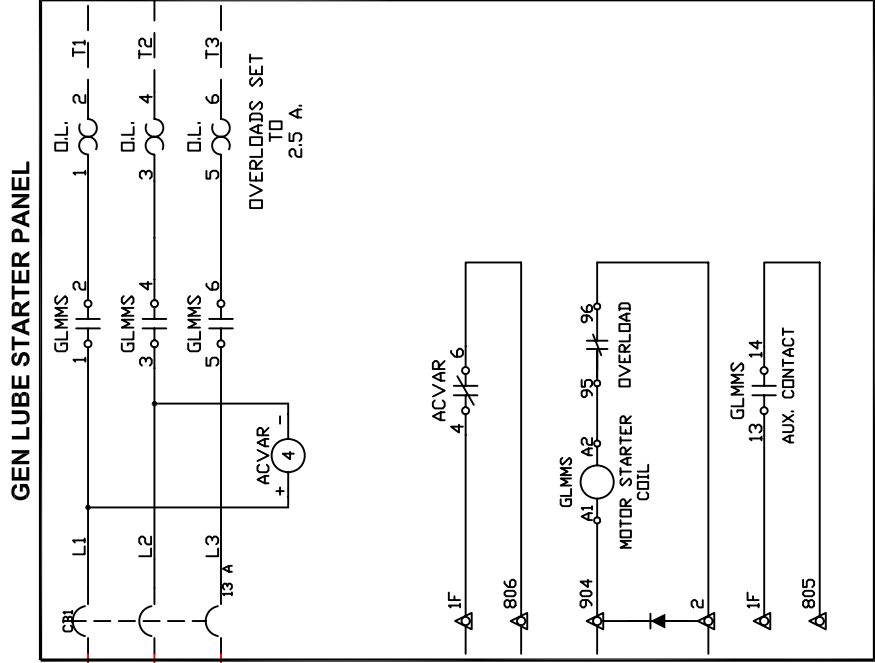
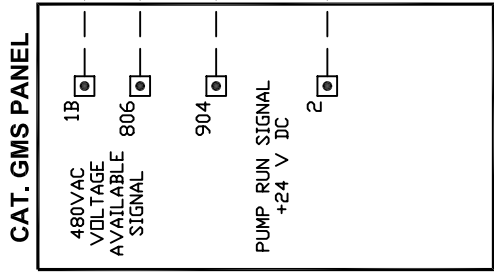
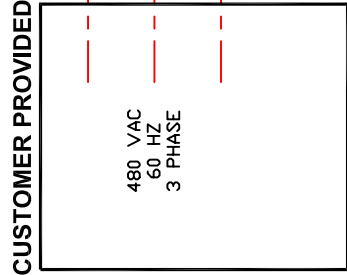
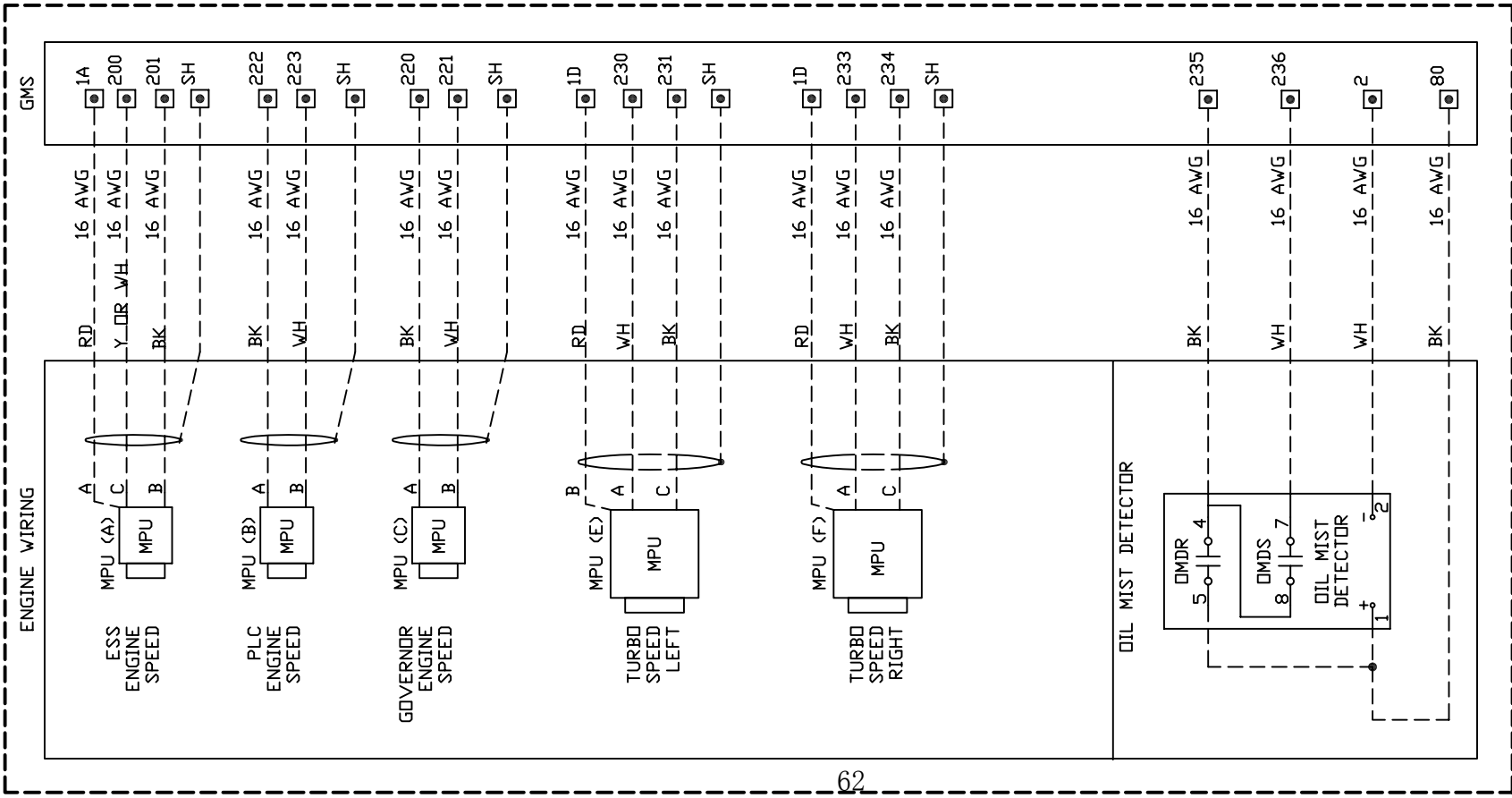








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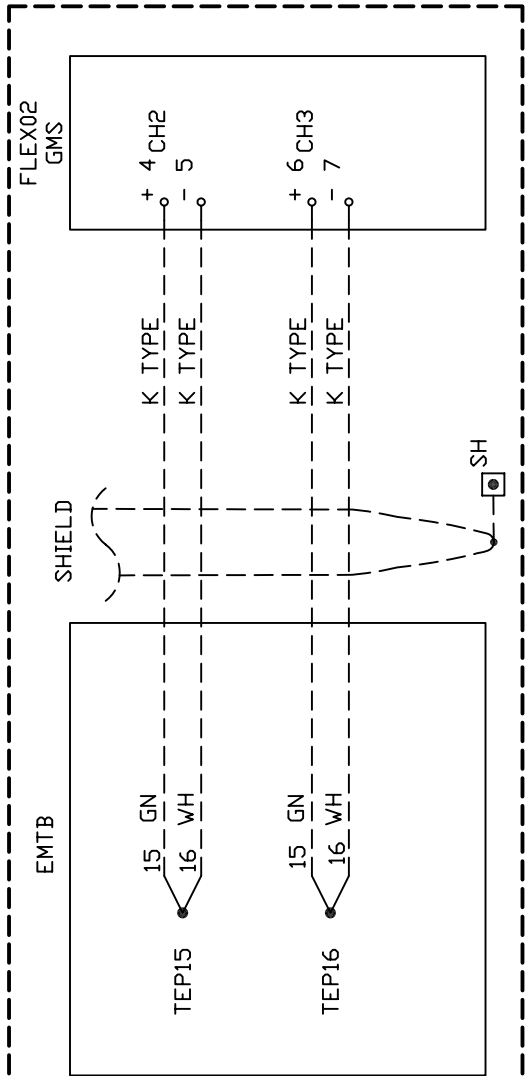
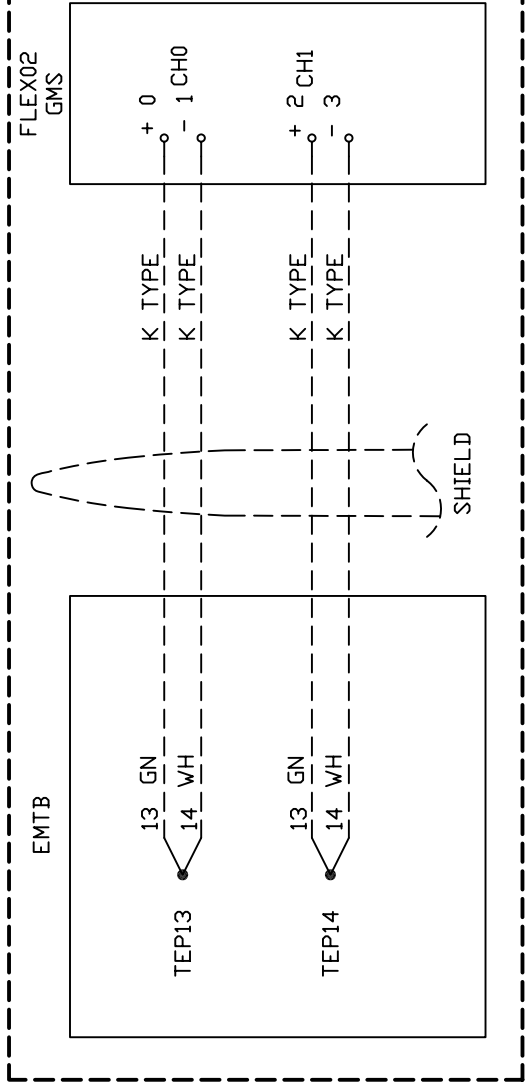
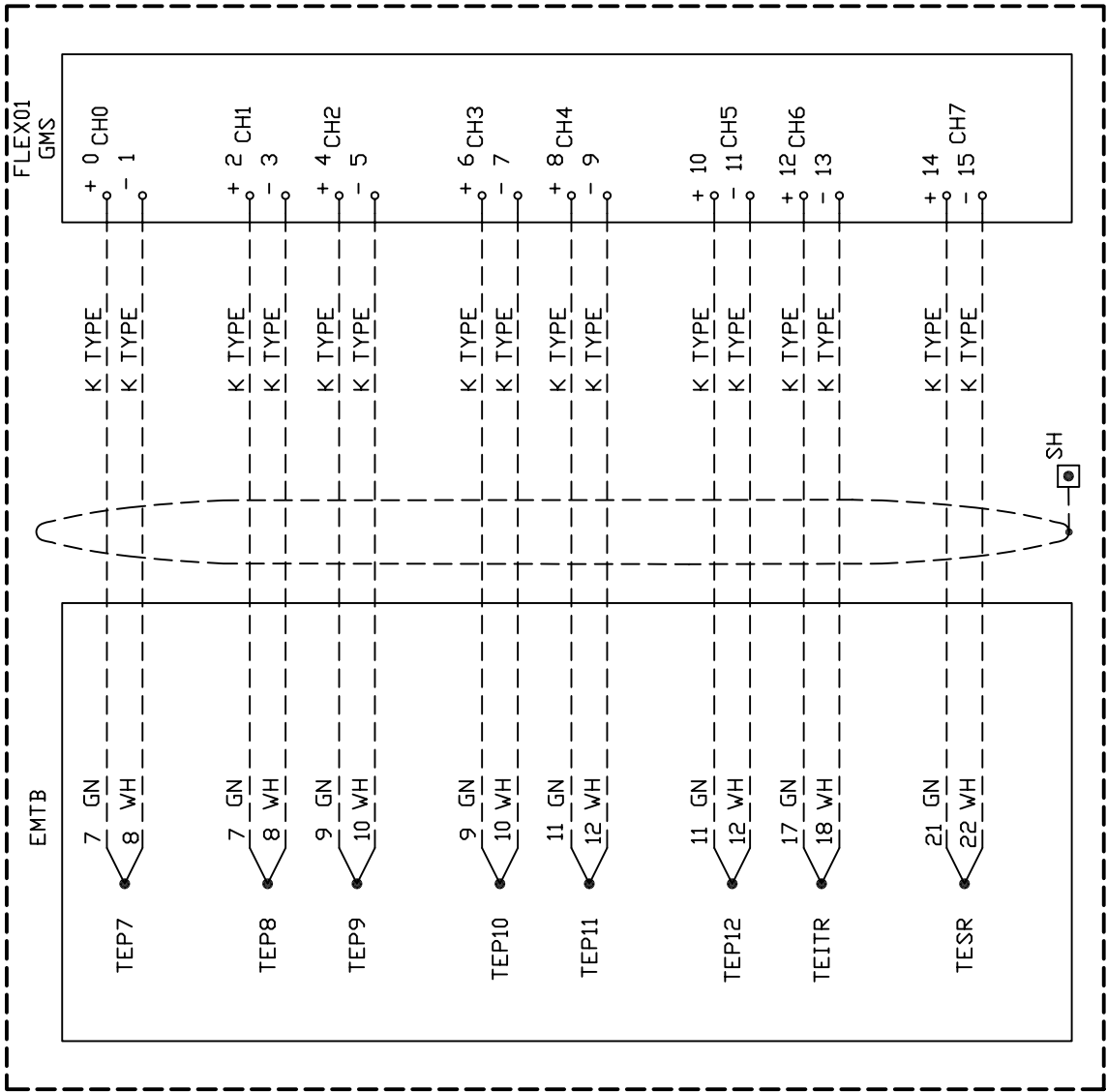
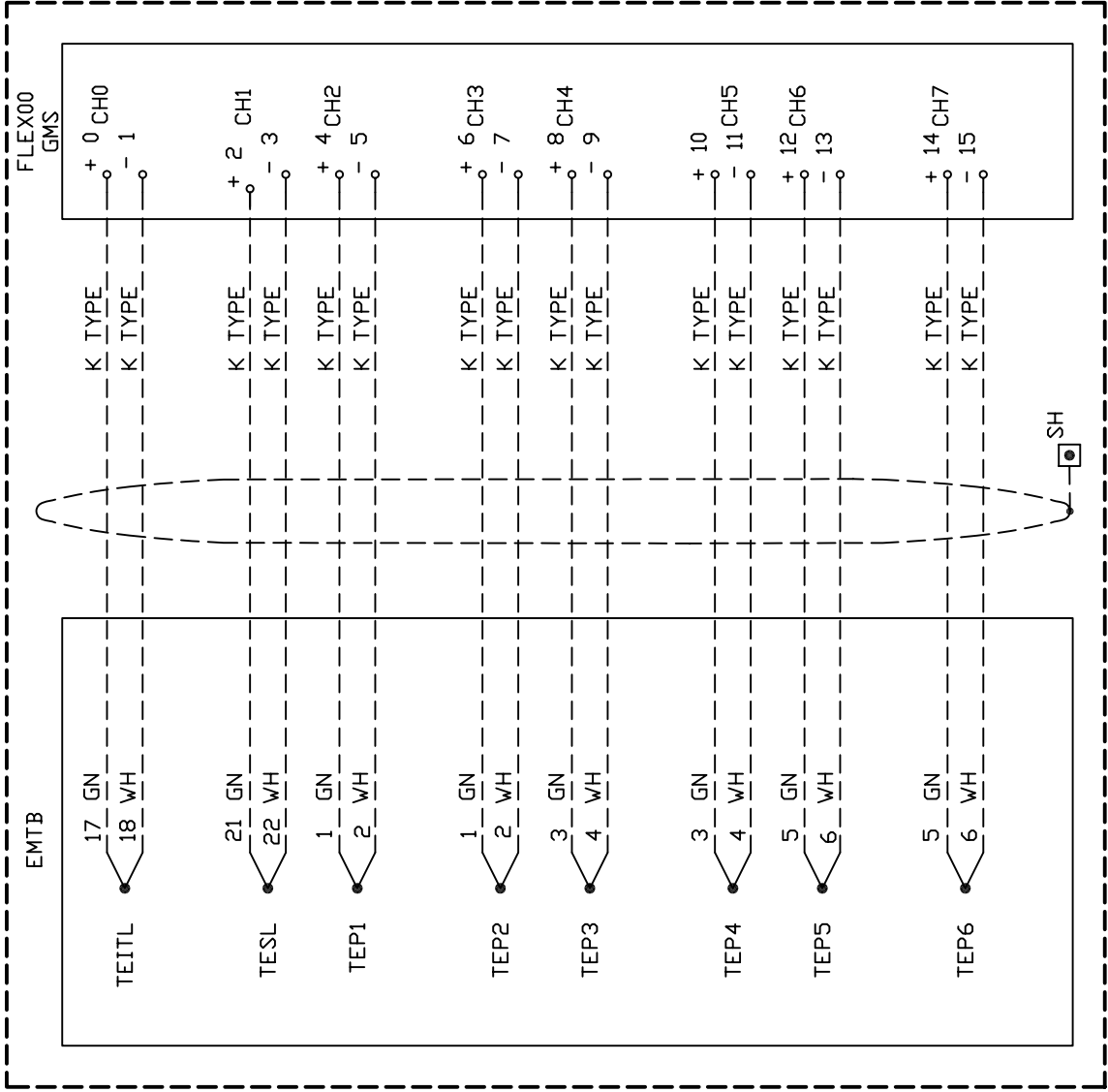
**CATERPILLAR INC.**

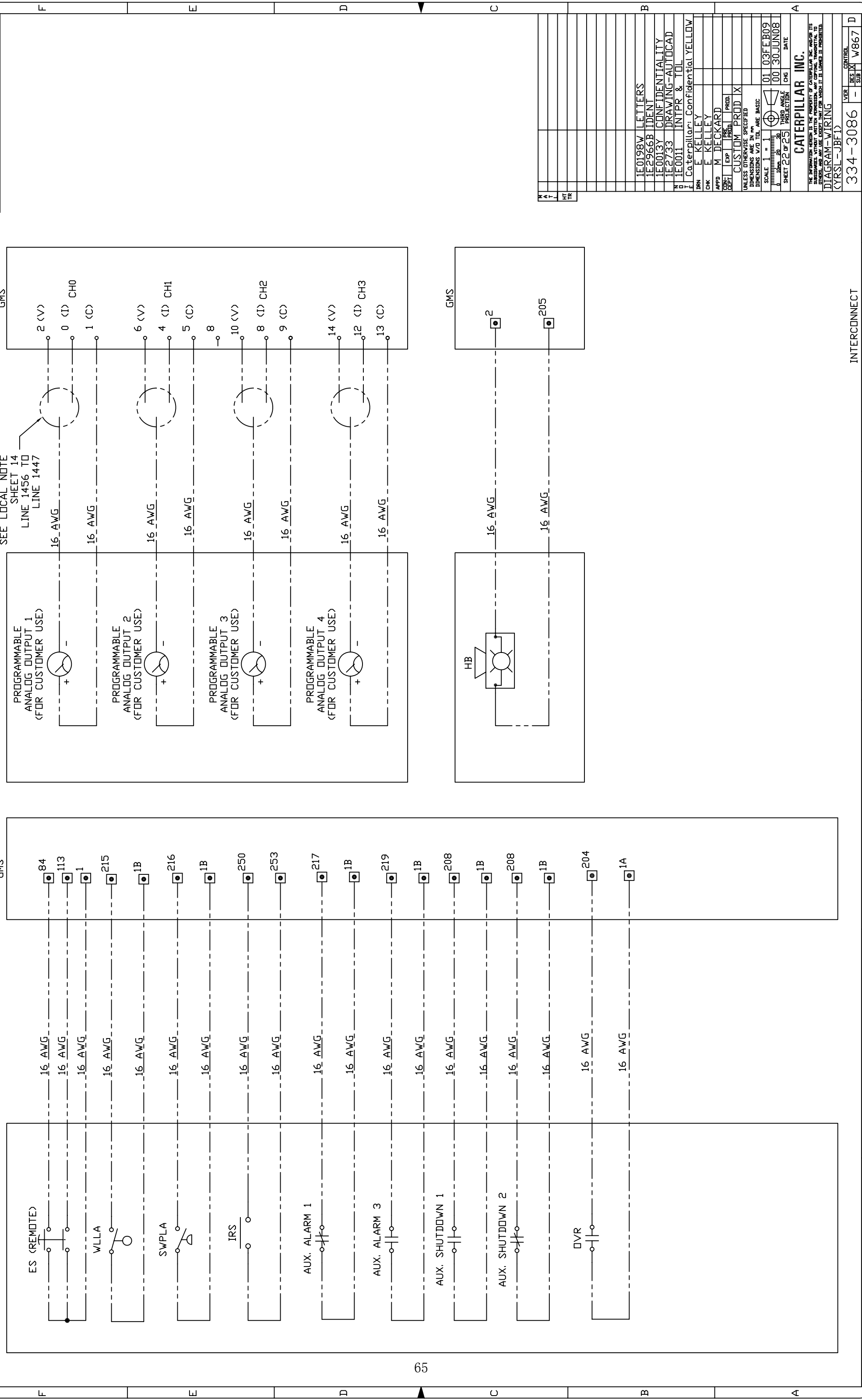
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**DIAGRAM-WIRING**

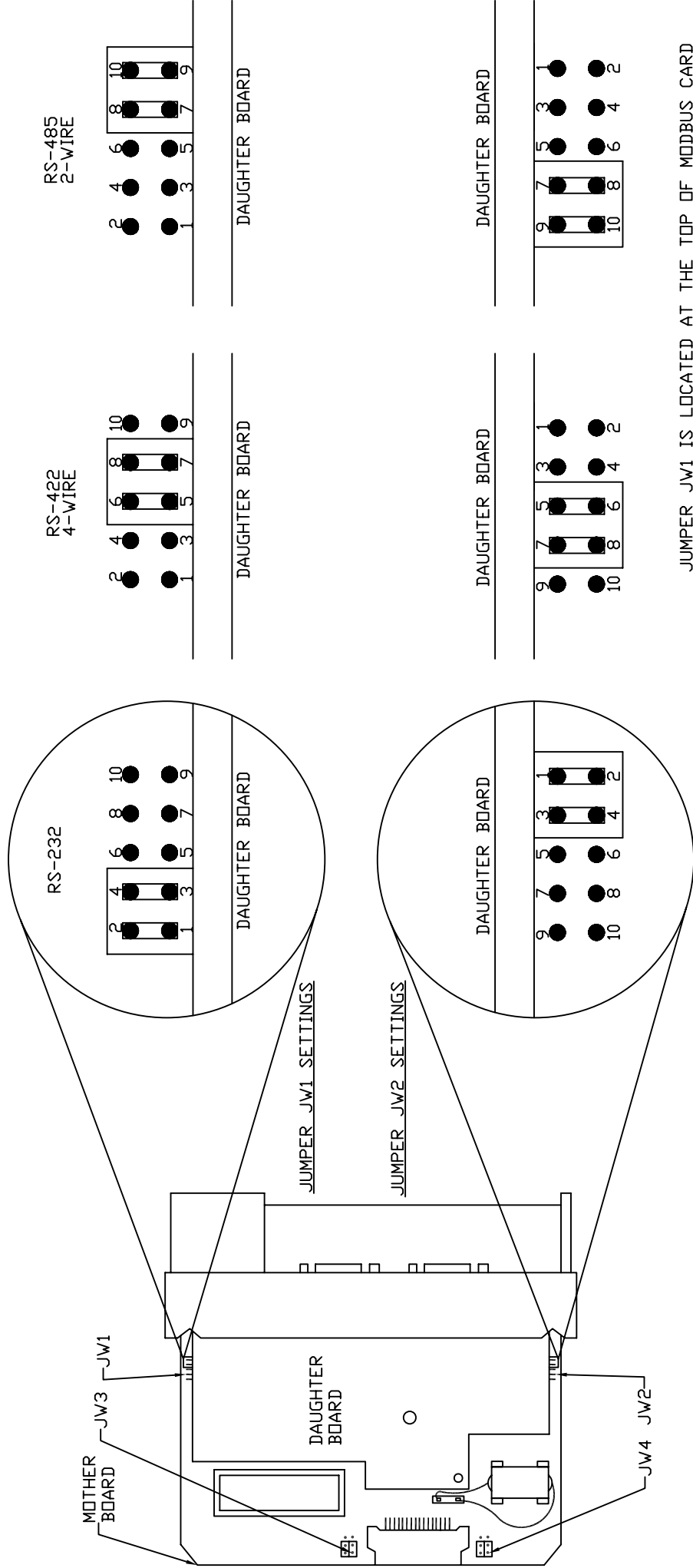
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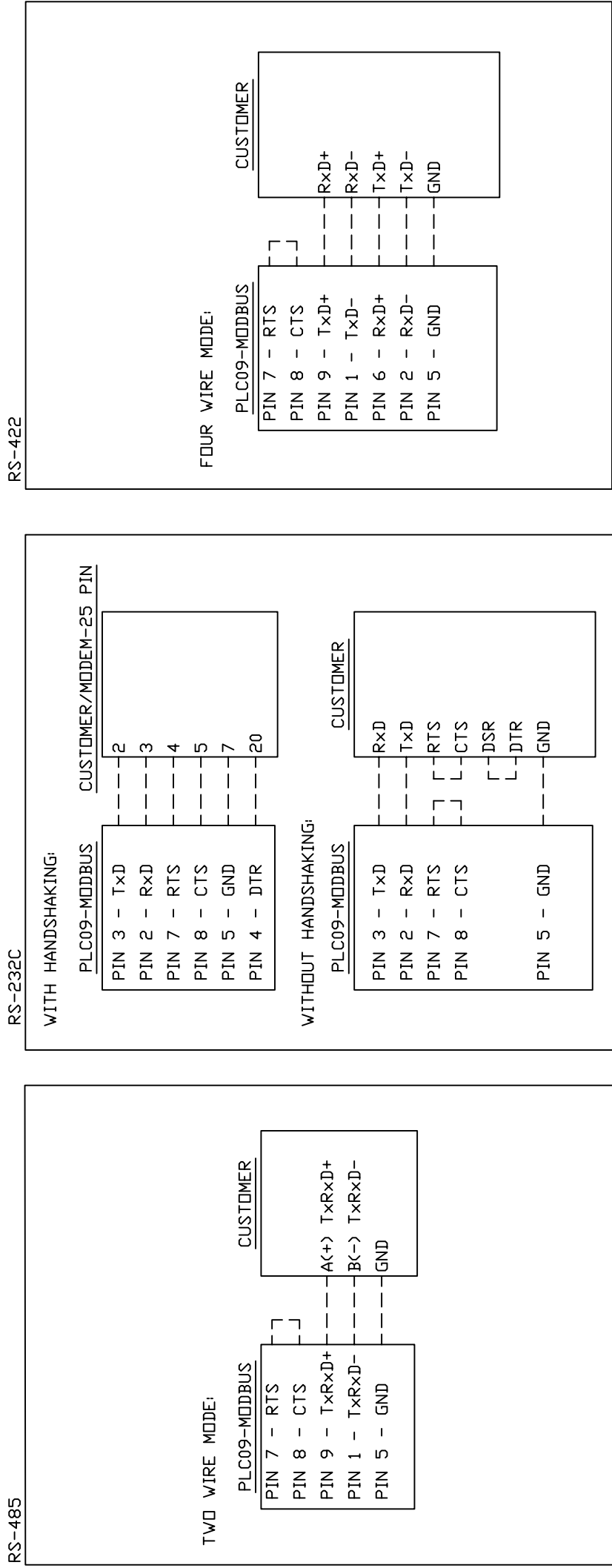
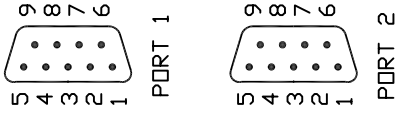
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## MODBUS JUMPER CONFIGURATION

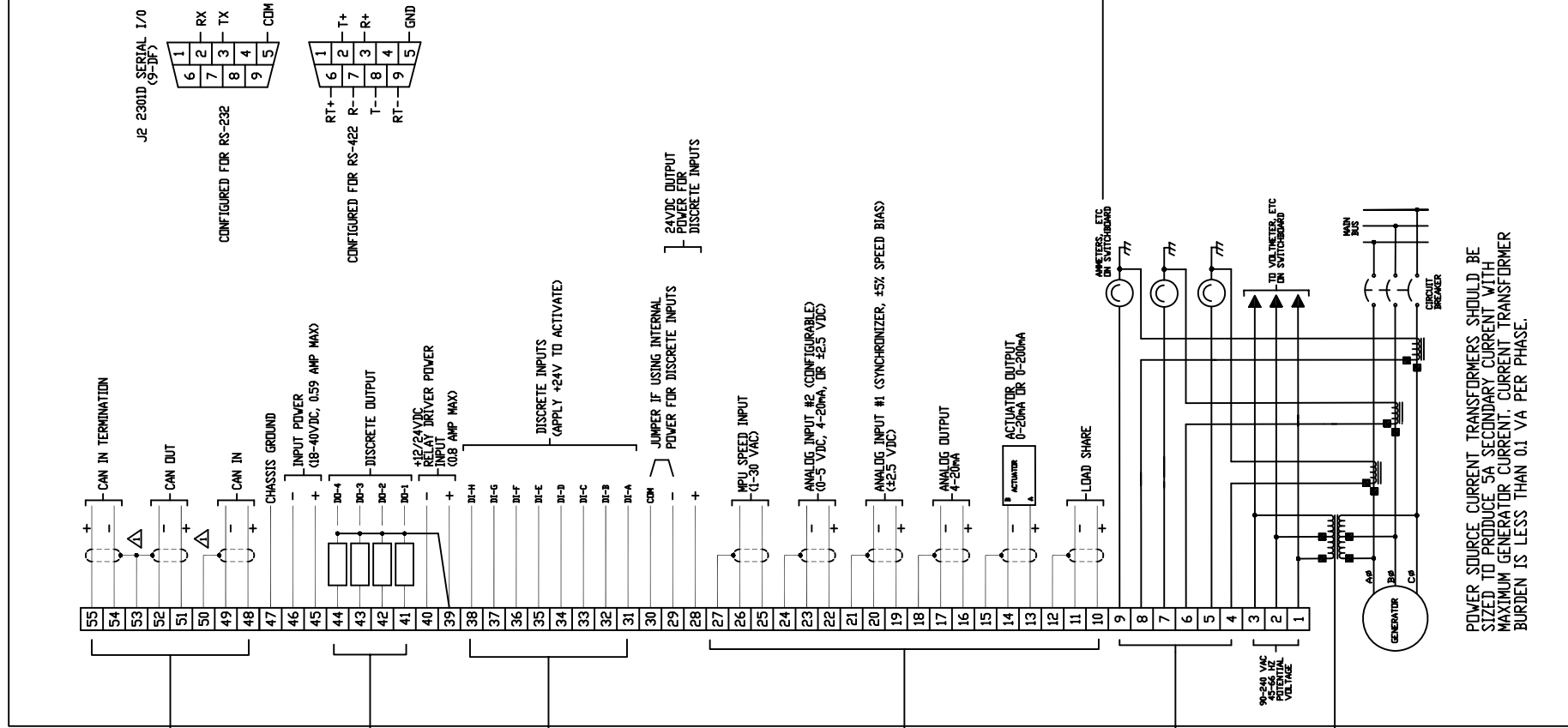


## MODBUS CABLE INFORMATION



IF COMMUNICATION IN RS-485 DOES NOT WORK, DESPITE ALL ATTEMPTS, TRY SWITCHING TERMINATION POLARITIES. SOME MANUFACTURERS INTERPRET (+) AND (-) DIFFERENTLY.

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SHIELDED WIRES TO BE TWISTED PAIRS, WITH  
SHIELD GROUNDED AT CONTROL END ONLY.

INDICATES RELAY COIL OR LAMP, 200 mA MAXIMUM PER CHANNEL.

**WARNING:** THE PRIME MOWER SHOULD BE EQUIPPED WITH SEPERATE OVERTEMPERATURE OR OVERPRESSURE SHUTDOWN DEVICES(S) TO PROTECT AGAINST RUNAWAY OR DAMAGE TO THE PRIME MOWER WITH POSSIBLE PERSONAL INJURY OR LOSS OF LIFE. DO NOT USE FOR EMERGENCY SHUTDOWN.

SHIELDED WIRES TO BE TWISTED PAIRS, WITH SHIELD GROUNDED AT CONTROL END ONLY.

INTERNAL CURRENT TRANSFORMER BURDEN MUST BE CONNECTED ACROSS POWER SOURCE CURRENT TRANSFORMER AT ALL TIMES, TO PREVENT LETHAL HIGH VOLTAGES. —

WITH A BALANCED THREE PHASE LOAD AND UNITY POWER FACTOR, THE CURRENT TRANSFORMERS SHOULD BE WIRED IN THE CORRECT POTENTIAL LEG AND MUST BE PHASED AT THE CONTROL AS FOLLOWS:

PHASE A: POTENTIAL TERMINAL 1, WITH RESPECT TO NEUTRAL, IN PHASE WITH CT TERMINALS 4 ( ) TO 5.

PHASE B: POTENTIAL TERMINAL 2, WITH RESPECT TO NEUTRAL, IN PHASE WITH CT TERMINALS 6 ( ) TO 7.

PHASE C: POTENTIAL TERMINAL 3, WITH RESPECT TO NEUTRAL, IN PHASE WITH CT TERMINALS 8 ( ) TO 9.

DIGITAL INPUT AND ANALOG OUTPUTS	
DI-4	MINIMUM FUEL, CLOSE TO RUN
DI-3	FAILURE OVERDIE
DI-2	DEGRADED
DI-1	LAND SWARM ENABLE, CR AUX
DI-E	SPEED TRIM RAISE
DI-F	SPEED TRIM LOWER
DI-G	LOAD
DI-H	SELECT BASE LOAD
DI-I	SPEED SWITCH #1
DI-2	SPEED SWITCH #2
DI-3	LOAD SWITCH #1
DI-4	LOAD SWITCH #2
ANALOG OUT	CONFIGURABLE (SEE SOFTWARE)
ANALOG IN #1	SYN IN
ANALOG IN #2	CONFIGURABLE (SEE SOFTWARE)
JP	R5232

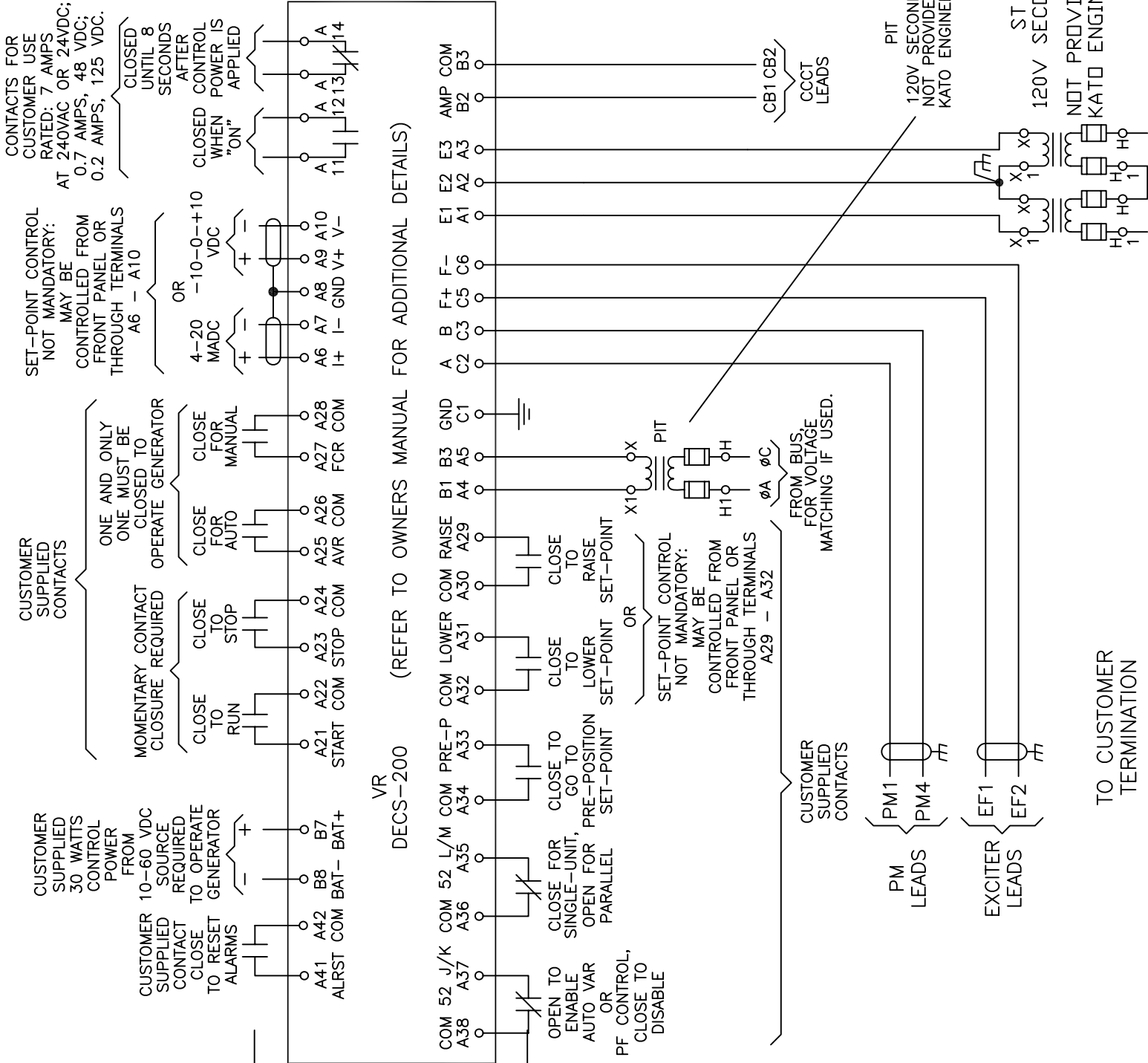
IF METERS ARE NOT USED, JUMPERS MUST BE  
INSTALLED IN PLACE OF METERS SHOWN.

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LEGEND

- CCCT - PARALLELING CURRENT TRANSFORMER
- DCT - DIFFERENTIAL CURRENT TRANSFORMER
- PIT - POTENTIAL ISOLATION TRANSFORMER WITH UNDERFREQUENCY LIMIT
- ST - SENSING TRANSFORMER
- VR - VOLTAGE REGULATOR WITH VAR/PF CONTROLLER, PARALLELLING AND VOLTAGE WATCHING

NOTE: ALL CUSTOMER-SUPPLIED CONTACTS SHOWN HERE MUST BE "DRY" (VOLTAGE-FREE) TO OPERATE IN A 12 VDC CIRCUIT.





**5.0 APPLICATION & INSTALLATION GUIDE****LIST OF GUIDE**

<b>No</b>	<b>Guide No</b>	<b>Guide Name</b>	<b>Total Pages</b>	<b>Remarks</b>
1		Guide for Diesel Fuel Selection	1	
2		Guide for fuel system Design	4	
3		Guide for Lube Oil System	2	
4		Guide for Pre-lube	3	
5		Guide for Cooling System	2	
6		Guide for Starting Air Systems	3	
7		Guide for Combustion Air System	13	
8		Guide for Exhaust System	16	
9		Guide for Crankcase Ventilation System	3	
10		Guide for Mounting	3	

- ~~Corrosion: A polished copper strip is immersed in fuel for three hours at 50°C (122°F).~~

~~Fuel imparting more than slight discoloration is rejected.~~

## Diesel Fuel Selection

The fuels recommended for use in Caterpillar diesel engines are normally No. 2-D diesel fuel and No. 2 fuel oil, although No. 1 grades are also acceptable. **Table 1** lists the worldwide fuel standards which meet Caterpillar requirements.

Standard	Name	Description
American	ASTM D975	No. 1-D & No. 2-D Diesel Fuel Oils
	ASTM D396	No. 1 & No. 2 Fuel Oils
	ASTM D2880	No. 1-GT & No. 2-GT Gas Turbine Fuels
British	BS 2869	Classes A1, A2 & B2 Engine Fuels
	BS 2869	Classes C2 & D Burner Fuels
West German	DIN 51601	Diesel Fuel
	DIN 51603	Heating Oil EI
Australian	AS 3570	Automotive Diesel Fuel
Japanese	JIS K2204	Types 1 (spl), 1, 2, 3, & 3 (spl) Gas Oil
U.S. Government	W-F-800C	DF-1, DF-2 Conus & DF-20 Conus Diesel Fuel
	W-F-815C	FS-1 & FS-2 Burner Fuel Oil
U.S. Military	MIL-L-16884G	Marine Oil

**Table 1**

**Table 2** lists acceptable aircraft jet fuels and kerosene type fuels.

Name	Description
ASTM D 1655-80	Aviation Turbine Fuel (JET A-1)
MIL-T-5624L	Aviation Turbine Fuel (JP-5) (NATO Code No. F-44)
MIL-T-8313B	Aviation Turbine Fuel (JP-8) (NATO Code No. F-34)

**Table 2**

---

## Diesel Fuel System Design Considerations

Diesel fuel supply systems must ensure continuous and clean supply of fuel to the engine's fuel system. The recommended diesel fuel supply system typically has three major components: a fuel storage system, a fuel transfer system and a fuel filtration system. The three component systems provide clean operating fuel to the engine.

### Fuel Storage Systems

Bulk fuel is usually stored in large main storage tanks and transferred to smaller auxiliary tanks (service tanks or day tanks) near engines by electric motor-driven pumps as shown in **Figure 3**.

If auxiliary tanks are not necessary, the main fuel tank must provide a ready fuel supply to the engine-mounted transfer pump.

### Main Fuel Tank

The main fuel tank represents the primary fuel reservoir in all applications, and must have adequate capacity for the intended

application. Rule of thumb for tank size is to find the fuel consumption rate at 100% load factor (depending on application: Prime, stand-by etc.) and multiply it with the number of hours between refills. Fuel consumption rates are shown on the Engine Technical Data Sheets for the specific engine. Additionally, 10% should be added to the result; 5% for expansion at the top of the tank, and 5% for sediment settlements at the bottom.

#### **Example:**

A power plant with one (1) 3516B diesel generator set, rated for 1145 bkW (1560 bhp) at 100% load. The fuel rate for the engine is 284 L/hr (75 G/hr) as found in TMI.

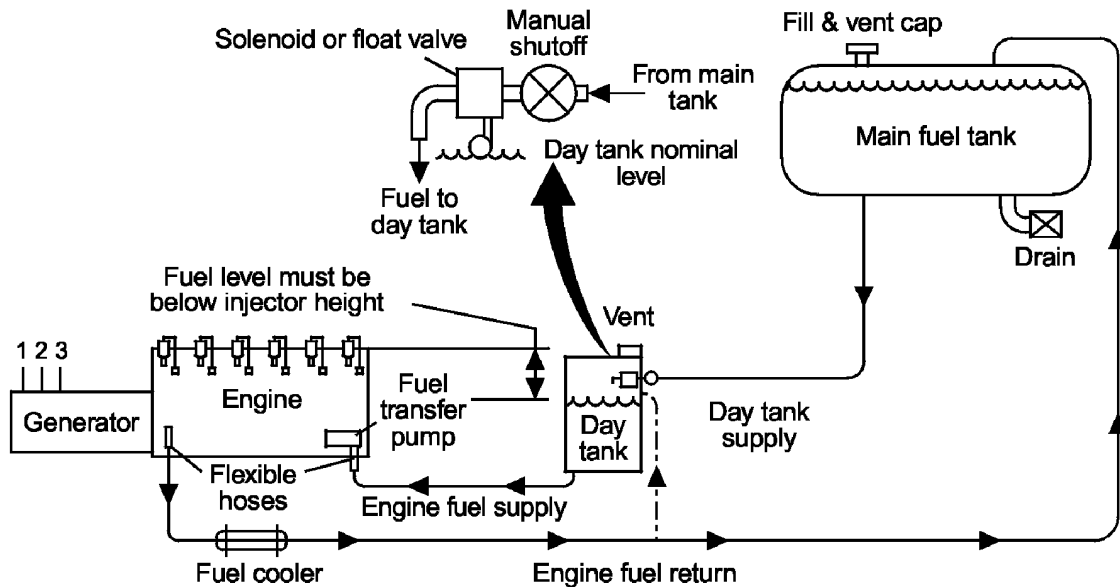
The time between tank refills is based on weekly fuel tanker truck deliveries, so refill time is 168 hours.

#### **Solution:**

$$\text{Tank vol.} = 284 \times 168 \times 1.1 = 52,583 \text{ L}$$

$$\text{Tank vol.} = 75 \times 168 \times 1.1 = 12,600 \text{ gal}$$

## Installation Example with Main and Auxiliary Fuel Tanks



**Figure 3**

### Auxiliary Fuel Tanks

Auxiliary fuel tanks, service tanks and day tanks are secondary fuel tanks located between the main fuel tank and the engine. These tanks are required in the following situations.

- The main fuel tank is located on the same level but more than 15 m (50 ft) away.
- The main fuel tank is located 3.7 m (12 ft) or more below the engine.
- The main fuel tank is located above the engine fuel injectors.

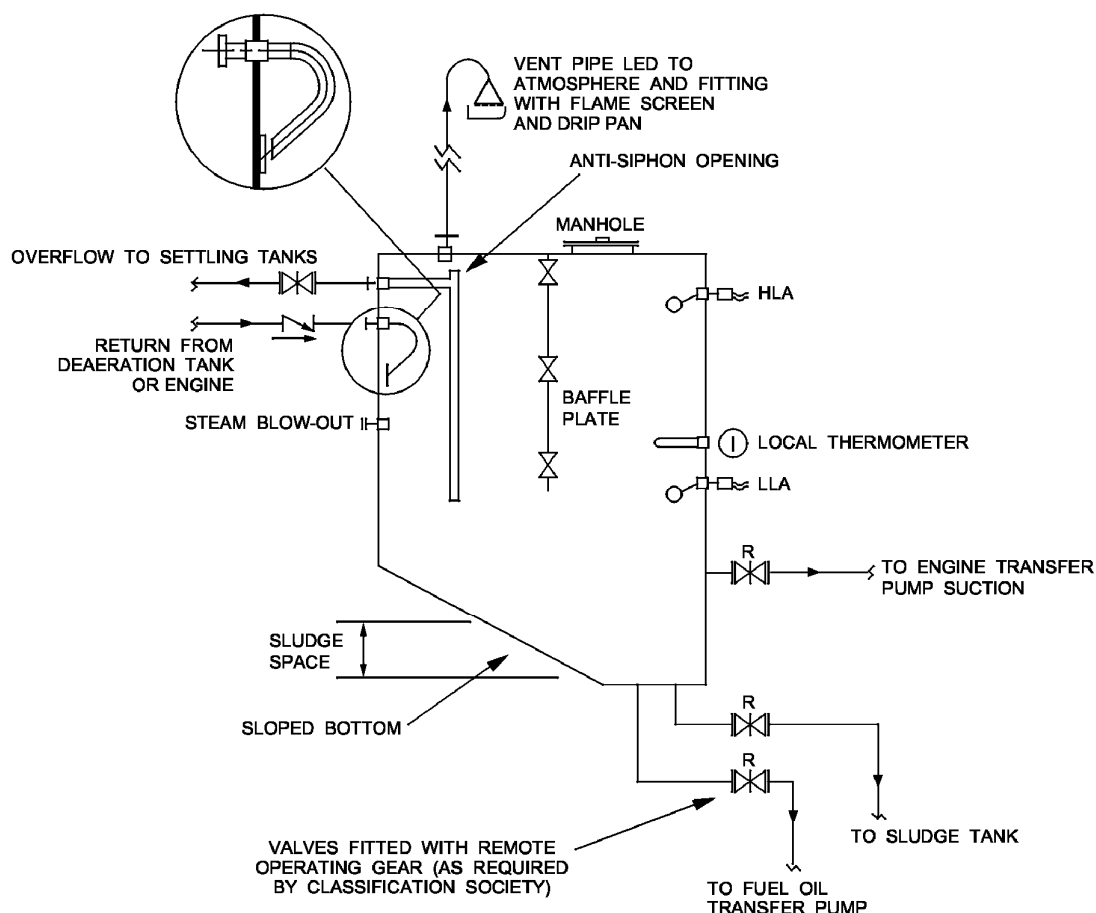
Any of the above conditions can cause unsatisfactory engine starting and operation. The purpose of an auxiliary tank is to relieve the fuel

pressure "head" from the fuel transfer pump and injection equipment for efficient fuel flow.

A manual fuel priming pump, offered as an attachment, or an electric motor-driver boost pump may allow operation under conditions more severe than those previously described; but where starting dependability is required, Caterpillar recommends the use of an auxiliary fuel tank.

Auxiliary tanks offer convenient and ready fuel storage while providing a settling reservoir for water, sediment and sludge. An example of an auxiliary fuel tank is shown in **Figure 4**.

## Typical Auxiliary Fuel Tank



**Figure 4**

### Fuel Service Tank or Day Tank

Auxiliary tanks such as fuel service tanks or day tanks can be quite simple. It usually consists of a small metal tank, either floor or wall mounted, in the immediate vicinity of the engine. The tank is usually sized to hold approximately eight hours of fuel, based on the engine's fuel consumption rate at full load.

Refilling can be accomplished by gravity, a hand pump, or with a motor-drive pump. Motor-drive pumps can be either manually or

automatically controlled. For convenience and safety, automatic control is usually employed using a float-actuated, electric motor-drive pump. The refilling pump can be positioned either at the auxiliary tank or at the main tank outlet. The performance capability of the pump must be considered during placement.

Features of the auxiliary tank, as shown in **Figure 5**, should include the following.

- Fill line - Located above the high fuel level, with outlet baffled to prevent agitation of sediment in the tank.
- Delivery line - Located near the bottom but not so low as to pick up collected sediment or condensation.
- Return line - To carry excess fuel back to the auxiliary tank. Should have its outlet baffled for the reason described above.
- Overflow line - Allows excess fuel to return to the main tank in event of overfilling of the auxiliary tank.
- Vent line - Allows air pressure to equalize as tank is drained or filled (vent cap should be located away from open flame or sparks).
- Drain valve - Allows removal of condensate and sediment.
- Sight glass or float-type gauge - Provides a positive check on fuel level.

To prevent damage to the fuel filter housings, the return line should have no valves or restrictions to allow dangerous pressure buildups.

Flexible rubber hoses, used as fuel return lines, should be supported to prevent closing off over time due to weight of the hose and fuel. Hard fuel lines prevent this problem, but a flexible connection is still required to isolate vibration between the line and the tank.

A nonflammable tank mounting will maximize fire protection.

The overflow line should be at least two pipe sizes larger than the

fill line. To simplify engine maintenance, a shut-off valve in the supply line is useful.

The delivery line, carrying the fuel to the engine-mounted fuel transfer pump, and the return line, carrying excess fuel back to the tank, should be no smaller in size than the respective fittings on the engine.

Larger fuel supply and return lines ensure adequate flow if the fuel tank supplies multiple engines over 9 m (30 ft.) away from the tank or when temperatures are low. Consult general dimension drawings for the sizes for each model.

It is important that the fuel return line is sloped down to the tank with no traps or obstructions in the line. If this is not done, the fuel system is prone to air-lock with consequent hard-starting.

The auxiliary tank should be located so that the level of the fuel when the tank is full is no higher than the injection valves. Static pressure would allow fuel to leak into the combustion chambers in the event of injection valve leakage. The tank should be close enough to the engine so that the total suction lift to the transfer pump with the fuel at low level, plus the line loss of the supply line, is less than the fuel pump's maximum suction lift capability. This figure should be minimized for better starting. A float valve or solenoid valve in this type of day tank regulates the fuel level to keep it below the level of the injectors.

## Lubrication Oil System



### General

The lube system is designed to supply 85°C (185°F) filtered oil at 430 kPa (62 psi) pressure under all engine operating conditions. For engines with front-mounted turbochargers, the lube oil system is designed to supply 70°C (158°F) filtered oil at 430 kPa (62 psi) under all engine operating conditions. The major feature of the 3600 lube system is a priority valve to regulate the oil pressure at the oil manifold rather than at the oil pump. The oil manifold pressure thus becomes independent of the oil filter and oil cooler under pressure drop.

The other major features of the 3600 lube system are as follows:

### Internal Lubrication System

#### Oil Coolers

The engines are equipped with a two-element lube oil tube cooler with the water flow arranged in series. As an option, a three-element oil cooler is available if additional cooling capacity is required. For engines with front-mounted turbochargers, a fresh water cooled plate-type cooler is shipped loose.

#### Thermostats

Four thermostats in the lube system control the oil inlet temperature to 85°C (185°F) for engines with rear-mounted turbochargers and 70°C (158°F) for engines with front-mounted turbochargers. The thermostat is mounted on the plate-type cooler for engines with front-mounted turbochargers.

#### Oil Filters

The oil pan is equipped with a 650-micron suction screen. The two final 20-micron lube oil filters can be changed while the engine is operating. The normal procedure specifies the filters to be changed at 100 kPa (14.5 psi) pressure drop across the filters.

#### Centrifugal Bypass Filters

Engine-mounted centrifugal bypass oil filters are installed as standard. The filters receive three to four percent of the oil pump flow to remove solid, micron size particles and can extend the oil filter change periods. The centrifugal filters each have a dirt capacity of 3.6 kg (8 lb). Cleaning intervals are discussed in the Caterpillar Operation & Maintenance Manual.

#### Oil Pumps

On-line connections are identical for all pumps. The pumps provide more than the required engine oil flow at rated conditions. This allows high oil pressure early in the operating speed range and provides flow margin.

## Prelubrication

All 3600 engines must be lubricated prior to starting. The pump is electrically driven. Prelubrication prior to starting may take several minutes depending upon oil viscosity, temperature, engine condition, and configuration. When the engine oil manifold pressure reaches 10 kPa (1.5 psi), the starter interlock allows the engine to be cranked. A prelubrication system supplied by Caterpillar is integrated with the engine and starting controls. An engine-mounted, air-driven prelube pump is available as an option.

## Oil Requirements

Caterpillar recommends the use of **Cat DEO CF** lubrication oil for the 3600 family engines operating on distillate diesel fuel. Caterpillar does not recommend other lubrication oils by brand name as the formulations are not controlled by Caterpillar and the lube oil properties may vary geographically.

To be acceptable in a 3600 engine, the oil must demonstrate satisfactory performance in the following areas:

1. **API category:** The oil must have an API classification of CF. Specifications Mil-L-2104D oils also meet this requirement.

**API CC oils are unacceptable.**

2. Caterpillar **Micro-Oxidation Test (MOT):** The oil must pass the Caterpillar Micro-Oxidation Test. For additional information on this test, contact a Caterpillar dealer. The test must be performed on samples from the supplier's facility supporting the engine.

3. **Field experience:** In addition to the above items, the oil must show its performance in the field with good results, demonstrated through a test program.

## Lubricant Viscosity

The primary recommendation for the 3600 family of engines is an SAE 40-grade oil. SAE 30 and some multigrade oils may be used if the application requires. SAE 30 is preferable to a multigrade oil.

## Total Base Number (TBN)

Always consult a Caterpillar dealer for the latest lubricant recommendations.

3600 engines operating on distillate Marine Diesel Oil require a TBN of 10 times the sulfur level measured in percent of weight. (Example: For a sulfur content of one percent weight, the TBN would be 10.) The minimum TBN level regardless of the sulfur content is 10. Excessively high TBN or high ash oils should not be used in 3600 family engines on distillate fuel, as these oils may lead to excessive piston deposits and loss of oil control. Successful operation of 3600 family engines has generally been obtained with new TBN levels between 10 and 15.

Ultra low sulfur content: Caterpillar has not experienced any detrimental effects operating on 0.05 percent sulfur fuel in the 3600 family engine.

Ambient Temperature Range	
Oil Viscosity	Ambient Temp. Range °C
SAE 40	+5 to +50
SAE 30	+0 to +40
SAE 20W/40	-10 to +40
SAE 15W/40	-15 to +40



Refer to the engine model specific price lists for the various options available.

**Note:** Remote mounted prelube pumps must be located and piped to prevent excessive inlet restriction.

Several automatic prelubrication systems available for Caterpillar engines are:

- Intermittent Prelube System
- Continuous Prelube System
- Redundant Prelube System
- Quick Start Prelube System

### **Intermittent Prelube System**

The intermittent prelube system provides suitable performance for applications not requiring quick start capability.

The intermittent prelube system uses an engine mounted pump and is engaged immediately prior to engine start-up.

**Figure 5** is a schematic of a prelube system for a 3600/G3600 engine. The type of prelube pump determines whether the system is intermittent or continuous.

Intermittent prelube time will vary with engine model as well as oil temperature. A well-designed system must include a prelube pump shutdown capability to prevent the pump from operating too long. Since the intermittent prelube pump operates at a higher flow rate and pressure than the continuous pump, operating the pump for extended time periods is not recommended. This can result in excessive oil in the cylinders and potentially cause hydraulic lock at start-up.

### **Continuous Prelube System**

Continuous prelubrication is for immediate starting applications and is typically used in conjunction with jacket water and lube oil heating.

Used mostly on the 3600 family of engines, a continuous prelube system eliminates the delay of waiting for the completion of an intermittent prelube cycle. This system is operating continuously when the engine is not running. This ensures that lube oil will be available at the bearings at all times, allowing immediate starting of the engine. The continuous prelube systems utilize pumps with lower flow rates than intermittent prelube systems. This system relies on an engine oil level start permissive, in lieu of the pressure switch permissive used with the intermittent prelube system. A minimum level of oil in the engine is required to fulfill the starting system interlock.

### **Redundant Prelube System**

The redundant prelube system combines the continuous and intermittent prelube systems, offering the benefits of both. Under normal circumstances, the continuous prelube pump keeps the engine ready for immediate start-up by maintaining the level of oil in the engine. The intermittent prelube pump will only operate if the continuous pump fails. This system is typically selected for black start or emergency generator applications, when it is critical that an engine is able to start.

When an engine equipped with a continuous prelube system shuts

down, the intermittent pump will postlubricate the engine. After postlubrication the oil pressure decreases and the pilot controlled spill valve opens and the prelube pressure switch opens. When the pressure switch opens the continuous pump will energize and maintain the oil level in the engine.

### **Quick Start Prelube System**

The quick start prelube system consists of two electric prelube pumps, a continuous pump and a booster pump. This system is similar to the redundant prelube system, except that the booster pump is not just a back-up for the continuous pump, it is an integral part of the system.

**Figure 6** and **Figure 7** are schematic examples of quick start prelube systems.

While the engine is not operating, the continuous pump maintains the oil level near the top of the cylinder block via the spill valve. When the engine is started, the continuous pump will stop and the booster pump will start in order to raise the pressure to a sufficient level to permit cranking.

When the engine shuts down, the booster pump will postlubricate. After postlubrication, the oil

pressure decreases and the pilot controlled spill valve opens and the prelube pressure switch opens. When the pressure switch opens the continuous pump will energize and maintain the oil level in the engine.

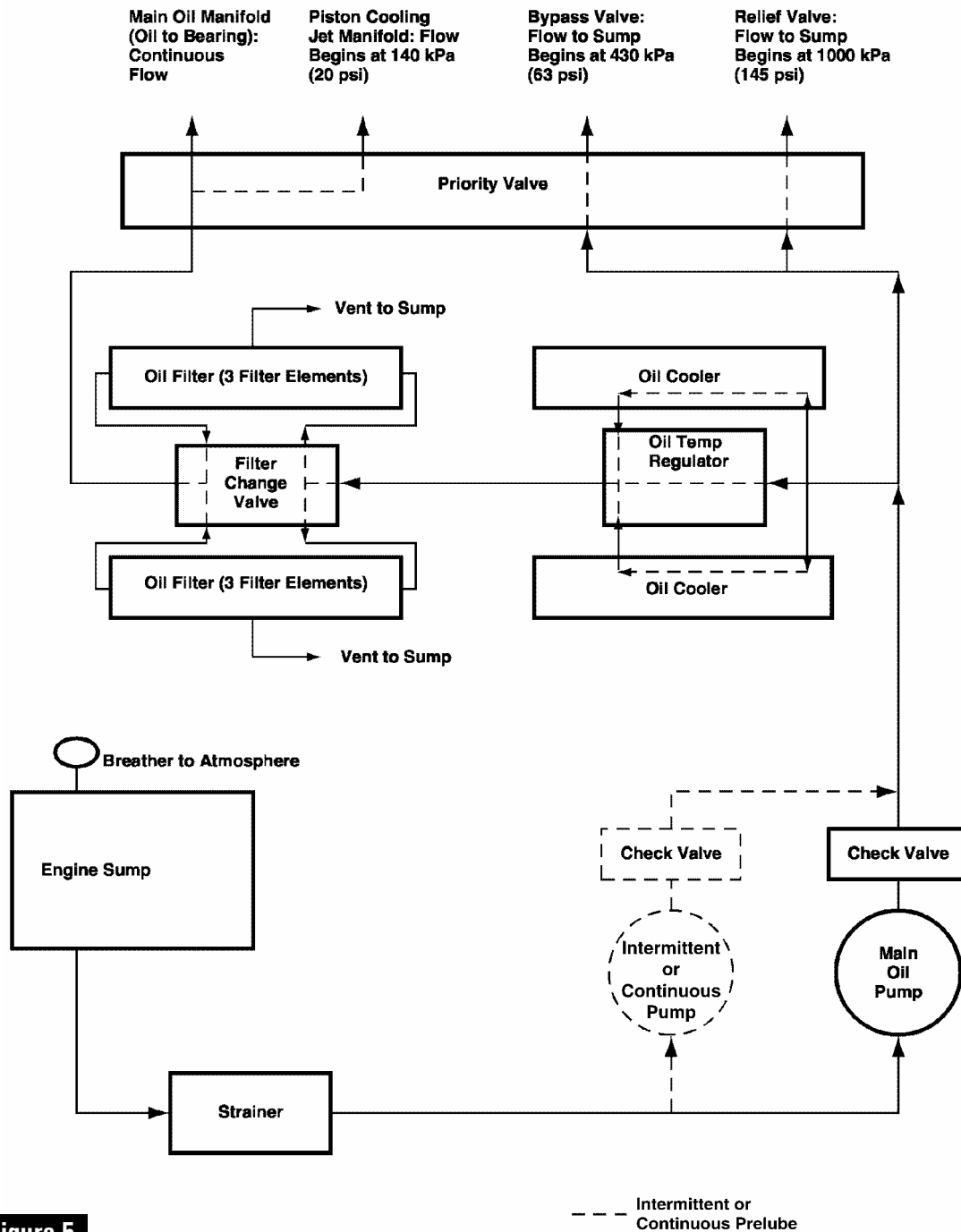
Quick start prelube time will vary little with oil temperature. Typical quick start prelube times, measured from the start initiate signal to starter engagement, are 5 to 7 seconds with 25°C (77°F) oil.

### **Postlubrication**

3600/G3600 engines have a standard postlubrication cycle. Postlubrication maintains oil flow after engine shutdown to protect the turbocharger bearings.

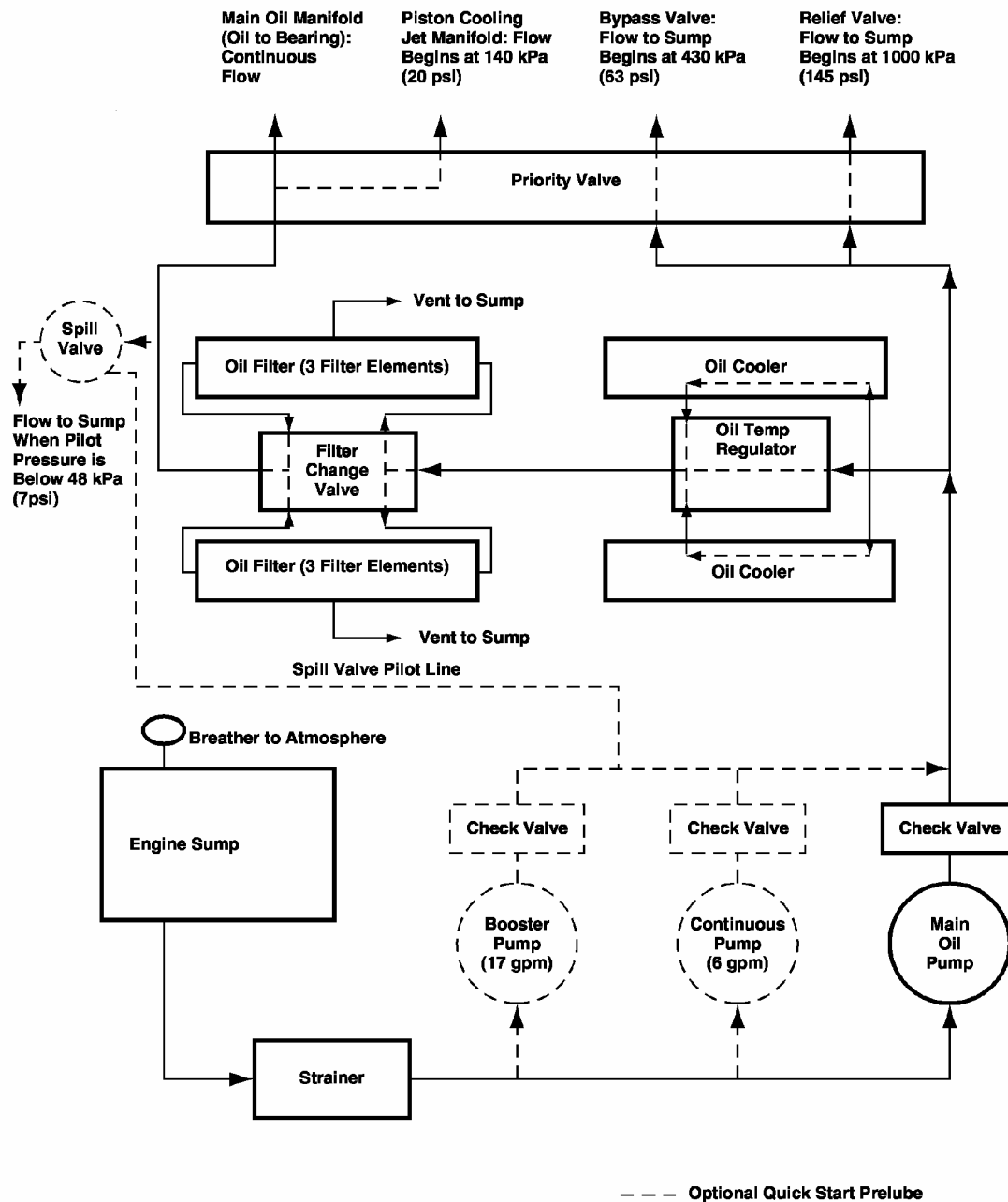
Engine postlubrication will not function if the Emergency Stop (E-Stop) button is used to shutdown the engine. Since an oil leak could potentially require the use of the E-Stop button, the postlubrication is disabled to stop oil flow to a possible leak. An E-Stop button is located on the control panel, junction box and the customer terminal strip. Since no postlubrication occurs with the use of the E-Stop button, the E-Stop should only be used for emergency shutdowns.

### 3600/G3600 Engine Lubricating Oil System with Intermittent or Continuous Prelube



**Figure 5**

### G3600 Engine Lubricating Oil System with Optional Quick Start Prelube


**Figure 6**

## 3600/G3600 Air Prelube

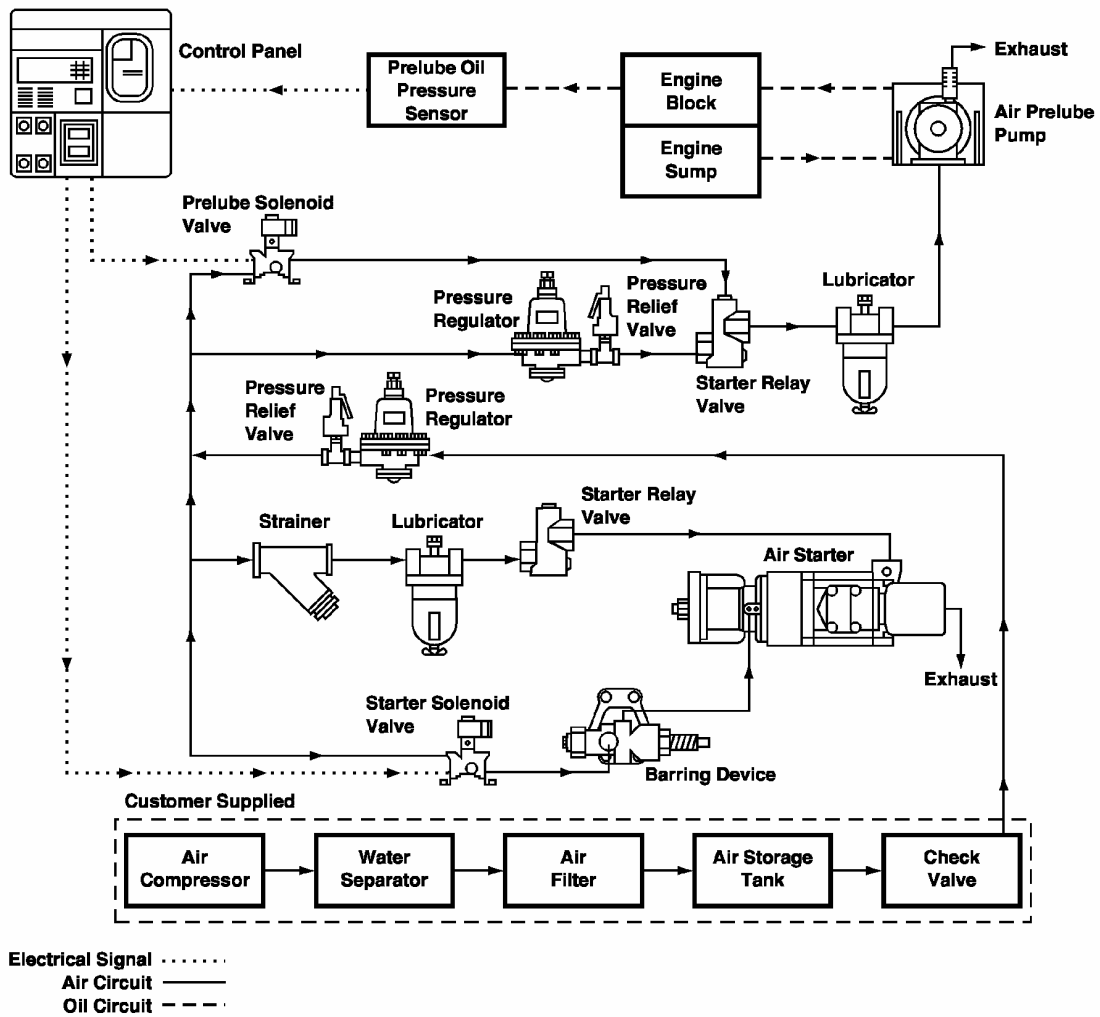
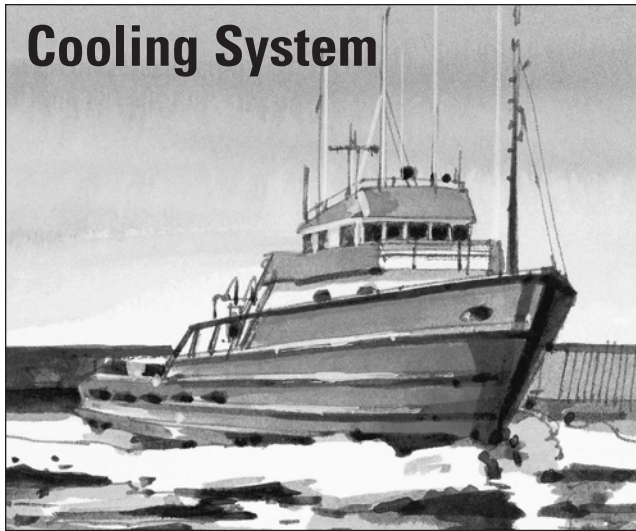


Figure 7



## Cooling System

### Introduction

The cooling system configuration for the 3612/3616 engine in marine propulsion applications is identified as a combined or single-circuit system. It is also referred to as the single-circuit – two-pump system.

The cooling system is laid out for the following temperature levels:

1. 32°C (90°F) nominal [38°C (100°F) maximum] water temperature to the aftercooler and oil cooler.
2. 90°C (194°F) nominal jacket water temperature to the cylinder block.
3. 85°C (185°F) nominal oil temperature to bearings.

The heat rejection data in this project guide are based on 32°C (90°F) nominal water [38°C (100°F) maximum] to the aftercooler and 25°C (77°F) air to the turbocharger inlet. The thermostatic valve for the low temperature control circuit has 32°C (90°F) nominal elements installed. However, with sea water at 32°C (90°F), the inlet temperature is allowed to go to 38°C (100°F) max. Thus a 6°C (10°F) approach temperature is available for sizing the heat exchanger.

### Coolant Flow Control

Correct coolant flows are obtained by factory installed orifices on the engine, combined with proper external circuit resistance set at each site during commissioning. The external circuit resistance setting establishes the total circuit flow by balancing total circuit losses with the characteristic pump performance curves. *Correct external resistance is very important.* Too high a resistance will result in reduced flows to the aftercooler and oil cooler, and their effectiveness will decrease. If there is too low a resistance, the fluid velocity limits may be exceeded, and cavitation/early wear could be the result.

The chart below shows the recommended external resistance maximum pressure drop for a 3612/3616 engine with combined circuit cooling systems.

### Coolant Temperature Control

The 3612/3616 engine uses fluid inlet control temperature regulators to provide uniform coolant temperature to the aftercooler, oil cooler, and cylinder block. For the combined system, the AC/OC circuit is externally regulated to provide a nominal 32°C (90°F) coolant temperature. The high temperature jacket water system uses the AC/OC outlet water to maintain 90°C (194°F) inlet water to the block. Jacket water temperature regulation is always required for all configurations and operating conditions.

### Fresh Water Pumps

The 3612/3616 engine has two identical gear-driven centrifugal water pumps mounted on the front housing. The right-hand pump (viewed from the flywheel end) supplies coolant to the block and heads. The left-hand pump supplies coolant to the aftercooler, oil cooler, and ultimately supplies the required cooling water needed to control the jacket water inlet temperature.

Engine Speed rpm	3612/3616 Differential Pressure kPa (psi) Full Cooler Flow	3612/3616 Differential Pressure kPa (psi) Full Bypass Flow
1000	85 (12)	130% of 85 (12)
900	66 (9.6)	130% of 66 (9.6)

## Heat Exchanger Sizing

The **minimum** acceptable heat exchanger configuration for a combined circuit system must provide 38°C (100°F) **maximum** coolant temperature at the AC/OC pump inlet at the following conditions:

1. **maximum** expected ambient temperature
2. **maximum** engine power capability (rack stop setting)
3. **maximum** expected sea water temperature
4. expected sea water fouling factor
5. anticipated coolant composition (i.e. 50% glycol).

In addition to the above criteria, the combined circuit heat exchanger should provide 32°C (90°F) coolant at normal load conditions, average summer ambient temperature, and normal sea water temperatures. See the technical data section of this manual for specific heat rejection figures.

Typical Heat Balance of the 3600 Engine	
Total energy input	100%
Mechanical work energy	44%
Exhaust gas	29%
Aftercooler cooling water	12%
Cyl block, heads cool water	9%
Lube oil cooling water	4%
Radiation	2%

## Jacket Water Heater Sizing

Caterpillar recommends using a separate heater and circulation pump. *An immersion-type heater is not recommended.* The pump should be sized to 150 L/min (39 gal/min) at 120 kPa (17 psi), and the heater is recommended to be minimum 24 kW or as required by the actual application.

## System Pressures and Velocities

The following pressure and velocity limits apply to 3600 family of engines:

Water Pump Pressures		
Maximum allowable static head	kPa (psi)	145 (21)
Minimum AC/OC inlet pressure (dynamic)	kPa (psi)	-5 (-0.7)
Minimum JW inlet pressure (dynamic)	kPa (psi)	30 (4)
Minimum sea water inlet pressure (dynamic)	kPa (psi)	-5 (-0.7)
Maximum Operating Pressures		
Engine cooling circuits	kPa (psi)	500 (73)
Caterpillar expansion tanks	kPa (psi)	150 (22)
Heat exchangers		type specific
Water Velocities		
Pressurized lines	m/s (ft/s)	4.5 (14.7)
Pressurized thin walled tubes	m/s (ft/s)	2.0-2.5 (6.6-8.2)
Suction lines (pump inlet)	m/s (ft/s)	1.5 (4.9)
Low velocity de-aeration line	m/s (ft/s)	0.6 (1.9)

## Weld Flange Sizes

All external cooling water connections of 100 mm (3.9 in.) diameter or above are supplied with DIN flanges, drilled to DIN 2501. Available in CAT standard,

ANSI standard, or DIN standard. CAT standard weld flanges are provided at every connection point; ANSI or DIN can be furnished.



## Air Starters

The Caterpillar supplied air starting system includes air starters, lubricator, air relay valve, strainer, shutoff valve, and pressure regulator, if required. The engine uses two vane-type starters as standard equipment. Turbine starters are optional.

## System Requirements

The standard vane air starters operate on air inlet pressures from 700 to 1550 kPa (102 to 225 psi). The turbine starters operate on pressures from 480 to 1040 kPa (70 to 151 psi). These pressures are required at the starter inlet port. An air tank pressure of 700 kPa (102 psi) [480 kPa (70 psi) for turbine] will not start the engine because of the pressure drop associated with the air supply lines. For initial system evaluation, assume a 200 kPa (29 psi) pressure drop between the tank and the air starter inlet.

A pressure regulator is necessary when the supply pressure exceeds the starter operating pressure. The pressure regulator should be set from 860 to 1550 kPa (125 to 225 psi). It should have the capacity to flow 300 L/sec (78 gal/sec) per starter at regulator inlet pressures above 860 kPa (125 psi).

The quantity of air required for each start and the size of the air receiver depend upon cranking time and air-starter consumption. A typical first start at 25°C (77°F) ambient will take five to seven seconds.

Restarts of warm engines normally take place in two to three seconds. The control system will shut off the air to the air starters at 170 rpm engine speed. At this firing speed, the governor is activated to allow fuel to the engine.

## Receiver Sizing

The chart on page 58 is for typical air receiver sizing. The chart shows the number of starts available with an initial starting air receiver pressure as shown on the curves. This chart assumes: Caterpillar vane-type starters, reduction valve outlet pressure of 860 kPa (125 psi), and a minimum pressure of 690 kPa (100 psi) at the starter inlet port. An average cranking time of three seconds has been used. The starting air receiver size is normally determined by the requirements of the classification society for the number of starts or start attempts.

The size of the air receivers should be increased if the starting air receiver also supplies air for purposes other than the main engine starting (e.g., engine air prelube, work air, auxiliary gen sets). The Caterpillar intermittent air prelube pump consumption rate is 28.2 L/sec (7.3 gal/sec) based on free air at 21°C (70°F) at 100 kPa (70°F at 14.5 psi). The pump motor operating pressure is 690 kPa (100 psi). The prelube pump will normally operate one to five minutes before the engine begins to crank.

## Customer Connections

Vane-type starters must be supplied with clean air. Deposits of oil-water mixture must be removed by traps installed in the lines. Lines should slope toward the traps and away from the engine. The air supply pipes should be short and at least equal in size to the engine inlet connection, which is 38 mm (1-1/2 in). Locate the pressure reducing valve as close to the engine as possible to minimize the air pressure reduction valve supply pipe diameter.

## Starting Air System Schematic

A typical starting air system is illustrated on page 59.

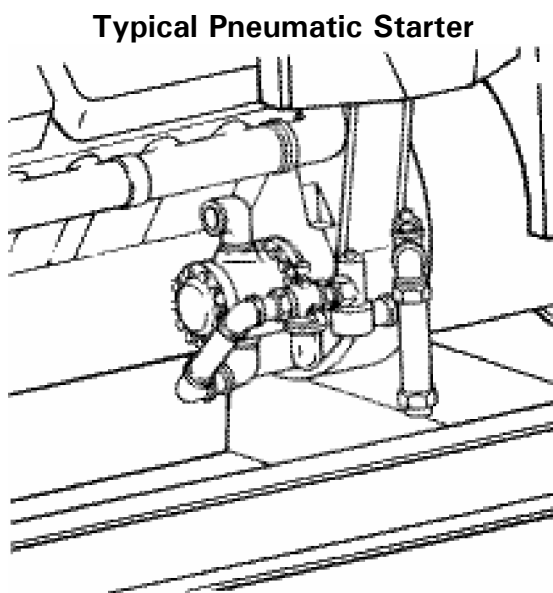


## Pneumatic Starting (Air)

Air starting, either manual or automatic, is highly reliable. Torque available from air motors accelerates the engine to twice the cranking speed in about half the time required by electric starters.

Pneumatic starting is generally applied to large engines in work-boats, on land where facilities have existing plant air, or where a combustible gases may be present in the atmosphere.

The air system can be quickly recharged; but air storage tanks are prone to condensation problems and must be protected against internal corrosion and freezing.



**Figure 7**

Air is usually compressed to 758 to 1723 kPa (110 to 250 psi) and is stored in storage tanks. Stored air is regulated to from 620 kPa (90 psi) to a maximum of 1550 kPa (225

psi) depending on the engine and starter and piped to the air motor. Consult the Owners manual for the recommended regulation pressure for the engine model you are using. A check valve between the compressor and the air receiver is good practice, to protect against a failure of plant air that might deplete the air receivers' supply.

The air compressors that supply pneumatic starters are driven by external power sources such as electric motors and diesel or gasoline engines. A small emergency receiver (not supplied by Caterpillar) can be hand pumped to starting pressure under emergency conditions. A more common emergency backup will include an auxiliary diesel engine-driven air compressor package.

### Air Tank

Air tanks are required to meet specific characteristics, such as the specifications of the American Society of Mechanical Engineers (ASME). Compressed air storage tanks must be equipped with a maximum pressure valve and a pressure gauge. Check the maximum pressure valve and pressure gauge often to confirm proper operation. A drain cock must be provided in the lowest part of the air receiver tank for draining condensation.

### Air Storage Tank Sizing

Many applications require sizing air storage tanks to provide a specified number of starts without recharging.

This is accomplished using the following formula:

$$V_T = \frac{V_S \times T \times P_A}{P_1 - P_{MIN}}$$

Where:

$V_T$  = Air storage tank capacity  
(cubic feet or cubic meters)

$V_S$  = Air consumption of the starter  
motor ( $m^3/sec$  or  $ft^3/sec$ ) —

Air Starting Requirements are in  
the TMI for model engine used.

If prelube is used its consumption  
must be added to  $V_S$  also.

$T$  = Total cranking time required  
(seconds): If six consecutive  
starts are required, use seven  
seconds for first start (while  
engine is cold), and two  
seconds each for remaining  
five starts, or a total cranking  
time of seventeen seconds.

$P_A$  = Atmospheric pressure (psi or  
kPa): Normally, atmospheric  
pressure is 101 kPa (14.7  
psi).

$P_T$  = Air storage tank pressure (psi  
or kPa): This is the storage  
tank pressure at the start of  
cranking.

$P_{MIN}$  = Minimum air storage tank  
pressure required to sustain  
cranking at 100 rpm (psi or  
kPa) — Air Starting  
Requirements are in TMI for  
model used.

**Example:**

A 3516 Diesel Engine with electric  
prelube has the following:

Maximum air tank pressure =  
1241 kPag (180 psig)

Minimum air to starter pressure =  
620.5 kPag (90 psig)

Expected air line pressure drop =  
207 kPag (30 psig)

Six consecutive starts. First start  
= 7 seconds the other 5 starts = 2  
seconds

Average barometric pressure at  
this location = 100 kPa (14.5 psi)

Preconditioned engine installation.  
( $cfm \times 0.02832 = m^3/min$ )

**Solution:**

$V_S = 0.40 m^3/sec$  (14.1  $ft^3/sec$ )

$T = 7 + (5 \times 2) = 17 sec$

$P_A = 100 kPa$  (14.5 psi)

$P_T = 1241 - 207 = 1034 kPag$   
(180 - 30 = 150 psig)

$P_{MIN} = 620.5 kPag$  (90 psig)

Therefore:

$$V_T = \frac{0.4 \times 17 \times 100}{1034 - 620.5} = 1.64 m^3$$

$$V_T = \frac{14.1 \times 17 \times 14.5}{150 - 190} = 57.93 ft^3$$

## Air Starting Motor

### Cranking Time Required

The cranking time depends on the  
engine model, engine condition,  
ambient air temperature, oil  
viscosity, fuel type, and design  
cranking speed. Five to seven  
seconds is typical for an engine at  
26.7°C (80°F). Restarting hot  
engines usually requires less than  
two seconds. Most marine societies  
require a minimum of six  
consecutive starts for propulsion

## Combustion Air Flow Requirements

Combustion air flow requirements will vary, depending on the specific engine model and rating. Specific air flow data for Caterpillar engines is given in both volumetric [ $\text{m}^3/\text{min}$  (cfm)] and mass [kg/hr (lb/hr)] flow terms, at standard reference conditions.

Reference conditions for temperature and pressure are used to provide a basis for consistent measure of combustion air quantities. However, different parts of the world subscribe to different standards, thus it is important to note that the metric and English conditions are not equivalent. Caterpillar practice is to use ISO "normal" conditions of  $0^\circ\text{C}$  ( $32^\circ\text{F}$ ) and 101.3 kPaa (14.7 psia) when providing values in metric units and ASME "standard" conditions of  $25^\circ\text{C}$  ( $77^\circ\text{F}$ ) and 101.3 kPaa (14.7 psia) when providing values in English units.

To convert from mass airflow to volumetric airflow at reference conditions, use the following formula:

$$\frac{M_R}{S_R} = Q_R$$

Where:

$M_R$  = Mass air flow at reference conditions (kg/hr), (lb/hr)

$Q_R$  = Volumetric air flow at reference conditions ( $\text{m}^3/\text{min}$ ), (cfm)

$S_R$  = Density of air at reference conditions ( $\text{kg}/\text{m}^3$ ), ( $\text{lb}/\text{ft}^3$ ).  
(Density of air =  $1.292 \text{ kg}/\text{Nm}^3$  ( $0.074 \text{ lb}/\text{ft}^3$ ))

To convert both mass airflow and volumetric air flow from reference conditions to site conditions, use the following formulas:

$$M_R \times \frac{T_S}{T_R} = M_S$$

$$Q_R \times \frac{T_S}{T_R} = Q_S$$

Where:

$M_R$  = Mass flow at reference conditions (kg/hr), (lb/hr).

$M_S$  = Mass flow at site conditions (kg/hr), (lb/hr).

$Q_R$  = Air flow at reference conditions ( $\text{m}^3/\text{min}$ ), (cfm).

$Q_S$  = Air flow at site conditions ( $\text{m}^3/\text{min}$ ), (cfm).

$T_R$  = Air temperature at reference conditions ( $^\circ\text{K}$ ), ( $^\circ\text{R}$ ).

$T_S$  = Air temperature at site conditions ( $^\circ\text{K}$ ), ( $^\circ\text{R}$ ).

$^\circ\text{K} = ^\circ\text{C} + 273.$

$^\circ\text{R} = ^\circ\text{F} + 460.$

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## Air Intake Ducting

### General

When ducting is necessary to obtain cooler or cleaner air, the filters should remain on the engine to prevent harmful dirt from leaking into the engine through ducting joints. When air cleaners must be remote-mounted it is extremely important that all joints be air tight to prevent ingestion of dirt.

When designing air intake ducting, consideration must be given to appropriate routing, duct support and system restriction, especially on the larger engines, where overhead cranes are used to service the engines. Proper support for duct work adjacent to the engine is critical, so that its weight is not borne by the turbocharger or other engine-mounted components.

Locate the air piping away from the vicinity of the exhaust piping so that the air provided to the engine is as cool as possible. Air temperature to the air inlet should be no more than 11°C (20°F) above ambient air temperature. Inlet air temperature should not exceed 45°C (113°F) for standard ratings.

Avoid abrupt transitions in the intake ducting to provide the smoothest possible air flow path. When unavoidable, transitions should be made as far upstream of the turbocharger as possible. Keep total duct head loss (restriction) below 0.5 kPa (2 in. H<sub>2</sub>O) for maximum filter life. Any additional restriction will reduce filter life.

To allow for minor misalignment due to manufacturing tolerances, engine-to-enclosure relative movement and isolate vibrations, segments of the piping should consist of flexible rubber fittings. These are designed for use on diesel engine air intake systems and are commercially available. These fittings include hump hose connectors and reducers, rubber elbows and a variety of special shapes.

Wire-reinforced flexible hose should not be used. Most material available is susceptible to damage from abrasion and abuse and is very difficult to seal effectively at the clamping points unless special ends are provided on the hose.

Inlet ducting should be designed to withstand a minimum vacuum of 12.5 kPa, (50 in. H<sub>2</sub>O) which is also the structural capability of the Caterpillar air cleaner filter element.

Piping diameter should be equal to or larger than the air cleaner inlet/outlet and the engine air inlet. A rough guide for pipe size selection is to keep maximum air velocity in the piping to 10 m/s (2,000 fpm). Higher velocities will cause high noise levels and excessive flow restrictions. Refer to the Air Intake Restriction section for guidance in determining required intake duct sizing.

All piping must be designed and supported to meet any local seismic requirements that may be in force.

The ducting should be of seamless or welded seam piping to minimize the flow restriction. The ducting should also be constructed of materials suitable for local environmental conditions such as offshore or marine applications.

Beaded pipe ends at hose joints are recommended. Sealing surfaces should be round, smooth and free of burrs or sharp edges that can cut the hose. The tubing should have sufficient strength to withstand hose clamping forces. Either T-bolt type or SAE type F hose clamps that provide a 360° seal should be used. High quality clamps must be used. Double clamps are recommended on connections downstream of the air cleaner.

PVC piping has a number of benefits. It is light-weight, provides a good seal without the chance of weld slag coming loose and will not rust. However, it is not well suited for high or low temperature environments. It can lose much of its strength when subjected to temperatures of 150C (300F) or above. It can also become brittle and shatter at low temperatures. If using

PVC pipe, use at least schedule 40 pipe and check that it meets local regulations for the area classification.

If ferrous material is necessary, it must be properly cleaned after fabrication and treated to prevent rust and scale from accumulating. Stainless steel ducting should be treated in the same manner. Flanged connections with gaskets are preferred over threaded connections. Fasteners such as rivets should not be used.

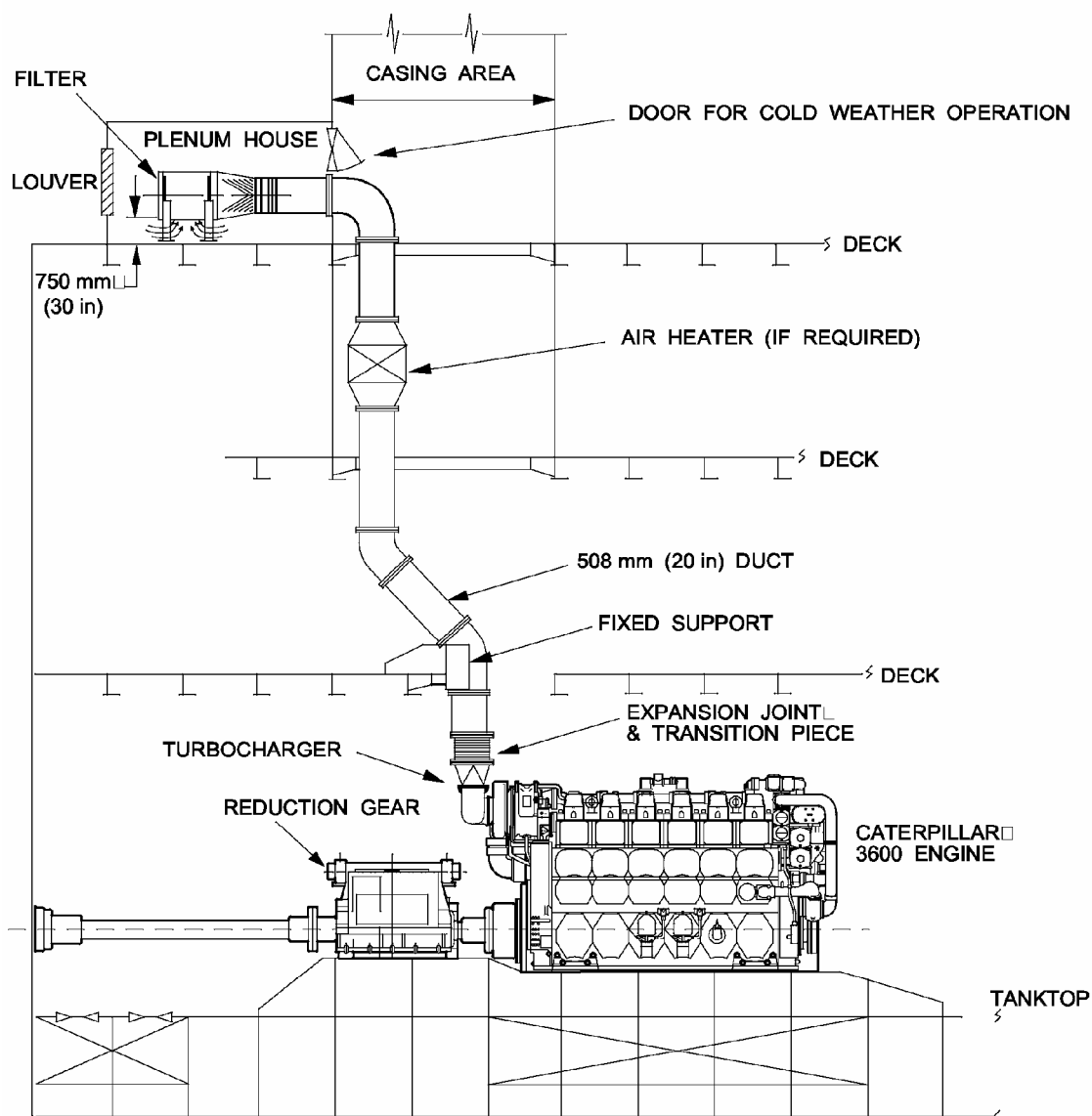
Unsupported weight on clamp-type joints should not exceed 1.3 kg (3 lb).

## **Marine Intake Air Piping Examples**

**Figure 4** shows a 3600 marine application configured to use a remote mounted air cleaner and outside air for combustion. An intake air heater may be required for cold weather operation.

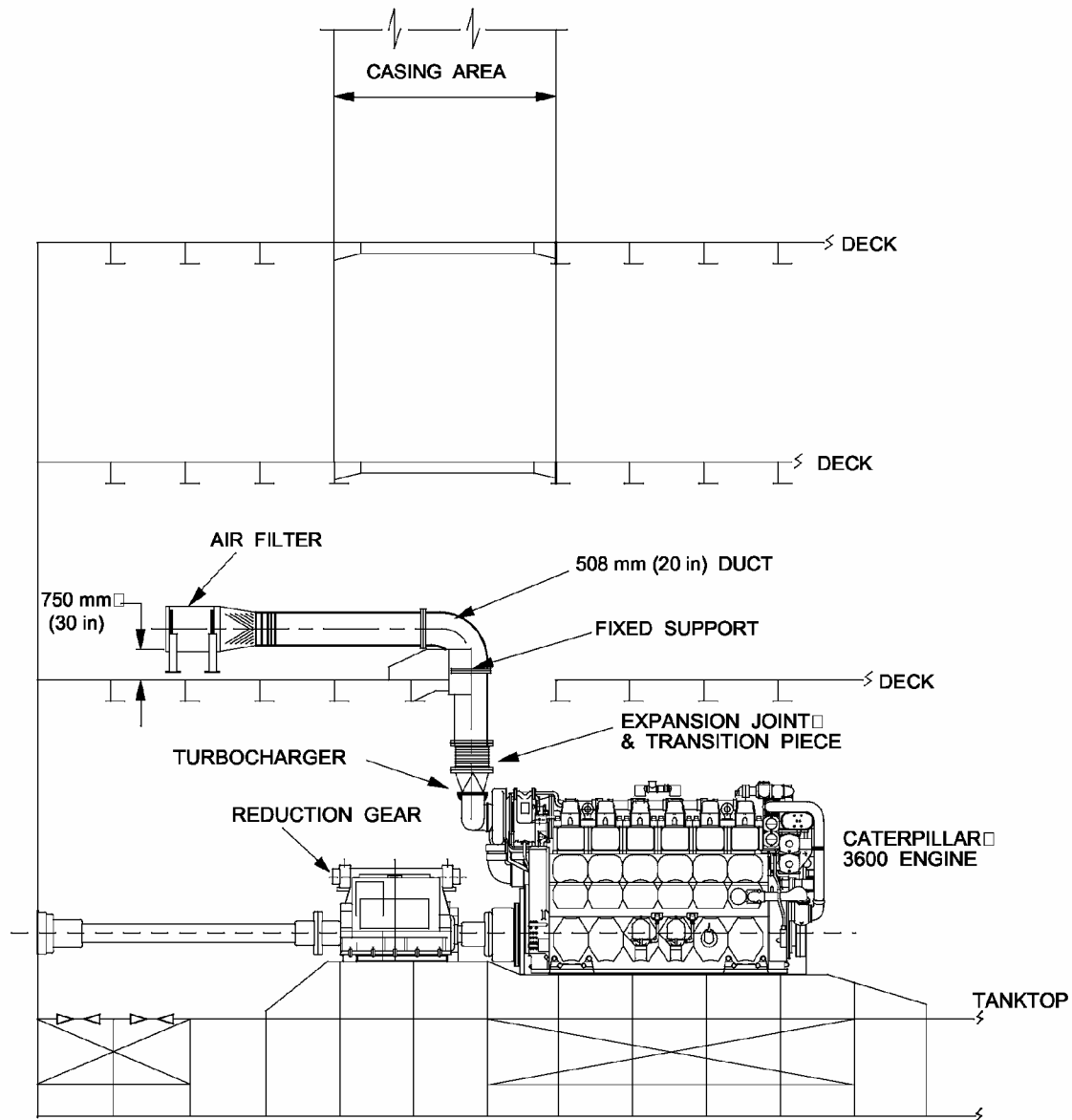
**Figure 5** shows a 3600 marine application configured to use a remote mounted air cleaner and engine room air for combustion.

### Remote Mounted Air Cleaner with Outside Air Intake



**Figure 4**

### Remote Mounted Air Cleaner with Engine Room Air Intake



**Figure 5**

### Air Inlet Adapters

Caterpillar offers various air inlet adapters for connecting to turbocharger air inlets. The adapters are part of the system to provide an efficient transition from the engine room intake air ducting to the engine turbocharger. Adapters are typically

shipped loose and include gaskets and mounting hardware.

**CAUTION:** Turbocharger performance may be adversely affected if appropriate air intake components are not used (They are designed to provide the proper air

flow pattern ahead of the turbocharger).

### **Connections to Inlet Adapters and Turbochargers**

The piping connected to the turbocharger inlet should be designed to ensure that air is flowing in a straight, uniform direction into the turbocharger compressor. This is typically achieved by installing a straight section of pipe, equal in length to at least two or three times pipe diameter, to the inlet. This arrangement reduces the possibility of premature compressor wheel failure due to pulsations created by air striking the compressor wheel at an angle. Transitional ducting

immediately preceding the straight section of pipe should consider the following guidelines:

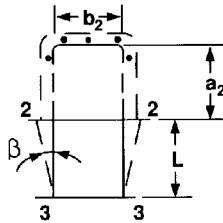
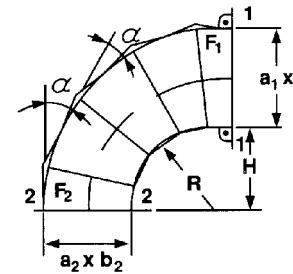
- The duct between the straight pipe and elbow cannot have protruding edges.
- The bend can be designed as a circular arc or with sections of mitered pipe with rectangular or round flow cross sections, or as transition from round to rectangular cross section.
- An accelerated flow is expected to occur in the bend. The flow area (F) should be:  $F_1 > 1.5 \times F_2$ , as shown in **Figure 6**.



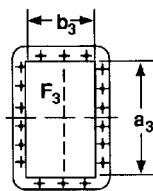
### Turbocharger Vertical Inlet Design Options

(a) Rectangular

$$Dh_1 = \frac{2 \times a_1 \times b}{a_1 + b_1}$$

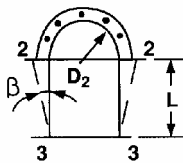
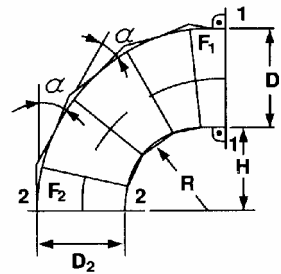


Intake Elbow Flange



(b) From Round to Rectangular

$$Dh_2 = D$$



F = flow area

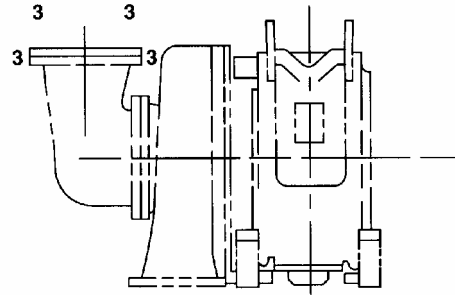
$R \geq 1.0 \cdot D_{h1}$

$H \geq 1.0 \cdot D_{h1}$

$L \geq 2.0 \cdot D_{h2}$

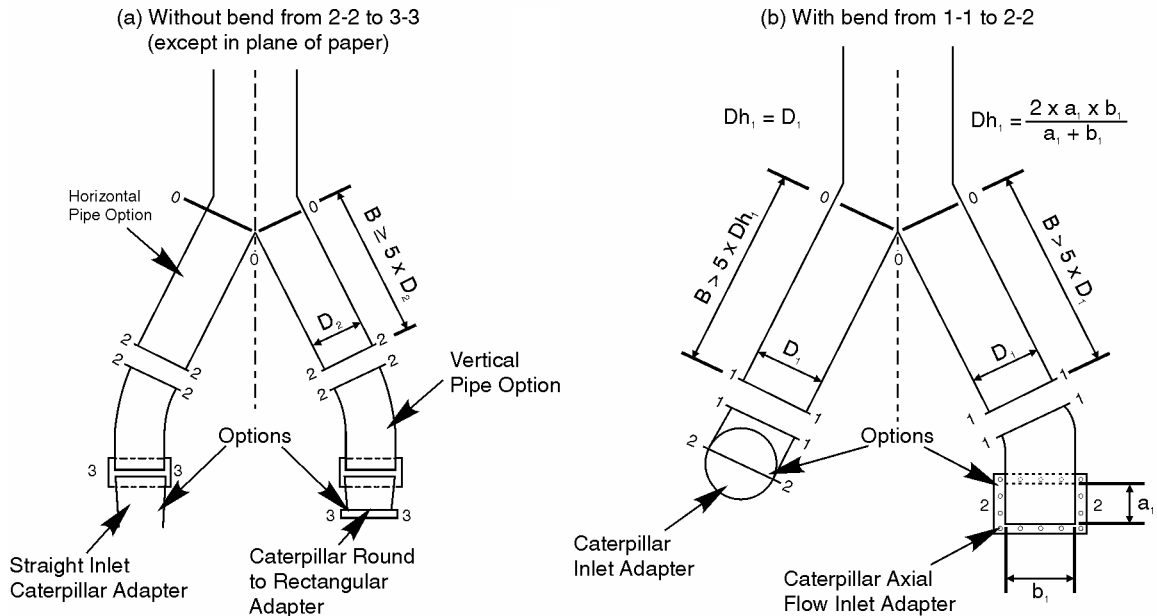
$\alpha \leq 22.5^\circ$

$\beta \leq 10^\circ$



**Figure 6**

### Inlet Pipe Design Joining Two Turbochargers



**Figure 7**

### Joining Two Turbochargers

When ductwork feeding two turbochargers is combined to form a single duct, a steadying zone must be provided after the dividing joint; as shown in **Figure 7**. The steadying zone B must be a minimum of 5 times the pipe diameter:

$$B > 5 \times Dh_1$$

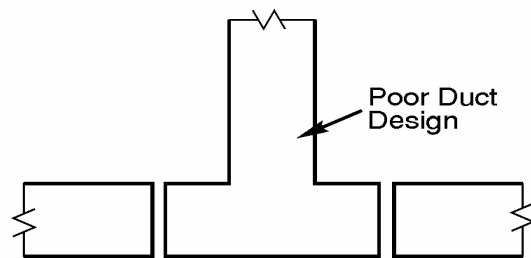
The flow area is:

$$F_0 = 1.0 \div 2.0 \times F_1$$

The transitions from Sections 0-0 to 1-1 and from 1-1 to 2-2 will have many circular or rectangular variations due to turbocharger hardware and installation site design. Regardless of the transition

selected, the steadying zone must be provided.

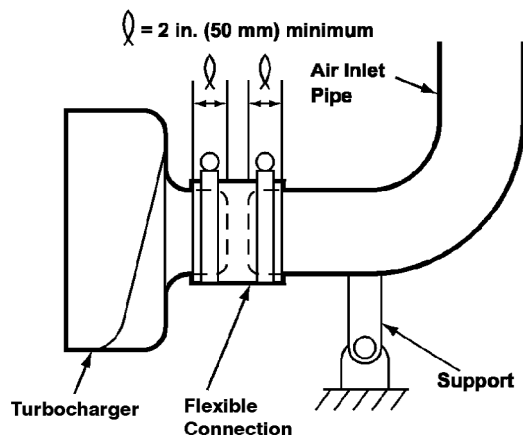
The transitions used to combine multiple ducts must also follow typical design standards described in this guide. Ducts should have smooth transitions and not cause disturbance in the air flow. Piping designs that use a Tee, as shown in **Figure 8**, should not be used to connect multiple ducts.



**Figure 8**

## Turbocharger Loading

When remote-mounted air cleaners are used, turbocharger loading from the weight of the air inlet components becomes a concern. The turbochargers are not designed to support any additional weight beyond standard factory attachments. When possible, make the flexible connection directly to the turbocharger air inlet, as shown in **Figure 9**. All duct work to that point must be supported off-engine.

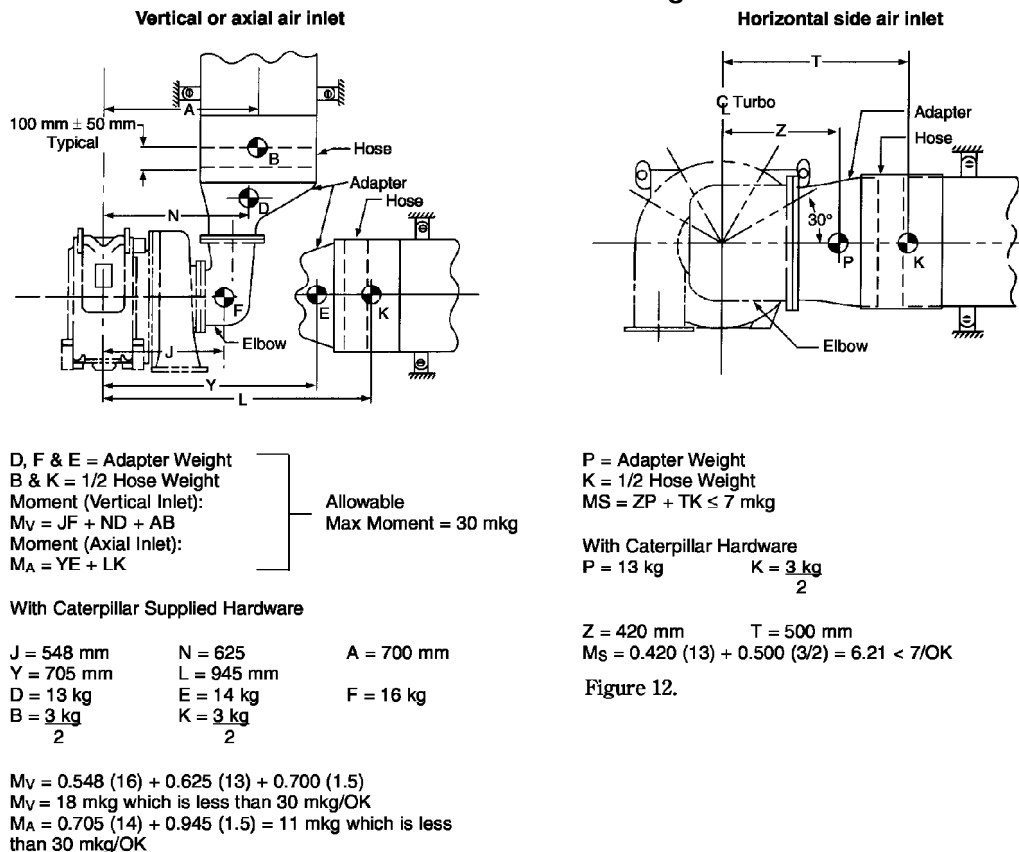


**Figure 9**

The maximum allowable turbocharger load will vary, depending on the engine model, the inlet adapter and the adapter orientation. In the example below, **Figure 10**, the 90° inlet adapter can be rotated in 30° increments. Turbochargers for 3600/G3600 engines are designed to withstand a maximum moment of 294 N•m (217 ft-lb). **Figure 10** shows how the moment can be calculated.

Models that provide a mounting bracket for the turbocharger inlet adapter, such as the C15, can support up to 11.3 kg (25 lb) of duct weight.

## Maximum Loads for Turbocharger Intake



**Figure 10**

## Flex Connections

Flexible connections are required to isolate engine vibration and noise from the ducting system. The connections should be configured for maximum allowable offset and compression to prevent early failure and excess forces on the turbocharger and air inlet components. The flexible connection should be as close to the engine as practical and installed in such a way to not induce stress on the ducting system. The flex engagement with the air intake duct should be a

minimum of 50 mm (2 in.) and a maximum of 200 mm (8 in.). Care must be used to prevent exhaust piping heat from deteriorating rubber flex connections.

## Cleanliness During Installation

The air intake ducting must be cleaned of all debris. Fabricated ducting, utilizing fasteners such as rivets, should not be used. The ducting should be made of material such that prolonged operation will not result in debris coming loose and entering the turbocharger.

An identifiable blanking plate should be installed ahead of the turbocharger to prevent debris from entering during initial installation of the unit. The plate should have a warning tag indicating it has to be removed prior to starting the engine. The Caterpillar supplied shipping cover can be used for this purpose.

Provisions should be made to inspect the ducting for cleanliness just prior to initial start up. If the piping is not clean, it must be cleaned before the engine is started at commissioning. This may require removal of the piping from its installed position.

### Inlet Air Duct Insulation

Insulation may be needed on the intake ducting for remote mounted air cleaners. Insulation reduces turbocharger noise emitted into the engine room and will minimize pre-heating of intake air.

### Air Intake Restriction

Excessive vacuum on the inlet side of the turbocharger (or the air inlet on naturally aspirated engines) can result in reduced engine power capability and degrade engine performance.

Air intake restriction is also an emissions critical parameter declared to obtain EPA non-road certification. Therefore, the air intake system's total restriction (including dirty filters, duct work, vents, silencers, etc.) is limited depending on engine model, rating and air configuration. The air intake restriction limits for Caterpillar engines can be found in

the Technical Information Appendix or TMI.

In order to maximize air filter life, it is important to keep total duct restriction below 0.5 kPa (2 in. H<sub>2</sub>O). Every additional restriction caused by the air inlet system subtracts from air filter life. Maximum filter life is partially dependent on the absolute pressure differential between the turbocharger compressor inlet and atmosphere.

Inlet air restriction includes the pressure losses between the air cleaner and the engine air inlet connection. For remote mounted air cleaners, the following formulas can be used to calculate duct restriction.

$$P(\text{kPa}) = \frac{L \times S \times Q^2 \times 3.6 \times 10^6}{D^5}$$

$$P(\text{in. H}_2\text{O}) = \frac{L \times S \times Q^2}{187 \times D^5}$$

Where:

P = Restriction (kPa), (in. H<sub>2</sub>O)

psi = 0.0361 x in. water column

kPa = 6.3246 x mm water column

L = Total equivalent length of pipe, measured in (m), (ft)

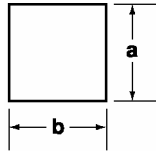
Q = Inlet air flow, measured in (m<sup>3</sup>/min), (cfm). - (found in TMI or performance book, and corrected for site conditions when necessary)

D = Inside diameter of pipe, measured in (mm), (inches)

If the duct is rectangular, as shown in **Figure 11**:

Then:

$$D = \frac{2 \times a \times b}{a + b}$$



**Figure 11**

$$S = \text{Density of air kg/m}^3 \text{ (lb/ft}^3\text{)}$$

$$S(\text{kg/m}^3) = \frac{352.5}{\text{Air Temperature} + 273^\circ\text{C}}$$

$$S(\text{lb/ft}^3) = \frac{39.6}{\text{Air Temperature} + 460^\circ\text{F}}$$

Use the following formulas to obtain equivalent lengths of straight pipe for various elbows.

$$\text{Standard Elbow (radius = diameter)} \quad L = \frac{33D}{X}$$

$$\text{Long Radius Elbow (radius = 1.5 diameter)} \quad L = \frac{20D}{X}$$

$$45^\circ \text{ Elbow (radius = 1.5 diameter)} \quad L = \frac{15D}{X}$$

$$\text{Square Elbow (radius = 1.5 diameter)} \quad L = \frac{66D}{X}$$

Where:

$$x = 1000 \text{ mm (12 in.)}$$

As shown above, if 90° bends are required, long radius elbows, with a radius of 1.5 times the pipe diameter, offer lower resistance than standard elbows.

### Example

Below is an example of an air intake duct restriction calculation.

A 3412 packaged genset has an inlet air flow of 36.7 m<sup>3</sup>/min (1292 cfm) with duct configuration consisting of 3 m (10 ft) of straight length duct along with 2 standard elbows and a long radius elbow. The pipe has a diameter of 152.4 mm (6 in) and the temperature of the air is 55°C (131°F).

First calculate the total equivalent length of the ducting.

$$L = \frac{3 + 2 \times (33 \times 152.4)}{1000 \text{ mm}} + \frac{20 \times 152.4}{1000 \text{ mm}}$$

$$L = 16.1 \text{ m}$$

$$L = \frac{10 + 2 \times (33 \times 6)}{12 \text{ in.}} + \frac{20 \times 6}{12 \text{ in.}}$$

$$L = 53 \text{ ft}$$

Next, calculate the density of the air.

$$S = \frac{352.5}{55 + 273^\circ\text{C}}$$

$$S = 1.075 \text{ kg/m}^3$$

$$S = \frac{39.6}{131 + 460^\circ\text{F}}$$

$$S = 0.067 \text{ lb/ft}^3$$

Lastly, insert the previous results into the duct restriction formula and calculate.

$$P = \frac{16.1 \times 1.075 \times 36.72 \times 3598805.2}{152^5}$$

$$P = 1.02 \text{ kPa or } 104 \text{ mm H}_2\text{O}$$

$$P = \frac{53 \times 0.067 \times 12922}{5184 \times 6^5}$$

$$P = 0.147 \text{ psi or } 4.07 \text{ in H}_2\text{O}$$

Total duct restriction should be below 0.5 kPa (2 in. H<sub>2</sub>O). The duct restriction in this example is above the desired value, therefore this duct configuration is unacceptable.

~~flow slows the turbocharger to avoid overspeed and excessive boost pressure.~~

~~On some natural gas engines, the wastegate may be manually adjusted for site conditions to optimize the throttle position for efficiency or improved response.~~

~~**Note:** The exhaust flow by passing the two turbochargers of a G3600 vee engine via the wastegate are plumbed together and exit on one exhaust outlet. Therefore, if measuring exhaust flow, you will notice an uneven exhaust gas flow through the two exhaust outlets when the wastegate is open. Typically, the right side flow will be 15% higher than the left side.~~

~~**CAUTION:** Tampering with the boost line to the wastegate will raise aftercooler heat rejection, increase turbocharger speed and peak engine cylinder pressure. This will negatively affect engine reliability, durability, stability, emissions and overall performance.~~

## Flexible Exhaust Connections

The exhaust piping system must be isolated from the engine with flexible connections, designed for zero leakage and flexible in all directions. Two types of flexible connections are normally used a flexible metal hose type and a bellows type.

## Flexible Metal Hose and Bellows

Flexible metal hose is commonly used for exhaust systems with a diameter of 150 mm (6 in) and smaller. Bellows are typically used for exhaust systems with a diameter of 200 mm (8 in) and larger.

Flexible connections should be installed as close as possible to the engine exhaust outlet. A flexible exhaust connection has three primary functions.

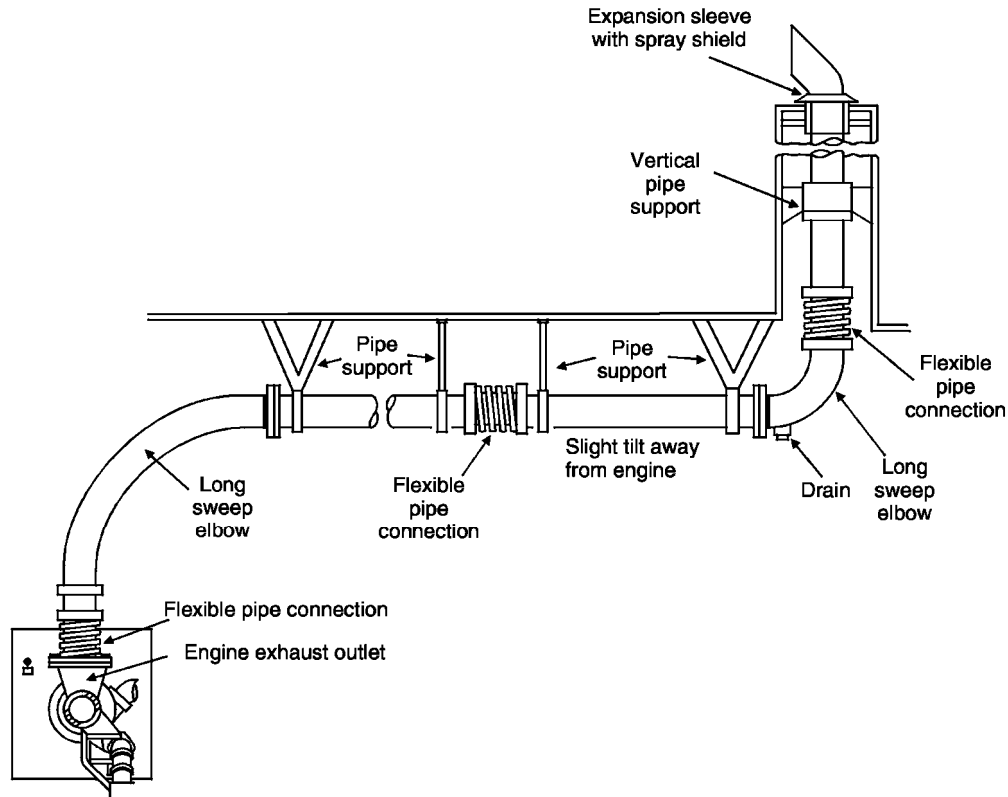
- To isolate the weight of the exhaust piping from the engine. The amount of weight which the exhaust outlet for each engine model can withstand varies.
- To relieve exhaust components of excessive vibrational fatigue stresses.
- To allow relative shifting of exhaust components. This has numerous causes. It may result from expansion and contraction due to temperature changes, by creep processes that take place throughout the life of any structure, or by torque reactions.

A typical piping layout with flexible connections is shown in **Figure 3**.

Flexible pipe connections, when insulated, must expand and contract freely within the insulation. This generally requires a soft material or insulated sleeve to encase the connection.



### Typical Exhaust Piping With Flexible Connection



**Figure 3**

The flexible connections should be pre-stretched during installation to allow for expected thermal growth. Four small straps can be tack-welded between the two end flanges to hold the engine exhaust flexible connections or bellows in a rigid position during exhaust piping installation. This will prevent the bellows from being installed in a flexed condition. Attach a warning tag to the bellows noting that the weld straps must be removed prior to starting the engine.

Any flexible connector must have good fatigue resistance. It should give acceptable service life while withstanding vibratory stress and it

should be soft enough to prevent transmission of vibration beyond the connection.

The installation limitations of Caterpillar supplied flexible exhaust fittings are shown in the table below. For maximum durability, allow the bellows to operate as close as possible to its free state.

#### Slip Joints

Slip joints are another method of handling the expansion and contraction of exhaust systems. Slip joints are designed to have controlled leakage when the system is cold. When the engine starts and the exhaust pipes warm up, the joints will expand and make a gas-

tight fit. The slip joints are flexible in only one direction and require good support on each side.

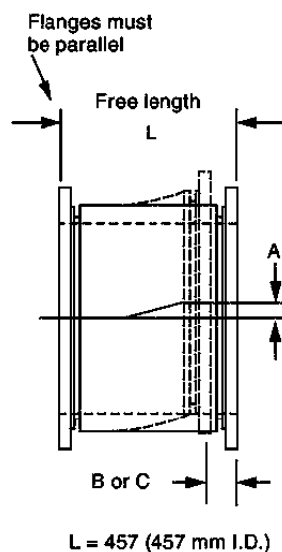
However, Caterpillar does not normally recommend the use of slip

joints due to disadvantages such as leaking exhaust fumes, exhaust slobber and the inability of the joint to flex in more than one direction.

Installation Limits of Flexible Exhaust Fittings – Flexible Metal Hose-Type						
Hose Diameter	A Maximum Offset Between Flanges		B Maximum Compression From Free Length		C Maximum Extension From Free Length	
	mm	in.	mm	in.	mm	in.
4 & 5 in.	25.4	1.0	6.25	.25	6.25	2.5
6 in.	38.1	1.5	6.25	.25	6.25	2.5

Installation Limits of Flexible Exhaust Fittings - Bellows-Type						
Hose Diameter	A Maximum Offset Between Flanges		B Maximum Compression From Free Length		C Maximum Extension From Free Length	
	mm	in.	mm	in.	mm	in.
8 & 12 in.	19.05	0.75	38.1	1.50	25.40	1.00
14 in.	19.05	0.75	76.2	3.00	25.40	1.00
18 in.	22.86	0.90	76.2	3.00	44.45	1.75

Spring Rate for Flexible Fittings - Bellows-Type		
Diameter	Spring Rate	
	kN/m	lb/in.
8 in.	29.7	170
12 in.	33.9	194
14 in.	68.5	391
18 in.	19.3	110



**Figure 4**

## Exhaust System Piping

The function of the exhaust piping is to convey the exhaust gases from the engine exhaust outlet to the silencer and other exhaust system components, terminating at the system outlet. Piping is a key feature in overall exhaust system layout.

### Exhaust System Design

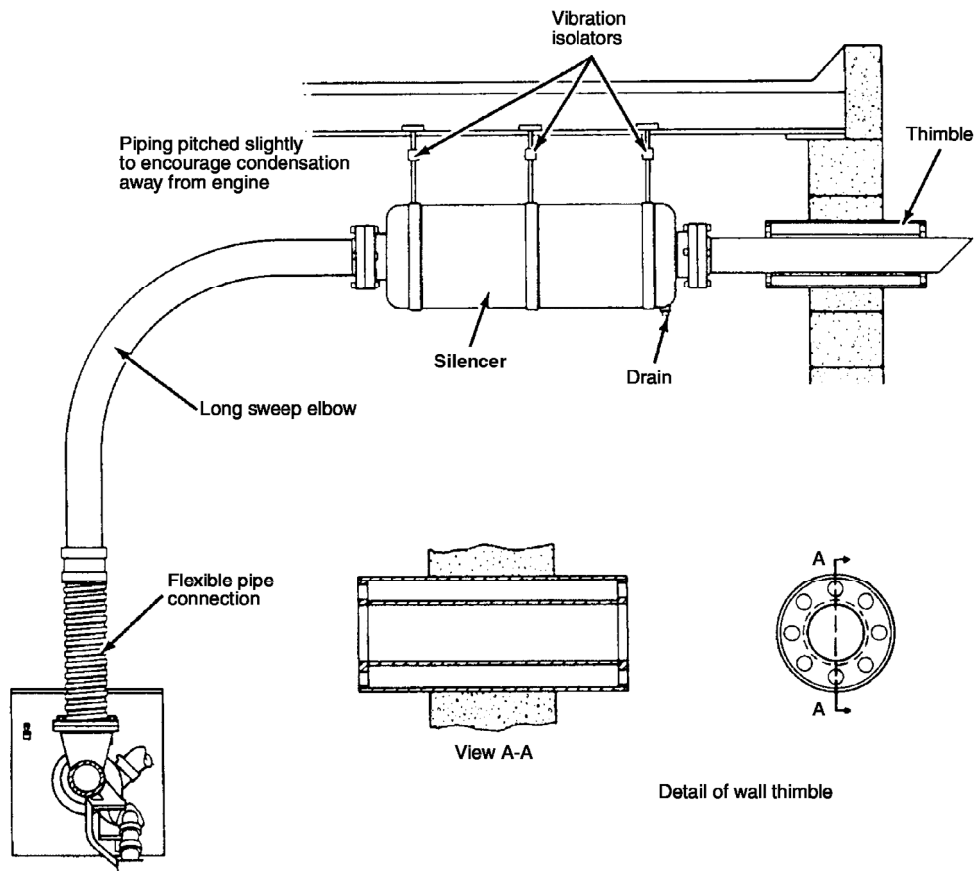
The physical characteristics of the engine room or engine bay will determine the exhaust system layout. Exhaust piping should be designed to minimize the exhaust backpressure while keeping engine serviceability in mind. The exhaust

piping should be securely supported. Allowances should be made for system movement and vibration isolation by using suitable flexible components such as rubber dampers or springs.

Piping must be designed with engine service in mind. In many cases, an overhead crane will be used to service the heavier engine components on the larger engines.

The following recommendations should be followed when designing an exhaust piping system.

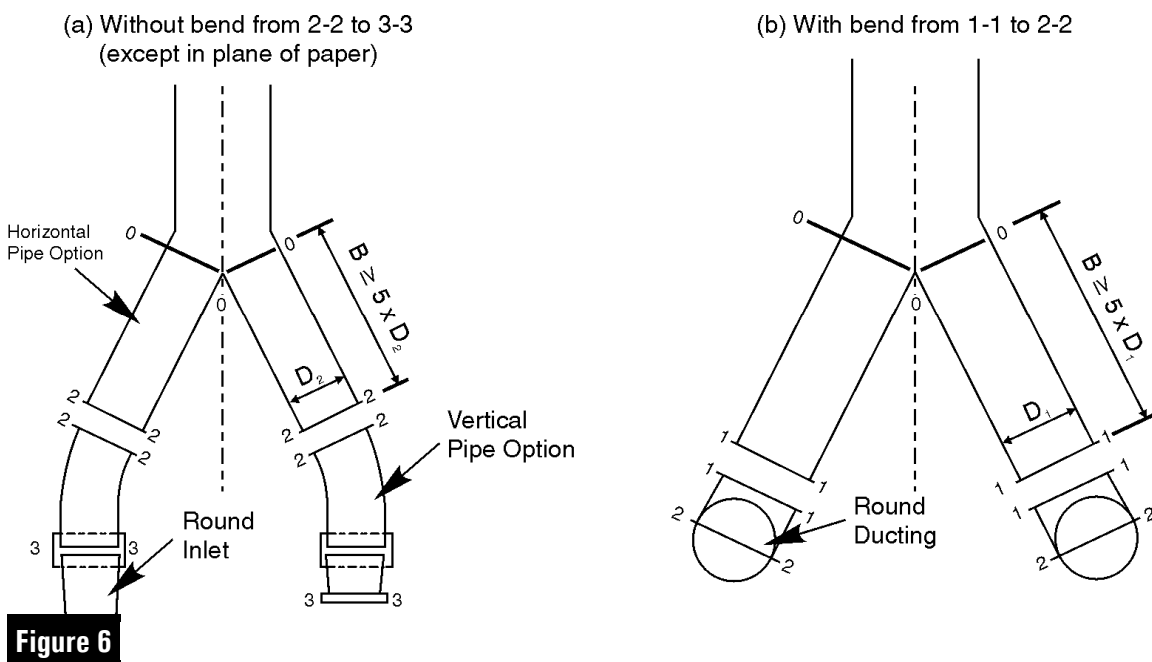
### Typical Silencer Details



**Figure 5**

- All piping should be installed with a minimum clearance of 229 mm (9 in) from combustible materials.
- The exhaust piping must be properly supported, especially adjacent to the engine, so that its weight is not borne by the engine or the turbocharger. This is discussed in more detail later in this section.
- Exhaust piping should be sized according to the maximum backpressure limit for the engine.
- Where appropriate, heat radiation may be reduced by covering the off-engine exhaust piping with suitable, high temperature insulation.
- Install metal thimble guards for exhaust piping passing through wooden walls or roofs. The thimble guards should be 305 mm (12 in) greater in diameter than the exhaust pipes, see **Figure 5**.
- When exhaust stacks are used, extend them upward and away from the engine room to avoid heat, fumes and odors.
- Locate the exhaust pipe outlets away from the air intake system. Engine air cleaners, turbochargers and aftercoolers contaminated with exhaust products can induce premature failures.
- Avoid routing exhaust piping close to fuel pumps, fuel lines, fuel filters, fuel tanks and other combustible materials.
- Exhaust pipe outlets cut at 30° to 45° angles, rather than 90° angles, will reduce exhaust gas turbulence and noise. Refer to **Figure 6**.
- Exhaust outlets should be arranged to keep water from entering the piping system. Rain caps forced open by exhaust pressure will accomplish this; however, they will also introduce additional backpressure into the system and should be carefully evaluated.

### Pipe Design Joining Two Turbochargers



The exhaust system can accumulate a considerable amount of condensed moisture. For example, gas engines burning natural gas create one pound of water for each 10 ft<sup>3</sup> of natural gas burned. For this reason, long runs of exhaust piping require traps to drain moisture. Traps should be installed at the lowest point of the line, near the exhaust outlet, to prevent rain water from reaching the engine. Exhaust lines should be sloped away from engine, toward the trap, so condensation will properly drain, see **Figure 3** in the previous section.

#### Exhaust Thimbles

Exhaust thimbles, as illustrated in **Figure 5**, are fabrications used for wall or ceiling penetrations. The thimble provides a separation of the exhaust pipe from walls or ceilings, in order to provide mechanical and thermal isolation. Single sleeve

thimbles must have diameters at least 305 mm (12 in) larger than the exhaust pipe. Double thimbles (inner and outer sleeve) should have outside diameters at least 152 mm (6 in) larger than the exhaust pipe.

#### Exhaust Pipe Insulation

No exposed parts of the exhaust system should be near wood or other combustible material. Exhaust piping inside the engine room or machinery space (and the silencer, if mounted inside) should be covered with suitable insulation materials to protect personnel and to reduce room temperature. A sufficient layer of suitable insulating material surrounding the piping and silencer and retained by a stainless steel or aluminum sheath may substantially reduce heat radiation to the room from the exhaust system. An additional benefit of the insulation is

that it provides sound attenuation to reduce noise in the room.

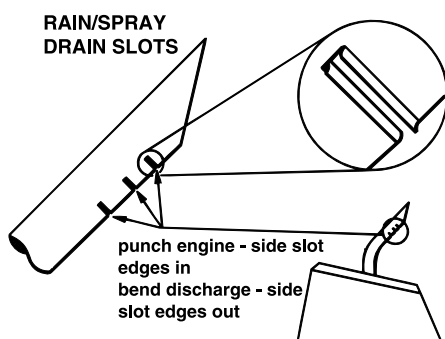
### Water Ingress Prevention

Exhaust system outlets must be provided with an appropriate means of preventing snow, rainwater or sea spray from entering the engine through the exhaust piping. This can be accomplished by several methods, but must be given careful consideration. The selected method can impose significant restrictions that must be taken into account when calculating system backpressure.

One simple method, used primarily with horizontal exhaust pipes, is to angle cut the end of the pipe as shown in **Figure 3**, **Figure 5** and **Figure 7**.

A common method used with vertical exhaust pipes is to angle the pipe at 45° or 90° from vertical using an appropriate elbow, then angle cutting the pipe end as previously described.

Another feature that may be used in conjunction with either of the above methods are Rain/Spray Slots as shown in **Figure 6**.



**Figure 7**

Slots are cut into the exhaust pipe to allow rain/spray to drain harmlessly. The edges of each slot are deformed as shown in the previous graphic. The engine side of the slot is bent inward and the downstream side of the slot is bent outward. No more than a 60° arc of the pipe circumference should be slotted in this way.

For applications where none of the above methods is possible, it may be necessary to fit some form of rain cap to the end of the vertical pipe section. This method can provide a positive means of water ingress prevention, but not without imposing a significant backpressure restriction.

### Exhaust System Backpressure

Excessive exhaust restriction can adversely affect performance, resulting in reduced power and increased fuel consumption, exhaust temperatures and emissions. It will also reduce exhaust valve and turbocharger life. It is imperative that exhaust backpressure is kept within specified limits for those engines subject to emissions legislation. To ensure compliance, exhaust system backpressure must be verified to be within the Caterpillar EPA declared maximum value for the engine configuration and rating. Values can be found in the "Systems Data" listed in the Caterpillar Technical Marketing Information (TMI) system.

Backpressure includes restrictions due to pipe size, silencer, system configuration, rain cap and other exhaust-related components.

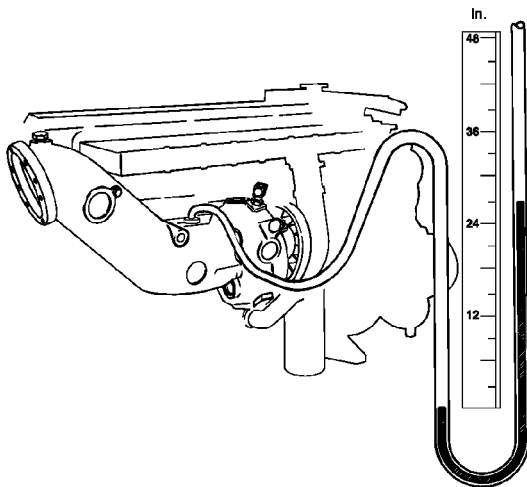
Excessive backpressure is commonly caused by one or more of the following factors:

- Exhaust pipe diameter too small.
- Excessive number of sharp bends in the system.
- Exhaust pipe too long.
- Silencer resistance too high.

Engines with a vee cylinder configuration should be designed so the exhaust piping gives equal backpressure to each bank.

### Measuring Backpressure

Exhaust backpressure is measured as the engine is operating under full rated load and speed conditions. Either a water manometer or a gauge measuring inches of water may be used. Refer to **Figure 8**.



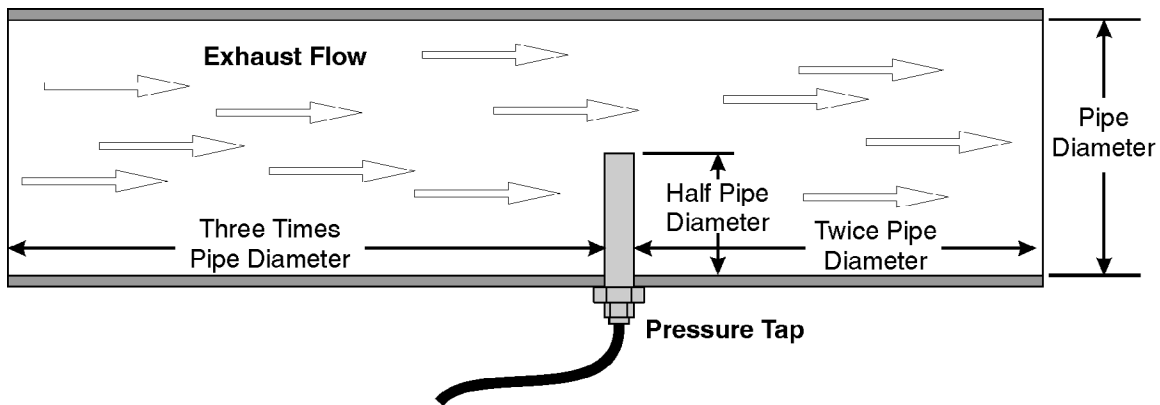
**Figure 8**

Many engine installations are already equipped with a fitting in the exhaust discharge for measuring backpressure. If the system is not equipped with such a fitting, use the following guidelines, **Figure 9** and **Figure 10** to locate and install a pressure tap.

- Locate the pressure tap in a straight length of exhaust pipe as close to the turbocharger as possible.
- Locate the tap three pipe diameters from any upstream pipe transition.
- Locate the tap two pipe diameters from any downstream pipe transition.

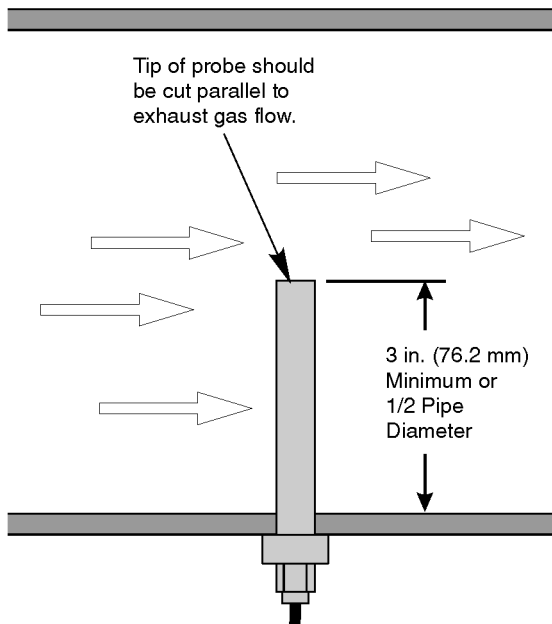
For example, in a 100 mm (4 in) diameter pipe, the tapping would be placed no closer than 300 mm (12 in) downstream of a bend or section change.

### Preferred Pressure Tap Location & Installation



**Figure 9**

If an uninterrupted straight length of at least five diameters is not available, care should be taken to locate the probe as close as possible to the neutral axis of the exhaust gas flow. This is necessary because measurements taken on the outside of a 90° bend at the pipe surface will be higher than a similar measurement taken on the inside of the pipe bend.



**Figure 10**

The pressure tapping can be made by using a 1/8 NPT "half coupling" welded or brazed to the desired location on the exhaust pipe. After the coupling is attached, a 3.05 mm (0.12 in) diameter hole is drilled through the exhaust pipe wall. Any burrs on the inside of the pipe wall should be removed so that the gas flow is not disturbed. The gauge or gauge hose can then be attached to the half coupling. The probe should be inserted to a depth equal to half the diameter of the pipe or a minimum of 3 in. (76.2 mm) as shown in **Figure 10**. The tip of the probe should be cut parallel to the exhaust gas flow.



**Calculating Backpressure**

Backpressure is calculated by:

$$P \text{ (kPa)} = \frac{L \times S \times Q^2 \times 3.6 \times 10^6}{D^5} + P_s$$

$$P \text{ (in. H}_2\text{O)} = \frac{L \times S \times Q^2 \times 3.6 \times 10^6}{187 \times D^5} + P_s$$

Where:

P = Back pressure (kPa), (in. H<sub>2</sub>O)

psi = 0.0361 x in. water column

kPa = 0.00981 x mm water column

L = Total Equivalent Length of pipe (m) (ft)

Q = Exhaust gas flow (m<sup>3</sup>/min), (cfm)

D = Inside diameter of pipe (mm), (in.)

S = Density of gas (kg/m<sup>3</sup>), (lb/ft<sup>3</sup>)

P<sub>s</sub> = Pressure drop of silencer/raincap (kPa), (in. H<sub>2</sub>O)

**Useful conversion factors:**

psi = 0.0361 x in. of water column

psi = 0.00142 x mm of water column

psi = 0.491 x in. of mercury column

kPa = 0.0098 x mm of water column

kPa = 0.25 x in. of water column

kPa = 3.386 x in. of mercury column

kPa = 0.145 psi

**Equivalent Length of Straight Pipe**

To obtain equivalent length of straight pipe for various elbows:

$$L = \frac{33D}{X} \quad \begin{array}{l} \text{Standard Elbow} \\ \text{elbow radius} = \text{pipe diameter} \end{array}$$

$$L = \frac{20D}{X} \quad \begin{array}{l} \text{Long Elbow} \\ \text{radius} = 1.5 \text{ diameter} \end{array}$$

$$L = \frac{15D}{X} \quad 45^\circ \text{ elbow}$$

$$L = \frac{66D}{X} \quad \text{square elbow}$$

Where X = 1000 mm or 12 in.

As shown by the equations, if 90° elbows are required, long radius elbows with a radius of 1.5 times the pipe diameter helps to lower resistance.

**Combined Exhaust Systems**

Although economically tempting, a common exhaust system for multiple installations is not acceptable. Combined exhaust systems with boilers or other engines allow exhaust gases to be forced into engines not operating. Water vapor created during combustion will condense in cold engines and quickly causes engine damage. Duct valves separating engine exhausts is also discouraged. High temperatures warp valve seats causing leakage.

Exhaust draft fans have been applied successfully in combined exhaust ducts, but most operate only whenever exhaust is present. To prevent turbocharger windmilling (without lubrication), draft fans

should not be operable when the engine is shut down. The exhaust system of non-running engines must be closed and vented.

3600 vee engines have two exhaust outlets, one for each bank. Combining these together with a Y-type fabrication, while possible, may result in unequal thermal growth and backpressure from one bank to the other. This unequal growth can put unwanted loading onto the turbocharger mounting or the flex bellows. The unequal backpressure can adversely affect the operation and performance of the engine. If the exhaust outlets are joined, these problems can be minimized by providing a flexible connection on each leg and by keeping each leg equal in length.

## Pipe Support Considerations

### Thermal Growth

Thermal growth of exhaust piping must be taken into account in order to avoid excessive load on supporting structures.

Steel exhaust pipe expands 1.13 mm/m (0.0076 in/ft) for each 100°C (100°F) rise of exhaust temperature. This amounts to 16.5 mm (0.65 in) expansion for each 3.05 m (10 ft) of pipe from 35° to 510°C (100° to 950°F).

Piping systems must be designed so thermal growth expands away from the engine.

A restraint member may be used to keep the ends of a long pipe run fixed in place, forcing all thermal growth towards the expansion joints.

Supports should also be located to allow expansion away from the engine. Supports can reduce strains or distortions to connected equipment and can allow component removal without additional support.

Flexible pipe connection, when insulated, must expand and contract freely within the insulation. This generally requires a soft material or insulated sleeve to encase the connection.

### Turbocharger Loading

Careful consideration must be given to the load external piping may induce on the turbocharger. To minimize the load carried by the turbocharger housing, a bellows should be placed as close as possible to the turbocharger outlet and downstream exhaust piping should be self supporting. The thermal growth of horizontal piping connected to the turbocharger exhaust must also be accounted for in the design.

Maximum allowable vertical load and bending moment limits are provided for each engine model. Consult the Technical Information Appendix for the appropriate information.

For 3600/G3600 series engines, the Caterpillar supplied bellows and adapter, or elbow and bellows options, account for the maximum allowable loading on the turbocharger. All other external piping must be self-supporting.

### Loading Calculations

The following example illustrates the type of calculation used to

determine vertical load and bending moment at the engine exhaust outlet due to the weight of the adapter mounted directly to the exhaust outlet.

**Figure 11** shows measurement points for the various distances when calculating the forces and moments on the turbocharger.

This example assumes a Gas 3516 with single exhaust outlet.

### **Vertical Exhaust**

W = Adapter Weight

I = 1/2 Bellows Weight

g = gravity = 9.82 m/s (32.2 ft)

### **With Caterpillar Hardware:**

W = 2.9 kg (6.4 lb)

I = 0.6 kg (1.4 lb)

### **Forces:**

$F_W = W \times g = 2.9 \times 9.82 = 28 \text{ N (6.4 lb)}$

$F_I = I \times g = 0.6 \times 9.82 = 6 \text{ N (1.4 lb)}$

### **Sum of Vertical Forces:**

$F_V = F_W + F_I = 28 + 6 = 34 \text{ N (7.8 lb)}$

### **Sum of Moments:**

$M = (h_1 \times F_W) + (h_2 \times F_I) = (0 \times 28) + (0 \times 6) = 0 \text{ N}\cdot\text{m (ft-lb)}$

Since for this particular engine model  $F_V < 111 \text{ N (25 lb)}$  and  $M < 120 \text{ N}\cdot\text{m (89.5 ft-lb)}$ , the exhaust system meets the load and moment requirements.

### **Horizontal Exhaust**

W = Adapter Weight

J = Elbow Adapter Weight

I = 1/2 Bellows Weight

### **With Caterpillar Hardware:**

W = 2.9 kg (6.4 lb)

I = 0.6 kg (1.4 lb)

J = 4.8 kg (10.7 lb)

$h_1 = 0$

$h_2 = 100 \text{ mm (3.9 in)}$

$h_3 = 580 \text{ mm (22.8 in)}$

### **Forces:**

$F_W = 28 \text{ N (6.4 lb)}$

$F_I = 6 \text{ N (1.4 lb)}$

$F_J = F \times g = 4.8 \times 9.82 = 47 \text{ N (10.7 lb)}$

### **Sum of Vertical Forces**

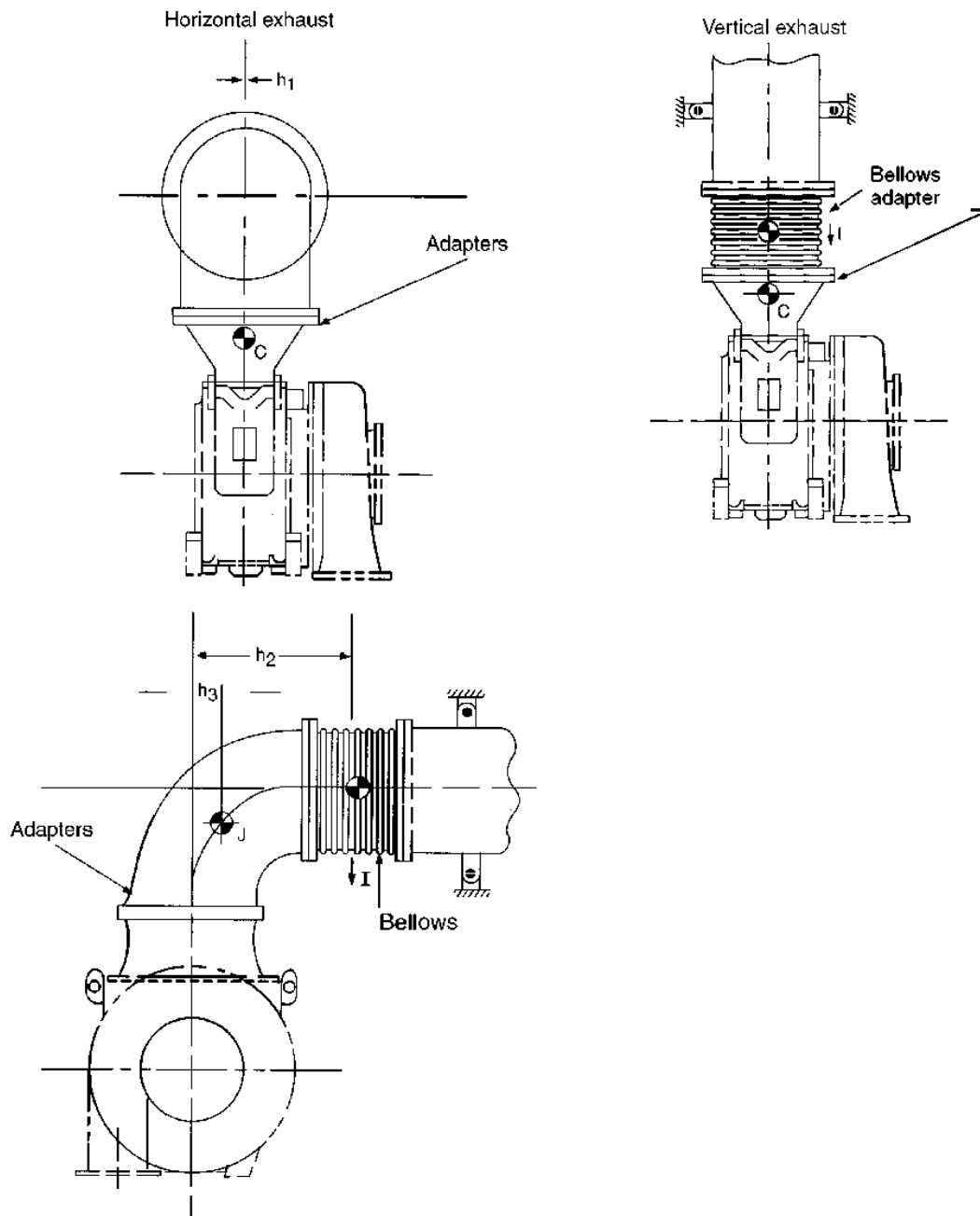
$F_V = F_W + F_I + F_J = 28 + 6 + 47 = 81 \text{ N (18.5 lb)}$

### **Sum of Moments**

$M = (h_1 \times F_W) + (h_2 \times F_I) + (h_3 \times F_J) = (0 \times 28) + (.100 \times 6) + (.580 \times 47) = 27.9 \text{ N}\cdot\text{m (20.8 ft-lb)}$

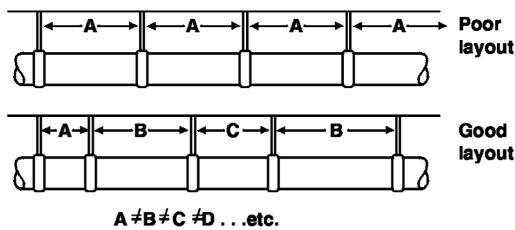
Since for this particular engine model  $F_V < 111 \text{ N (25 lb)}$  and  $M < 120 \text{ N}\cdot\text{m (89.5 ft-lb)}$ , the exhaust system meets the load and moment requirements.

### Vertical and Horizontal Exhaust

**Figure 11**

### Vibration Transmission

Piping connected to stationary engines requires isolation, particularly when resilient mounts are used. Without isolation, pipes can transmit vibrations long distances. Isolator pipe supports should have springs to attenuate low frequencies and rubber or cork to minimize high frequency transmissions.



**Figure 12**

To prevent build up of resonant pipe vibrations, support long piping runs at unequal distances as shown in **Figure 12**.

### Exhaust Discharge

Exhaust outlets, whether via an exhaust pipe or stack, must be designed to ensure that engine exhaust is discharged in such a manner that exhaust gas will not recirculate and be drawn back into the engine's environment. Engine air cleaners, turbochargers and aftercoolers may become contaminated with spent combustion products such as hydrocarbons and soot. This contamination can lead to various modes of failure.

Recirculation of hot exhaust gas can also adversely affect the ambient capability of the installation.

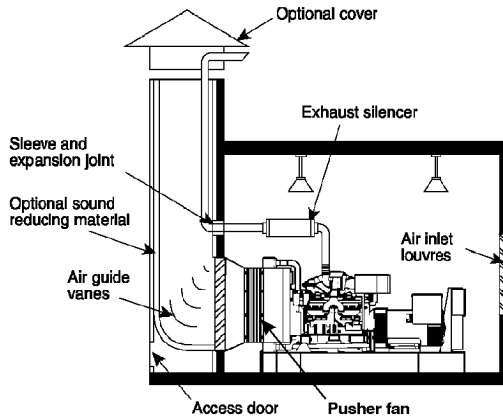
This can occur when air, significantly above ambient, is drawn through radiator equipped cooling systems. **Figure 13** through **Figure 17** show typical exhaust piping systems that are arranged to avoid recirculation.

### Common Exhaust Stack

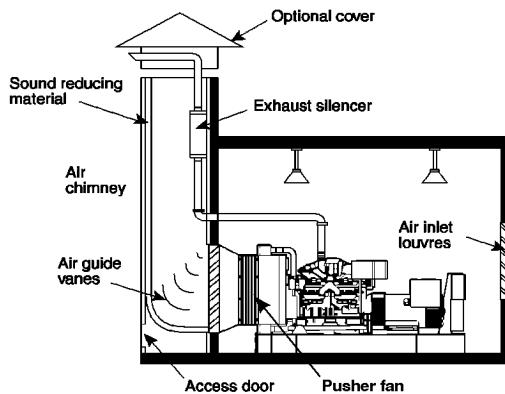
The exhaust can be directed into a special stack that also serves as the outlet for radiator discharge air and may be sound-insulated. In such instances the radiator discharge air enters below the exhaust gas inlet so that the rising radiator air tends to cool exhaust system components within the stack. Refer to the following two graphics.

The silencer may be located within the stack or in the room with its tail pipe extending through the stack and then outward. Air guide vanes should be installed in the stack to turn radiator discharge airflow upward and to reduce radiator fan air flow restriction. Alternatively the sound insulation lining may have a curved contour to direct air flow upward.

An exhaust stack will remain cooler and cleaner if the engine exhaust is contained within the exhaust piping throughout its run through the stack. If the exhaust pipe terminates short of the stack outlet, the discharged ventilation air will tend to cool the exhaust stack downstream of the point where it mixes with the exhaust gases.

**Figure 13**

**Figure 13** is an example of a horizontally mounted exhaust silencer with the exhaust pipe and radiator air utilizing a common stack.

**Figure 14**

**Figure 14** is an example of a vertically mounted exhaust silencer with the exhaust pipe and radiator air utilizing a common stack.

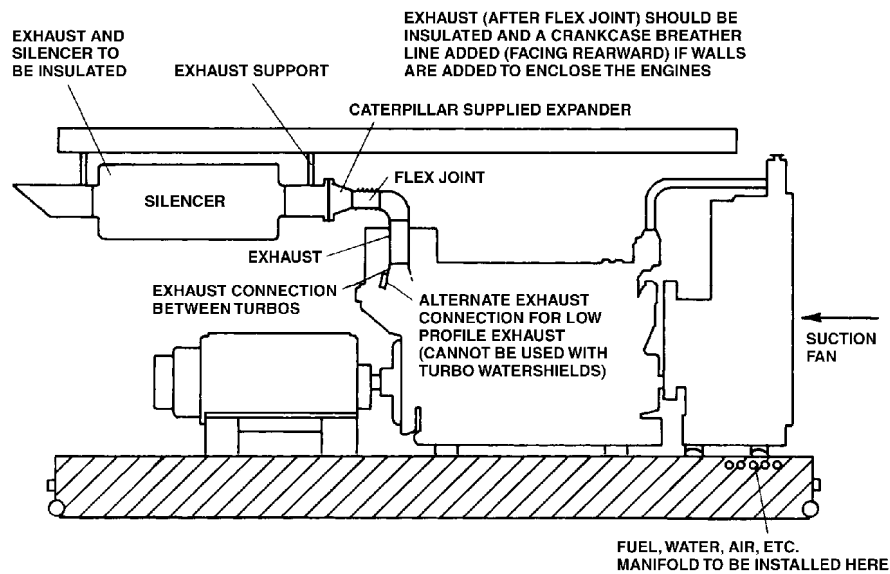
### Power Module or Drop-Over Enclosure

For a generator set enclosed in a power module or drop-over enclosure, the exhaust and radiator discharges should flow together, either above or below the enclosure without a stack.

This arrangement, as shown in **Figure 15** and **Figure 16**, will prevent the recirculation of exhaust gases back into the module or enclosure. Sometimes, for this purpose, the radiator can be mounted horizontally and the fan driven by an electric motor to discharge air vertically.

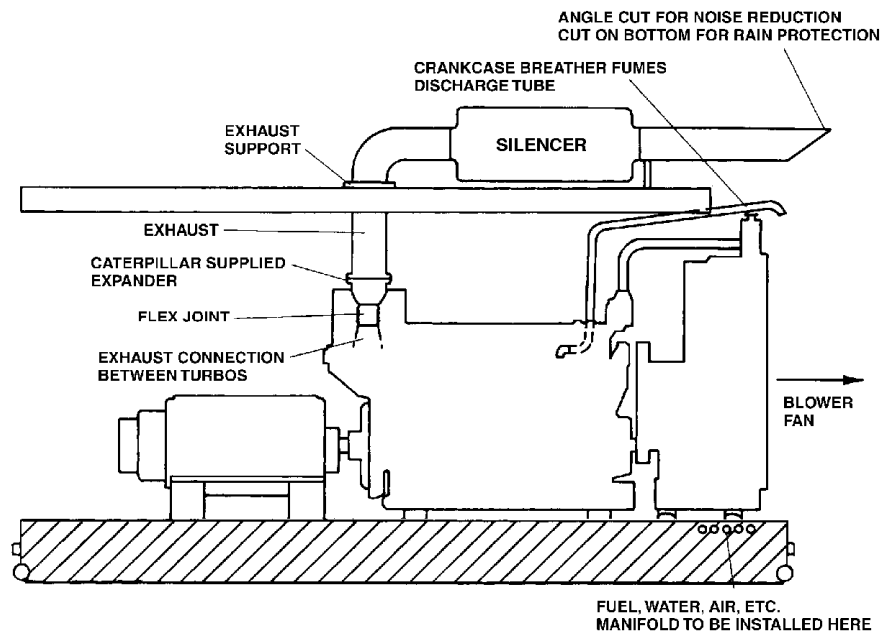
~~Note that the suction fan arrangement in **Figure 17** does not provide adequate cooling air to the generator. A separate source of generator cooling air is required for this configuration.~~

### Typical Power Module or Drop-Over Enclosure Exhaust System with Internal Silencer



**Figure 15**

### Typical Power Module or Drop-Over Enclosure Exhaust System with External Silencer



**Figure 16**

Crankcase pressures should not vary more than 25.4 mm H<sub>2</sub>O (1.0 in. H<sub>2</sub>O) from ambient barometric pressure for G3300, G3400 and G3500 engines. Some of the engines that monitor crankcase pressure will shutdown if excessive vacuum is sensed.

Restrictions higher than the limit on passive systems will encourage oil leaks. A powered system should draw no more than a 25.4 mm H<sub>2</sub>O (1.0 in. H<sub>2</sub>O) vacuum, or dirt and dust could be drawn into the engine past the main seals. Measurement should be made at the engine dipstick location with the engine at operating temperature, speed, and load.

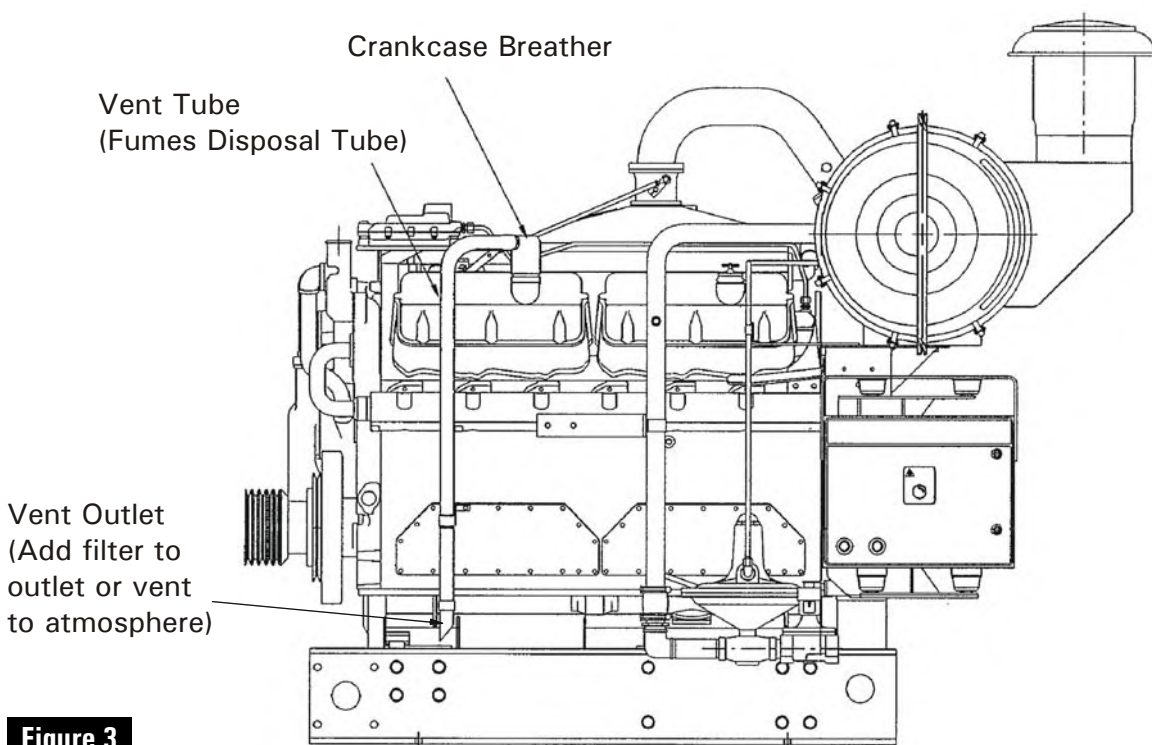
### Non-Ingestive

In regions that do not include crankcase emissions as part of the

total emissions for an engine, a non-ingestive crankcase ventilation system is an acceptable solution. In some situations customers can reduce cost and potential engine threats by venting the blow-by to the atmosphere. **Figure 3** shows a typical non-ingestive crankcase vent arrangement.

One of the goals of a PCV system is to increase the oil life of the engine. The removal of crankcase emissions can reduce the amount of oil degradation. It has been shown that a non-ingestive PCV system can double the oil life of an engine. However, the affects of PCV on oil life will vary with engine size, load, engine hours and ambient conditions.

### Typical Non-Ingestive Crankcase Ventilation Arrangement



**Figure 3**



Although a non-ingestive system is less complex than an ingestive system, there are a number of requirements and considerations for the effective venting of crankcase emissions.

Do not vent emissions directly into the engine room without filtration.

- Emissions can clog air filters, consequently causing engine damage.
- Exposure to crankcase emissions can cause problems in electrical equipment.
- Emissions can be a health hazard if discharged in a poorly ventilated room.
- Use a crankcase ventilation system to properly filter and vent emissions to the atmosphere.
- Multiple engines at a site require a separate vent lines for each engine. This prevents fumes and moisture produced by a running engine from entering an idle engine. The addition of moisture into an engine can cause corrosion and buildup of harmful deposits.
- Use appropriately sized vent pipes.

Crankcase vent pipes must be large enough to minimize back pressure. Through the entire life of the engine, blow-by may vary dramatically depending on engine operating temperature and type of oil used. As a general rule, blow-by on a new engine is approximately 0.02 m<sup>3</sup>/hr bkW (0.5 ft<sup>3</sup>/hr bhp).

Adequately size the vent pipes to accommodate a worn engine, with a blow-by rate of 0.04 m<sup>3</sup>/hr bkW (1 ft<sup>3</sup>/hr bhp). Size the vent pipe with a maximum of 13 mm H<sub>2</sub>O (0.5 in. H<sub>2</sub>O) pressure drop at full load.

The following formulas allow the crankcase ventilation designer to calculate a pipe diameter that will provide a back pressure of less than 13 mm H<sub>2</sub>O (0.5 in. H<sub>2</sub>O).

Use the following formulas to calculate back pressure:

$$P \text{ (kPa)} = \frac{L \times S \times Q^2 \times 3.6 \times 10^6}{D^5}$$

$$P \text{ (in. H}_2\text{O)} = \frac{L \times S \times Q^2}{187 \times D^5}$$

Where:

P = Back pressure (kPa), (in. H<sub>2</sub>O)

psi = 0.0361 x in. water column

kPa = 0.00981 x mm water column

L = Total Equivalent Length of pipe (m), (ft)

Q = Exhaust gas flow (m<sup>3</sup>/min), (cfm)

D = Inside diameter of pipe (mm), (in.)

S = Density of gas (kg/m<sup>3</sup>), (lb/ft<sup>3</sup>)

S (kg/m<sup>3</sup>) = 1.08

S (lb/ft<sup>3</sup>) = 0.067

To obtain equivalent length of straight pipe for various elbows:

Standard Elbow  
(radius = diameter)  $L = \frac{33D}{X}$

Long Radius Elbow  
(radius = 1.5 diameter)  $L = \frac{20D}{X}$

45° Elbow  
(radius = 1.5 diameter)  $L = \frac{15D}{X}$

Square Elbow  
(radius = 1.5 diameter)  $L = \frac{66D}{X}$

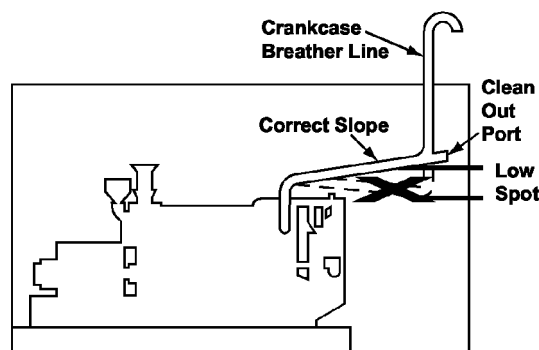
Where:

$$X = 1000 \text{ mm (12 in)}$$

Calculate the pipe diameter according to the formula, then choose the next larger commercially available pipe size.

As evidenced in the formulas, if 90° bends are required, then using long radius elbows, with a radius of 1.5 times the pipe diameter, helps lower resistance.

Loops or low spots in a crankcase vent pipe must be avoided to prevent condensation from building up in the pipe and restricting the emissions discharge. Where horizontal runs are required, install the pipe with a gradual slope from the engine as shown in **Figure 4**. The slope should be approximately 41.7 mm/m (1/2 in/ft).



**Figure 4**

The weight of the vent pipes will require separate off-engine supports as part of the installation design. Any horizontal or vertical run of pipe that cannot be disassembled for cleaning should have clean-out ports installed.

Crankcase emissions from non-ingressive crankcase ventilation systems must not discharge into the air ventilation ducts or exhaust pipes. Ducts and pipes on these engines will become coated with oily deposits that create a fire hazard.

Vent the crankcase pipe directly into the atmosphere and direct it to keep rain or spray from entering the engine. Give consideration to equipment located near the discharge area as well as to the building itself. Over a period of time, very small amounts of oil carry-over can accumulate and become unsightly, even harmful, to auxiliary equipment.

If the mounting beams will be subjected to external bending forces, such as frame flexing of the chassis in mobile applications, the engine and driven equipment must be mounted to the beams using a three point mounting system. This type of mounting system, similar to **Figure 6**, supports the engine at a single point at the front and at two points, one on each side at the rear of the engine or driven unit. This mounting system is capable of efficiently allowing large amounts of frame deflection without imparting stresses to the mounting or engine and driven unit.

### Engines and Packaged Units with Mounting Rails

Standard Caterpillar mounting rails must be used for engines and packaged units requiring mounting rails. Two types of standard mounting rails are available, depending on the application.

#### Three Point Mounting

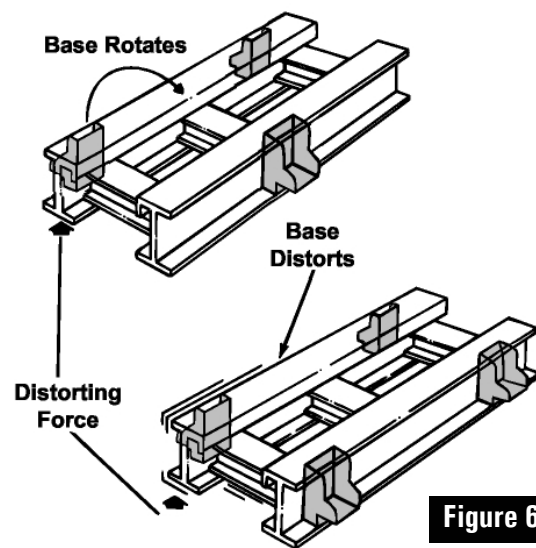
Three-point mounting rails are used if the rails are to be secured to a base or foundation that may subject the mounting rails to distorting forces.

Suspending the power unit on three points isolates the unit from deflection of the substructure, thus maintaining proper relationship and alignment of all equipment and preventing distortion of the engine block.

More than three mounting points can cause base distortion as shown in **Figure 6**. The three point mounting rails can only be used for

close-coupled driven units where the rails are extended for the mounting of these units.

Objectionable vibration can occur if the power unit is not mounted on well-supported structures or is not anchored securely. In addition to the three-point mounting, vibration isolators may be required to isolate objectionable vibrations.

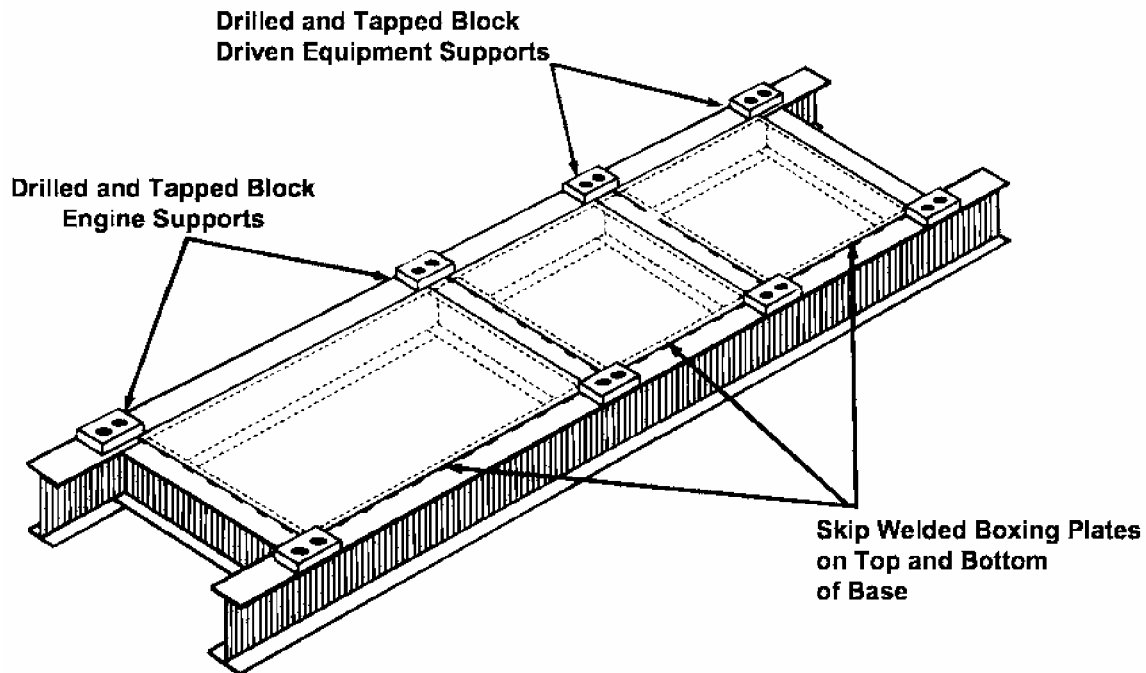


**Figure 6**

#### ~~Four Point Mounting~~

~~Four point mounting rails are used if the rails are to be secured to a base or foundation that will not subject the rails to distorting forces. Refer to Figure 7. These rails can be used for either remote mounted or close coupled driven units. For close coupled driven units, extended mounting rails are available so the driven equipment can be fastened directly to the rails.~~

### Base for Engines with Remote Mounted Driven Equipment



**Figure 42**

#### Bases for Engines with Remote-Mounted Loads

The design requirements for bases used on engines with close-coupled units also apply to bases used for engines with remote-mounted units. Bases for use with remote-mounted units must be more rigid. The base must absorb the full-load torque between the engine and driven unit without causing excessive deflection in the coupling.

The base shown in **Figure 42** is a boxed beam design that provides a torsionally rigid base.

Boxing consists of welding steel plates on top and bottom surfaces of machinery base girders. Skip-weld the plates to prevent excessive base

distortion during welding. Boxing is done to make the base structure stiffer.

The additional stiffness is necessary to resist torque loads between the engine and remote-mounted driven equipment and to resist possible vibration loads. Vibration-induced base loads are difficult to predict.

Experience has shown boxing is effective in preventing base cracking and misalignment.

#### Caterpillar Special Duty Bases

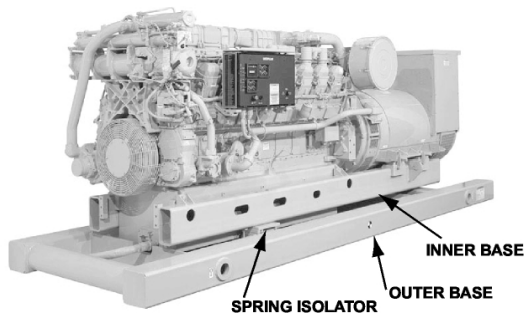
##### Offshore Power Modules

The Caterpillar petroleum offshore base consists of a base-within-a-base. The inner base is three-point mounted, with integral spring

isolators and limit stops, to the outer base. The outer base can be welded to the rig support structure. The inner base structure is not the same as engine rails used in other applications. Refer to **Figure 43**.

The outer base must be supported by large girders, and can either be welded or bolted to the rig structure. Inadequate support may result in power module vibrations.

### Offshore Power Module



**Figure 43**

### ~~Land Rig Power Modules~~

The Caterpillar petroleum land rig base uses 460 mm (18 in) wide flange beams, available in lengths of 7.85 m, 9.37 m and 12.42 m (25 ft 9 in, 30 ft 9 in and 40 ft 9 in), to allow matching base length to equipment needs. As with the offshore power modules, alignment integrity is provided by using a base

~~within a base design as shown in Figure 44 and Figure 45.~~

~~The 12.42 m (40 ft 9 in) base has no decking provided from the rear of the generator to the rear of the base. Customer supplied auxiliary equipment is to be mounted here, necessitating customer supplied decking and reinforcement.~~

~~Site preparation is generally required for land rig power modules in order to provide level, firm soil. In some cases, planking or concrete surfacing may be necessary. Certain types of soil, such as fine clay, loose sand, sand near the ground water level, or soil that is freezing or thawing, are particularly unstable under dynamic loads. Loose planking under the power modules may also cause power module vibration.~~

~~Because ground conditions may vary from well location to well location, vibrations may result which are not due to misalignment or unbalanced parts. Unstable ground, as previously mentioned, may be reacting to normal forces within the engine/generator combination, whereas at another well location no such reaction may occur.~~

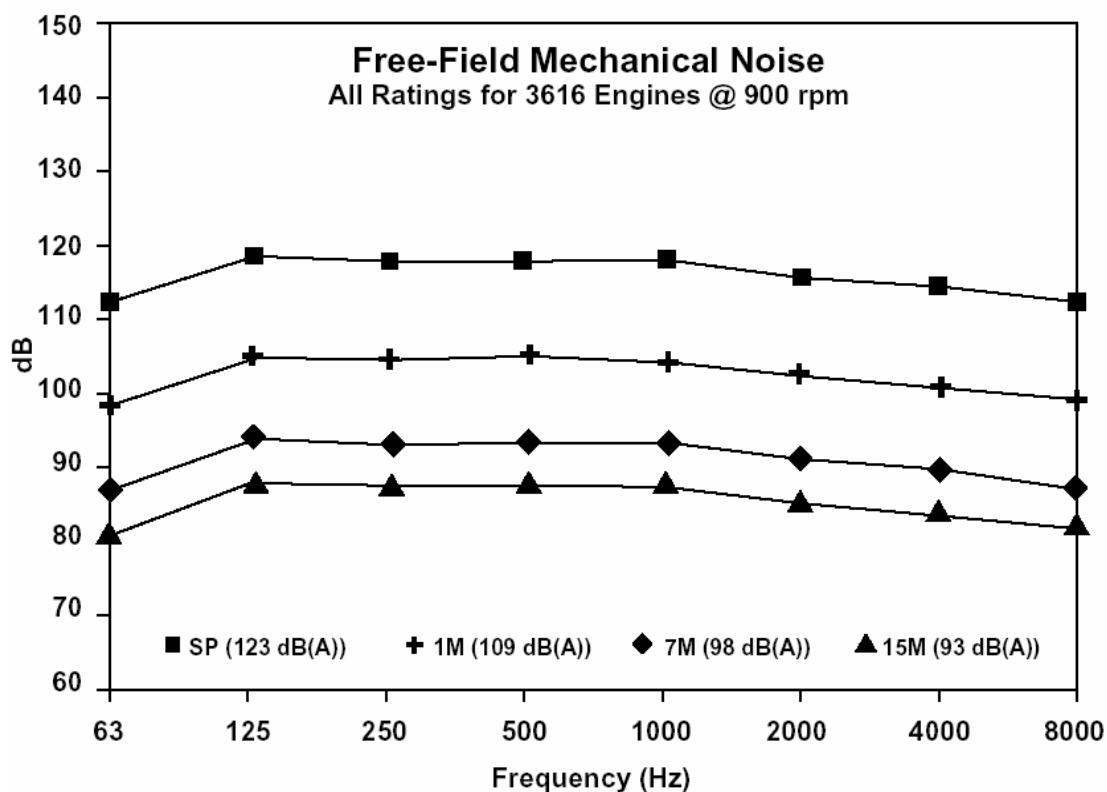
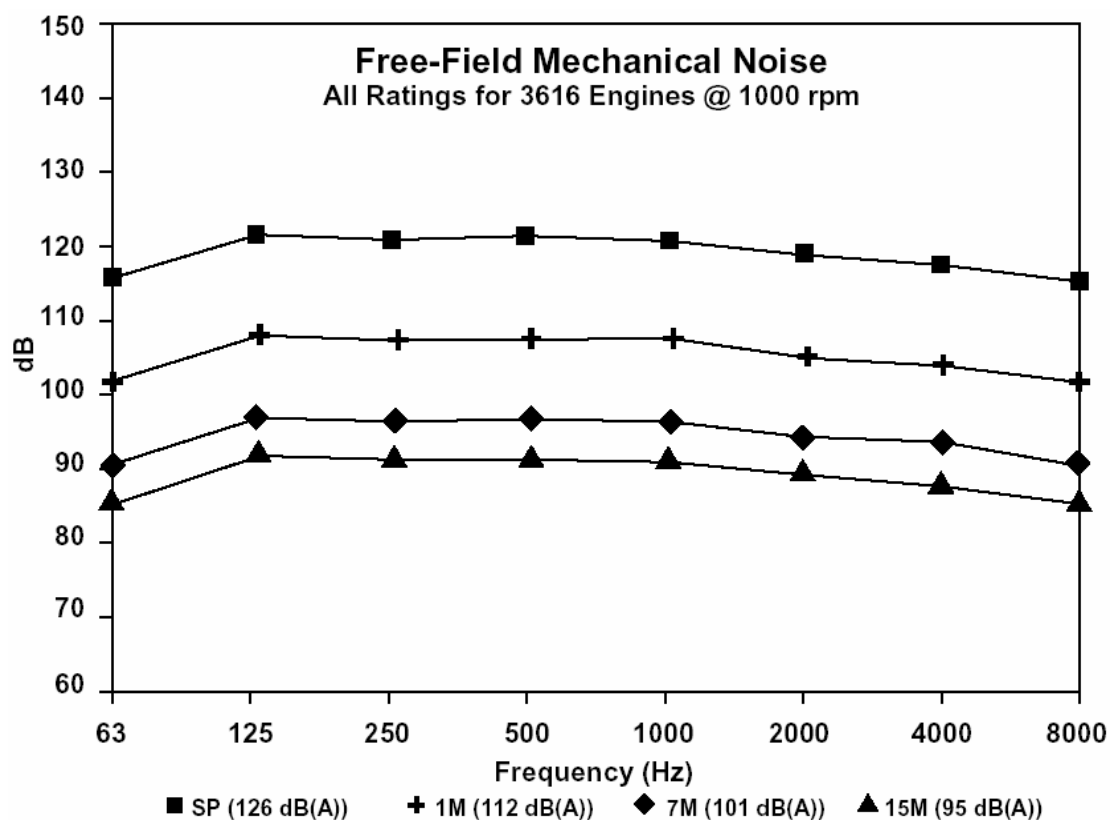
~~In addition, rough handling may occur during rig moves, so alignment should be checked after every move.~~

**6.0 APPENDIX****APPENDIX LIST**

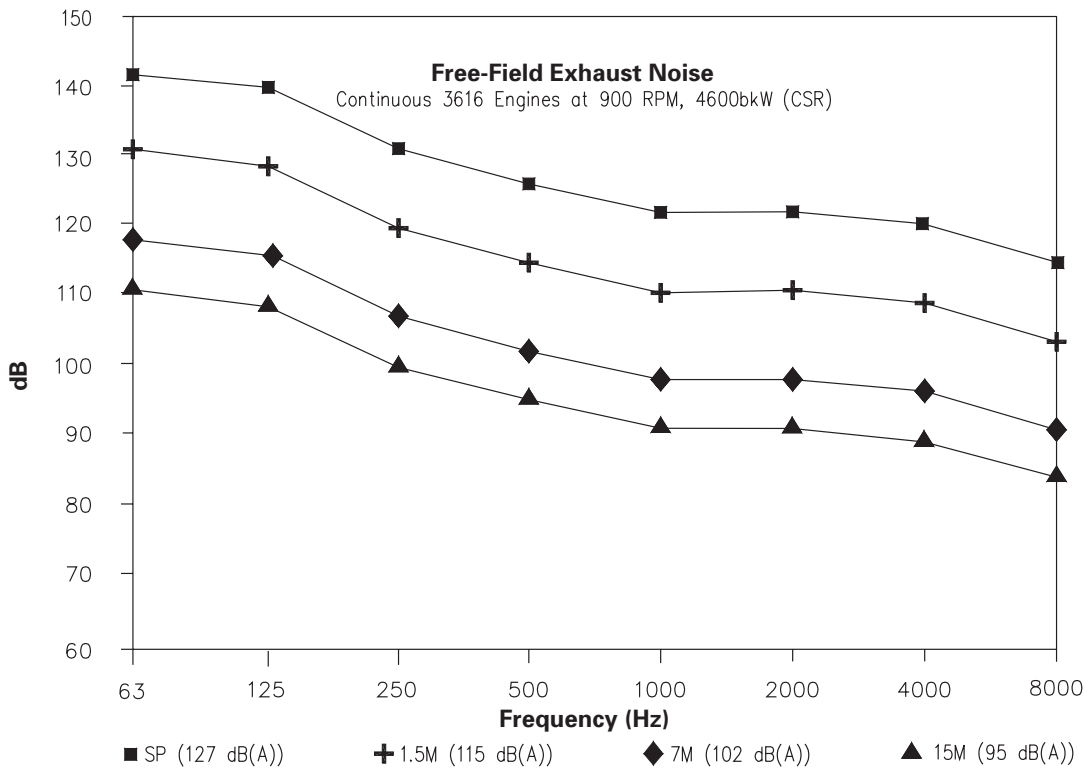
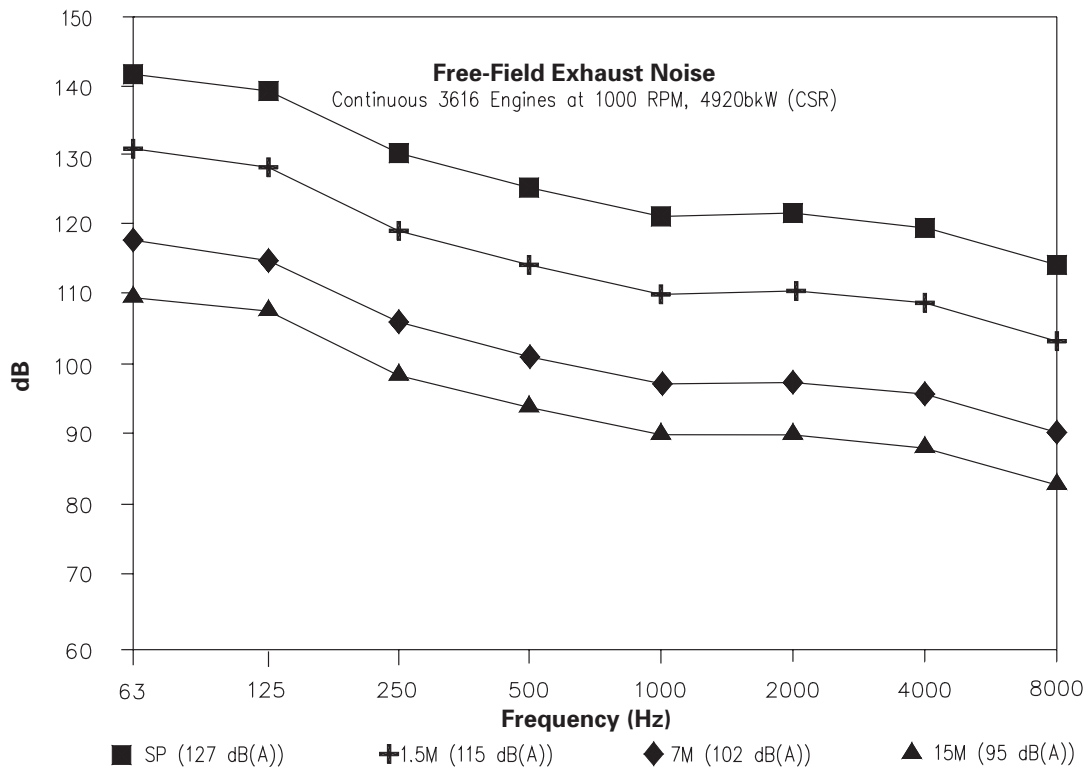
<b>No</b>	<b>Sheet No</b>	<b>Contents</b>	<b>Total Pages</b>
1		3600 diesel Engine General Data	1
2		Engine Mech. Noise Curves	1
3		Engine Exhaust Noise Curves	1
4		Engine Service Envelopes / Components Weights	4
5		Engine Maintenance Interval Schedule	2

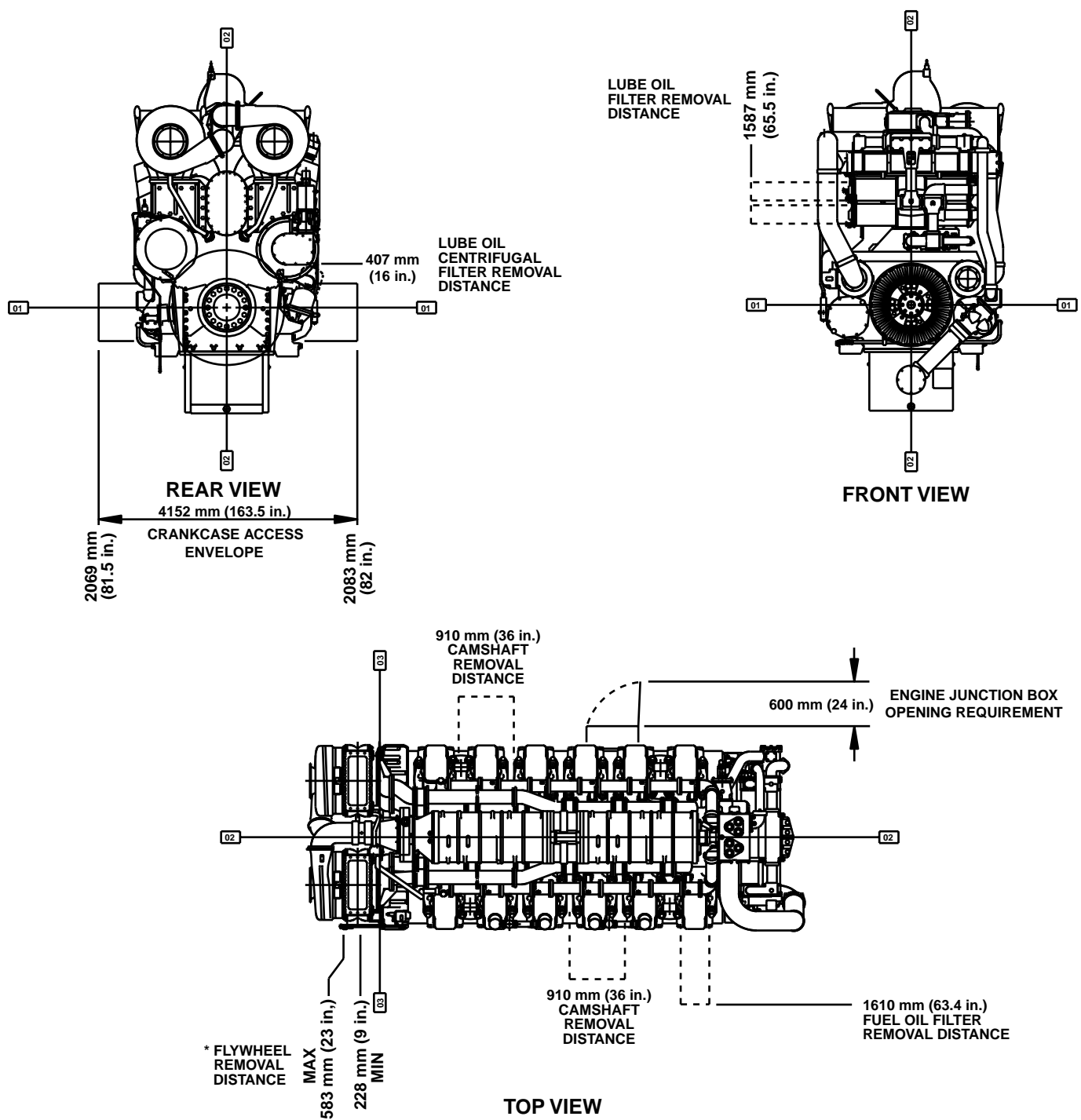
## General Data

System Description Metric (English)	3606	3608	3612	3616
Cylinder Bore mm (in)	280 (11)	280 (11)	280 (11)	280 (11)
Stroke mm (in)	300 (11.8)	300 (11.8)	300 (11.8)	300 (11.8)
Displacement/Cylinder L (in <sup>3</sup> )	18.5 (1127)	18.5 (1127)	18.5 (1127)	18.5 (1127)
Firing Pressure, maximum (Continuous/CSR) kPa (psi)	16,200 (2350)	16,200 (2350)	16,200 (2350)	16,200 (2350)
Firing Pressure, maximum (Prime Power/MCR) kPa (psi)	17,334 (2514)	17,334 (2514)	17,334 (2514)	17,334 (2514)
Firing Pressure, maximum (Standby) kPa (psi)	18,500 (2684)	18,500 (2684)	18,500 (2684)	18,500 (2684)
Rated Speed rpm	700 to 1000	700 to 1000	700 to 1000	700 to 1000
Mean Piston Speed m/s (ft/s)	7.2 – 10.0 (23.6 – 32.8)	7.2 – 10.0 (23.6 – 32.8)	7.2 – 10.0 (23.6 – 32.8)	7.2 – 10.0 (23.6 – 32.8)
Idle speed (low) rpm	300 to 400	300 to 400	300 to 400	300 to 400
Idle speed (high) rpm	720 to 1000	720 to 1000	720 to 1000	720 to 1000
Firing Order – CCW	1-5-3-6-2-4	1-6-2-5-8-3-7-4	1-12-9-4-5-8-11-2-3-10-7-6	1-2-5-6-3-4-9-10 -15-16-11-12-13-14-7-8
Firing Order – CW	1-4-2-6-3-5	1-4-7-3-8-5-2-6	1-6-7-10-3-2-11-8-5-4-9-12	1-8-7-14-13-12-11-15 -16-10-9-4-3-6-5-2
Wet weight kg (lb)	16,804 (36,775)	20,221 (44,486)	26,848 (59,065)	32,104 (70,489)
Dry weight kg (lb)	15,680 (34,500)	19,000 (41,800)	25,140 (55,300)	29,950 (65,900)
Center of Gravity Distance from Cylinder Block Rear Face mm (in)	1290 (50.8)	1700 (66.9)	1411 (55.6)	1858 (73.1)
Vertical Distance Above Crankshaft Centerline mm (in)	350 (13.8)	350 (13.8)	380 (14.9)	380 (14.9)
Transverse Distance from Crankshaft Centerline	On Crank Center	On Crank Center	On Crank Center	On Crank Center
NOTE: Center of gravity locations apply to dry runable engines.				





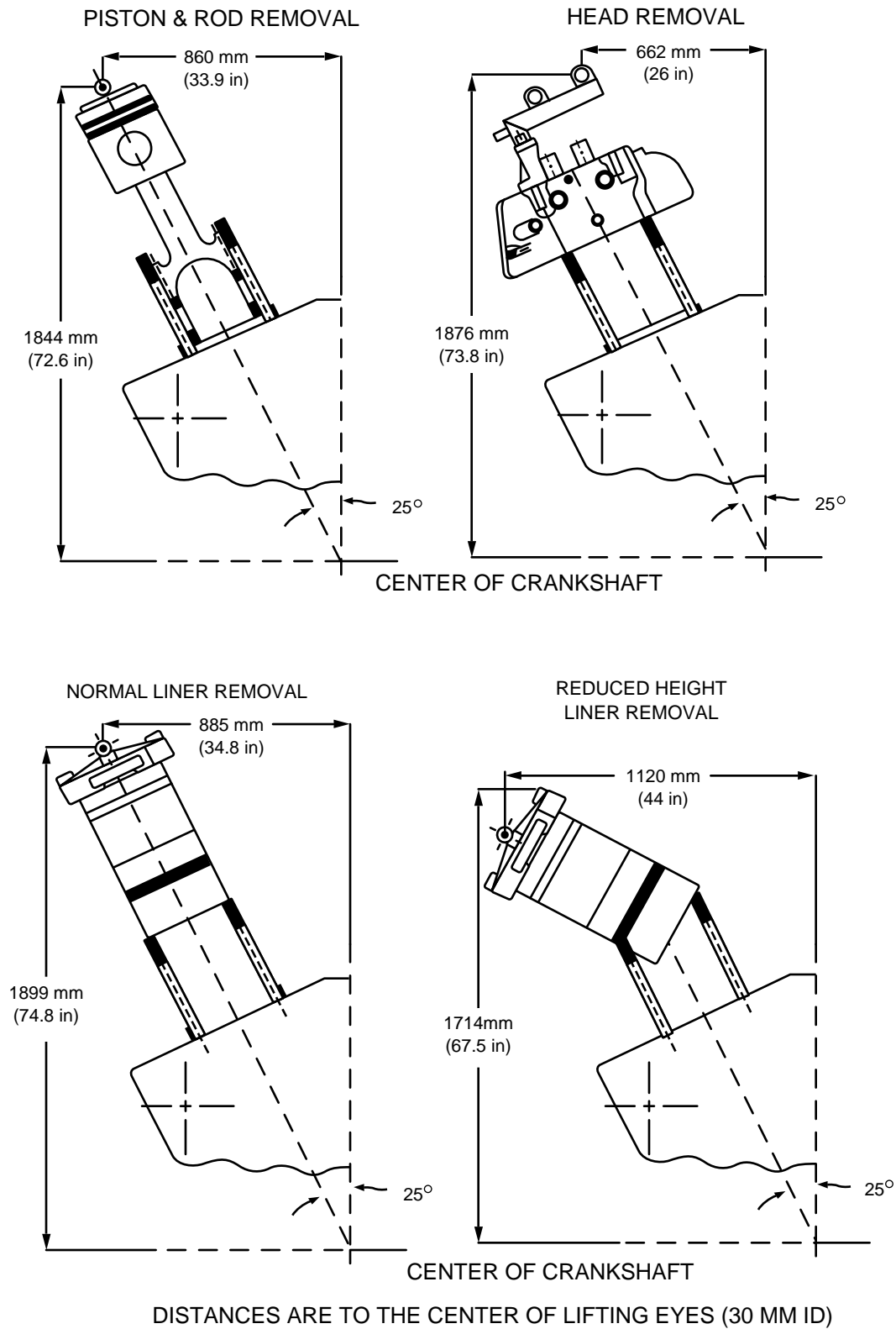




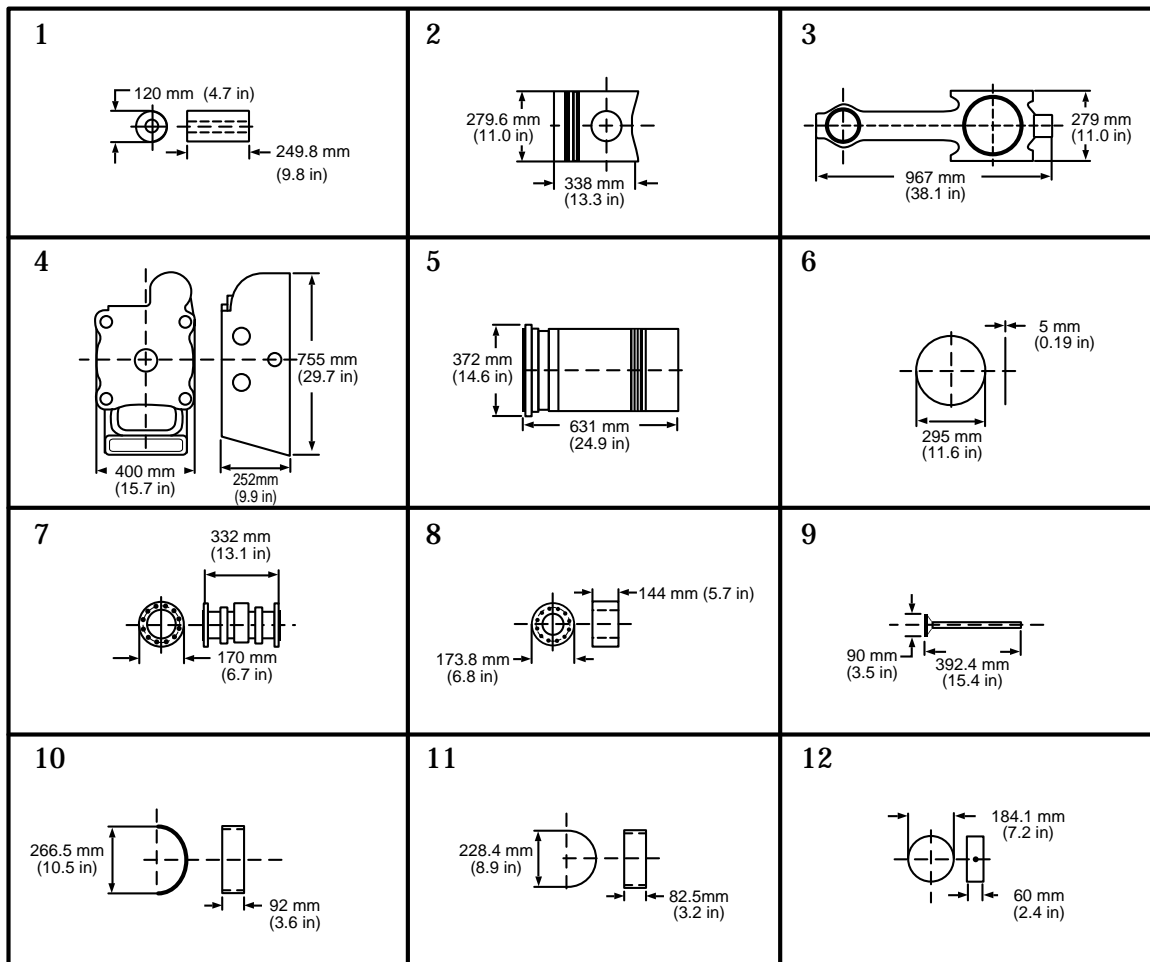
\* If Caterpillar service tool is used.  
355 mm is required if service tool is not used.

**Figure 3**

## Service Removal Distances Vee Engine

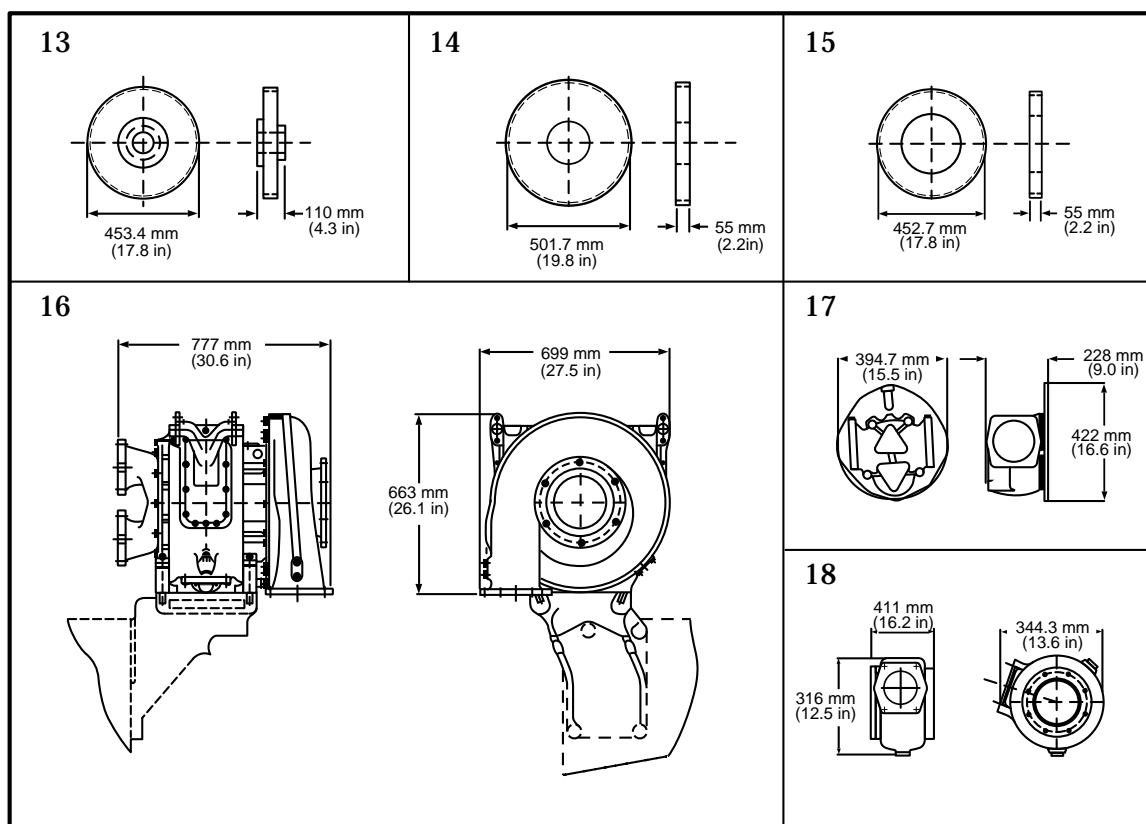


**Figure 4**



Shipping Weights							
Item	Description	kg	(lb)	Item	Description	kg	(lb)
1.	Piston Pin	19.1	(42)	7.	Camshaft Segment	30	(66)
2.	Piston	35.2	(77)	8.	Cam Journal	13.6	(30)
3.	Connecting Rod	56.2	(124)	9.	Valve	1.4	(3)
4.	Cylinder Head Assembly	235	(517)	10.	Main Crankshaft Bearing	4.5	(10)
5.	Cylinder Liner	127.9	(282)	11.	Connecting Rod Bearing	2.3	(5)
6.	Piston Ring	0.3	(0.7)	12.	Cam Bearing	1.4	(3.1)

**Figure 5**



Shipping Weights							
Item	Description	kg	(lb)	Item	Description	kg	(lb)
13.	Camshaft Drive Gear	43.5	(96)	16.	Turbocharger (VTC 254)	500	(1100)
14.	Idler Gear	35.4	(78)	17.	Oil Pump	120	(264)
15.	Crankshaft Gear	37.6	(83)	18.	Water Pump	95	(210)

**Figure 6**

## Maintenance Interval Schedule — CSR and MCR Ratings

### Daily Maintenance

- Check oil level for the air starting motors
- Drain air tank moisture and sediment
- Check the cooling system coolant level
- Check the engine oil level
- Drain the fuel filter/water separator
- Inspect engine air cleaner service indicator
- Clean engine air precleaner
- Check governor actuator oil level
- Inspect instrument panel
- Perform walk-around inspection

### 250 Hour Maintenance (or 6 weeks)

- Clean the air starting motor lines screen
- Test the coolant in the cooling system
- Obtain engine oil sample for analysis
- Check the oil mist detector
- Test air shutoff
- Check governor actuator linkage

### 500 Hour Maintenance (or 3 months)

- Clean the oil mist detector
- Replace the filter for the oil mist detector
- Inspect the engine mounts

### 6 Month Maintenance

- Obtain Level II cooling system coolant analysis (ELC)

### Initial 1,000 Hour Maintenance

- Inspect the engine valve rotators
- Inspect/Adjust engine valve lash

### 1,000 Hour Maintenance (or 6 months)

- Check the engine mounts
- Lubricate barring device
- Inspect engine air cleaner service indicator
- Change engine oil filter
- Inspect exhaust piping
- Replace fuel system primary filter/water separator element
- Replace fuel system secondary filter
- Clean/inspect magnetic pick-ups
- Lubricate prelube pump

### 2,000 Hour Maintenance (or 1 year)

- Drain aftercooler condensation
- Clean/Replace oil mist detector
- Inspect/Adjust engine valve lash

### Annual Maintenance

- Obtain Level II cooling system coolant analysis (ELC)

### 4,000 Hour Maintenance (or 1 year)

- Clean and inspect the aftercooler core
- Clean the lubricator bowl for the air starting motors

### 8,000 Hour Maintenance (or 1 year)

- Calibrate engine protection devices

### 8,000 Hour Maintenance (or 3 years)

- Change the coolant in the cooling system (DEAC)
- Replace the water temperature regulator
- Inspect camshaft roller followers
- Add cooling system coolant extender (ELC)
- Inspect crankshaft vibration damper
- Replace engine oil temperature regulator
- Inspect exhaust shields
- Replace governor actuator oil
- Inspect turbochargers
- Inspect water pumps

### 16,000 Hour Maintenance

- Change cooling system coolant (ELC)

### 16,000 to 24,000 Hour Maintenance

#### Top End Overhaul

#### *Remanufacture or Rebuild*

- Cylinder heads
- Cylinder head exhaust inserts
- Cylinder head dowels
- Cylinder head inlet inserts
- Cylinder head inner valve springs
- Cylinder head outer valve springs
- Cylinder head valve locks
- Cylinder head valve rotocoils
- Cylinder head valve spring guides
- Exhaust valves
- Inlet air valves

If the valves and seats are not replaced, grind the valves and lap the valves for full face contact.

## *Replace*

- Connecting rod bearings
- Cylinder head gaskets
- Exhaust manifold gaskets
- Fuel transfer pump seals
- Inlet air lines seals
- Oil cooler seals
- Oil pump bearings and seals
- O-ring seals and plugs
- Oil temperature regulators and seals
- Turbocharger bearings, bushings, and seals
- Valve lubricator pump
- Water pump bearings and seals
- Water temperature regulators and seals
- Unit injectors

## *Clean and Inspect*

- Oil cooler core
- Oil suction screen

## *Inspect/Replace*

- Cylinder sleeves
- Exhaust shields
- Starting motor

## **36,000 to 44,000 Hour Maintenance**

### Major Overhaul

#### *Remanufacture or Rebuild*

- Centrifugal oil filter bearings
- Cylinder heads
- Starting motor

#### *Replace*

- Accessory group bearings
- Cylinder head valves and valve guides
- Cylinder head valve spring guides
- Exhaust manifold seals and bellows
- Exhaust shields
- Front gear train bearings
- Inlet air lines seals
- Oil temperature regulators and seals
- Unit injectors
- Water temperature regulators and seals
- Grind the valves and lap the valves for full face contact.

#### *Inspect/Replace*

- Aftercooler core
- Alarm and shutoff controls
- Camshafts

- Camshaft bearings
- Connecting rod bearings
- Crankshaft
- Cylinder liners
- Cylinder sleeves
- Exhaust manifolds
- Front gear group
- Main bearings
- Oil pump bushings and seals
- O-ring seals and plugs
- Pistons and piston rings
- Priority valve
- Rear gear train bearings and seals
- Rocker arm bearings
- Thrust bearings
- Turbocharger bearings, bushings, and seals
- Valve lubricator pump
- Valve mechanism group

If the crankshaft or the camshaft is removed for any reason, use the magnetic particle inspection process to check the component for cracks.

#### *Clean and Inspect*

- Oil cooler core
- Oil suction screen

#### *Replace gaskets and seals*

*Replace the gaskets and seals for the following components:*

- Camshaft front covers
- Camshaft drive gear covers
- Central structure covers
- Crankcase side covers
- Crankshaft
- Crankshaft vibration damper
- Front housing group
- Fuel lines
- Fuel transfer pump
- Gear inspection group
- Oil cooler
- Oil lines
- Power take-off covers
- Priority valve group
- Rear housing group
- Rear structure covers
- Valve covers
- Water lines